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Modeling of a Heat Pump Charged With a Non-Azeotropic Refrigerant Mixture

Piotr Domanski

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Modeling of a Heat Pump Charged With a Non-Azeotropic Refrigerant Mixture

Piotr Domanski

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ABSTRACT

An analysis of the vapor compression cycle and the main components of an air-to-air heat pump charged with a binary non-azeotropic mixture has been performed for steady-state operation. The general heat pump simulation model HPBI has been formulated which is based on independent, analytical models of system components and the logic linking them together. The logic of the program requires an iterative solution of refrigerant pressure and enthalpy balances, and refrigerant mixture and individual mixture component mass inventories.

The modeling effort emphasis was on the local thermodynamic phenomena which were described by fundamental heat transfer equations and equation of state relationships among material properties. In the compressor model several refrigerant locations were identified and the processes taking place between these locations accounted for all significant heat and pressure losses. Evaporator and condenser models were developed on a tube-by-tube basis where performance of each coil tube is computed separately by considering the cross-flow heat transfer with the external air stream and the appropriate heat and mass transfer relationships. A constant flow area expansion device model was formulated with the aid of Fanno flow theory. Equation of state for mixtures is described and equation constants for a R13B1/R152a mixture are given.

The developed heat pump model was validated by checking computer results against laboratory tests data of one heat pump at two cooling and two heating rating points.

Program HPBI can be used to evaluate potentials of non-azeotropic mixtures working in a split residential heat pump. User's Manual and listing of the program is included in the report.

Key words: air conditioner; capillary tube; coil; compressor; condenser; expansion device; heat pump; modeling; mixture; non-azeotropic refrigerant; vapor modeling cycle

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DISCLAIMER

In view of the presently accepted practice of the building industry in the United States and the structure of the computer software used in this project, common U.S. units of measurement have been used in this report. In recognition of the United States as a signatory to the General Conference of Weights and Measures, which gave official status to the SI system of units in 1960, appropriate conversion factors have been provided in the table below. The reader interested in making further use of the coherent system of SI units is referred to: NBS SP330, 1972 Edition, 'The International System of Units,' or E380-72, ASTM Metric Practice Guide (American National Standard 2210.1).

METRIC CONVERSION FACTORS

Length	1 inch (in) = 25.4 millimeters (mm) 1 foot (ft) = 0.3048 meter (m)
Area	1 ft ² = 0.092903 m ²
Volume	1 ft ³ = 0.028317 m ³
Temperature	F = 9/5 C + 32
Temperature Interval	1°F = 5/9°C or K
Mass	1 pound (lb) = 0.453592 kilogram (kg)
Mass Per Unit Volume	1 lb/ft ³ = 16.0185 kg/m ³
Energy	1 Btu = 1.05506 kilojoules (kJ)
Specific Heat	1 Btu/[(lb)(°F)] = 4.1868 kJ/[(kg)(K)]
Gallon	1 gallon = 0.0037854 m ³

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LIST OF SYMBOLS

A	= flow cross section area
A_f	= fin surface area
A_h	= heat transfer surface area
$A_{n,i}$	= coefficients in equation (89)
$A_{p,i}$	= pipe inside surface area
$A_{p,m}$	= pipe mean surface area
A_o	= pipe total outside surface area
a_1, a_2, a_3	= coefficients in equation (45)
CP	= pressure drop parameter
CQ	= heat transfer parameter
C_c	= correction factor in equation (34)
C_e	= clearance volume, in fraction of compressor stroke volume
C_p	= specific heat at constant pressure
C_v	= specific heat at constant volume
D	= tube inside diameter
D_o	= tube outside diameter
D_f	= fin tip diameter
d_1, d_2	= dimensions representing arrangement of tubes in a coil, as per figure 21
E	= electrical power input
F_1, F_2	= dimensionless parameters used in equation (83)
f	= Fanning friction factor
$f_{tp,L}$	= friction factor for the liquid portion of two-phase flow, flowing alone in the tube
G	= $\frac{m}{A}$, mass flux, or Gibbs free energy
G_{max}	= air mass flux between two adjacent fins
H	= liquid refrigerant level in an accumulator
h	= heat transfer coefficient, h_c refers to air-side forced convection heat transfer coefficient for wet air, $h_{c,o}$ refers to air-side

forced convection heat transfer coefficient for dry air, h_i refers to inside tube convection heat transfer coefficient

$h_{D,o}$	= air-side mass transfer coefficient
hp	= horse power
i	= enthalpy
i_{fg}	= latent heat of evaporation or condensation
J	= the mechanical equivalent of heat
K	= flow contraction coefficient
K_f	= $J \cdot i_{fg} \cdot \Delta x/L$, boiling number
k	= thermal conductivity
L	= tube length
Le	= $\frac{h_{c,o}}{h_{D,o} C_{p_a}}$, Lewis number
M	= mass of refrigerant, or molecular weight
m	= mass flow rate
Nu	= $\frac{h \cdot D}{k}$, Nusselt number
n	= polytropic index
P	= pressure
Pr	= $\frac{\mu \cdot C_p}{k}$, Prandtl number
Q	= heat transfer rate
R	= rate of moisture removal per unit area, or universal gas constant
R'	= rate of moisture removal per unit width of a fin
Re	= $\frac{G \cdot D}{\mu}$, Reynolds number
$Re_{tp,L}$	= Reynolds number for the liquid portion of two-phase, flowing alone in the tube
RPM	= compressor number of revolutions per minute
S	= tube perimeter
s	= entropy
T	= temperature

$T_{f,b}$	= fin base temperature
$T_{f,m}$	= mean fin temperature
$T_{r,g}$	= refrigerant saturation temperature
t	= fin thickness
U	= overall heat transfer coefficient
V	= velocity
V_s	= compressor swept volume
\forall	= volume
v	= specific volume or molar volume
W_c	= mechanical power available for compression process
W_e	= mechanical power input
w_a	= humidity ratio of air, $w_{a,i}$ refers to tube row inlet, $w_{a,e}$ refers to tube row outlet
w_w	= humidity ratio of saturated air at temperature of wetted water film
XL	= molar composition of liquid phase
XM	= molar composition of mixture
XV	= molar composition of vapor phase
XW	= weight composition of mixture
x_{tt}	$= \left(\frac{1-x}{x} \right)^{0.9} \left(\frac{v_L}{v_V} \right)^{0.5} \left(\frac{\mu_L}{\mu_V} \right)^{0.1}$, Lockhart-Martinelli parameter
x	$= \frac{m_V}{m_V + m_L}$, quality
x_p	= pipe wall thickness
y	= fin height
Z_{tp}	= fraction of the tube in the two-phase region for the tube with liquid and two-phase flow
Z_V	= fraction of the tube in the superheated vapor region for the tube with two-phase and superheated vapor flow
z	= distance between adjacent fins
α	= void fraction

β	= exponent in equation (83)
γ	= isentropic index
σ	= liquid layer (frost) thickness or Stefan Boltzman constant
ϵ	= surface emissivity
η_e	= electric motor efficiency
η_m	= mechanical efficiency of compressor
η_p	= polytropic efficiency of compressor
η_v	= volumetric efficiency of compressor
μ	= absolute viscosity
ρ	= density
τ	= skin friction factor
Φ	= correction factor for two-phase pressure drop
ϕ	= fin efficiency

Subscripts:

a	= air, a,d refers to dry air
e	= exit
f	= frost or fin
i	= inlet
L	= liquid
m	= mean value
P	= constant pressure process
r	= refrigerant
s	= constant entropy process
t	= total value
V	= vapor
v	= constant volume process
w	= water
1 to 13	= refrigerant key locations in a heat pump, as per figure 4, unless otherwise explained in the text

1. INTRODUCTION

Among equipment providing thermal comfort for indoor spaces, the heat pump has gained in recent years substantial popularity for residential applications. The heat pump is unique since it is the only device, which can provide both heating and cooling. Heat pumps have become very competitive economically and are considered to be a very good investment by homeowners.

A heat pump works on the vapor compression cycle principle. The most important heat pump components are two heat exchangers, a compressor and an expansion device. The heat pump thermodynamic cycle can be explained by analysing the processes that the refrigerant undergoes in these four components. The most convenient diagram for such an explanation and performance analysis is that of a pressure vs. enthalpy coordinate system, as shown in figure 1. The compressor receives low pressure and temperature refrigerant at state 1 and compresses it to a high pressure. This compression process is associated with an increase of refrigerant temperature. At state 2, the high pressure and high temperature vapor enters the condenser. The refrigerant passing through the condenser rejects heat to the high temperature reservoir and changes, usually to a subcooled liquid at state 3. Then, the refrigerant flows through the expansion device undergoing a drop in pressure and temperature. Finally, the low pressure, low temperature, and low quality refrigerant at state 4 enters the evaporator, where it picks up heat from the low temperature reservoir, reaching a superheated (or high quality) vapor state 1 at the evaporator exit. In the explanation above, the low and high temperature reservoirs are the indoor and the outdoor environment, when the heat pump is operating in the cooling mode.

Besides the compressor, two heat exchangers, and the expansion device, there are, for practical reasons, many other components in an actual heat pump system. For modeling consideration, the most important are: tubes connecting basic elements, 4-way valve enabling refrigerant flow reversal in the unit to operate in the heating or cooling mode, and an accumulator (not used in all systems), which acts as a protective device for the compressor by storing excess refrigerant during part load operation and preventing liquid refrigerant from entering the compressor.

There are many concepts and designs of heat pump components, however, certain types are predominant. A reciprocating hermetic compressor is usually used for vapor compression in small systems. Heat exchangers are usually in the form of staggered tubes with closely packed wavy fins (spine-fins or bristle-fins are sometimes used). There are basically two types of expansion devices in application today. Constant flow area restrictors prevail in small capacity units such as household refrigerators, window-type air conditioners, and central residential air conditioners and heat pumps. Larger, more expensive residential units and commercial units are usually equipped with a variable flow area device called a thermostatic expansion valve (TXV).

All the above mentioned components make up an actual vapor compression system. Configuration of such a system and the thermodynamic cycle are illustrated in figure 2 and figure 3, respectively. Refrigerant states in figure 3 correspond to particular locations in the system marked in figure 2, which are:

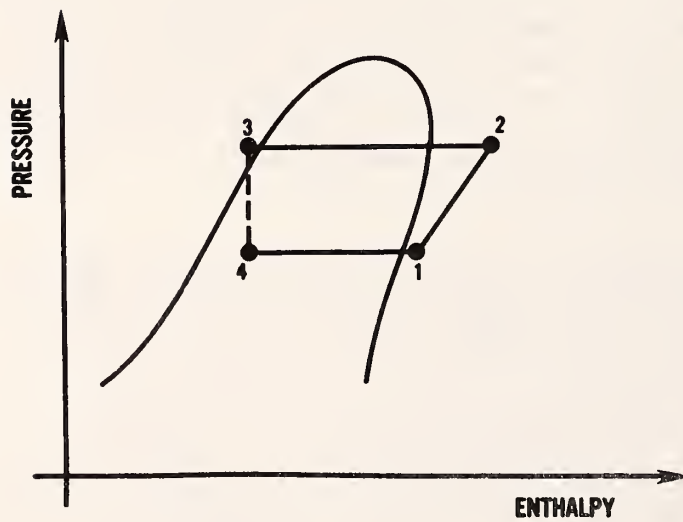
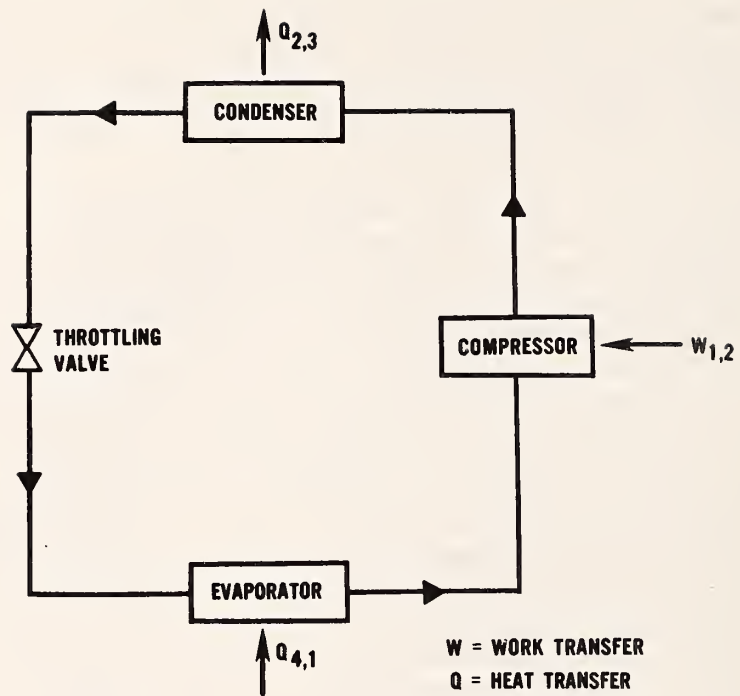
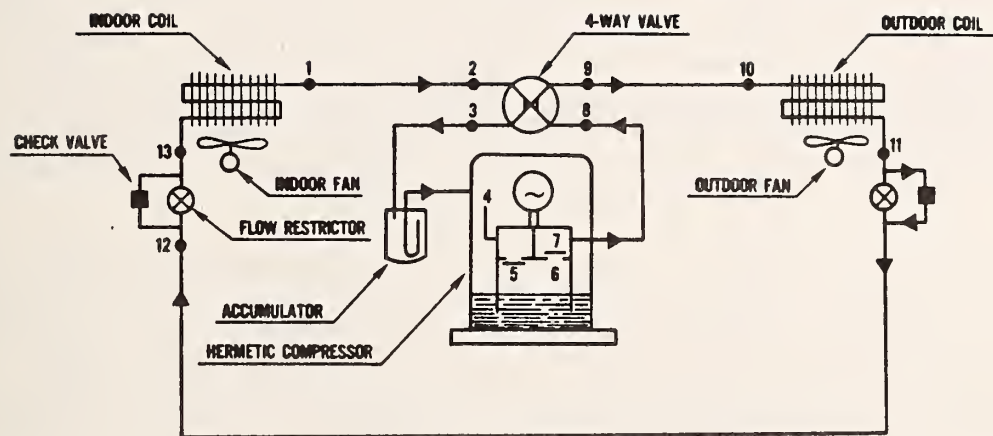


Figure 1. Oversimplified thermodynamic cycle of a heat pump.



Note: 1. Refrigerant flow direction is marked for cooling operation.

2. Numbers 5 and 6 situated in the compressor correspond to refrigerant state before and after the compression process.

Figure 2. Schematic of a heat pump.

- 1 - evaporator exit, suction pipe inlet
- 2 - suction pipe outlet, low pressure 4-way valve inlet
- 3 - low pressure 4-way valve exit, compressor enclosure inlet
- 4 - inside compressor enclosure
- 5 - compressor cylinder at suction
- 6 - compressor cylinder at discharge
- 7 - discharge manifold
- 8 - compressor enclosure exit, high pressure 4-way valve inlet
- 9 - high pressure 4-way valve exit, discharge pipe inlet
- 10 - discharge pipe outlet, condenser inlet
- 11 - condenser outlet, liquid line inlet
- 12 - liquid line outlet, expansion device inlet
- 13 - evaporator inlet

In the heat pumping process refrigerant is the carrier of heat from a low temperature reservoir to a high temperature reservoir. There are many kinds of refrigerants of different properties available and they are being used depending on the application. For residential air-to-air heat pumps, refrigerant 22 has been generally accepted as the most advantageous. However, in recent years interest has intensified in non-azeotropic binary mixtures as potential working fluids in heat pumps. Gliding evaporation/condensation temperatures and the possibility of heat pump capacity control via mixture composition shift are often cited as factors giving mixtures a theoretical performance edge over single component refrigerants.

The objective of this study is the development of a mathematical model of an air-to-air heat pump working with a non-azeotropic mixture as the refrigerant in order to be able to investigate performance potentials of mixtures. The type of heat pump considered here is the one most commonly commercially available with a reciprocating hermetic compressor, fin-tube heat exchangers and a constant flow area expansion device. The mixture considered in this study consists of refrigerants 13B1 and 152a.

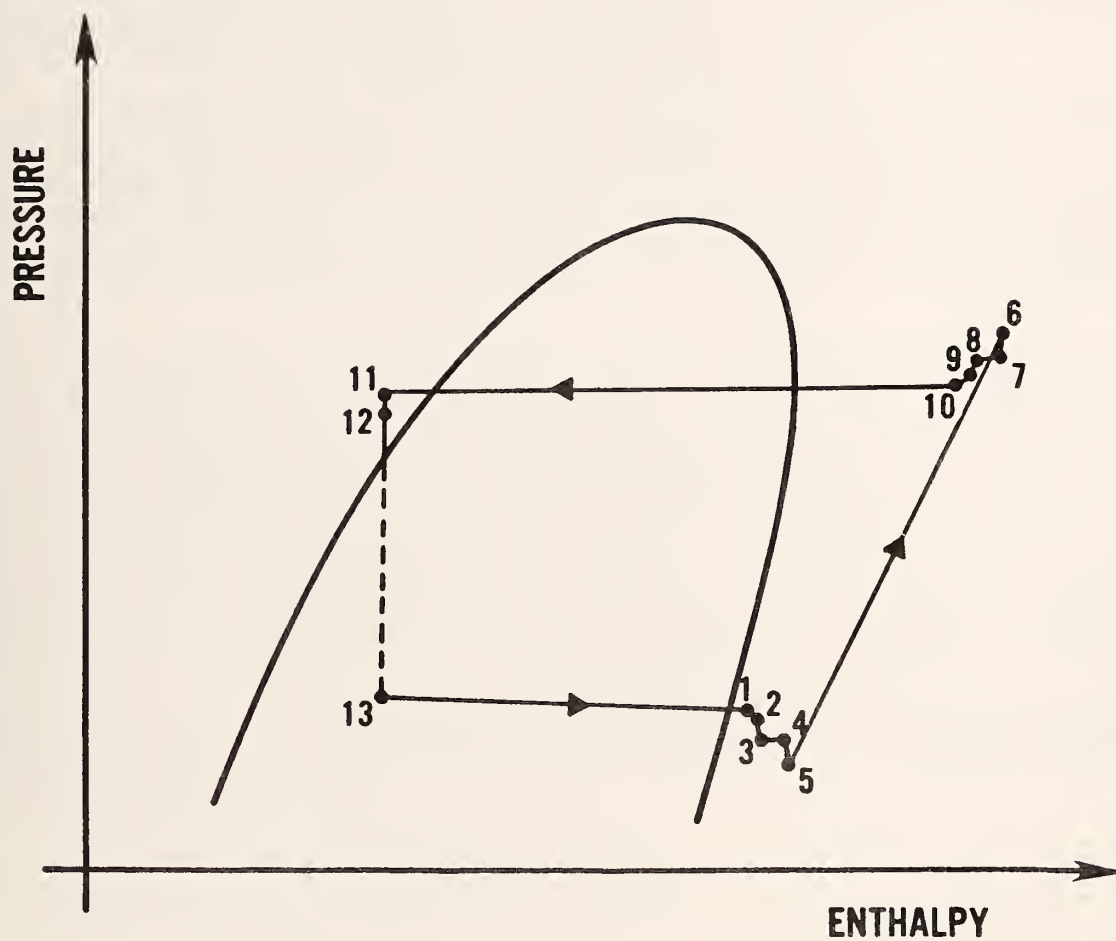


Figure 3. Thermodynamic cycle realized by a heat pump (for number location refer to figure 2).

2. GUIDELINES FOR MODEL DEVELOPMENT

Before beginning the modeling effort, general guidelines for model development have to be established. These guidelines should evolve from future model applications as they are broadly perceived today. It is understood that the model is planned to be used for parametric studies by different users, applied to different hardware and possibly charged with different non-azeotropic refrigerants. The model should help prospective users to evaluate potential benefits of non-azeotropic mixtures, as applied to heat pumps, resulting from changing refrigerant temperature during evaporation/condensation and change of mixture composition which influences heat pump capacity.

It has been stated already that a heat pump equipped with a reciprocating compressor, compact staggered tube and flat fin heat exchangers, and a constant flow area expansion device will be modeled. This choice has been made since, in addition to the fact that components of this design type are most commonly encountered, results of initial research indicated a possibility of inherent to the system capacity modulation in systems employing a capillary tube [1].

The model should be a first principle one. Regression analysis curve fits should be avoided so the program could be applied to simulate systems of different characteristics, size and performance level. Modular structure of the program should allow for later model upgrades on the local basis. Since heat exchangers and changing refrigerant mixture temperatures during change of phase are going to be investigated, a tube-by-tube approach to modeling of an evaporator and condenser was adapted. Finally, an appropriate equation of state is desirable that could be universally used for accurate predictions of thermodynamic properties for a variety of possible non-azeotropic mixtures.

During the past, a heat pump modeling effort has been underway at NBS [2,3]. The latest model [3] fulfills the basic postulates of the guidelines and is adopted here for expansion to the simulation model of a heat pump charged with a non-azeotropic mixture.

3. EVALUATION OF THERMODYNAMIC AND TRANSPORT PROPERTIES OF A NONAZEOTROPIC MIXTURE, R13B1/R152a

The refrigerant mixture, R13B1/R152a, discussed in this chapter showed promising performance in experiments performed by Cooper [1]. The choice of this mixture for development of this heat pump model does not mean this mixture to be optimum and precludes other mixtures to be more suitable for heat pump application.

3.1. Evaluation of Thermodynamic Properties

Evaluation of thermodynamic properties for a non-azeotropic mixture represents an additional complication as compared to single component refrigerants. The employed equation of state has to be able to provide predictions for a broad range of compositions that the circulating mixture may have during heat pump operation. This range of possible compositions is even further expanded by the fact that in the two-phase region, saturated liquid and saturated vapor in equilibrium have different compositions.

Connon and Drew [4] presented application of Redlich-Kwong-Soave (R-K-S) equation to generate R13B1/R152a mixture thermodynamic data. The R-K-S equation successfully generated all vapor thermodynamic information, however, prediction of liquid density of the mixture was based on densities of pure components and application of a mixing rule, since the R-K-S equation did not provide the desired accuracy.

It has been shown by Morrison et al. [5] that it is possible to describe both the liquid and the vapor properties of pure-refrigerants as well as refrigerant mixtures with a single equation of state. They have shown that this approach has a significant advantage over the traditional methods, that of using a vapor equation of state and a library of liquid properties. This advantage is that it will accurately predict property values in the region near or above the critical point of either of the binary components [6]. Because of this advantage, Morrison's equation was adopted in this modeling effort. The basic features of this equation are given below, though for full review, the source report is recommended [5]. The equation of state has the following form:

$$\frac{Pv}{RT} = \frac{1 + y + y^2 - y^3}{(1 - y)^3} - \frac{a}{RT(v + b)} \quad (1)$$

where $y = \frac{b}{4v}$

For single component refrigerants, parameters a and b can be determined by second degree polynomials:

$$a = a_0 + a_1T + a_2T^2 \quad (2)$$

$$b = b_0 + b_1T + b_2T^2 \quad (3)$$

are a_0 , a_1 , a_2 , b_0 , b_1 , and b_2 are constants which are based on empirical data for a given refrigerant. For mixtures, these parameters may be determined by:

$$a = w_I^2 a_I + 2w_I w_{II} a_{I,II} + w_{II}^2 a_{II} \quad (4)$$

$$b = w_I^2 b_I + 2w_I w_{II} b_{I,II} + w_{II}^2 b_{II} \quad (5)$$

Parameters a_I , a_{II} , b_I , and b_{II} are for pure refrigerants and are obtained by equations (2) and (3).

Remaining parameters are:

$$a_{I,II} = (1 - f_{I,II}) (a_I a_{II})^{0.5} \quad (6a)$$

$$b_{I,II} = [(b_I^{1/3} + b_{II}^{1/3})/2]^3 \quad (6b)$$

$$f_{I,II} = d - cT \quad (6c)$$

(c and d constants are obtained from mixture measurements to allow for the interactions of the different molecular species.)

w_I, w_{II} = molar composition of the mixture, fraction of I and II components, respectively.

Numerical values of the above explained constants are given in Table 1 for refrigerants R13B1 and R152a.

Having defined the mixture thermodynamic state, by specifying temperature, pressure and mixture composition, in order to evaluate other state thermodynamic properties it is essential to determine if the mixture is liquid, vapor, or two-phase. In the latter case, evaluation of the compositions of liquid and vapor in equilibrium is also required.

The starting point to this calculation procedure is pure component analysis to determine saturation pressure for a given temperature. Since equation (1) also contains molar volume, an iterative process is mandatory for both the liquid and vapor phases. The criteria for verifying that the saturation pressure and molar volume guesses are correct is the equality of the Gibbs free energy and pressure of the two phases. The logic of this procedure is illustrated in the flow diagram in figure 4, and Gibbs free energy is calculated by the equation:

$$G = G^{Pg}(p^*, T) + RT \ln \frac{RT}{p^* v} + (pv - RT) - \frac{a}{b} \ln \frac{v+b}{v} + \frac{4RTy}{v-y} + \frac{RTy}{(v-y)^2} \quad (7)$$

where G^{Pg} = perfect gas Gibbs free energy
 p^* = equilibrium vapor pressure of the pure liquid at temperature T

Table 1. Equation of State Constants for Refrigerants R13B1 and R152a

	R13B1	R152a
a_0	25.4145	27.39273
a_1	-0.063368	-0.059421169
a_2	4.140051 E-5	3.3176956 E-5
b_0	0.1353977	0.1239878
b_1	-1.50409 E-4	-1.445514 E-4
b_2	-1.354434 E-7	-1.9022381 E-8
c	-2.241 E-4	
d	0.1466	

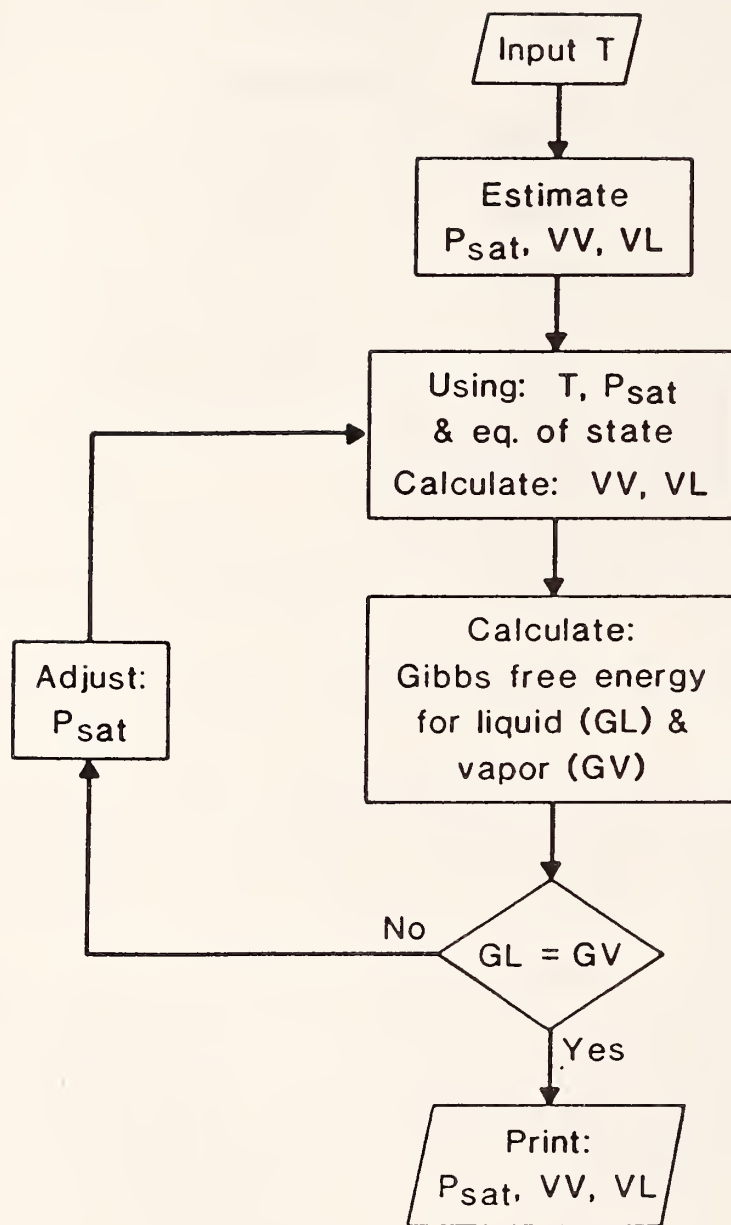


Figure 4. Flow chart for calculating the saturation pressure of single component refrigerant

The evaluation of compositions of the liquid and vapor phases in equilibrium follows the rule that phase equilibrium in mixtures is defined by equality of pressure, temperature and the chemical potential between phases of each of the species in the mixture. Determination of the equilibrium composition is a rather lengthy procedure and it is not practical to use a full algorithm in a large vapor compression cycle simulation program where property values are requested on the order of hundred of thousands of times. For this reason, equation of state was used to generate sufficient data to create a polynomial spline where saturated liquid and saturated vapor compositions are functions of temperature, pressure and saturation pressures of pure components at mixture temperature. Generation of polynomial splines is discussed in [7] and [8]. The following equations were obtained for R13B1/R152a mixture:

$$RP = \frac{P1 - P}{P1 - P2}$$

$$S1 = 12.58741 - 0.06465226 * TK - 9.56391 * 10^{-5} * TK^2$$

$$S2 = -48.9799 - 0.289414 * TK - 4.504945 * 10^{-4} * TK^2$$

$$S3 = 47.7509 - 0.273993 * TK + 4.14487 * 10^{-4} * TK^2$$

$$S = S1 + S2 * RP + S3 * RP^2$$

$$Z1 = -0.34065414 - 3.337532 * 10^{-3} * TK + 9.66115 * 10^{-6} * TK^2$$

$$Z2 = -10.3139754 + 0.0653856 * TK - 9.9899162 * 10^{-5} * TK^2$$

$$Z3 = 10.035917 - 0.05527259 * TK + 7.6821554 * 10^{-5} * TK^2$$

$$Z = 1 + Z1 * (RP - 1) + Z2 * (RP^2 - 1) + Z3 * (RP^3 - 1)$$

$$XL = \frac{(1 + S) * RP}{(1 + S * RP)} \quad (8)$$

$$XV = XL * Z \quad (9)$$

where TK = mixture temperature (K)
 P = mixture pressure (std atm)
 P1 = saturation pressure of pure R13B1 at temperature TK (std atm)
 P2 = saturation pressure of pure R152a at temperature TK (std atm)
 XL = molar composition of liquid phase (fraction of R152a)
 XV = molar composition of vapor phase (fraction R152a)

Knowing saturated liquid and vapor compositions and overall mixture composition, evaluation of mixture quality, XQ, is readily available:

$$XQ = \frac{XL - XM}{XL - XV} \quad (10)$$

where XM = mixture molar composition (fraction of R152a)

Phase equilibrium compositions calculated by the spline agree to the third decimal point with compositions obtained using the equation of state.

Equation of state does not provide us directly with information about the bubble point or dew point. However, the bubble point for a mixture of given composition and pressure can be found by iteration using the criteria that, at bubble point the liquid composition (XL) is equal to a known mixture composition (XM). The solution logic is illustrated in the flow diagram of Figure 5, where the parameters are graphed in the phase diagram of Figure 6. Similar logic is used for calculating the dew point temperature; in this case, searching for the unique temperature at which the saturated vapor composition equals to composition of the mixture.

Evaluation of molar enthalpy is straight forward:

$$h = h^{pg} + \frac{a'bTK - ab'TK - ab}{b^2} \ln \frac{v+b}{v} + \frac{ab'T - ab}{b(v+b)} + \frac{8RTv(8v-b)}{(4v-b)^3} (b - b'T) \quad (11)$$

where prime denotes a temperature derivative.

The molar enthalpy of a perfect gas of a mixture, h^{pg} , is the integral of a linear weighting of the heat capacities of the component perfect gases.

$$h^{pg} = \int_{T_{ref}}^T C_p dt = \int_{-40F}^T (w_I C_{pI} + w_{II} C_{pII}) dT \quad (12)$$

where C_{pI} and C_{pII} are perfect gas heat capacities of the respective components, functions of temperature, whose coefficients are determined empirically.

$$C_{pI} = C_I(1) + C_I(2) \cdot T + C_I(3) T^2 \quad (13)$$

$$C_{pII} = C_{II}(1) + C_{II}(2) \cdot T + C_{II}(3) \cdot T^2 \quad (14)$$

Specific entropy can be calculated by the following expression:

$$S_m = w_I \int_{-40}^T \frac{C_{vI}}{T} dT + R \ln \frac{v}{w_I \cdot v_I^*} - \Delta S_I^{pg}(v_I^*, -40^\circ C) + w_{II} \int_{40}^T \frac{C_{vII}}{T} dT + R \ln \frac{v}{w_{II} \cdot v_{II}^*} - \Delta S_{II}^{pg}(v_{II}^*, -40^\circ C) + \Delta S_m^{pg}(v, T) \quad (15)$$

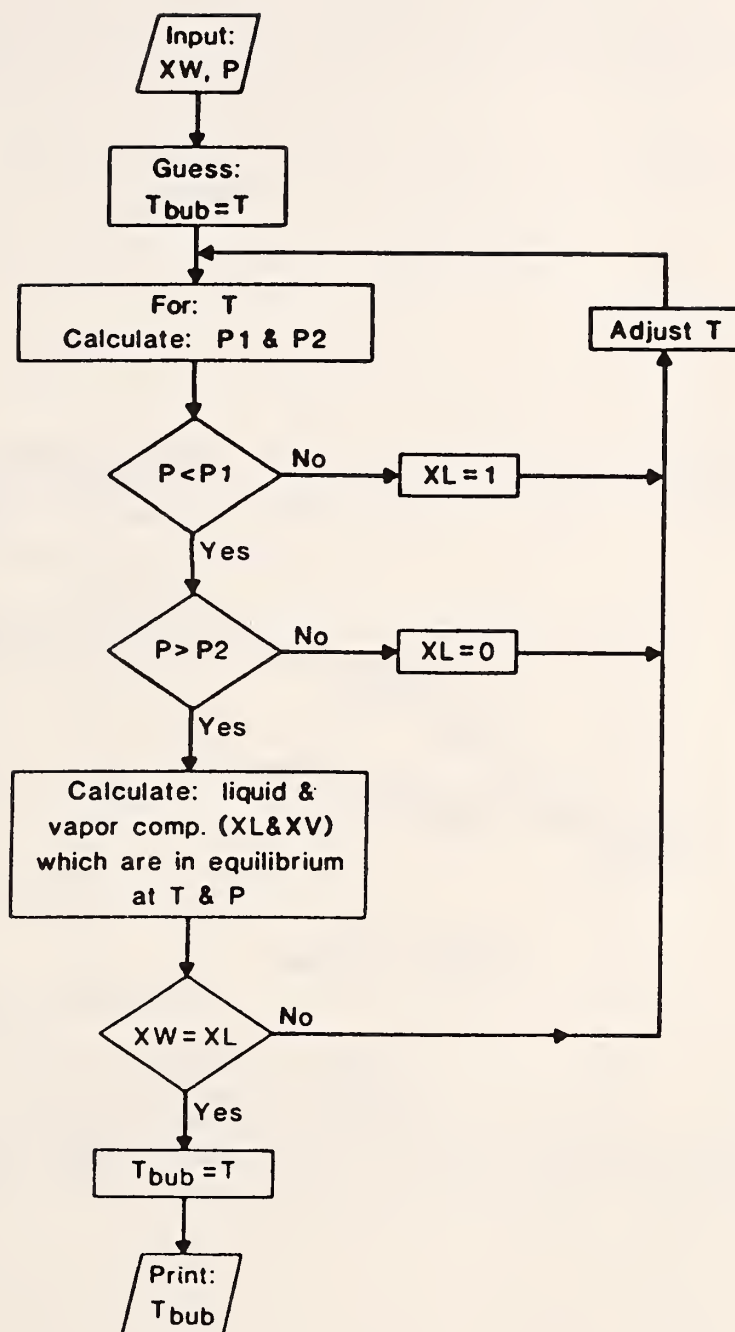


Figure 5. Flow chart for calculating the bubble point temperature of a non-azeotropic binary mixture.

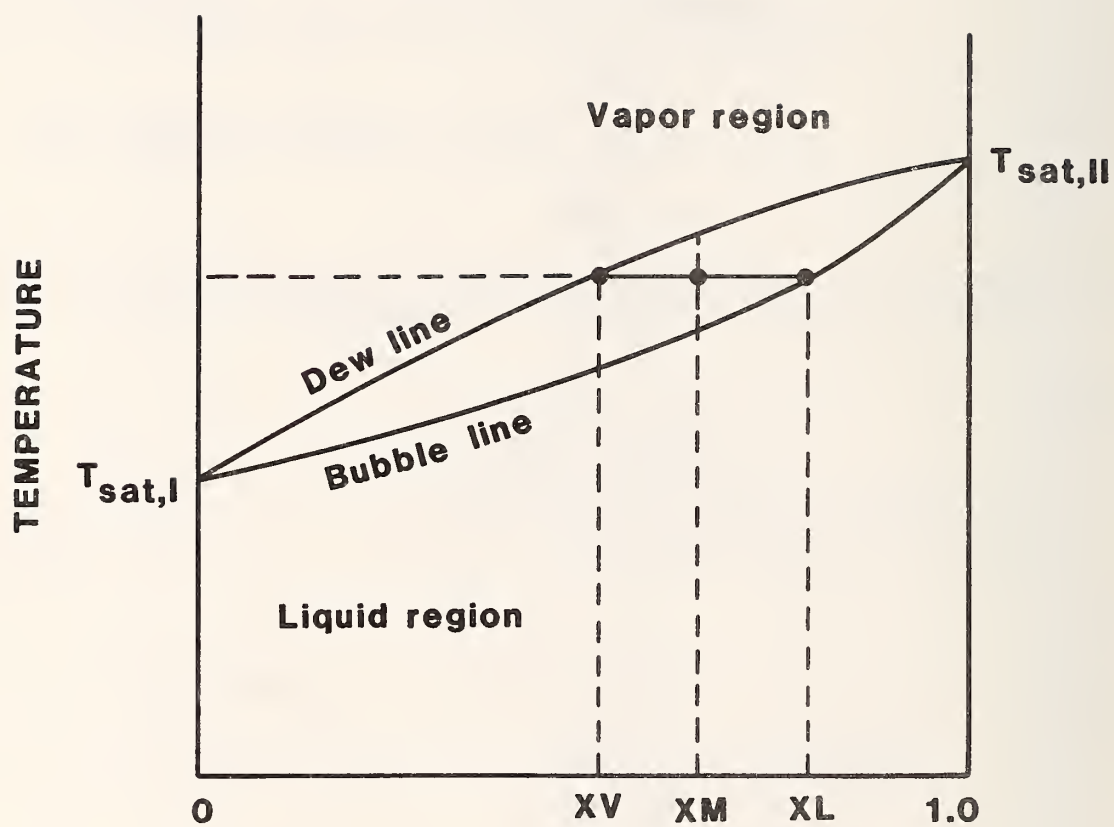


Figure 6. Temperature-composition diagram for a non-azeotropic binary mixture

where:

v_I^* and v_{II}^* are reference state (-40°C) molar volumes of liquid for each of the pure components, ΔS_I^{pg} and ΔS_{II}^{pg} are the differences of entropy between the perfect and the real gas having a molar volume of saturated pure component liquid I and II, respectively, at the reference temperature (-40°C), ΔS_m^{pg} is the difference of entropy between perfect and real mixture of the actual molar volume v and at the actual mixture temperature T .

In either case, ΔS^{pg} can be evaluated by the equation (16) with v put to v_I^* or v_{II}^* , if pure components are considered.

$$\Delta S^{pg} = \frac{a'b - ab'}{b'} \ln \frac{v + b}{v} + \frac{ab'}{b(v + b)} - \frac{Rb(16v - 3b)}{(4v - b)^2} - \frac{8RTbv(8v - b)}{(4v - b)^3} \quad (16)$$

Application of the equation of state requires estimated values of the following pure component properties at saturation as function of temperature; pressure, molar volume of vapor, and molar volume of liquid. Correlations to provide these guesses are given below:

$$P = \exp \sum_{I=1}^3 a(I) \cdot T^{1-I} \quad (17)$$

$$v_V = \sum_{I=1}^3 b(I) \cdot T^I / P \quad (18)$$

$$v_L = \sum_{I=1}^3 c(I) \cdot T^{I-1} \quad (19)$$

where P = saturation pressure
 T = temperature
 v_V = molar volume of saturated vapor
 v_L = molar volume of saturated liquid

The equation coefficients $a(I)$, $b(I)$, $c(I)$ for refrigerants 13B1 and 152a are given in Table 2 along with applicable units of refrigerant state property parameters.

Table 2. Coefficients to be Used in Equations (17), (18) and (19)

Coefficient	R13B1	R152a
a(1)	1.00522804E+1	1.06410518E+1
a(2)	-2.2045632E+3	-2.6428994E+3
a(3)	9.6365313E+3	4.6087585E+2
b(1)	-5.05060051E-2	2.18192958E-2
b(2)	1.11455764E-3	5.416820778E-4
b(3)	-2.56392871E-6	-1.24731336E-6
c(1)	2.749422E-1	1.023688715E-1
c(2)	1.702569E-3	-4.0752759E-4
c(3)	3.71008313E-6	1.0409447E-6

Applicable units of the state parameter in equations (17), (18) and (19) are

pressure - std. atm

temperature - $^{\circ}\text{K}$

molar volume - L/mol

3.2 Evaluation of Transport Properties

In addition to the thermodynamic state equation, algorithms have to be used to calculate thermal conductivity and absolute viscosity of the liquid and vapor. These algorithms consist of curve-fitted correlations for the prediction of properties of pure components, and of some kind of a mixing rule. Reid, Prausnitz and Sherwood [9] gave a comprehensive review of mixing rules that could be used depending on mixture component characteristics and component property data availability. Following their recommendation, appropriate correlations were selected and are listed below along with the correlations used for evaluation of the transport properties of pure components. Correlations for property calculations are given here as presented in the referenced documents. However, for use in the mixing rule they were converted to common dimensions.

Liquid Thermal Conductivity, k

(The Filippov Equation is used. The error should not exceed 4% [9].)

$$k = k_1 + (k_2 - k_1) \cdot XW \cdot (0.72 \cdot XW + 0.28) \quad (20)$$

where: k_1, k_2 = thermal conductivity of the pure components of R13B1 and R152a, respectively.

$$k_1 = 0.35 - 1.5 \cdot 10^{-4} \cdot TF \quad (\text{Btu/h ft F}) [8]$$

$$k_2 = 0.11650 - 0.000497 \cdot TC \quad (\text{w/m} \cdot \text{K}) [8]$$

$$TC = \text{temperature} \quad (\text{C})$$

$$TF = \text{temperature} \quad (\text{F})$$

$$XW = \text{weight composition} \quad (\text{fraction of R152a})$$

Vapor Thermal Conductivity, k

(The Wassiljewa Equation is used with the Lindsay and Bromley Modification. The error rarely exceeds 5% [9].)

$$k = \frac{(1 - XM) \cdot k_1}{(1 - XM) \cdot A_{11} + XM \cdot A_{12}} + \frac{XM \cdot k_2}{(1 - XM) \cdot A_{21} + XM \cdot A_{22}} \quad (21)$$

where: k_1, k_2 = thermal conductivity of the pure components of R13B1 and R152a, respectively.

$$k_1 = 8.2982 \cdot 10^{-3} - 5.1971 \cdot 10^{-5} \cdot TK + 1.8413 \cdot 10^{-7} \cdot TK^2 \quad (\text{w/m} \cdot \text{K}) [8]$$

$$k_2 = -8.357 \cdot 10^{-3} + 6.32 \cdot 10^{-5} \cdot TK + 4.257 \cdot 10^{-8} \cdot TK^2 \quad (\text{w/m} \cdot \text{K}) [9]$$

$$A_{ij} = 0.25 \left[1 + \left(\frac{\mu_i}{\mu_j} \left(\frac{M_j}{M_i} \right)^{0.75} \frac{TK + S_i}{TK + S_j} \right)^{0.5} \right]^2 \frac{TK + S_{ij}}{TK + S_i}$$

μ_1, μ_2 = absolute viscosity of the pure components, R13B1 and R152a, respectively.

M_1, M_2 = molecular weight of the pure components, R13B1 and R152a, respectively.

TK = absolute temperature (K)

$$\begin{aligned}
S_i &= 1.5 \cdot T_{bi} \\
T_{bi} &= \text{normal boiling point of 'i' component (K)} \\
S_{ij} &= 0.73 \cdot (S_i \cdot S_j)^{0.5} \\
XM &= \text{composition (mole fraction of R152a)}
\end{aligned}$$

Liquid Absolute Viscosity, μ

(The Lobe correlation is used. The prediction error should be less than 15%, [9].)

$$\mu = \frac{1}{v_m} (\phi_1 \cdot \gamma_1 e^{\phi_2 \alpha_2^*} + \phi_2 \gamma_2 e^{\phi_1 \alpha_1^*}) \quad (22)$$

v_m = mixture specific volume

ϕ_1, ϕ_2 = volume fraction of the pure components, R13B1 and R152a, respectively.

γ_1, γ_2 = kinematic viscosity of the pure components, R13B1 and R152a, respectively. (obtained, as shown below, through evaluation of the absolute viscosities, μ_1 and μ_2)

$$\begin{aligned}
\mu_1 &= (e^{-4.22529 + 710.843/TK}) \cdot 10^{-3} \quad (\text{N} \cdot \text{s/m}^2) \quad [10] \\
\mu_2 &= (e^{-4.28224 + 753.013/TK}) \cdot 10^{-3} \quad (\text{N} \cdot \text{s/m}^2) \quad [10]
\end{aligned}$$

$$\alpha_1^* = 1.7 \ln \frac{\gamma_2}{\gamma_1}$$

$$\alpha_2^* = 0.27 \ln \frac{\gamma_2}{\gamma_1} + (1.3 \cdot \ln \frac{\gamma_2}{\gamma_1})^{0.5}$$

TK = temperature (K)

Vapor Absolute Viscosity, μ

(The equation derived from the vigorous kinetic theory of Chapman-Enskog is used. The error seldom exceeds 4% [9].)

$$\mu = \frac{(1 - XM)\mu_1}{(1 - XM) + XM \cdot \phi_{12}} + \frac{XM\mu_2}{XM + (1 - XM)\phi_{21}} \quad (23)$$

where

$$\phi_{12} = \frac{\left[1 + \left(\frac{\mu_1}{\mu_2} \right)^{0.5} \left(\frac{M_2}{M_1} \right)^{0.25} \right]^2}{\left[8 \left[1 + \frac{M_1}{M_2} \right] \right]^{0.5}}$$

$$\phi_{21} = \phi_{12} \cdot \frac{\mu_2}{\mu_1} \frac{M_1}{M_2}$$

X_M = composition (mole fraction of R152a)

μ_1, μ_2 = absolute viscosity of the pure components, R13B1 and R152a, respectively.

$$\mu_1 = 0.67329 \cdot 10^{-3} + 7.60593 \cdot 10^{-6} \cdot TK - 2.81108 \cdot 10^{-8} TK^2 + 3.47410 \cdot 10^{-11} TK^3 \quad (N \cdot s/m^2) \quad [9]$$

$$\mu_2 = 3.205 \cdot 10^{-5} \cdot \frac{k_2 M_2}{FE} \quad (N \cdot s/m^2) \quad [11]$$

k_2 = thermal conductivity of pure R152a vapor (w/m · K)

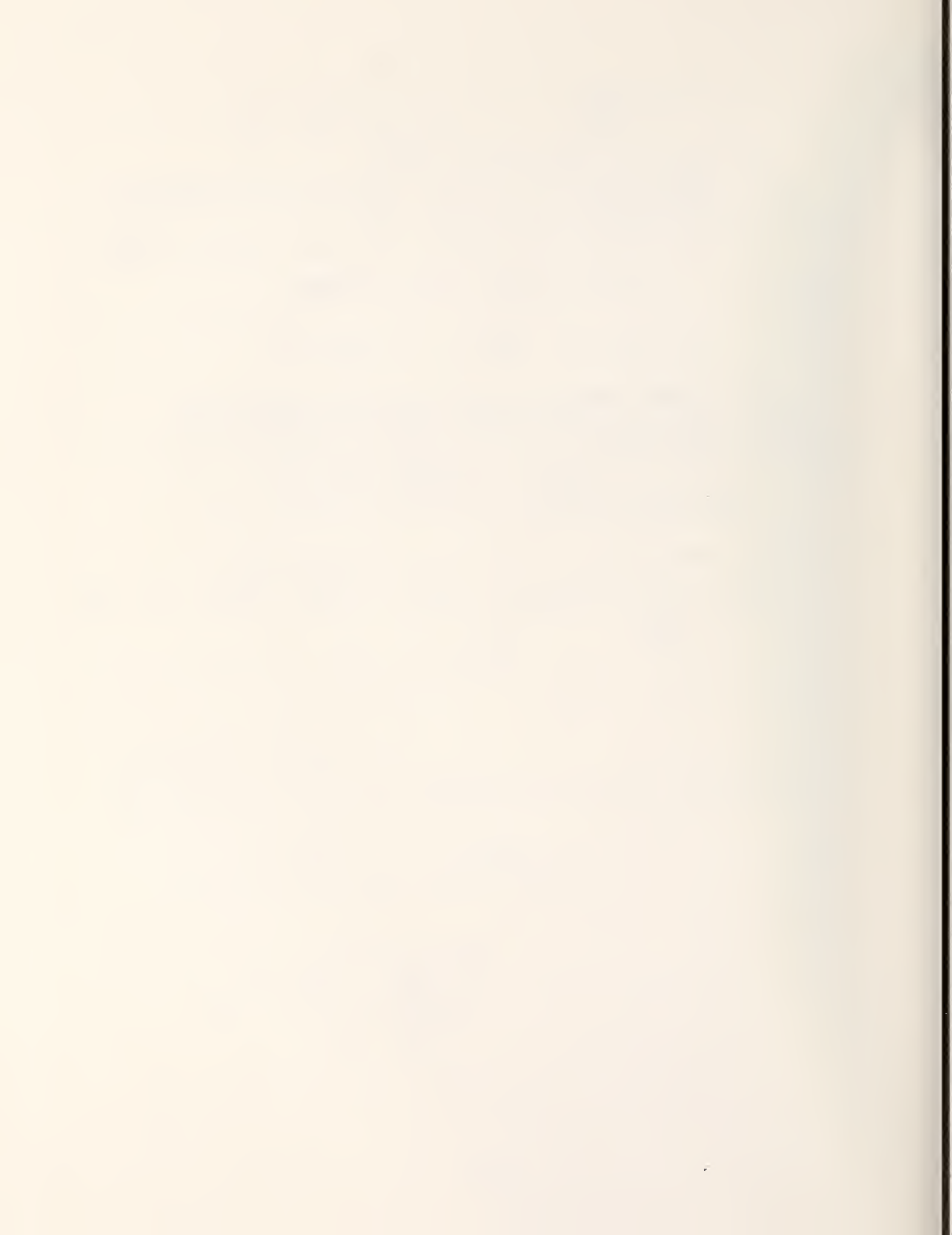
M_1, M_2 = molecular weight of the pure components, R13B1 and R152a, respectively.

$$FE = 0.115 + 0.354 \frac{C_{pII}}{R}$$

C_{pII} = specific heat at constant pressure of R152a vapor.

$$C_{pII} = -7.33704 + 0.093438 TK - 3.61094 \cdot 10^{-4} TK^2 + 4.80449 \cdot 10^{-7} \cdot TK^3$$

$$\left(\frac{kJ}{kg \cdot K} \right)$$



4. OPERATION OF A HEAT PUMP WITH A CONSTANT FLOW AREA EXPANSION DEVICE

In a heat pump with a capillary tube, vapor superheat at the compressor inlet will vary with changing operating conditions. As a matter of fact, no single refrigerant parameter stays constant when indoor or outdoor air conditions are altered and the system seeks equilibrium under new conditions. To understand the phenomena that cause refrigerant state changes, consider a heat pump working in the cooling mode at constant indoor air conditions subjected to a step increase of the outdoor temperature. For the purpose of the analysis, it is assumed that the overall heat transfer coefficient for both heat exchangers does not vary. It is also assumed, as a first approximation, that the latent heat of evaporation, i_{fg} , is constant under the conditions discussed.

A step increase in the outdoor air temperature first affects the performance of the condenser. Since the condenser environmental temperature increases, the mean temperature difference between the air and the refrigerant decreases. Less heat is rejected by the refrigerant to the air and a smaller enthalpy change is realized in the coil. Consecutively, higher enthalpy, less subcooled liquid refrigerant enters the capillary tube. The capillary tube is sensitive to the amount of subcooling and with less subcooling the mass flow rate through the capillary tube decreases. Since the compressor capacity stays unchanged, pressure builds up in the condenser. Thus point 11 on the $p - h$ diagram (figure 3) moves in the direction of higher enthalpy and higher pressure.

The environment of the evaporator was not altered. However, refrigerant parameters in the evaporator change in response to the change of refrigerant state at location 11 as well as to the change in refrigerant mass flow rate. For pressure drop considerations, as a first approximation, the liquid line, the capillary tube, and the evaporator are viewed as one tube, in which a given flow experiences a certain pressure drop. As pressure of the refrigerant at location 11 increases, it pulls up the refrigerant pressure in the evaporator. Conflicting phenomena affect the change of refrigerant enthalpy at point 1. Increased enthalpy at point 11 and reduced refrigerant mass flow rate work against the impact of the smaller temperature difference between the air and the refrigerant causing refrigerant enthalpy to slightly decrease. The significance of the change of the refrigerant state at location 1 is a move towards smaller vapor specific volume. Since the compressor, as a first approximation, is a constant intake volume pump, increase in gas density results in a higher mass flow rate.

The indicated phenomena will balance themselves further until the refrigerant in the key locations of the system acquires the thermodynamic states that satisfy simultaneously the equilibrium of all the heat pump components. Since under steady-state operation, the refrigerant mass flow rate through the compressor and the capillary tube have to match, the pressure in the condenser will rise to an appropriate level. Again, higher pressure in the condenser implies some increase of pressure (and saturation temperature) in the evaporator. The smaller temperature difference between the indoor air and the refrigerant, and higher refrigerant mass flow rate result in a smaller refrigerant enthalpy increase in the evaporator. The refrigerant state at point 1 with respect to point 11 is determined by this enthalpy change and the appropriate pressure drop. Finally, when steady-state conditions are reached the following changes in refrigerant thermodynamic states can be noted:

higher refrigerant pressure in the condenser, higher refrigerant pressure in the evaporator, less superheat at the evaporator exit, and less refrigerant liquid subcooling at the condenser exit. The change of saturation temperature in the condenser corresponds approximately to the change in the outdoor air temperature. The change of pressure in the evaporator is a fraction (approximately 15 percent) of the pressure change in the condenser. Though the refrigerant is circulated in the system at a higher mass flow rate, the capacity of the heat pump is decreased due to the smaller refrigerant enthalpy change in the evaporator. The efficiency also drops since, in addition to a smaller cooling effect, the energy input to the compression process increases due to the higher refrigerant mass flow rate and the higher compression rate. A decrease of the outdoor air temperature would result in the opposite trends.

The change of refrigerant parameters is illustrated in figure 7, where results of three tests of a heat pump charged with refrigerant 22 are plotted on the pressure-enthalpy diagram (the plot is based on measured condenser and evaporator pressures and temperatures only). In these tests, the indoor air conditions were held constant while the outdoor air dry bulb temperature was changed producing modifications of the thermodynamic cycle.

For the heating mode operation, refrigerant flow is reversed by the action of the four-way valve. The flow direction of the refrigerant in the system is the opposite to that in the cooling mode, with the exception of the compressor. The indoor coil becomes the condenser while the outdoor coil serves as the evaporator. It is worthwhile to note the effect this change has on the system. Since pressures in heat pump heat exchangers are functions of environment temperatures, the pressure in the evaporator is now much lower than during cooling operation. Consequently, the density of the refrigerant at the evaporator exit is smaller and less refrigerant is being pumped by the compressor. Also note that an indoor coil is usually much smaller than an outdoor coil. The need for dehumidification is the primary reason. The indoor coil working as a condenser cannot condense and hold as much refrigerant as the outdoor coil during cooling operation. In heat pumps equipped with a capillary tube the excess of liquid refrigerant is stored in an accumulator.

A schematic of an accumulator is given in figure 8. The accumulator is a closed container with two tubes. The longer bent tube has a small diameter hole on the bottom side, and is connected to the compressor. The other connects the tank with the evaporator exit.

If superheated vapor leaves the evaporator, only vapor is contained in the accumulator and no special function is fulfilled by the accumulator. If wet vapor enters the accumulator, liquid droplets accumulate on the bottom of the tank, while saturated vapor enters the tube leading towards the compressor. Some liquid refrigerant enters the vapor stream through the hole in the tube, driven by the static pressure difference in the liquid-vapor stream interface. It should be noted that during steady-state operation with liquid in the accumulator, qualities of the vapor entering and leaving the accumulator must be equal.

In a heat pump charged with a non-azeotropic mixture, incomplete evaporation in the evaporator and collection of liquid refrigerant in the accumulator affect the actual mixture circulating composition. As shown on the temperature-composition diagram (figure 6) the liquid and vapor phases in

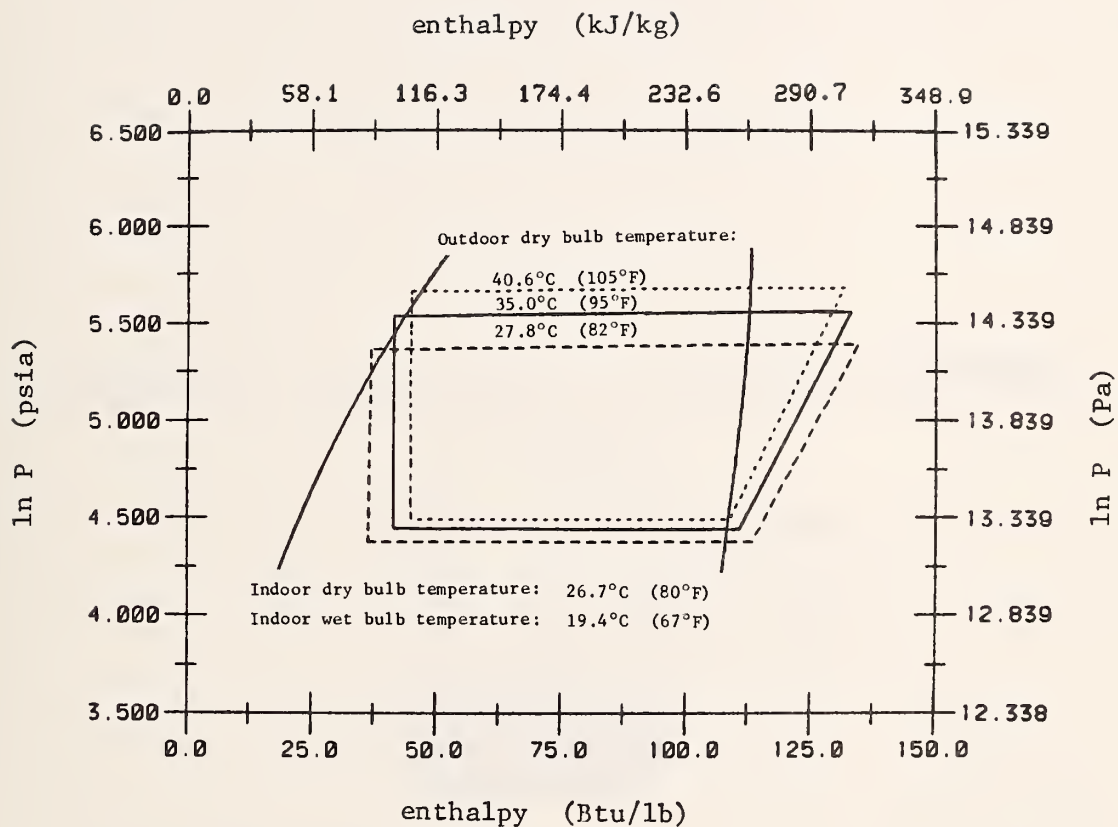


Figure 7. Simplified thermodynamic cycles realized by a heat pump charged with refrigerant 22 in the cooling mode at different outdoor temperatures.

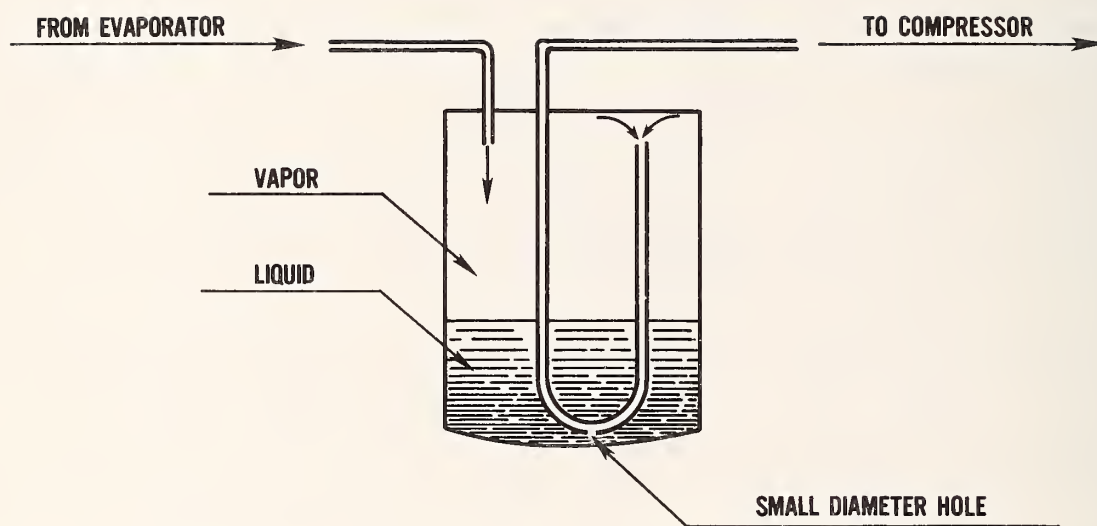


Figure 8. Schematic of an accumulator.

equilibrium at a given temperature and pressure have different compositions, the liquid being richer in less volatile component. Consequently, with the first liquid drop stored in the accumulator, circulating mixture composition becomes richer in the lower boiling point refrigerant. This composition shift increases with the increase of accumulated liquid.

5. MODELING OF A HEAT PUMP

5.1. The Logic of a Model of a Heat Pump Charged with a Non-Azeotropic Refrigerant Mixture

During heat pump operation, refrigerant parameters at any location are established at a level which satisfies the respective components of the system. There is a one-to-one relationship between working medium parameters and the operating conditions, i.e., for given outdoor and indoor air conditions, there is just one set of refrigerant parameters at any location within the system which satisfy steady state operation. This unique set of refrigerant parameters has to be determined by the heat pump simulation program in an iterative process.

In order to set up a heat pump iteration process (Fig. 9), balances taking place during steady state operation have to be recognized. From the fact that a thermodynamic cycle is a closed loop, an analysis of a heat pump cycle on a pressure-enthalpy diagram allows the statement of two balances:

- Enthalpy Balance

The net refrigerant enthalpy change in all components of the system has to equal zero.

- Pressure Balance

This balance implies that the increase of refrigerant pressure during the compression process has to be equal to the total pressure drop during the other processes forming the cycle. As pressure drop and mass flow rate are strictly related, this balance may be restated that pressure drop through each component has to be such that the mass flow rate in each component is the same.

The enthalpy balance and pressure balance can be supplemented by two additional balances resulting from the law of mass conservation, as expressed by the following equations:

$$\frac{D}{Dt} \int_V \rho_m dV = 0 \quad (24)$$

$$\frac{D}{Dt} \int_V \rho_i dV = 0 \quad (25)$$

where ρ_m = mean density of the mixture
 ρ_i = mean density of 'i' compone
 V = system internal volume

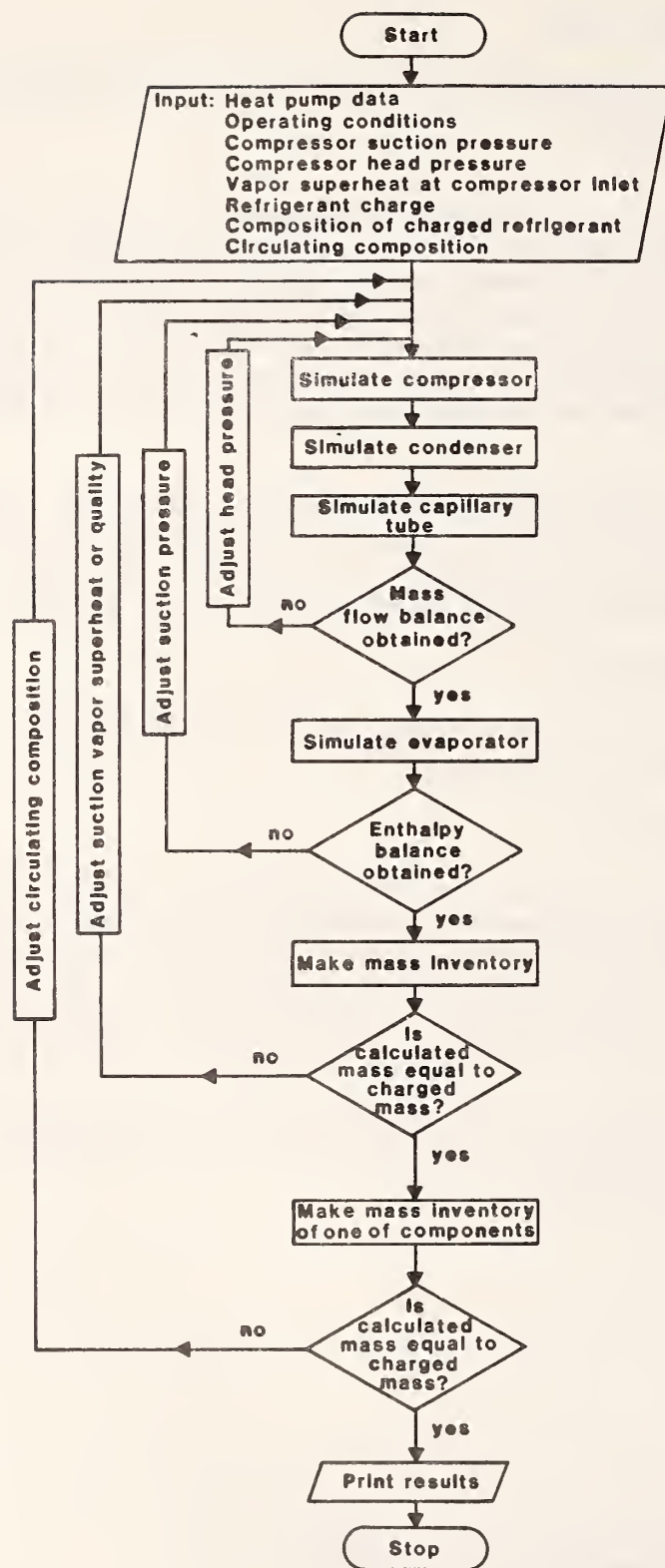


Figure 9. Overall logic of a program simulating performance of a heat pump employing a constant flow area expansion device and charged with a non-azeotropic binary refrigerant mixture.

Equation (24) implies, that the amount of working medium in the system is the same at all times, for any operating conditions. Equation (25) states the same conservation principle for each of the mixture components. Though for a binary mixture the last equation provides us with two balances, we can actually use only one of them with conjunction with equation (24).

It should be noted that the first three balances are sufficient to set up the logic of a simulation program of a heat pump charged with a single component refrigerant. These three balances were utilized in [3] allowing for iteration of the refrigerant thermodynamic states at key system locations without imposed restrictions on the refrigerant state anywhere in the system. In a heat pump charged with a non-azeotropic binary mixture, the circulating composition may change if some liquid refrigerant is collected in the accumulator. The actual composition of the circulating mixture has to be evaluated during simulation. Since this change of composition represents an additional degree of freedom, an additional equation (equation (25)) is used in the solution logic.

The logic of the program is based on these four balances. It provides means for calculation of vapor superheat (quality) at the compressor can inlet and allows for heat pump simulation at the imposed operating conditions. Explanation of the logic, for sake of clarity, is done considering only four main components of a heat pump, i.e., a compressor, a condenser, a capillary tube, and an evaporator.

The required input consists of outdoor and indoor conditions, heat pump data and composition of the charged and circulating mixture. The simulation process begins with estimated refrigerant pressure and vapor superheat at the compressor inlet, and the compressor discharge pressure. Using these data, the compressor performance is simulated yielding the refrigerant mass flow rate. Next, the condenser and the capillary tube are simulated and a mass flow balance is sought by comparing the refrigerant mass flow rates through the compressor and capillary tube. If the compared mass flows are not equal, simulation of the compressor, the condenser, and the capillary tube is redone with an unchanged refrigerant state at the compressor can inlet and a modified estimate of compressor discharge pressure. Increasing this pressure reduces refrigerant mass flow rate through the compressor. Then the condenser works with this smaller mass flow rate and at a higher saturation temperature. Consequently, refrigerant reaching the expansion device inlet is at a higher pressure and has more subcooling. Both factors promote greater mass flow rate through the capillary tube. Thus, increase in the discharge pressure has a clear and opposite impact on mass flow rates through the compressor and through the capillary tube and the appropriate discharge pressure can be found for which mass flow balance exists.

Once mass flow balance is reached, simulation of the evaporator is performed with the known refrigerant state at the evaporator exit and the refrigerant mass flow rate. Since the thermodynamic process in a capillary tube may be viewed as adiabatic, refrigerant enthalpy at the evaporator inlet should be equal to the enthalpy at the condenser outlet. If these enthalpies are not equal (energy balance is not reached), a new calculation starts from the beginning with a modified estimate of the refrigerant pressure at the compressor can inlet. From the condenser operation point of view, a change in compressor suction pressure induces a change in the refrigerant mass flow rate

and a modification of the condenser saturation temperature resulting from the system flow balance search. These two changes have opposite effects on refrigerant enthalpy at the condenser exit, leaving it only slightly altered. On the other hand, the same change of compressor suction pressure has a strong effect on the refrigerant enthalpy at the evaporator inlet due to a change in the evaporator saturation temperature and refrigerant mass flow rate, both working to change the enthalpy in the same direction. Thus an appropriate suction pressure at the compressor inlet can always be found when an energy balance exists.

Once energy and pressure balances are established, two out of four refrigerant parameters estimated at the outset are determined. However, these two parameters, compressor suction and discharge pressures, were obtained for imposed refrigerant superheat (quality) at compressor can inlet and assumed circulating composition which still have to be verified. For refrigerant superheat (quality) verification, the refrigerant mass inventory is made. It is based on refrigerant states in the system which are found to satisfy energy and pressure balances with assumed vapor superheat at the compressor can inlet. The amount of refrigerant obtained from mass inventory calculations is compared to the actual refrigerant mass input (known design parameter). If the amount of refrigerant calculated is smaller than refrigerant input into the heat pump, the superheat (quality) estimate has to be decreased and all calculations have to be repeated from the outset.

Once, in addition to satisfied pressure and energy balances, the refrigerant superheat (quality) at the compressor can inlet is verified by means of mixture mass inventory, the ultimate verification of iterated refrigerant states has to be done by checking if the circulating mixture composition for which all results were obtained is in fact the actual circulation composition. This is done by performing a mass inventory of one of the mixture components. It should be mentioned that a composition shift of circulating mixture can occur only as a result of accumulation somewhere in the system of some of the mixture of composition different from the original, i.e., accumulation of liquid in the accumulator. If such accumulation does not take place (refrigerant entering the compressor is a superheated vapor) the circulation composition equals the original composition of the charged refrigerant. If refrigerant entering the compressor has quality less than 1, the program has to proceed with mass inventory for one mixture component. This mass inventory is based on refrigerant states in the system which are found to satisfy enthalpy and pressure balances, and for which the total mass of refrigerant in the system is conserved. If the amount of calculated mixture component is not equal to the original amount charged into the system, the estimate of composition of the circulating refrigerant has to be adjusted and all calculations have to be repeated from the outset. The solution logic described here is presented graphically in figure 9. The actual implementation of this logic in the main program of the model is presented in Appendix D.

5.2 Modeling of a Reciprocating, Hermetic Compressor

5.2.1 Compressor Operation

The compressor is mechanically the most complex component of a heat pump. A reciprocating, hermetic compressor is most commonly used in heat pump systems. A schematic of this type of compressor is shown in figure 10. The compressor consists of a shell containing an electric motor, a cylinder/piston assembly with valves and manifolds and tubes. The electric motor is coupled to the compressor eccentric. Lubricating oil is collected on the bottom of the can and is in free contact with the refrigerant.

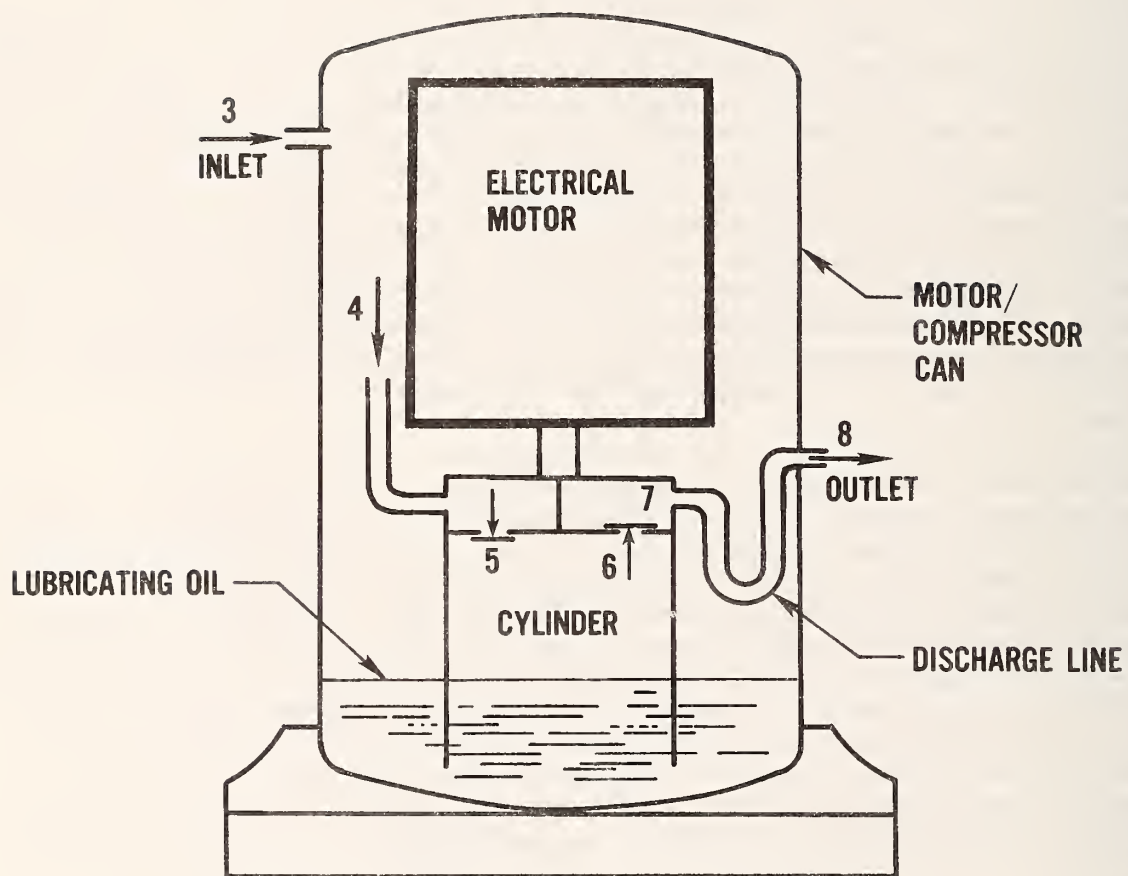
Flow of the refrigerant is from location 3 to location 8, as marked in figure 10. Low pressure refrigerant enters the compressor can and is directed towards the electric motor to cool motor windings. The enthalpy of the refrigerant changes due to this and other heat transfer with other surrounding surfaces, namely, the compressor can, manifolds, the discharge tube and the cylinder body. Then the refrigerant at state 4 undergoes the process in the cylinder, which in case of positive displacement compressors consists of: expansion through the suction valve and mixing the residual gas, compression, expansion through the discharge valve, and re-expansion of residual gas. The compressed, high temperature and high pressure refrigerant vapor leaves the compressor through the discharge manifold and discharge line after giving up some heat to the low pressure refrigerant at state 4.

The numbers marked in figure 10 identify key compressor locations and the corresponding refrigerant states. Even during steady state operation refrigerant parameters in the compressor can are not steady, since pressure changes in a series of rapid pulses. This pulsation is initiated by the periodic compression and suction processes and valve behavior. Because a certain pressure difference is needed for valve opening, the pressure in the cylinder at the end of compression is considerably above the discharge manifold pressure. Similarly, pressure during intake falls below suction manifold pressure. Particularly strong pressure peaks are observed in case of spring equipped valves. These pressure differences vary from compressor to compressor because of valve and manifold design, but also vary for the same compressor with compressor speed, compression pressure ratio, refrigerant, and mass flow rate.

Since processes taking place in a compressor are dynamic in nature, the most correct way of modeling a compressor would require dynamic simulation of valve motion, cylinder-manifold pressure interaction, and cylinder heat transfer. Such models exist and are described in [12], [13], and [14]. However, these models usually require some very detailed experimental data or design information, not necessarily readily available for the prospective user of a heat pump simulation program. That is why a relatively simpler approach to compressor modeling was taken in this study.

5.2.2 Theory and Governing Relations

Several assumptions are made for model formulation. The basic assumption is that the highly dynamic process in the compressor results in steady gas parameters throughout the can. The pressure and temperature in the cylinder and in suction and discharge manifolds are assumed to have constant values.



Note: Numbers 5 and 6 situated in the compressor correspond to the refrigerant state before and after the compression process.

Figure 10. Schematic of a hermetic compressor.

The refrigerant is considered to have uniform thermodynamic properties throughout the space assigned to the particular location. Refrigerant flow through the compressor is assumed to be one-dimensional, so one-dimensional steady flow equations can be used. Though the model does not simulate valve dynamic behavior, it allows for valve representation by a steady difference between manifold and cylinder pressures.

In order to derive equations governing compressor balances, the thermodynamic irreversibilities taking place in the hermetically sealed compressor have to be identified. These losses can be put into four categories:

1. Those associated with incomplete conversion of electric energy into a mechanical energy available for vapor compression.
2. Those associated with the non-isentropic conversion process inside the cylinder.
3. Those due to heat transfer at the different locations.
4. Those due to pressure drop at the different locations.

All of these losses contribute to the overall compressor efficiency and have to be considered in the modeling effort.

Conversion of Electrical to Mechanical Energy

Electric energy is supplied to an electric motor to be converted into mechanical energy. This conversion has losses due to windage, friction, winding resistance, and hysteresis, which are accounted for by a motor efficiency, η_e . By definition

$$\eta_e = \frac{W_e}{E} \quad (26)$$

where E = electric power input
 W_e = mechanical power output

Electric motor efficiency, η_e , depends on load and is customarily given as a function of the fraction of actual mechanical power output to the maximum power output. In heat pumps with power requirements below 5 hp, single phase electric motors are usually used. They are permanent split-capacitor or capacitor-start capacitor-run types. A typical efficiency versus load curve for a permanent split-capacitor 2 pole electric motor is shown in figure 11 [15].

The electric motor is coupled to the compressor eccentric. Moving compressor parts experience friction and a power loss accounted for by the compressor mechanical efficiency, η_m . By definition

$$\eta_m = \frac{W_c}{W_e} \quad (27)$$

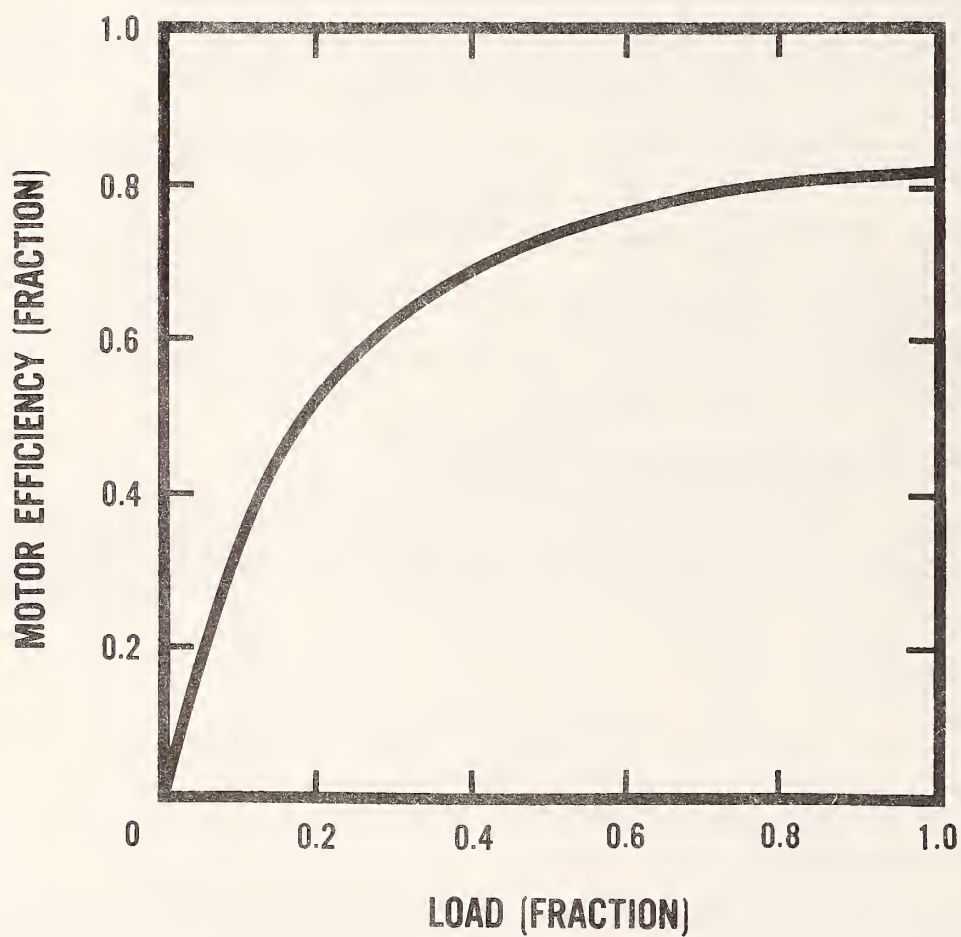


Figure 11. Typical speed (RPM) versus load curve for a permanent split-capacitor 2 pole electrical motor [15].

where W_c = mechanical power available for compression

A common value of compressor mechanical efficiency falls in range of 0.95 to 0.98.

Compression Process

A simplified indicator diagram is shown in figure 12. It shows idealized constant suction and discharge pressures in a cylinder. In modeling the cylinder process, both compression and re-expansion processes are assumed to be polytropic with the same polytropic index, n , following the equation [16]:

$$P \cdot v^n = \text{const} \quad (28)$$

where n = polytropic index
 P = pressure
 v = refrigerant vapor specific volume

Since constant pressure and temperature are assumed during the discharge process, this implies no change in specific volume between points C and D (figure 12); consequently, the compression and re-expansion polytropic curves will coincide. This further means, that the net work required for compression of the residual gas is zero.

The refrigerant enthalpy increase during polytropic compression, $i_6 - i_5$ (refer to figure 3), can be evaluated by the equation derived from the expressions for isentropic and polytropic work of compression at the same compression ratio. Equating these expressions results in the equation:

$$i_6 - i_5 = (i_{6s} - i_5) \frac{\frac{n}{\gamma - 1} \left(\frac{P_6}{P_5} \right)^{\frac{n-1}{n}} - 1}{\frac{\gamma}{\gamma - 1} \left(\frac{P_6}{P_5} \right)^{\frac{\gamma-1}{\gamma}} - 1} \quad (29)$$

where i_5 = refrigerant enthalpy before compression
 i_6 = refrigerant enthalpy after compression
 i_{6s} = refrigerant enthalpy after isentropic compression, defined by pressure p_6 and entropy s_5
 n = polytropic index
 P_5 = suction pressure
 P_6 = discharge pressure
 γ = isentropic index

The isentropic index, γ , and the polytropic index, n , are related by the polytropic efficiency of the compressor:

$$\eta_p = \frac{\frac{\gamma - 1}{n}}{\frac{\gamma}{n - 1}} \quad (30)$$

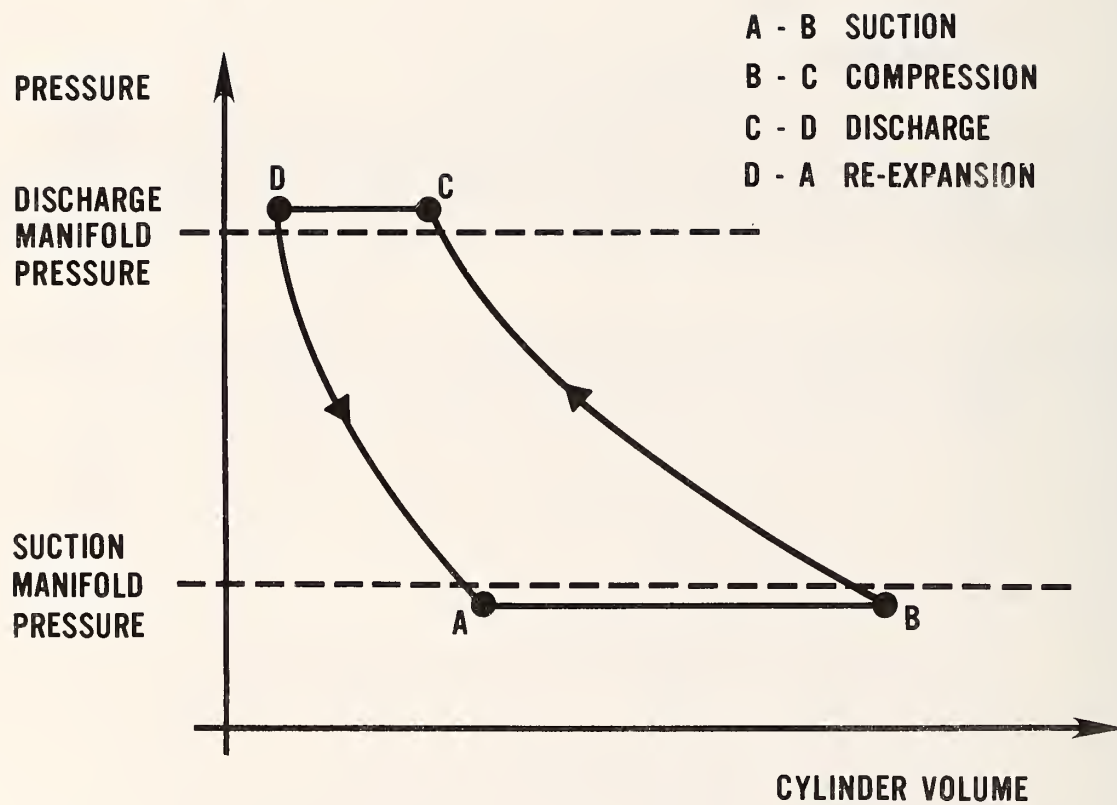


Figure 12. Simplified indicator diagram for a reciprocating compressor.

The isentropic index, γ , equals the ratio of specific heats, as in the following formula:

$$\gamma = \frac{C_p}{C_v} \quad (31)$$

where C_p = specific heat at constant pressure
 C_v = specific heat at constant volume

The specific heats of ideal gases are constant, thus the isentropic index of an ideal gas is also constant. For real gases such as refrigerant vapors, specific heats vary along the compression path. To accommodate this fact, the isentropic index can be evaluated by taking the average of the respective specific heat ratios at points 5 and 6s.

The refrigerant enthalpy increase during polytropic compression could also be calculated using the isentropic efficiency and enthalpy increase during isentropic compression at the same compression ratio. However, isentropic efficiency is sensitive to the compression ratio while polytropic efficiency is more consistent from one application to another and provides a more consistent representation of average compressor performance [17]. The imperfection of using isentropic efficiency may be traced to the general thermodynamic relation:

$$\left(\frac{\partial i}{\partial s} \right)_P = T \quad (32)$$

where i = enthalpy
 P = pressure
 s = entropy
 T = temperature

which requires pressure lines to diverge on a Mollier chart.

The refrigerant mass flow rate pumped by a compressor can be calculated by the following formula:

$$m_r = \frac{60 \cdot \text{RPM} \cdot V_s}{v_5} \eta_v \quad (33)$$

where m_r = refrigerant mass flow rate
 RPM = compressor speed (revolutions per minute)
 V_s = compressor displacement per revolution
 v_5 = refrigerant specific volume in the cylinder before compression
 η_v = volumetric efficiency

The RPM of a compressor is equal to that of an electric motor and is a function of load on the motor. A typical speed (RPM) versus load curve for a permanent split-capacitor 2-pole electric motor is shown in figure 13.

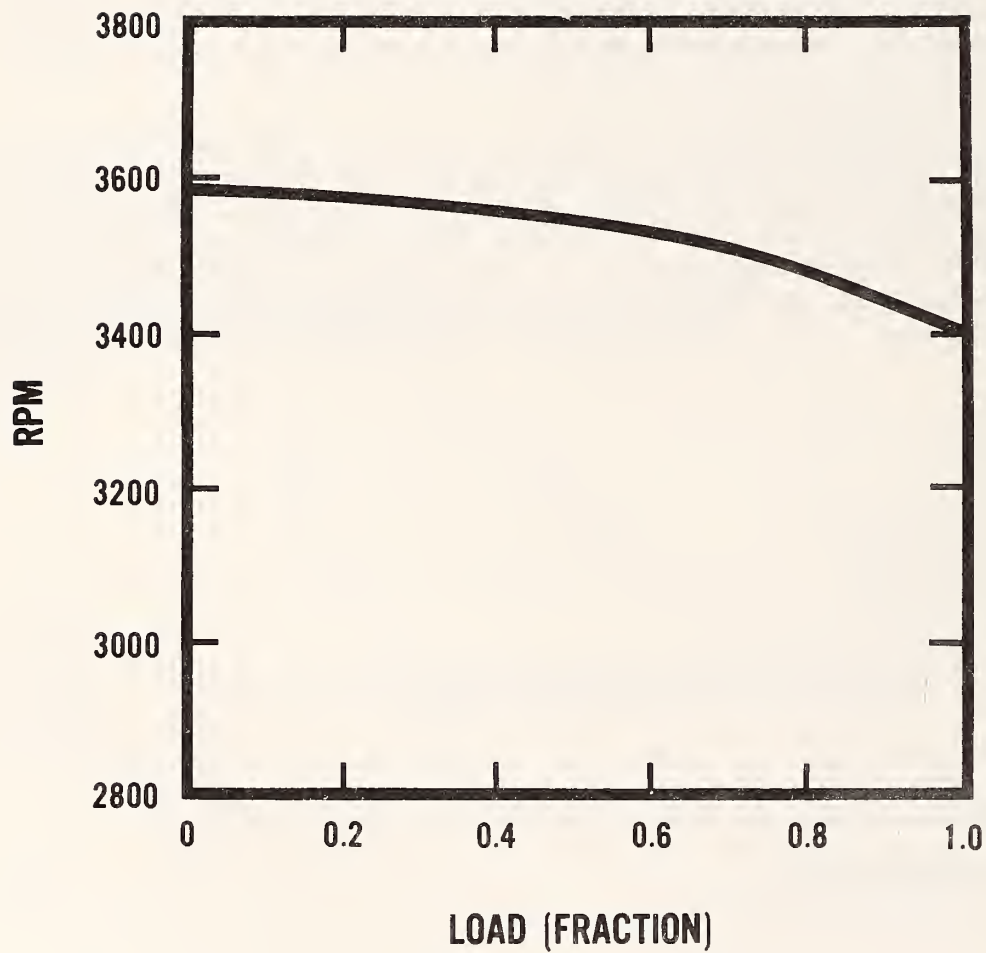


Figure 13. Typical efficiency versus load curve for a permanent split-capacitor 2 pole electric motor [15].

With assumptions stated so far for the compression and re-expansion processes, the following formula for volumetric efficiency of a compressor can be derived [16]:

$$\eta_v = C_c \left[1 - C_e \left[\left(\frac{P_6}{P_5} \right)^{\frac{1}{n}} - 1 \right] \right] \quad (34)$$

where C_c = correction factor for leakage from the piston, valves and for the throttling effect, assumed to be 0.96, [18]
 C_e = clearance volume, fraction of displacement

Heat Transfer Relations

Electrical energy supplied to an electric motor is in part transferred to the refrigerant and in part is dissipated to the compressor environment. This compressor heat balance can be expressed by the following equation:

$$E + m_r (i_3 - i_8) - Q_{C,A} = 0 \quad (35)$$

where E = electric energy input
 i_3, i_8 = refrigerant enthalpy at respective locations (refer to figure 10)
 m_r = refrigerant mass flow rate
 $Q_{C,A}$ = rate of heat rejected to ambient air

Reference is made to figure 10, where key locations of refrigerant in the hermetic compressor are marked. In order to solve the compressor heat balance represented by equation (35), the following heat transfer losses are considered in the compressor model:

1. Heat transfer between the compressor can and ambient air, $Q_{C,A}$
2. Heat transfer between the refrigerant in the compressor can and the compressor can, $Q_{4,C}$
3. Heat transfer between the inlet refrigerant and the suction manifold and valve, $Q_{4,5}$
4. Heat transfer between the discharge refrigerant and the discharge valve and manifold, $Q_{6,7}$
5. Heat transfer between the refrigerant in the compressor can and the refrigerant in the discharge line, $Q_{7,8}$

The heat transfer between the compressor can and ambient air (item 1) is governed by free convection and radiation. Respective heat transfer coefficients, h_c and h_r , are calculated by the following equations [19]:

$$h_c = 0.18(T_c - T_a)^{0.33} \quad (\text{Btu}/(\text{h} \cdot \text{ft}^2 \cdot \text{F})) \quad (36)$$

$$h_r = \sigma \cdot \varepsilon \cdot \frac{(T_c + 459.67)^4 - (T_a + 459.67)^4}{T_c - T_a} \quad (\text{Btu}/(\text{h} \cdot \text{ft}^2 \cdot \text{F})) \quad (37)$$

where T_a = ambient air temperature (F)
 T_c = compressor shell temperature (F)
 $\sigma = 0.1714 \times 10^{-4} \frac{\text{Btu}}{\text{h} \cdot \text{ft}^2 \cdot \text{R}^4}$ = Stefan-Boltzman constant
 ε = surface emissivity ($\varepsilon = 0.9$ is used)

The heat transfer rate between the compressor shell and the ambient air, $Q_{C,A}$ can be calculated by the equation:

$$Q_{C,A} = CQ_{C,A} \cdot (h_c + h_r) \cdot (T_c - T_a) \quad (38)$$

where $CQ_{C,A}$ = heat transfer parameter

The heat transfer inside the compressor shell between the shell and refrigerant vapor (item 2) is governed by forced convection, for which the non-dimensional heat transfer parametric expression in terms of Nusselt, Reynolds, and Prandtl numbers is in the form [20]:

$$Nu \propto Re^{0.8} \cdot Pr^{0.333} \quad (39)$$

where $Nu = \frac{h \cdot L}{k}$ = Nusselt Number

$Pr = \frac{\mu \cdot Cp}{k}$ = Prandtl Number

$Re = \frac{G \cdot L}{\mu}$ = Reynolds Number

$G = \frac{m_r}{A}$ = refrigerant mass flux

L = characteristic length

k = refrigerant thermal conductivity

μ = refrigerant dynamic viscosity

Using equation (39) the forced convection heat transfer coefficient, h , can be expressed as:

$$h = C \cdot m_r^{0.8} \cdot k^{0.666} \cdot Cp^{0.333} \cdot \mu^{-0.467} \quad (40)$$

where C = constant of proportionality, a function of wetted surface geometry

Combining equations (36) and (40) yields the following expression for heat transfer rate between the compressor can and refrigerant:

$$Q_{4,C} = C \cdot A_h \cdot m_r^{0.8} \cdot k_4^{0.667} \cdot Cp_4^{0.333} \cdot \mu_4^{-0.467} (T_4 - T_c) \quad (41)$$

or

$$Q_{4,C} = CQ_{4,C} \cdot m_r^{0.8} \cdot k_4^{0.667} \cdot Cp_4^{0.333} \cdot \mu_4^{-0.467} (T_4 - T_c) \quad (42)$$

where $CQ_{4,C}$ = heat transfer parameter

Obviously, heat transfer rates $Q_{C,A}$ and $Q_{4,C}$ are equal. Derivations of equations for the heat transfer rates between the inlet refrigerant and the suction manifold and valve, $Q_{4,5}$, and between the discharge refrigerant manifold valve, $Q_{6,7}$, are similar to the derivation just performed for $Q_{4,C}$, since in these two cases heat transfer is also by forced convection. Resulting expressions for these heat transfer rates are:

$$Q_{4,5} = CQ_{4,5} \cdot m_r^{0.8} \cdot k_{4,5}^{0.666} \cdot Cp_{4,5}^{0.333} \cdot \mu_{4,5}^{-0.467} (T_6 - T_4) \quad (43)$$

$$Q_{6,7} = CQ_{6,7} \cdot m_r^{0.8} \cdot k_{6,7}^{0.667} \cdot Cp_{4,5}^{0.333} \cdot \mu_{6,7}^{-0.467} (T_7 - T_4) \quad (44)$$

The heat transfer between the refrigerant in the compressor can and the refrigerant in the discharge line (item 5) is modeled as forced convection heat transfer between the fluids separated by a barrier non-resistant to heat flow. Assuming that the temperature of refrigerant in the shell does not change (as a result of other heat transfers in the can and mixing) and applying the logarithmic mean temperature difference, the following expression for heat transfer rate $Q_{7,8}$ can be derived [2]:

$$Q_{7,8} = CQ_{7,8} \cdot m_r^{0.8} \frac{a1}{a2 + a3} (T_7 - T_8) / \ln \frac{T_7 - T_4}{T_8 - T_4} \quad (45)$$

where

$$a1 = (Cp_4 \cdot Cp_{7,8})^{0.333} (k_4 \cdot k_{7,8})^{0.667}$$

$$a2 = \mu_4^{0.467} \cdot Cp_{7,8}^{0.333} \cdot k_{7,8}^{0.667}$$

$$a3 = \mu_{7,8}^{0.467} \cdot Cp_4^{0.333} \cdot k_4^{0.667}$$

Subscripts 3 through 8 refer to refrigerant key locations in the compressor in figure 10. If the subscripts are separated by a comma, the average value is implied.

The heat transfer to/from the flowing refrigerant changes the refrigerant enthalpy according to the equation:

$$Q = m_r \cdot \Delta i \quad (46)$$

where Δi = refrigerant enthalpy change

Combining equations (43), (44), (45), and (46) yields the following expressions for the refrigerant enthalpy change during flow between respective locations of the compressor:

$$i_5 - i_4 = CQ_{4,5} \cdot k_{4,5}^{0.667} \cdot Cp_{4,5}^{0.333} (T_6 - T_4) / (m_r^{0.2} \cdot \mu_{4,5}^{0.467}) \quad (47)$$

$$i_6 - i_7 = CQ_{6,7} \cdot k_{6,7}^{0.667} \cdot Cp_{6,7}^{0.333} (T_7 - T_4) / (m_r^{0.2} \cdot \mu_{6,7}^{0.466}) \quad (48)$$

$$i_7 - i_8 = CQ_{6,7} \frac{a_1}{a_2 + a_3} (T_7 - T_8) / (m_r^{0.2} \cdot \ln \frac{T_7 - T_4}{T_8 - T_4}) \quad (49)$$

where a_1, a_2, a_3 are as in equation (45)

The derived heat transfer relations contain heat transfer parameters $CQ_{C,A}$, $CQ_{4,5}$, $CQ_{6,7}$, and $CQ_{7,8}$. These parameters are primarily functions of heat transfer surface geometry and have to be found empirically. If required laboratory test data for a given compressor are not available, typical compressor test data can be used. A large number of experimental compressor measurements have been published. The summary of these data and a list of references can be found in [21].

Pressure Drop Relations

Total pressure drop ΔP_{tot} experienced by a flowing fluid results from pressure drops due to friction, momentum change, and gravity, i.e.,

$$\Delta P_{tot} = \Delta P_{friction} + \Delta P_{accel} + \Delta P_{gravity} \quad (50)$$

Pressure drop due to gravity in the hermetic compressor may be disregarded based on an order of magnitude analysis. On the same grounds, pressure drop of the flowing refrigerant between certain compressor locations may be attributed to either dynamic effect or viscous effect.

Pressure drop due to the dynamic effect, ΔP_{accel} , is proportional to velocity head, i.e.,

$$\Delta P_{accel} \propto \rho \cdot V^2 \quad (51)$$

which can be written, using the continuity equation:

$$\Delta P_{accel} = CP \cdot m_r^2 / \rho \quad (52)$$

where CP = pressure drop parameter
 m_r = refrigerant mass flow rate
 V = refrigerant velocity
 ρ = refrigerant density

The relation for the pressure drop due to the viscous effect, P_{friction} , can be derived from the classical Fanning equation for pressure drop in a tube:

$$\Delta P_{\text{friction}} = 2f \cdot \rho \cdot V^2 \cdot L/D \quad (53)$$

where f = friction factor
 D = tube diameter
 L = tube length

The friction factor, f , in the equation above is approximately proportional to the Reynolds number to the -0.2 power for the Reynolds number greater than 2000 [22]. Considering this and applying the equation of continuity, equation (53) becomes:

$$\Delta P_{\text{friction}} = CP \cdot \mu^{0.2} \cdot m_r^{0.8}/\rho \quad (54)$$

The pressure drop of the refrigerant in the compressor can be modeled by evaluating individual pressure drops between refrigerant key locations indicated in figure 10. Based on equations (52) and (54), and attributing pressure drop between particular compressor locations either to the viscous or friction effect, the following pressure drop relations are proposed:

$$P_3 - P_4 = CP_{3,4} \cdot m_r^2/\rho_{3,4} \quad \text{for the compressor can inlet} \quad (55)$$

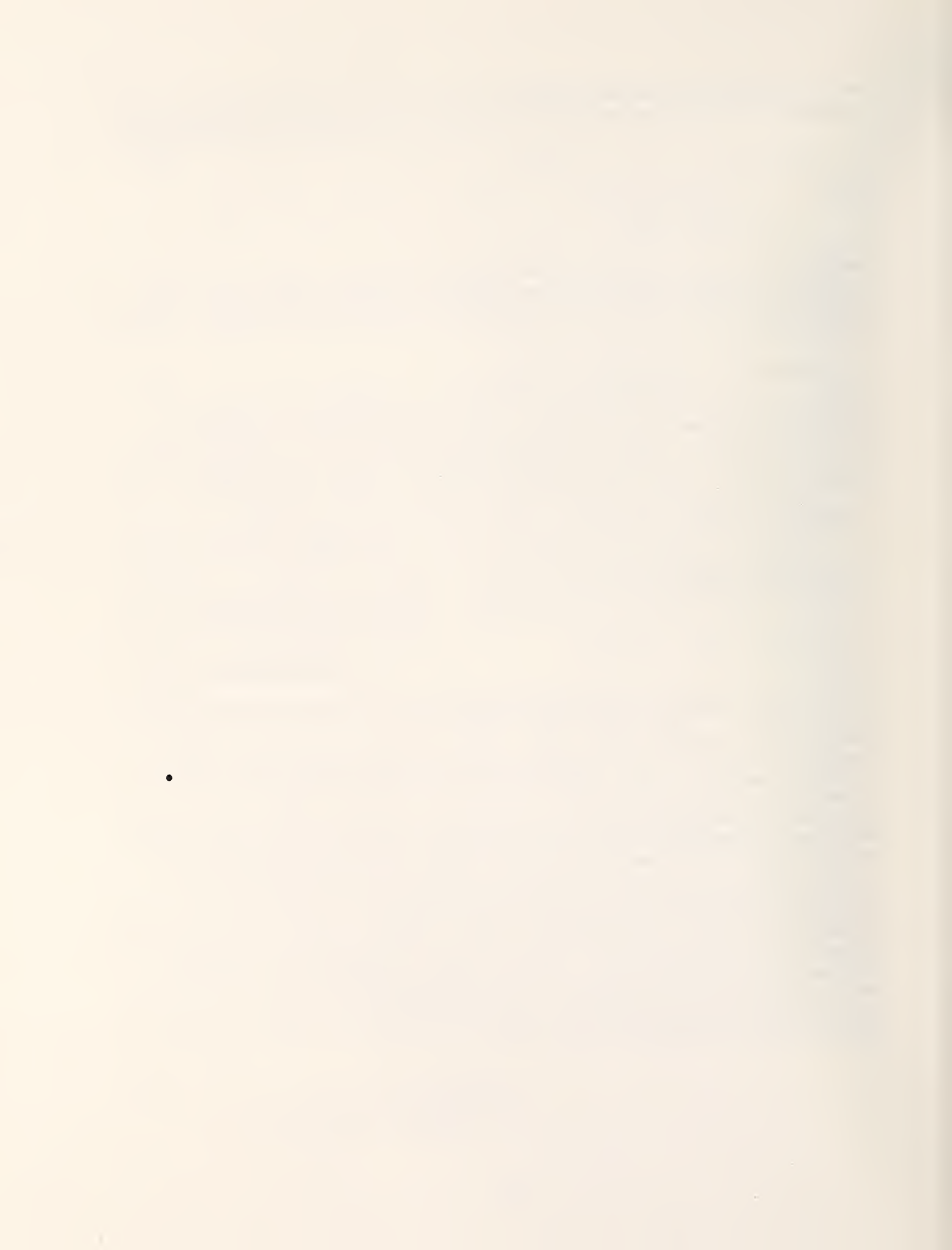
$$P_4 - P_5 = CP_{4,5} \cdot m_r^2/\rho_{4,5} \quad \text{for the suction manifold and valve} \quad (56)$$

$$P_6 - P_7 = CP_{6,7} \cdot m_r^2/\rho_{6,7} \quad \text{for the discharge valve and manifold} \quad (57)$$

$$P_7 - P_8 = CP_{7,8} \cdot m_r^{1.8} \cdot \mu_{7,8}^{0.2}/\rho_{7,8} \quad \text{for the discharge line} \quad (58)$$

Pressure drop parameters $CP_{3,4}$, $CP_{4,5}$, $CP_{6,7}$, and $CP_{7,8}$ have to be found experimentally for a given compressor or can be calculated using test data for a similar compressor. For sources of such data, refer to [21].

Equations derived in this section are used in a compressor subroutine, COMPRE, described in Appendix E. These equations make it possible to carry out an energy balance of a hermetic compressor using an iterative process for electrical energy input, for heat lost to the ambient air, and for refrigerant parameters in compressor key locations. The variety of designs of refrigerant flow passages in the compressor caused the modeling of pressure drop and heat transfer to be done in an approximate manner in this general compressor model. In spite of several assumptions that were made to simplify the modeling process and reduce computing time, the model still retains sufficient details of the underlying physical principles to allow designers to determine which specific changes in the compressor design will lead to increased efficiency of the compressor and the heat pump system.



5.3 Modeling of a Constant Flow Area Expansion Device

A constant flow area expansion device, as used in heating/air conditioning systems, is commonly called a capillary tube or a refrigerant flow restrictor. Usually it is a small bore tube of length as short as one-half inch up to a few feet, connecting the outlet of the condenser (or a liquid line) to the inlet of the evaporator. The main task of the constant flow area expansion device is to maintain the minimum pressure at the condenser at which all the flowing refrigerant can condense. Many researchers have investigated flow of a single component refrigerant through a capillary tube and a bibliography on the subject can be found in [15].

There is no experimental data known to the authors which would refer to the flow of a non-azeotropic mixture through a capillary tube or orifice. Because of lack of data, single component refrigerant flow experience has been extrapolated to the non-azeotropic binary situation and the model developed accordingly.

5.3.1 Available Capillary Tube Performance Data

The capillary tube is a traditionally accepted name for a constant flow area expansion device used in heat pump systems. This name is inadequate and misleading since for tube diameters in the neighborhood of 1/16 of an inch, capillary forces are negligible. The pressure drop consists primarily of:

- the loss due to sudden contraction at the entrance
- the loss due to flow in the tube
- the loss due to sudden enlargement at the exit to the evaporator

The flow of refrigerant through a constant bore tube is more complex than the geometric simplicity of the device would first indicate. The pressure and temperature distribution along a typical capillary tube is shown in figure 14. Bolstad and Jordan's description of the flow is as follows:

At the entrance to the tube, section 0-1, there is a slight pressure drop which was usually unreadable on the gages. From point 1 to point 2 the pressure drop is linear. In the portion of the tube 0-1-2 the refrigerant is entirely in the liquid state and at point 2 the first bubble of vapor forms. From point 2 to the end of the tube the pressure drop is not linear, the pressure drop per unit length increasing as the end of the tube is approached. For this portion of the tube, both the saturated liquid and saturated vapor phases are present, the percent and volume of vapor increasing in the direction of flow . . .

With a saturation temperature scale corresponding to the pressure scale superimposed along the vertical axis, it is possible to plot the observed temperatures in a more meaningful way than if a uniform temperature scale were used. The temperature is constant for the first portion of the tube 0-1-2. At point 2, the pressure has dropped to the saturation pressure corresponding to this temperature. Further pressure drop beyond point 2 is accompanied by a corresponding drop in temperature, the

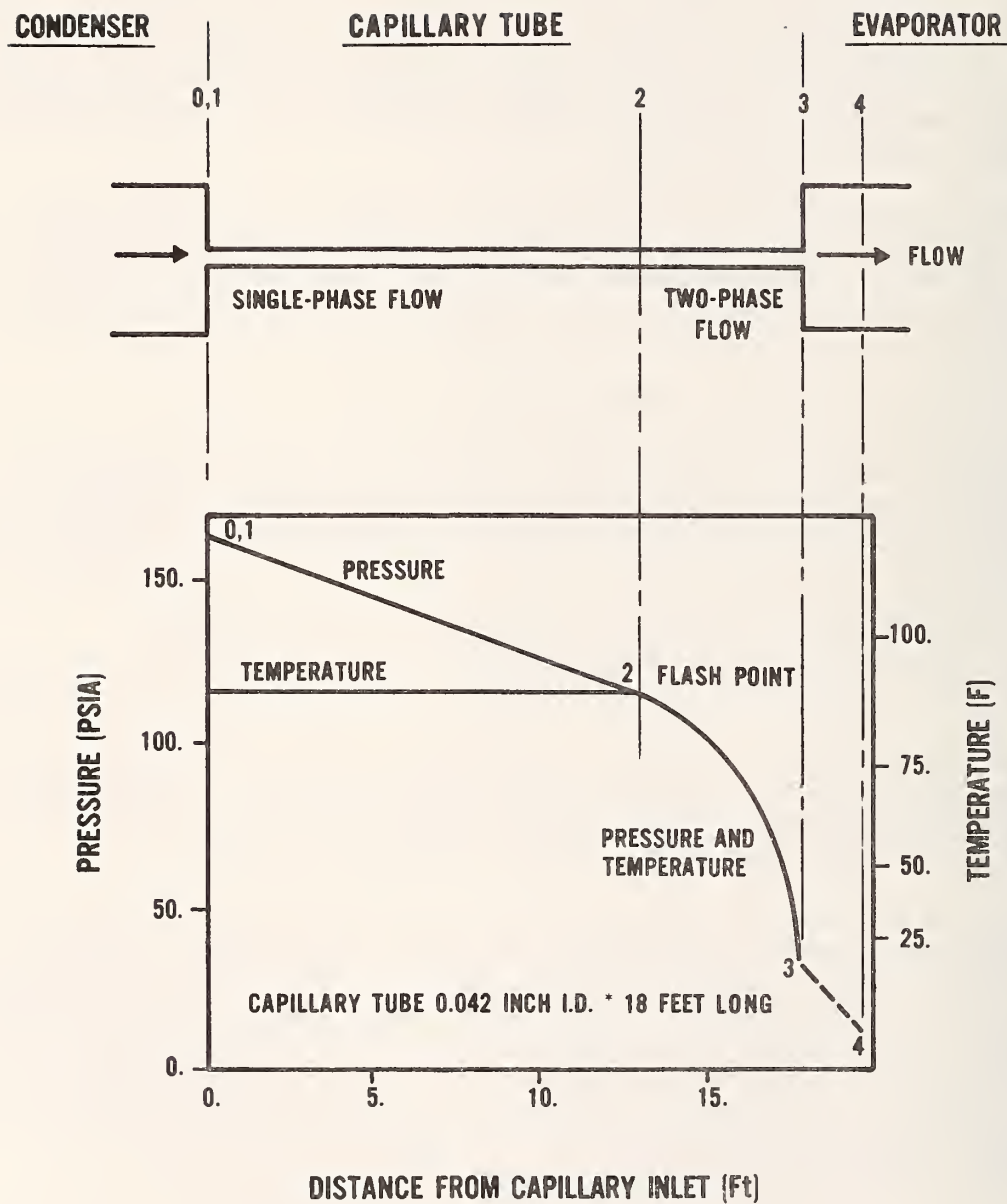


Figure 14. Pressure and temperature distribution along typical capillary tube [23].

temperature being saturation temperature corresponding to the pressure. As a consequence, the pressure and temperature lines coincide from point 2 to the end of the tube [23].

The point of the tube where the first bubble forms is called a bubble point or a flash point. The pressure at the point is called the flash pressure. Bolstad also presented an analytical method of solution for adiabatic flow through the capillary based on the Fanno flow theory.

Mikol [24] performed a capillary tube investigation from which his conclusions can be summarized as follows:

- Fluid flow through small bore tubes conforms to continuous flow as established for large bore tubes and pipes.
- The friction factor correlation of Moody and any others consistent with Moody's correlation [25] is applicable to single-phase flow in small bore tubes.
- The phenomena of metastability, persistence of the liquid state at pressures less than the saturation pressure corresponding to its local temperature, has been found to occur.
- The phenomenon of choked flow in two-phase flow occurs in the same way and for the same reasons as in the case of gaseous flow. Sonic velocity occurs at the tube exit.

One of Mikol's findings, existence of superheated liquid in a small portion of a tube, was not observed by Bolstad and Jordan [23]. However, it was reported by Cooper et al. [26] and Rezk [27]. Investigators have found that delayed evaporation is affected by initial disturbances and flow agitation, but there is not enough data in the current literature to assess all the factors promoting or eliminating this phenomena and making it possible to consider metastability in a capillary tube model at this time.

The pressure and temperature distributions along a capillary tube as shown in figure 14 occur at design operating conditions of a long (a few feet) capillary. Part of the capillary is filled with flowing liquid, while two-phase flow exists in the other part. However, there are also other possible modes of operation, i.e., with only two-phase flow in the capillary (the case of incomplete condensation in the condenser) or with only liquid flow (the case of short restrictor). All these cases are observed in practice and have to be simulated by a general model of the constant flow area expansion device.

Based on the experimental evidence and the theory of large tube fluid mechanics, the following assumptions were used for model formulation:

1. The capillary tube is straight, horizontal, and has a constant inner diameter.
2. Flow in the capillary is one-dimensional and homogeneous.
3. Flow in the capillary is adiabatic.

4. Flow resistance in the capillary tube can be subdivided into
 - a. Resistance due to the entrance effect
 - b. resistance due to flow in a tube which consists, in the general case, of single-phase liquid flow resistance from the entrance to the flash point, and two-phase mixture flow resistance in the rest of the tube. The existence of the delayed evaporation phenomena is neglected. Resistance due to the exit effect is neglected as meaningless for a choked flow and insignificant for a non-choked flow [23].
5. Choked flow phenomenon for two-phase flow of a non-azeotropic mixture is governed by the same laws as for single-phase flow of a single component fluid and can be modeled accordingly.

5.3.2 Available Short Tube Restrictor Performance Data

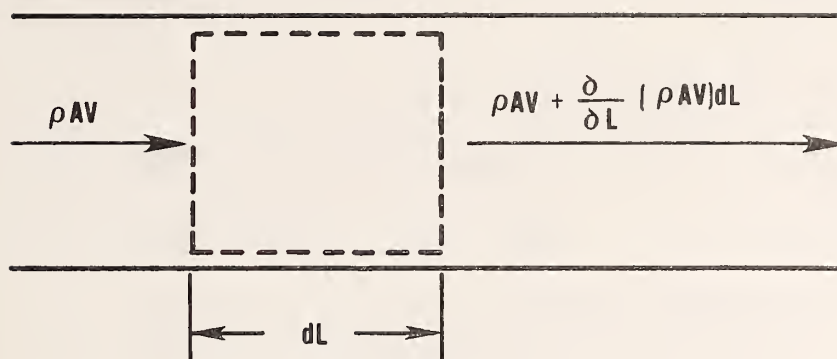
An experimental study on the flow of refrigerant 12 through short tubes was performed by Pasqua [28]. From his visual observations he found that the refrigerant flowed through the restrictor in the form of a metastable liquid core surrounded by a vapor annulus. Based on his experiment, Pasqua also determined flow characteristics of subcooled and saturated liquid through a short tube restrictor. A study of short tube restrictors applied to refrigerant 22 was performed by Mei [29]. He tested five restrictors of a length/diameter ratios from 7.5 to 11.9. He confirmed occurrence of first-stage choking but reported that second-stage choking did not take place at conditions obtainable in his test facility. Mei provided two correlations for evaluation of refrigerant mass flow rate which, however, are limited to the tested refrigerant 22.

As experimentally obtained information available in the literature is not sufficient for a development of a general simulation model of a short restrictor applicable to different refrigerants, such models have to be developed using fundamental equilibrium fluid mechanics. Using this approach, the analysis and basic assumptions made in the previous section regarding a capillary tube would apply to a short tube.

5.3.3 Critical Flow

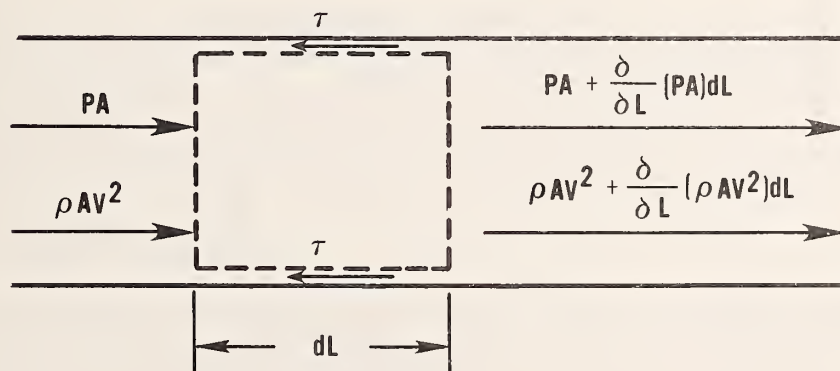
Refrigerant mass flow rate through a given flow restrictor will depend on the inlet refrigerant state and on the pressure that will be established at the tube outlet. This pressure may equal the evaporator pressure or may have a higher value if the flow is choked. Since pressure at the tube exit is one of the parameters affecting the flow, it has to be known for accurate refrigerant mass flow rate prediction.

The assumption that the flow in the flow restrictor is one-dimensional and homogeneous enables the two-phase flow in the tube to be treated as single-phase flow with uniform properties at any cross-section of the flow and allows use of the single-phase, one-dimensional form of the governing equations as presented in figure 15, 16 and 17.



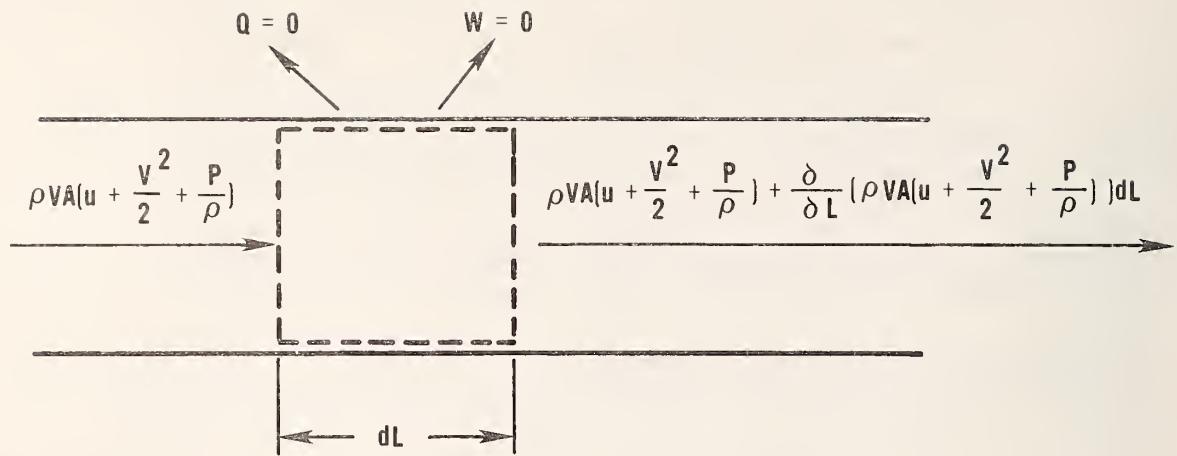
CONTINUITY EQUATION: $d(\rho V) = 0$

Figure 15. Mass balance for an element of fluid in one-dimensional steady flow in a constant area duct.



MOMENTUM EQUATION: $AdP + \rho AVdV + \tau SdL = 0$

Figure 16. Momentum balance for an element of fluid in one-dimensional steady flow in a horizontal, constant area duct.



ENERGY EQUATION: $d(u + \frac{V^2}{2} + \frac{P}{\rho}) = d(i + \frac{V^2}{2}) = 0$ $i + \frac{V^2}{2} = i_0$

Figure 17. Energy balance for an element of fluid in one-dimensional adiabatic, steady flow in a horizontal, constant area duct.

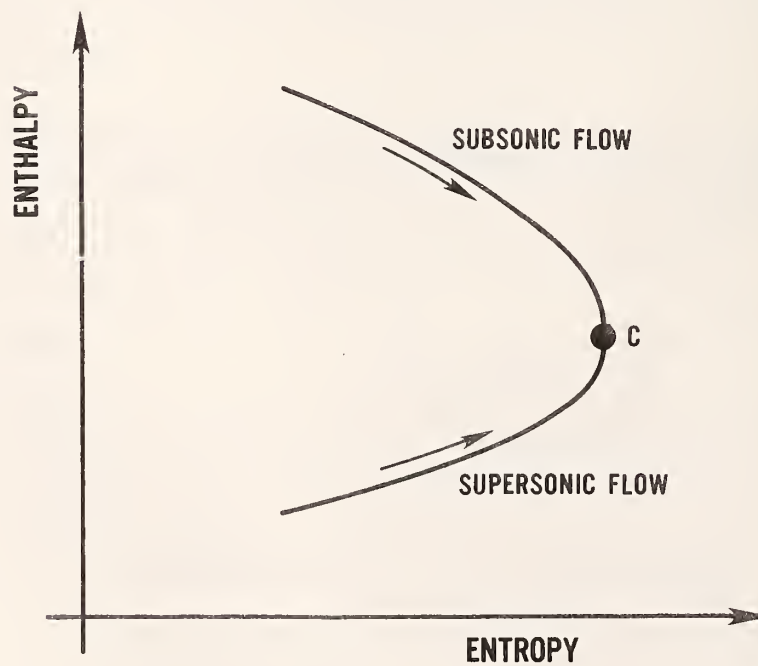


Figure 18. Fanno line.

Adiabatic flow through a capillary tube is a classical example of so called Fanno flow; adiabatic flow with friction in a constant area duct. The energy equation for such flow has the form:

$$di + V \cdot dV = 0 \quad \text{or} \quad i_o = i + V^2/2 = \text{const} \quad (59)$$

which, when combined with the equation of continuity yields:

$$i_o = i + \frac{m_r^2}{A} \cdot v^2/2 = i + G^2 \cdot v^2/2 \quad (60)$$

where A = tube cross-sectional area

$$G = \frac{m_r}{A}, \text{ refrigerant mass flux}$$

i = refrigerant enthalpy
 i_o = refrigerant stagnation enthalpy
 m_r = refrigerant mass flow rate
 V = flow velocity
 v = refrigerant specific volume

Graphical representation of equation (60) on the enthalpy-entropy diagram, as shown in figure 18, is called a Fanno line. Fanno flow, as an irreversible adiabatic process, can sequentially exist only in the direction of increasing entropy. The upper branch of the Fanno line corresponds to subsonic, accelerating flow while the lower branch is the supersonic, decelerating flow. Both flows tend towards point C where the sonic velocity is reached.

The Fanno line implies that fluid cannot reach the velocity of sound inside a constant area duct, because, if this happened, further flow in the duct would have to be associated with a decrease of entropy (figure 18), and that would be in violation of the Second Law of Thermodynamics. Thus, the only location of a tube at which sonic velocity can be reached is the tube exit, and choking will never occur inside the capillary regardless of external outlet pressure.

5.3.4 Model Formulation

Refrigerant flow from the liquid line into the capillary tube experiences a pressure drop due to sudden contraction. This pressure drop consists of an acceleration loss and entrance friction loss and is usually expressed by a decrease in the Bernoulli head and a contraction coefficient referred to the kinetic energy of the flow in the section of smaller flow area:

$$\frac{P_0 - P_1}{\rho_{0,1}} + \frac{V_0^2 - V_1^2}{2} = K \frac{V_1^2}{2} \quad (61)$$

Subscripts in eq. (61) refer to sections shown in figure 14. Combining with the equation of continuity, the above can be rearranged to:

$$P_0 - P_1 = \Delta P = (1 + K) \rho_{0,1} \cdot \frac{V_1^2}{2} \quad (62)$$

The contraction coefficient, K , given in the literature is strictly empirical and is represented as a function of the contraction area ratio. Several sources are in disagreement about its value. The value of $K = 0.15$, used here is from a derivation based on Kays' general formula [30]. It was calculated for a normal range of contraction area ratios for capillary tubes with slightly beveled entrances.

The equation of motion for steady flow in a constant cross-section area pipe has the following form:

$$\rho \cdot A \cdot V \cdot dV + A \cdot dP + \tau \cdot S \cdot dL = 0 \quad (63)$$

The skin friction coefficient, τ , can be expressed in terms of the friction coefficient, f , and the velocity head:

$$\tau = \frac{1}{2} f \cdot \rho \cdot V^2 \quad (64)$$

The flow velocity term, V , can be eliminated by means of the equation of continuity:

$$d(V \cdot \rho) = 0 \quad (65)$$

Substituting and rearranging, the equation of motion assumes the following form:

$$\left(\frac{A}{m_r}\right)^2 \cdot \int_{P_i}^{P_i + 1} \rho \cdot dP + \frac{2}{D} \int_{L_i}^{L_i + 1} f \cdot dL + \ln \frac{\rho_i}{\rho_i + 1} = 0 \quad (66)$$

As discussed before, flow in a capillary tube, in the general case, can be subdivided into two parts separated by a flash point: the liquid flow part and the two-phase mixture part. The same equation will be applicable for both flows though in the case of liquid flow, it can be simplified on grounds of incompressibility. In fact, it reduces to the Fanning pressure drop formula in the following form:

$$\Delta P = \frac{2f \cdot G^2 \cdot L}{\rho \cdot D} \quad (67)$$

where the friction factor, f , can be evaluated by the empirical formula:

$$f = 0.046 \cdot Re^{-0.2} \quad (68)$$

for the Reynolds number, Re , greater than 2000 [22].

For the two-phase mixture flow, equation (66) has to be solved in its full form. This was done by Whitesel [31,32] for refrigerants 12 and 22, but with significant oversimplified approximations for the refrigerant properties. In

solving equation (42), difficulty arises with evaluating the first term because it depends directly on the pressure-density relation along the path of flow. However, the relation can be obtained by considering the adiabatic flow case. The specific volume at a given pressure can be expressed in terms of the property values for saturated liquid and vapor and in terms of quality:

$$v = v_L + x(v_V - v_L) \quad (69)$$

where v = specific volume
 x = quality
 Subscripts L and V are for liquid and vapor, respectively

The quality of the flow in the Fanno path can be found as explained in Appendix C. Thus integration of refrigerant density over a given pressure interval can be done numerically. Still another problem is faced in evaluating the second term of equation (66), which includes a two-phase friction factor as a function of tube length. Erth [33] made an effort to correlate two-phase average friction factor in a capillary tube for refrigerant 12 and refrigerant 22. His regression analysis, based on four sets of data from four different experiments, yielded the following correlation for a two-phase flow mean friction factor, f_m , as a function of the inlet conditions only:

$$f_m = \frac{0.775}{Re^{0.5}} \exp [(1 - x_i^{0.25})/2.4] \quad (70)$$

where x_i = quality of refrigerant entering capillary tube

$$Re = \frac{G \cdot D}{\mu_L + x_i (\mu_V - \mu_L)} \quad (71)$$

Using this mean friction factor, f_m , equation (66) may be written for the two-phase portion of the flow in the following form:

$$m_r = A \left[\frac{\frac{P_3}{P_2} - \int_{P_2}^{P_3} \frac{1}{\rho} dP}{\frac{2}{D} f_m \cdot (L_3 - L_2) + \ln (\rho_2/\rho_3)} \right]^{0.5} \quad (72)$$

where the numbers used as subscripts denote location consistent with figure 14.

A subroutine, CAPIL, modeling a constant flow area expansion device is based on the equations presented above. These equations have to be solved in a highly iterative process since choking pressure, friction factor, fraction of capillary tube length with liquid and two-phase flow, and the velocity head used to correct enthalpy are functions of refrigerant mass flow rate which has to be determined. Additional information about the subroutine CAPIL is given in Appendix F.

5.4 Modeling of an Evaporator and a Condenser

5.4.1 Modeling Methodology

There are two heat exchangers in a heat pump: an indoor coil and an outdoor coil. Both coils are made in a similar way and both serve as an evaporator or condenser depending on the heat pump operation mode. A schematic of a typical heat pump heat exchanger is shown in figure 19. It consists of a set of finned tubes connected in a specifically designed circuit configuration. The refrigerant flows through the tubing while air flows over the outside of the coil. Various schemes of circuiting the tubes together can be used. An example of the coil circuitry is illustrated in figure 20.

The tube-by-tube modeling technique is applied here to model the coil. This technique depends on imaginary isolation of one tube with appropriate fin surfaces from the coil assembly and calculating the performance independently. The heat transfer to and from a tube is calculated with the aid of the heat exchanger cross-flow theory. Input for calculations consists of finned tube design data, refrigerant and air mass flow rates, and inlet refrigerant and air thermodynamics states. These are uniquely evaluated by the model for each particular tube. Performance calculations are conducted for each tube independently in proper sequence and their summation results in total coil capacity.

In order to perform heat transfer calculations, four surfaces associated with the tubes must be defined. Following Carrier and Anderson [34], it was assumed that the fin area served by each tube is equivalent in performance to a circular-plate fin of equal area. Thus a single tube is considered with a circular fin of diameter, D_t , as shown in figure 21.

5.4.2 Heat Transfer Rate for a Tube in a Cross-Flow Arrangement

Usually a heat pump coil employs some form of cross-flow arrangement. If a separate tube is considered, the problem is one of pure cross flow. Fortunately, this kind of arrangement has received much attention in theoretical investigations. According to the general heat transfer equation:

$$Q = U \cdot A_h \cdot \Delta T \quad (73)$$

where A_h = heat transfer surface area
 U = overall coefficient of heat transfer
 ΔT = temperature difference

In the case of a pure cross-flow arrangement with changing temperatures of both fluids during heat exchange, the following equation for mean temperature difference between fluids applies [16]:

$$\Delta T_m = \frac{t_2 - t_1}{\frac{T_1 - T_2}{\ln \frac{t_2 - t_1}{T_1 - T_2} + \ln \frac{T_2 - t_1}{T_1 - t_1}}} \quad (74)$$

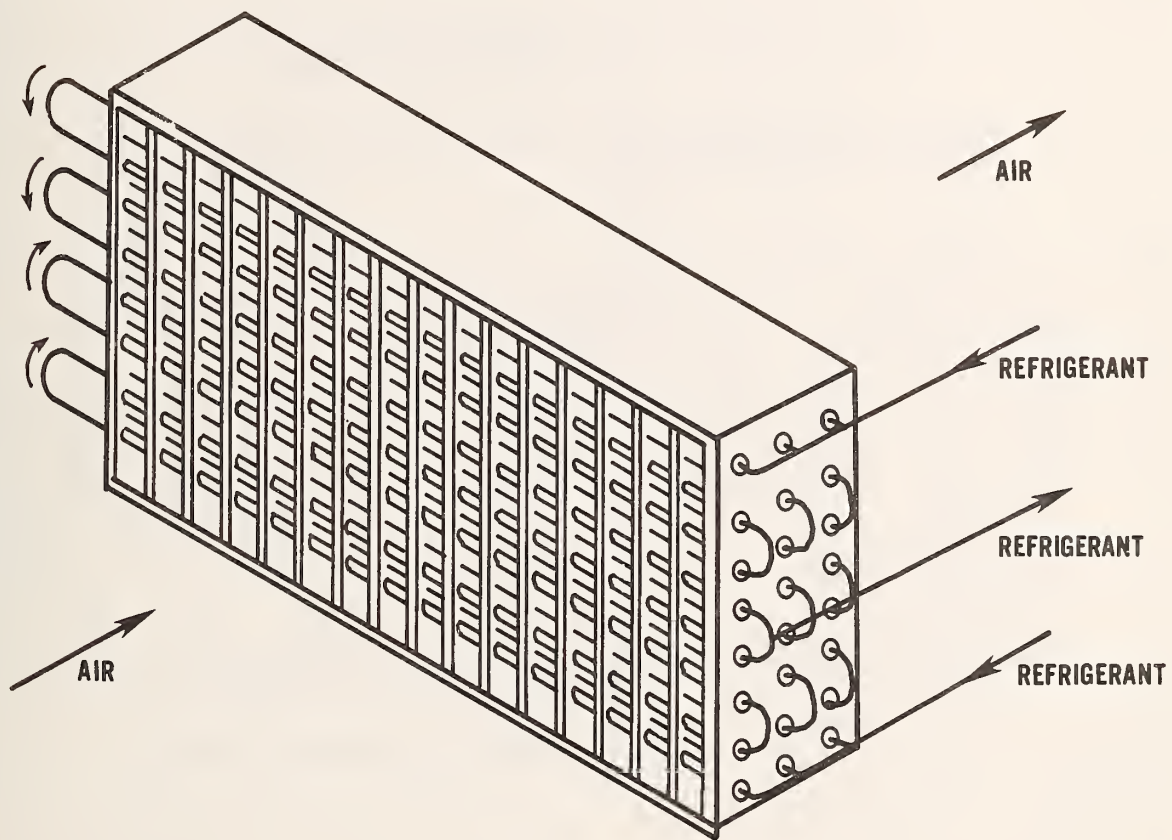
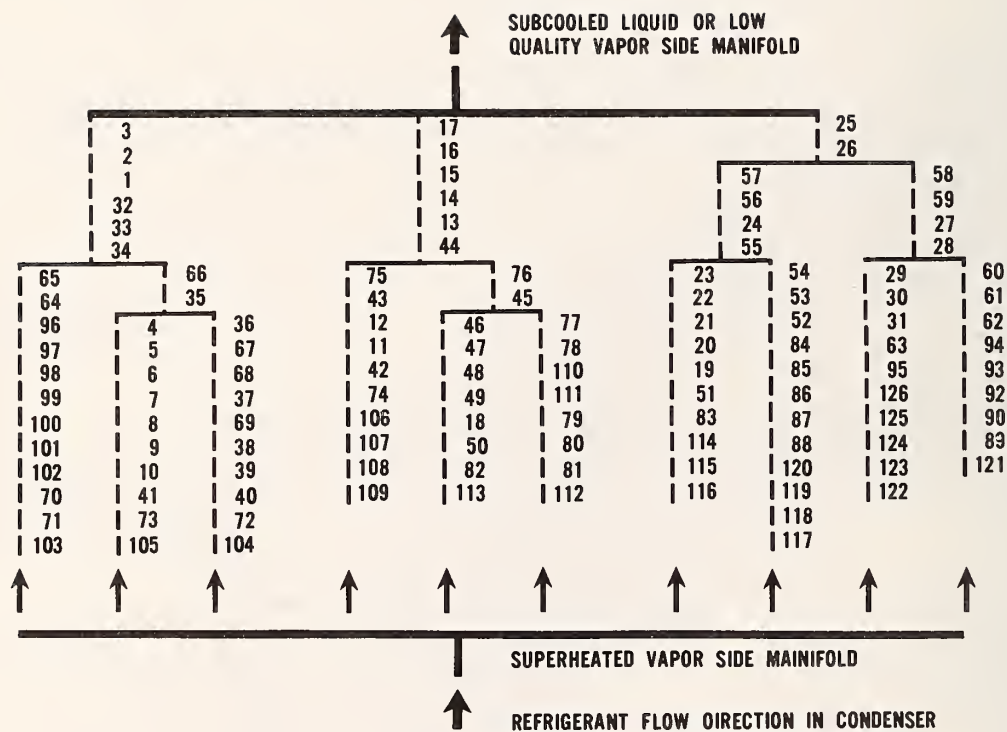


Figure 19. Schematic of a heat pump heat exchanger.



Numbers in the figure represent a location of the particular tube counting them left to right in each depth row starting with the row facing the incoming air.

Figure 20. Example of coil circuitry.

$$D_t = 2 \left(\frac{d1 \cdot d2}{\pi} \right)^{0.5}$$

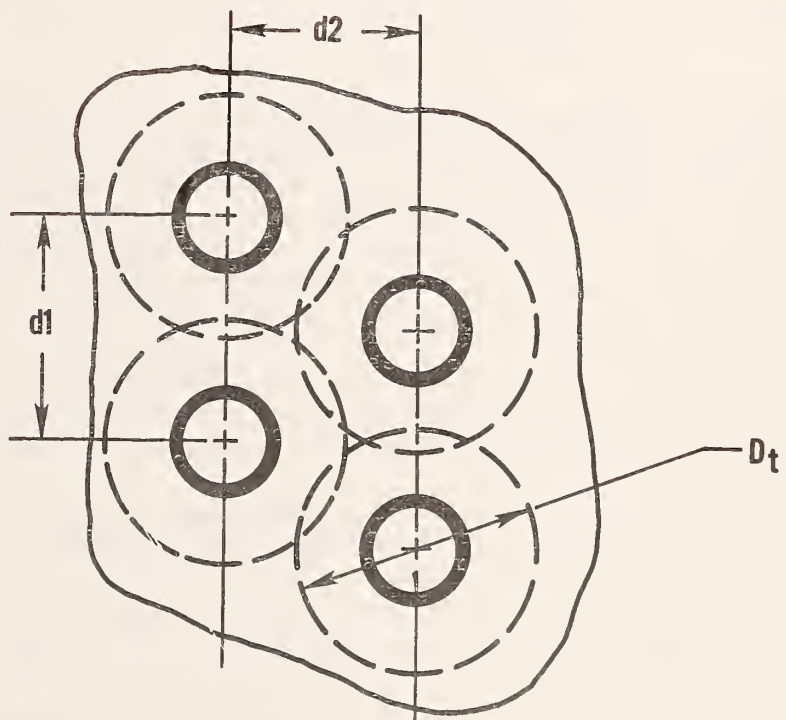


Figure 21. Approximation method for treating a rectangular-plate fin of uniform thickness in terms of a flat circular-plate fin of equal area.

where T = temperature of one fluid
 t = temperature of another fluid
 ΔT_m = mean temperature difference
subscripts 1 and 2 refer to tube inlet (upstream) and outlet (downstream) conditions, respectively.

The heat exchanged with the fluid can be calculated by the equation:

$$Q = m \cdot (i_2 - i_1)$$

or

$$Q = m \cdot C_p \cdot (T_2 - T_1) \quad (75)$$

where C_p = average specific heat of fluid at constant pressure
 i = enthalpy
 m = mass flow rate

Looking at any vapor compression cycle P-h or T-s diagram, it can be realized that both single-phase and two-phase refrigerant flow usually exists in a given heat exchanger. Also, both flow patterns can actually exist in one tube. That means that the rate of change of temperature of refrigerant flowing in the tube will not be uniform over tube length. Not only the mean temperature between fluids is affected by the flow pattern inside the tube but also the refrigerant pressure drop.

Equations presented below, derived from equations (73), (74), and (75), allow for detailed consideration of these problems. The heat transfer rate for each mentioned flow condition can be calculated as follows:

- single-phase or two-phase flow only, refrigerant is superheated, subcooled or in two-phase at both inlet and outlet

$$Q = C_{pr} \cdot m_r (T_{r,i} - T_{a,i}) \left(1 - \exp \left(- \frac{C_{pa} \cdot m_a}{C_{pr} \cdot m_r} \left(1 - \exp \left(\frac{-U \cdot A_o}{C_{pa} \cdot m_a} \right) \right) \right) \right) \quad (76)$$

- superheated vapor at tube inlet, two-phase at the tube outlet

$$Q = m_r (i_{r,i} - i_{r,v}) + C_{pr} \cdot m_r \cdot (T_{r,v} - T_{a,i}) \quad (77)$$

$$\left(1 - \exp \left(\frac{C_{pa} \cdot m_a (1 - Z_V)}{C_{pr} \cdot m_r} \left(1 - \exp \left(- \frac{U \cdot A_o}{C_{pa} \cdot m_a} \right) \right) \right) \right)$$

where Z_V = fraction of the tube length in the superheated region which can be calculated by the equation:

$$Z_V = \frac{-Cp_r \cdot m_r \cdot \ln(1 - \frac{i_{r,i} - i_{r,V}}{Cp_r(T_{r,i} - T_{a,i})})}{Cp_a \cdot m_a(1 - \exp(-\frac{U \cdot A_o}{Cp_a \cdot m_a}))} \quad (78)$$

- two-phase at tube inlet, subcooled liquid at the tube outlet

$$Q = Cp_r \cdot m_r(T_{r,L} - T_{a,i})(1 - \exp(-\frac{Cp_a \cdot m_a(1 - Z_{tp})}{Cp_r \cdot m_r}(1 - \exp(-\frac{U \cdot A_o}{Cp_a \cdot m_a})))) + m_r(i_{r,i} - i_{r,L}) \quad (79)$$

where Z_{tp} = fraction of the tube length in the two-phase region which can be calculated by the equation:

$$Z_{tp} = \frac{-Cp_r \cdot m_r \cdot \ln(1 - \frac{i_{r,i} - i_{r,L}}{Cp_r(T_{r,i} - T_{a,i})})}{Cp_a \cdot m_a(1 - \exp(-\frac{U \cdot A_o}{Cp_a \cdot m_a}))}$$

In equations (76) through (79) the following nomenclature was used:

- A_o = total exterior surface area associated with the tube wetted by air
- Cp_a = air specific heat at constant pressure
- Cp_r = refrigerant specific heat at constant pressure (in the two-phase region the specific heat is assumed to be a ratio of enthalpy change to temperature change, i.e., $Cp_r = \Delta i_r / \Delta T_r$)
- $i_{r,i}$ = refrigerant enthalpy at the tube inlet
- $i_{r,L}$ = enthalpy of refrigerant saturated liquid
- $i_{r,V}$ = enthalpy of refrigerant saturated vapor
- m_a = air mass flow rate associated with the tube
- m_r = refrigerant mass flow rate in the tube
- $T_{a,i}$ = air temperature upstream of the tube
- $T_{r,i}$ = refrigerant temperature at tube inlet
- $T_{r,L}$ = refrigerant bubble-point temperature
- U = overall tube heat transfer coefficient

5.4.3 Refrigerant and Air Mass Flow Rates Associated with a Tube

Refrigerant Mass Flow Rate

During flow through a heat pump heat exchanger, the refrigerant undergoes a change of phase in the course of evaporation or condensation. The change of phase is associated with a dramatic change of density which affects the

velocity and pressure drop of the working fluid. In order to prevent a high pressure drop, tubes in some heat pump coils are connected to form branched circuits. An example of such coil circuitry is shown in figure 20.

Refrigerant flow direction marked in figure 20 is for the coil working as a condenser. Superheated vapor enters the vapor side manifold and is distributed into 10 tubes. On its flow path, refrigerant merges several times and finally merges in the liquid manifold to enter the liquid line. For the coil operating as an evaporator, the direction of flow is opposite to that marked in the figure. Low quality refrigerant enters the coil and the flow is subdivided into three circuits. On its way through the coil the refrigerant evaporates and splits several times on its way towards the exit where it is finally collected into one larger diameter vapor line. The mass flow rates through the particular circuits of the coil are self adjusting so the pressure at merging (splitting) tubes is the same.

To perform a simulation of a coil by the tube-by-tube method, refrigerant mass flow rate for each tube has to be known. Since total refrigerant mass flow rate supplied to the coil is known, the problem reduces to the determination of refrigerant distribution. This could be determined by the model itself, however, at expense of going through iterative calculations. Another approach was tried to determine refrigerant flow distribution. Since most of the coil total pressure drop may be expected to result from superheat vapor and two-phase flow, it was assumed that the refrigerant flow is uniformly distributed among tubes connected to the vapor side manifold and that mass flow rates in other tubes may be found by following the refrigerant path with direction of flow as marked in figure 20. The resulting distribution was checked by examining calculated refrigerant (R22) pressures at the ends of circuit branches. These pressures should be equal if the assumed distribution is correct. The maximum pressure discrepancy found was equal to 0.3 psi which represented less than a 0.2°F variation in the saturation temperature of refrigerant 22 between merging tubes. This was considered satisfactory and the method was adopted for determining refrigerant mass flow rate distribution in a coil.

Air Mass Flow Rate

Air mass flow rate was assumed to be distributed uniformly over the whole coil face regardless of the coil and fan respective locations, so each tube in particular depth row was associated with the same air mass flow.

The temperature of the inlet air for a given tube was assumed to be equal to the temperature of air exiting from the upstream tube and not to be affected by mixing with air leaving neighboring tubes.

5.4.4 Overall Heat Transfer Coefficient for a Dry Finned Tube

Dry finned tube analysis is applicable to a condenser and also to an evaporator if no dehumidification takes place. The overall heat transfer coefficient, U , for a dry finned tube can be derived by summing the individual resistances between the refrigerant and the air, [16]:

$$U = \left[\frac{A_o}{A_{p,i} h_i} + \frac{A_o x_p}{A_{p,m} k_p} + \frac{1}{h_{c,o} (1 - \frac{A_f}{A_o} (1 - \phi))} \right]^{-1} \quad (81)$$

where A_f = fin surface area
 A_o = total exterior surface area exposed to air
 $A_{p,i}$ = pipe inside surface area
 $A_{p,m}$ = pipe mean surface area
 $h_{c,o}$ = convection heat transfer coefficient at the exterior surface
 h_i = inside tube convection heat transfer coefficient
 k_p = thermal conductivity of pipe material
 x_p = thickness of pipe wall

$$\phi = \frac{T_{f,m} - T_a}{T_{f,b} - T_a}, \text{ fin efficiency}$$

T_a = air temperature
 $T_{f,b}$ = fin base temperature
 $T_{f,m}$ = mean fin temperature

The second term of equation (81) can be evaluated if the heat exchanger material and geometry are known. Terms 1 and 3, which refer to the inside and outside convection resistance respectively, required considerably more analysis to establish the proper algorithm for determining the heat transfer coefficient.

5.4.5 Forced Convection Heat Transfer Inside a Tube

Analyzing the problem for both an evaporator and a condenser, the following modes of forced convection are encountered:

- single-phase forced convection
- two-phase forced convection with condensation
- two-phase forced convection with evaporation

The physics of these phenomena are very much different and forced convection in each mode have to be considered separately.

Single-Phase Forced Convection

Single-phase forced convection takes place in a condenser, at the entrance section where the superheated vapor is being cooled, and at the exit section where a subcooled liquid is being cooled. It is also applicable in the evaporator, as the superheated vapor passes through the exit tubes. The non-dimensional heat transfer parameter describing this phenomena, Nusselt number, is related to the non-dimensional Reynolds and Prandtl numbers in the following form [19]:

$$Nu = 0.023 \cdot Re^{0.8} \cdot Pr^a \quad (82)$$

where $Nu = \frac{h \cdot D}{k} = \text{Nusselt Number}$

$Pr = \frac{\mu \cdot C_p}{k} = \text{Prandtl Number}$

$Re = \frac{G \cdot D}{\mu} = \text{Reynolds Number}$

$\alpha = 0.3$ for cooling; 0.4 for heating
 C_p = specific heat at constant pressure
 D = inside diameter of a tube
 G = refrigerant mass flux
 h = convection heat transfer coefficient
 k = refrigerant thermal conductivity
 μ = refrigerant absolute viscosity

Two-Phase Forced Convection with Condensation

The predominant flow pattern during condensation in a heat pump condenser is annular flow with liquid refrigerant flowing on the pipe wall and vapor refrigerant flowing in the core. To the author's knowledge there are no data or correlations available in the literature on the forced convection condensation heat transfer coefficient of non-azeotropic mixtures flowing inside a tube. The only option left here is to use one of the correlations developed for condensing heat transfer coefficient for single component refrigerant. The correlation proposed by Traviss, Baron and Rohsenow [35] was chosen as the most theoretically derived, thus having most chances to provide reasonable predictions for other than the tested refrigerants 12 and 22. The correlation proposed in [35] and adopted in this modeling effort without modification is expected to overpredict condensing heat transfer coefficient for binary mixtures.

The theoretical background for the Traviss et al. correlation is as follows: the von Karman universal velocity distribution in the condensate film was assumed (like on a flat plate), pressure was calculated using the Lockhart-Martinelli method [22], and the momentum and heat transfer analogy was applied. The proposed correlation has the following form:

$$Nu = \frac{Re_L^{0.9} \cdot Pr_L \cdot F1^\beta}{F2} \quad (83)$$

where $Nu = \frac{h \cdot D}{k_L}$

h = condensation heat transfer coefficient
 D = tube inside diameter
 k_L = thermal conductivity of liquid refrigerant

$$Re_L = \frac{G(1-x)D}{\mu_L}$$

G = refrigerant mass flux

x = quality

μ_L = liquid refrigerant absolute viscosity

$$Pr_L = \frac{\mu_L \cdot Cp_L}{k_L}$$

$$\beta = 1 \text{ for } F1 \leq 1, \beta = 1.15 \text{ for } F1 > 1$$

$F1$ and $F2$ in equation (83) are dimensionless parameters expressed as follows:

$$F1 = 0.15 (X_{tt}^{-1} + 2.85 X_{tt}^{0.524})$$

$$F2 = 0.707 \cdot Pr_L \cdot Re_L^{0.5} \quad \text{for } Re_L < 50$$

$$F2 = 5 \cdot Pr_L + 5 \cdot \ln(1 + Pr_L(0.09636 \cdot Re_L^{0.585-1})) \quad \text{for } 50 < Re_L < 1125$$

$$F2 = 5 \cdot Pr_L + 5 \cdot \ln(1 + Pr_L) + 2.5 \cdot \ln(0.00313 \cdot Re_L^{0.812})$$

$$\text{for } Re_L < 1125$$

Parameter, X_{tt} , formulated by Lockhart-Martinelli [22] with the assumption of no radial pressure gradient and a smooth pipe, has the following form:

$$X_{tt} = \left(\frac{1-x}{x} \right)^{0.9} \left(\frac{v_L}{v_V} \right)^{0.5} \left(\frac{\mu_L}{\mu_V} \right)^{0.1} \quad (84)$$

Parameter X_{tt} is inversely proportional to flow quality and refers to turbulent vapor and turbulent liquid flow. Physically it is equal to the square root of the ratio of the frictional pressure drop of the liquid phase to the frictional pressure drop of the vapor phase if each of these phases was flowing alone in the tube, i.e.,:

$$X_{tt} = \left[\frac{\left(\frac{dP}{dL} \right)_L}{\left(\frac{dP}{dL} \right)_V} \right]^{0.5} \quad (85)$$

Equation (83) is applicable where conditions for annular condensation in a tube exists. Such conditions may be assumed to exist for flow qualities ranging from 0.1 to 0.9. At qualities larger than 0.9, the whole tube inner surface is not covered by a liquid film and part of the heat transfer is just that of single-phase convection. At qualities less than 0.1, flow was observed to be in the slug regime [35]. It is assumed, that in the quality range 0.0 to 0.1 and 0.9 to 1.0, the heat transfer coefficient changes linearly from a two-phase flow value to a single-phase flow value and is calculated using linear interpolation between values obtained from equations (82) and (83).

Two-Phase Forced Convection with Evaporation

Refrigerant enters an evaporator from an expansion device at a quality of about 20 percent and forms an annular flow instantly. The quality increases with the proceeding flow and the annular flow pattern is maintained until the quality reaches about 0.90, at which point refrigerant vapor has enough kinetic energy to gradually destroy the liquid layer and patches of dry wall appear.

Many experiments were performed and correlations published for calculating the forced convection evaporative heat transfer coefficient for R12 and R22, however, no data are available in the literature on non-azeotropic mixtures. Simultaneous to the development of this model, evaporative heat transfer coefficient measurements were performed at NBS on mixtures of R13B1/R152a at a variety of compositions. The results of preliminary tests with heat balances within 10% were correlated for use in this model in the following form:

$$h_{ev,m} = h * 3.22 X_{tt}^{-0.3} \quad (86)$$

where $h_{ev,m}$ = evaporative heat transfer coefficient of the refrigerant mixture
 h = forced convection heat transfer coefficient of the liquid film calculated by equation (82) for liquid flow rate equal to two-phase refrigerant flow rate in the tube.
 X_{tt} = Lockhart-Martinelli parameter, equation (84)

It is important to note that all liquid and vapor properties used in X_{tt} parameter have to be evaluated for liquid and vapor phases being in equilibrium based on mixture temperature, pressure and composition of mixture in the tube.

Predictions of the above correlation agree within accuracy of $\pm 10\%$ for 50 percent of experimental data, $\pm 20\%$ for 89% of data, and within $\pm 31\%$ for all data. This correlation is applicable for annular flow at qualities from 10% to 90%.

5.4.6 Forced Convection Heat Transfer at the Air-side of a Flat-Finned Tube

In order to evaluate the forced convection heat transfer outside a flat-finned tube (term 4 of equation (81)), the total exterior surface area, A_o , the fin area, A_f , the air-side heat transfer coefficient, $h_{c,o}$, and fin efficiency, , have to be known.

From the number of air-side heat transfer correlations available in the literature, the one proposed by Briggs and Young [36] is most applicable here. This correlation was developed after extensive tests on 18 tube banks of different fin geometry. A regression analysis of the test data for the air Reynolds number range from 1000 to 20000 yielded the following equation:

$$Nu = \frac{h_{c,o} \cdot D_o}{K_a} = 0.134 \cdot Re_a^{0.681} \cdot Pr_a^{0.333} \cdot \left(\frac{z}{y}\right)^{0.2} \cdot \left(\frac{z}{t}\right)^{0.1134} \quad (87)$$

where D_o = outside tube diameter
 $h_{c,o}$ = air-side mean convective heat transfer coefficient for dry air

k_a = air thermal conductivity

$$Pr_a = \frac{\mu_a \cdot C_{p_a}}{k_a}, \text{ Prandtl number}$$

$$Re_a = \frac{G_{\max} \cdot D_o}{\mu_a}, \text{ Reynolds number}$$

G_{\max} = air mass flux at minimum cross section

t = fin thickness

y = fin height

z = distance between adjacent fins

The geometric parameters affecting the heat transfer are illustrated in figure 22. The accuracy of equation (87) was further verified by Jones and Russell [37].

The addition of fins to the tubes greatly increases the outer heat transfer area but at the expense of decreasing the mean temperature difference between the surface and the air stream. The parameter, fin efficiency, ϕ , is used to rate the thermal effectiveness of a fin. As mentioned in Section 5.4.1, it is assumed in this study for heat transfer analysis, that each tube is served by a circular-plate fin of equivalent surface area, as in figure 21. Gardner [38] solved the differential equation for describing the temperature distribution in a circular fin and presented fin efficiency curves in terms of two parameters.

$$D_o/D_t \text{ and } y \left[\frac{2 \cdot h_c}{k_f \cdot t} \right]^{0.5} \quad (88)$$

The theoretical results are correlated well by the following equation (see also figure 23) [2]:

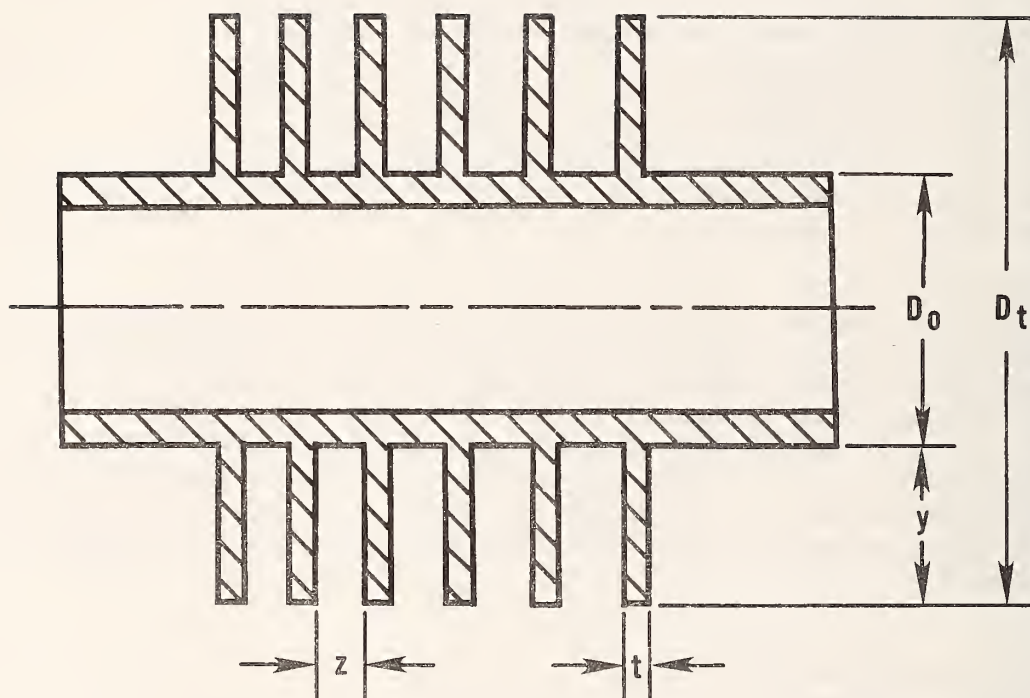
$$\phi = \sum_{i=1}^8 (A_{1,i} + A_{2,i} \frac{D_o}{D_t} + A_{3,i} \frac{D_o^2}{D_t^2}) (y \left(\frac{2 \cdot h_c}{k_f \cdot t} \right)^{0.5})^{i-1} \quad (89)$$

where h_c = air-side convective heat transfer coefficient
 k_f = fin material thermal conductivity

The geometric parameters are indicated in figure 22. The coefficients, $A_{n,i}$, are given in Table 3.

5.4.7 Overall Heat Transfer Coefficient for a Wet Finned Tube

Wet finned tube analysis is applicable to an evaporator when its temperature is below the dew point temperature of ambient air. In such a case, moisture is being removed from an air stream and is transferred to the evaporator external surface. If the evaporator temperature is above 32°F, a water film flows down the fin under force of gravity. If the exterior evaporator temperature is below 32°F, frost is accumulated.



D_o = TUBE OUTSIDE DIAMETER
 D_t = FIN TIP DIAMETER
 y = FIN HEIGHT
 t = FIN THICKNESS
 z = DISTANCE BETWEEN ADJACENT FINS

Figure 22. Cross section of a flat-finned tube indicating parameters which affect the air-side heat transfer coefficient.

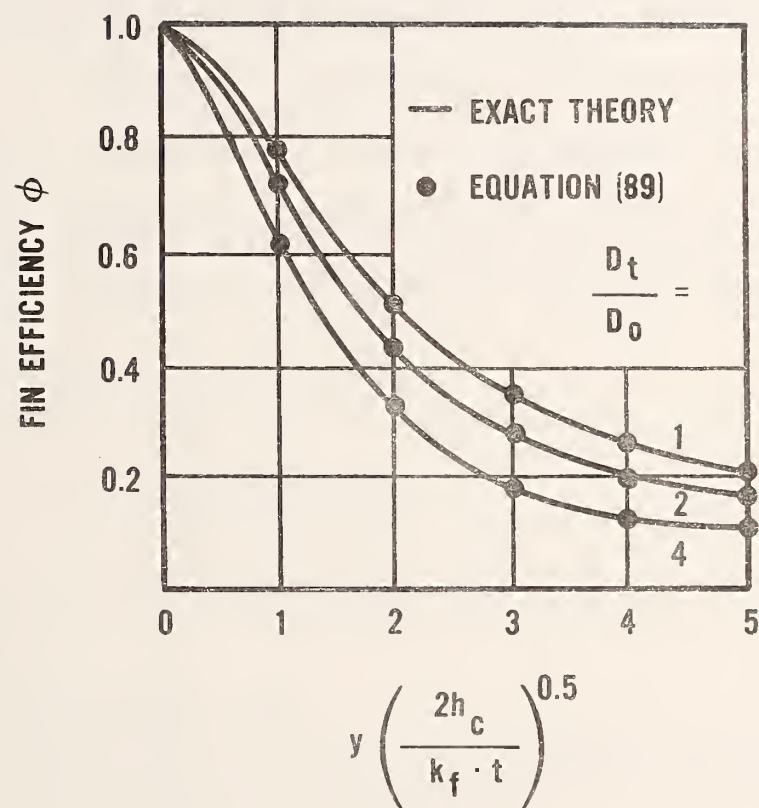


Figure 23. Efficiency for a circular-plate fin of uniform thickness. Comparison of exact theory results with those obtained by equation (89).

Table 3. Coefficients to be Used in Correlation for Fin Efficiency (Equation (89)).

i	$A_{1,i}$	$A_{2,i}$	$A_{3,i}$
1	1.0	0.0	0.0
2	-0.22920E-01	-0.13755E+00	0.20130E-01
3	0.16106E+00	0.81890E-01	-0.11440E-01
4	-0.64975E+00	-0.55500E-01	-0.28753E-01
5	0.53491E+00	0.18040E-01	0.42477E-01
6	-0.19286E+00	0.36494E-03	-0.20335E-01
7	0.32564E-01	-0.10660E-02	0.40947E-02
8	-0.20972E-02	0.12410E-03	-0.29673E-03

Considering the heat transfer from the refrigerant to the air, one can realize that the dehumidification process will alter the heat transfer situation on the external surface of the finned tube, while other processes in the tube and refrigerant stay unaffected and are governed by relations already proposed in previous sections. Thus only processes connected with dehumidification need to be discussed.

The heat transfer rate between the air stream and the water surface is described by the following equation:

$$dQ = (h_{c,o}(T_a - T_w) + h_{D,o}(w_a - w_w)i_{fg,w})dA_o \quad (90)$$

The first term accounts for sensible heat transfer and the second term accounts for latent heat transfer. For air at atmospheric pressure the Lewis number,

$$Le = \frac{h_{c,o}}{h_{D,o}Cp_a} \quad (91)$$

is close to 1 [16]. Therefore, equation (90) assumes the following form for a tube with flat fins:

$$dQ = h_c(1 - \frac{A_f}{A_o}(1 - \phi))(T_a - T_w)dA_o \quad (92)$$

where
$$h_c = h_{c,o}(1 + \frac{i_{fg,w}(w_a - w_w)}{Cp_a(T_a - T_w)})$$

Symbols used in equations (90), (91) and (92) denote:

- A_f = fin area
- A_o = total external area
- Cp_a = specific heat of air
- h_c = air-side forced convection heat transfer for wet air
- $h_{c,o}$ = air-side forced convection heat transfer coefficient for dry air
- $h_{D,o}$ = air-side mass transfer coefficient
- $i_{fg,w}$ = latent heat of condensation for water (frost sublimation)
- T_a = temperature of liquid water (frost)
- Q = heat transfer rate
- w_a = humidity ratio of air
- w_w = humidity ratio of saturated air at T_w temperature

$$\phi = \frac{T_{w,m} - T_a}{T_{w,b} - T_a}, \text{ fin efficiency}$$

- $T_{w,m}$ = mean temperature of water film (frost)
- $T_{w,b}$ = temperature of water film (frost) at fin base

The one-dimensional heat conduction across the condensate (frost) film can be expressed by the equation:

$$dQ = h_L \cdot \Delta T_L \cdot dA_O \quad (93)$$

where $h_L = \frac{k_w}{\delta}$, heat transfer coefficient for the condensate (frost) film

k_w = thermal conductivity of water (frost)

δ = thickness of condensate (frost) film (for evaluation of water and frost properties see Appendix B)

ΔT_L = temperature difference across the condensate (frost) film

Using equations (92) and (93) and referring to equation (81), the following relation for overall heat transfer coefficient for a wet finned tube can be derived.

$$U = \left[\frac{A_O}{h_i A_{p,i}} + \frac{A_O x_p}{A_{p,m} k_p} + \frac{1}{h_L} + \frac{1}{h_{c,o} \left(1 + \frac{i_{fg,w}(w_a - w_w)}{C_{p_a}(T_a - T_w)} \right) \left(1 - \frac{A_f}{A_O}(1 - \phi) \right)} \right]^{-1} \quad (94)$$

where symbols used are defined as in equations (81) and (92).

In the above formulation of the overall heat transfer coefficient, it is assumed that the temperature difference across the liquid film (frost) is uniform. The calculated value for wet fin efficiency is affected by the change of the air-side heat transfer coefficient caused by the air passage geometry alteration by liquid (frost) accumulation, and by the released latent heat of condensation (sublimation). The effect of water (frost) conductance on fin efficiency is neglected. In summary, the heat transfer phenomena that occurs during dehumidification on the air-side may be itemized as follows:

- (1) the layer of wet (frost) offers additional heat flow resistance
- (2) the air-side heat transfer resistance is decreased due to effect of condensation.
- (3) the air-side heat transfer coefficient, h_c , has an increased value since it is sensitive to external surface geometry and the Reynolds number (see equation (87)).
- (4) fin efficiency decreases as h_c is increased (see figure 23).
- (5) the cross sectional area of the air flow passage between the fins has decreased, decreasing the flow rate.

In order to evaluate water (frost) layer thickness, consider the mass transfer equation:

$$m_{a,d} \cdot dw_a = - h_{D,o}(w_a - w_w) dA_O \quad (95)$$

For the Lewis number equal to 1 equation (95) assumes the following form:

$$m_{a,d} \cdot dw_a = - \frac{h_{c,o}}{Cp_a} (w_a - w_w) \cdot dA_o \quad (96)$$

The change in the air humidity ratio can be calculated by integrating equation (96), which yields:

$$w_{a,e} = w_{a,i} - (w_{a,i} - w_w) \left(1 - \exp \frac{-h_{c,o} \cdot A_o}{Cp_a \cdot m_{a,d}}\right) \quad (97)$$

The rate of moisture removal per unit area, R, can now be calculated:

$$R = m_{a,d}(w_{a,i} - w_{a,e})/A_o \quad (98)$$

where $m_{a,d}$ = mass flow rate of dry air
 $w_{a,e}$ = humidity ratio of air at tube row exit
 $w_{a,i}$ = humidity ratio of air at tube row inlet

If the evaporator temperature is below the freezing point, moisture removed from the air stream accumulates on the evaporator external surface in the form of frost. Its thickness, δ_f , can be evaluated by integrating the rate of moisture removal with respect to time, i.e.,:

$$\delta_f = \int_0^t \frac{R}{\rho_f} dt \quad (99)$$

where t = time
 R = rate of moisture removal per unit area
 δ_f = frost layer thickness
 ρ_f = frost density

In case of evaporator temperature above 32°F, condensate flows down on the fin. Assuming no air drag on the liquid layer, its local velocity is expressed by the closed solution of the Navier-Stokes equation for a viscous flow on a vertical wall [39]:

$$V_z = \frac{\rho g \delta^2}{2\mu} \left[1 - \left(\frac{y}{\delta}\right)^2\right] \quad (100)$$

where V_z = local liquid layer velocity
 ρ = liquid density
 g = gravitational acceleration
 y = distance from the wall
 δ = liquid layer thickness
 μ = liquid absolute viscosity

Applying to the liquid film the continuity equation:

$$m(z) = \rho \int_0^{\delta} V_z dy \quad (101)$$

and assuming uniform condensation rate on the fin (i.e., $m(z) = m * z/h$, where: $m(z)$ = mass flow rate of condensate at elevation z , m = water condensation rate by a fin of height h), the average condensate layer thickness can be obtained by integrating a local layer thickness over the fin height and dividing the obtained expression by the height. The resulting expression is:

$$\delta_f = 1.082 \left[\frac{\mu_w \cdot R'}{g \cdot \rho_w^2} \right]^{1/3} \quad (102)$$

where g = gravitational acceleration
 R' = condensation rate per unit width of a fin
 μ_w = water dynamic viscosity
 ρ_w = water density

5.4.8 Pressure Drop in a Tube

As expressed by equation (50), the total pressure drop experienced by a flowing fluid results from pressure drops due to friction, momentum change, and gravity. In an actual heat pump heat exchanger, pressure drop due to gravity effect is very small and may be neglected. Only pressure drop due to friction and due to momentum change will be considered for the different flow patterns in a tube.

Single-Phase Flow

Frictional pressure drop for a single-phase turbulent flow in a tube can be calculated by the Fanning equation with the Fanning friction factor, equations (67) and (68):

$$\Delta P = \frac{2f \cdot G^2 \cdot L}{D \cdot \rho} \quad (103)$$

$$f = 0.046 Re^{-0.2} \quad (\text{for } Re > 2000, [22]) \quad (104)$$

Pressure drop due to momentum change can be calculated by the following equation:

$$\frac{dP}{dL} = - G^2 \frac{dv}{dL} \quad (105)$$

where G = refrigerant mass flux
 L = tube length
 v = refrigerant specific volume

Two-Phase Flow with Condensation

The frictional pressure drop for two-phase flow with condensation can be calculated by the method proposed by Lockhart and Martinelli [22]. They

performed a semi-empirical study of adiabatic two-phase flow with air and different liquids including benzene, kerosene, water, and various oils in tubes varying in diameter from 0.586 to 1.017 inch. They related the pressure drop of two-phase flow to the pressure drop of the liquid portion of the flow flowing alone in the pipe, by a dimensionless parameter X_{tt} , i.e.,:

$$\frac{\Delta P_{tp}}{\Delta P_L} = f(X_{tt}) = \Phi \quad \text{or} \quad \Delta P_{tp} = \Delta P_L \cdot \Phi \quad (106)$$

where ΔP_L = frictional pressure drop of the liquid portion of two-phase flow flowing alone in the tube
 ΔP_{tp} = frictional pressure drop of two-phase flow
 Φ = correction factor for two-phase pressure drop
 X_{tt} = as given by equation (84)

The pressure drop, ΔP_L , is calculated by the single-phase pressure drop relation with the liquid Reynolds number and friction factor calculated as follows:

$$Re_{tp,L} = \frac{(1-x)G \cdot D}{\mu_L} \quad (107)$$

$$f_{tp,L} = 0.046 \cdot Re_{tp,L}^{-0.2} \quad (\text{for } Re_{tp,L} > 2000) \quad (108)$$

where $f_{tp,L}$ = friction factor for the liquid portion of two-phase flow flowing alone in the pipe
 $Re_{tp,L}$ = Reynolds number for the two-phase liquid portion flowing alone in the pipe
 x = quality
 μ_L = liquid dynamic viscosity

A correction factor for two-phase pressure drop, Φ , was correlated by the following equation:

$$\Phi = \exp \left(\sum_{i=1}^5 A_i \cdot X_{tt}^{-0.25 \cdot i} \right)^2 \quad (109)$$

where $A_1 = -0.418956$
 $A_2 = 1.47330$
 $A_3 = 0.668583$
 $A_4 = -0.321168$
 $A_5 = 0.0408167$

Combining equations (106), (107), (108), and (109), the two-phase pressure drop equation assumes the form:

$$\Delta P_{tp} = 2 f_{tp,L} \cdot G^2 (1-x)^2 L \cdot \Phi / (D \cdot \rho_L) \quad (110)$$

The pressure drop due to momentum change for separated two-phase flow can be estimated by the following equation:

$$\frac{dp}{dL} = - G^2 \frac{d}{dL} \left(\frac{v_V \cdot x^2}{\alpha} + \frac{v_L(1-x)^2}{(1-\alpha)} \right) \quad (111)$$

where. x = quality
 v_L = specific volume of liquid
 v_V = specific volume of vapor
 α = void fraction

Void fraction, α , percent of tube filled with vapor, was shown by Lockhart and Martinelli to be a function of X_{tt} under any flow conditions for separated flow with both phases turbulent. Wallis [40] correlated their results in the following form:

$$\alpha = (1 + X_{tt}^{0.8})^{-0.378} \quad (112)$$

This expression was found to correlate well with data presented in [22] for values of $X_{tt} \leq 10$. For X_{tt} greater than 10, another curve fitted formula is used:

$$\alpha = 0.823 - 0.157 \cdot \ln X_{tt} \quad (113)$$

Two-Phase Flow With Evaporation

The Lockhart-Martinelli method for pressure drop calculation of two-phase flow is widely used for adiabatic and condensing flows of single component refrigerants. However, it does not yield accurate prediction for evaporative flow. Instead of the Lockhart-Martinelli correlation, Anderson, Rich and Geary [41] recommended a method proposed by Pierre [42]. In order to evaluate the accuracy of this correlation for the R13B1/R152a mixture, pressure drop predictions were compared with laboratory data of evaporator tests performed in NBS environmental chambers. It was found that this correlation underpredicted pressure drop by about 40 percent in a consistent manner. Until pressure drop of non-azeotropic mixtures is fully investigated Pierre's correlation will be used for calculation of pressure drop with a correction factor of 1.4.

The correlation of Pierre based on experiments with refrigerants 12 and 22 has the following form:

$$\Delta P = \left(f \frac{L}{D} + \frac{\Delta x}{x_m} \right) G^2 \cdot v_m \quad (114)$$

where D = inner tube diameter
 L = tube length
 f = friction factor (calculated by equation (115))
 x_m = mean quality
 Δx = quality change
 $v_m = v_L + x_m(v_V - v_L)$, mean specific volume

The friction factor to be used in equation (114) was correlated by Pierre from his experimental data by the following empirical equation valid for $Re/K_f > 1$:

$$f = 0.0185(K_f/Re)^{0.25} \quad (115)$$

where $K_f = \frac{J \cdot i_{fg} \cdot \Delta x}{L}$, boiling number

$$Re = \frac{G \cdot D}{\mu_L} \text{ , Reynolds number}$$

J = mechanical equivalent of heat

The correlation proposed by Pierre is in the conventional format for the single pressure drop formula. The first term of equation (114) is for frictional pressure drop while the second is for pressure drop due to change of momentum. The formula for the friction factor contains the Reynolds number divided by the boiling number, making the friction factor sensitive to vapor generation rate at the vapor-liquid interface.

5.5 Modeling of Additional Heat Pump Components

In the previous sections, modeling of the main heatpump components has been discussed. The analysis included a compressor, a constant flow area expansion device, a condenser, and an evaporator. Additional components that have to be considered for a more accurate heat pump performance prediction by the model are: the vapor suction and discharge lines, the liquid line, and the four-way valve. Their modeling is briefly explained below.

Vapor Lines

A suction line connects the evaporator with the compressor. A compressor and a condenser are connected by a discharge line. Usually, single-phase flow exists in the vapor lines in the form of either saturated or superheated vapor. Heat transfer rates between the refrigerant vapor flowing in the vapor lines and ambient air can be calculated by a general heat transfer equation for a circular duct with insulation [19]. Single-phase forced convection is assumed inside the tube (equation (82)) and free convection is assumed outside the tube. The following equation is used for calculation of the free convection heat transfer coefficient for a horizontal tube [43]:

$$h = 0.27 \left(\frac{\Delta T}{D_o} \right)^{0.25} \quad (116)$$

for Grashof numbers from 10^3 to 10^9 .

where ΔT = temperature difference between tube wall and air
 D_o = tube outside diameter

Pressure drop in vapor lines can be calculated by the single-phase pressure drop equation (67).

Liquid Line

A liquid line connects the evaporator and the condenser. This line is filled with a subcooled liquid or low quality two-phase flow. The pressure drop can be calculated by equations (67) or (108), depending on the flow pattern. Heat loss from the liquid line to the ambient air is neglected.

Four-Way Valve

The main function of a four-way valve is to direct refrigerant flow from the compressor to the indoor or outdoor coil depending on the mode of operation (heating or cooling). The side effects of flow through a four-way valve are changes in the refrigerant thermodynamic state due to the pressure drop and heat exchange. Assuming for simplicity an adiabatic exterior wall, all heat lost by the discharge refrigerant is gained by the suction refrigerant. The heat transfer rate and refrigerant pressure drop in the valve can be evaluated by formulas similar to equations (45) and (55), respectively. Heat transfer and pressure drop parameters in these formulas have to be found using a subroutine VALPAR and a four-way valve (heat pump) one test data as explained in Appendix E.

Accumulator

Simplified schematic of an accumulator is shown in Figure 8. It is assumed that refrigerant experiences no pressure drop while flowing through the accumulator, and the accumulator is adiabatic. With these assumptions the refrigerant state at the accumulator inlet is equal to the state at the outlet during steady-state operation. The main purpose of the simulation of the accumulator is the calculation of mass of refrigerant contained in it. If superheated vapor is entering the accumulator, the accumulator is filled with vapor and calculation is straight forward. If wet vapor is entering the accumulator, refrigerant liquid is collected in it and the liquid level has to be evaluated. The liquid level is found by evaluating the hydrostatic pressure that along with the dynamic pressure exerted at the oil return hole would cause refrigerant liquid to flow through the hole at the rate that would change saturated vapor in the accumulator tube into a wet vapor of the inlet vapor quality.

The basic equations used in the accumulator subroutine, WACCUM, are:

$$m(1 - x) = A \cdot K \cdot \sqrt{2 \cdot \Delta P \cdot \rho_L} \quad (117)$$

$$\Delta P = H \cdot \rho_L \cdot g + 0.5 \left(\frac{x \cdot m}{A} \right)^2 v_v \quad (118)$$

where K = orifice flow coefficient assumed $K = 0.585$ [44])
 A = tube cross section area
 H = liquid refrigerant level in the accumulator
 g = gravitational acceleration
 m = refrigerant mass flow rate through the accumulator
 Δp = pressure drop through the oil return hole
 v_v = specific volume of vapor
 x = refrigerant quality
 ρ_L = liquid density

Equation (117) is the orifice equation applied to the oil return hole. In equation (118) the first term represents the hydrostatic pressure while the second term expresses the dynamic pressure exerted at the hole (velocity of the liquid is neglected).

Some accumulators have two holes in the accumulator suction line. In addition to the regular oil return hole at the bottom of the suction line, they have another hole located some distance (approximately 1.5 inch) above. The accumulator subroutine is capable of simulating both types of accumulators. In either case the subroutine is solving for the liquid level, H, using equations (117) and (118).

5.6 Refrigerant Mass Inventory in a Heat Pump

The mass, M, of a substance occupying a known volume, V, may be determined by:

$$M = \int_V \rho \cdot dV \quad (119)$$

where ρ = local density

In reality, the mass of refrigerant in the system can be found by estimating the masses of the refrigerant in each system component and adding them up. For this purpose, equation (119) can be written in the form:

$$M = \sum M_i \quad (120)$$

$$M_i = V_i \cdot \rho_{m,i}$$

where M_i = mass of refrigerant in particular component i
 V_i = internal volume of component i
 $\rho_{m,i}$ = mean fluid density in component i

In order to make mass inventory of individual mixture components, the following equation can be applied with the known mixture weight composition, XW, of the considered refrigerant:

$$M_{i,I} = M_i \cdot XW \quad (121)$$

where $M_{i,I}$ = mass of refrigerant 'I' in heat pump component, 'i'

The heat pump components taken into account in the mass inventory calculations are shown in figure 24. The refrigerant phase in each of the components as indicated in the figure, is based on the following considerations:

Discharge line - receives and is filled with superheated vapor from the compressor.

Condenser - receives superheated vapor from a discharge line. In the course of passage through the condenser tubes, vapor temperature is brought to the dew point temperature. Starting at this point, a thin condensed liquid layer forms on the tube walls. Depending on the mass flux, this liquid film may be swept and entrained within the vapor as a mist forming a dispersed flow. With more condensed vapor,

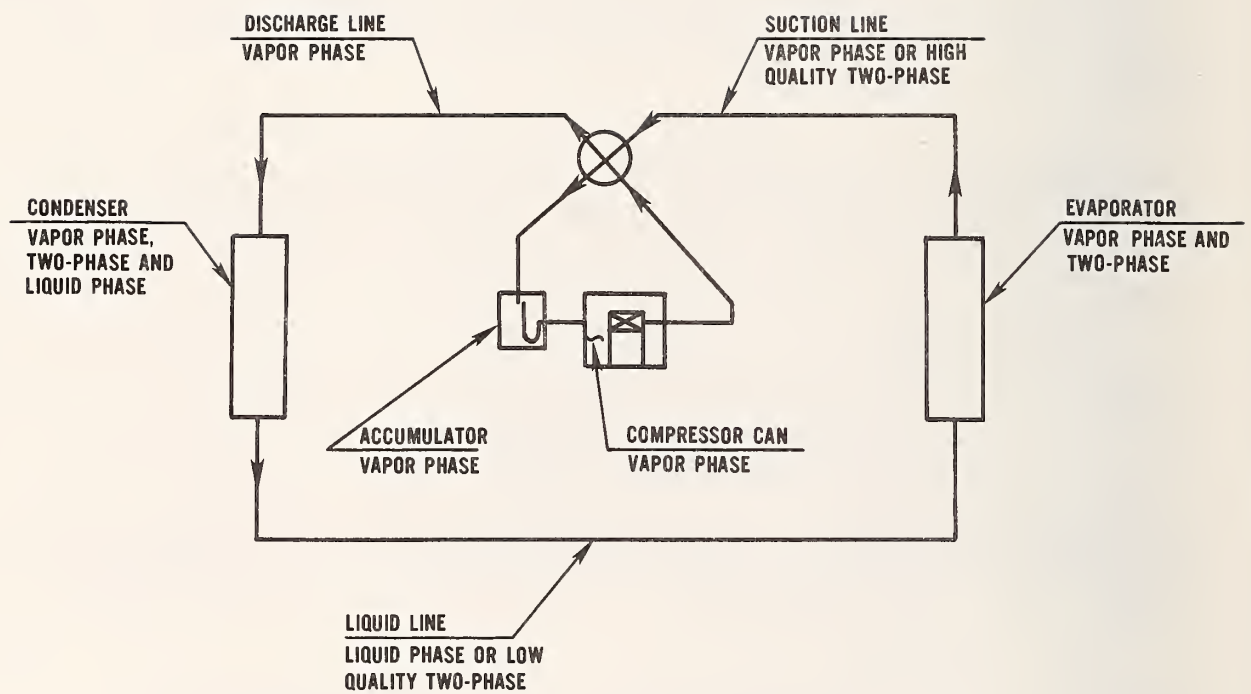


Figure 24. Refrigerant phase in the heat pump components.

the velocity of vapor decreases, a permanent liquid film is formed on the wall, and the flow proceeds further as the annular or semiannular flow. The quality decreases and the flow slows down in the direction of flow. Gravity forces result in a stratifying effect and the liquid flowing in the bottom of the tube periodically wets the upper tube wall (slug flow). If full condensation is reached, there will be subcooled liquid at the condenser exit.

Thus there are four two-phase flow patterns in the condenser, i.e., dispersed flow, annular flow, semi-annular flow, and slug flow as identified by Traviss and Rohsenow [45]. Soliman and Azer [46] identified nine flow regimes, however, five of them may be considered as transition regimes and can be put into the above basic four categories. The effect of return bends connecting condenser tubes was investigated in [47]. It was found that this effect is insignificant and that the annular flow pattern 'recovers' almost immediately in a tube after a return bend.

Liquid line - receives refrigerant from a condenser and delivers it to an expansion device. Entering the expansion device at the end of the liquid line, the flow experiences a sudden contraction. The entrance pressure drop depends upon the mass flow rate and the fluid density. This, along with the capillary tube sensitivity to the amount of subcooling, provides a flow controlling mechanism for the system. It is common practice to design a capillary tube so the refrigerant mass flow rate through it would balance with the mass flow rate through the compressor when a liquid seal exists at the capillary tube entrance. Temporary existence of vapor at the capillary tube would decrease the mass flow rate and build up the pressure. This would provide better conditions for condensation in the condenser and thus cause a return to a subcooled liquid or very low quality two-phase flow in the liquid line.

Expansion device - due to its small internal volume it may be disregarded for mass inventory purposes.

Evaporator - receives refrigerant from a capillary tube in the form of homogeneous flow of about 20 percent quality, at sonic or close to sonic velocity. This flow experiences a sudden deceleration and is assumed to form an annular or semi-annular flow very quickly having a void fraction about 0.8 [48]. The liquid phase flows along the tube in the form of an annulus and the vapor phase flows in the core. This type of flow prevails in most of the evaporator. With increasing quality, the velocity of the vapor increases and becomes eventually high enough to entrain small liquid drops and tear the liquid film apart. This results in a mist flow at qualities close to 1. Finally, the flow may reach the evaporator exit in a superheated vapor state.

Suction line - receives refrigerant from the evaporator as a mist flow or superheated vapor flow and the same flow pattern exists at the exit.

Accumulator - receives the refrigerant from a suction line after some heat is added in the four-way valve. Superheated, saturated or wet vapor may be expected. With wet vapor entering, liquid will be accumulated on the accumulator sump.

Compressor can - receives refrigerant from the accumulator as a saturated or superheated vapor which is further superheated due to heat transfer from an electric motor, discharge line, and compressor cylinder body.

According to the above analysis, refrigerant in the compressor can, suction line, and discharge line is in (or close to) a homogeneous vapor state. In the liquid line, homogeneous liquid flow or two-phase flow exists but in a state so close to saturation that for simplicity of mass inventory no slip is assumed and the flow is considered to be homogeneous. The mass of refrigerant in these components can be then calculated straight forward in a similar manner since refrigerant densities in the components described above will be known as a result of the simulation program iteration process.

In order to calculate the mass of refrigerant in the accumulator, it is necessary to determine the portion of the internal volume filled with liquid. Once this is found, total refrigerant mass in the accumulator can be calculated based on refrigerant state parameters.

In the case of both coils, the refrigerant flow is partly homogeneous, however, in most of the coil some type of annular flow prevails. For a separated flow regime, like an annular flow, density of the flowing fluid is determined from vapor and liquid densities and the fractions of tube volume occupied by the liquid and vapor phase:

$$\rho_m = \frac{\rho_L \cdot V_L + \rho_V \cdot V_V}{V_t} = \rho_L(1 - \alpha) + \rho_V \cdot \alpha \quad (122)$$

where M = mass
 V = occupied volume
 ρ = density
 $\alpha = V_V/V_t$, void fraction

subscripts refer to:

L = liquid phase
 m = mean value
 t = total value
 V = vapor phase

It should be pointed out that for a two-phase flow of a given quality, mean density depends not only on the thermodynamic parameters affecting densities of the liquid and vapor, but also on the ratio of mean velocities of the vapor and liquid referred to often as the slip ratio. This can easily be noticed by examining the equation for the void fraction, which can be derived, considering mass flow rates of each phase:

$$\alpha = \frac{1}{1 + \frac{1 - x}{x} \frac{V_V \rho_V}{V_L \rho_L}} \quad (123)$$

where V = velocity

Calculation of the void fraction has received much less attention than the calculation of heat transfer or pressure drop, although all three of these quantities are undoubtedly inter-related. The most often referenced method for void fraction calculation is that of Lockhart and Martinelli [22] based on their experimental void fraction data reported along with pressure drop results. They correlated void fraction with the dimensionless parameter, X_{tt} . Their experiment dealt with adiabatic flow, however, in the conclusions of their paper they suggested that their pressure drop correlation could be used for prediction of pressure drop during evaporation and condensation as well. On similar grounds, their void fraction data should be applicable beyond the adiabatic case.

There is another, earlier, method available for void fraction calculation by Martinelli and Nelson [49], derived with the same assumptions for water/steam evaporating flow. The method is similar to the Lockhart-Martinelli method so the Lockhart-Martinelli method was used in this study for calculation of the void fraction in both the evaporator and the condenser coils.

Correlations for void fraction based on [22] have already been given in section 5.4.8 (equations (112) and (113)).

$$\alpha = (1 + X_{tt}^{0.8})^{-0.378} \quad \text{for } X_{tt} \leq 10 \quad (124)$$

$$\alpha = 0.823 - 0.157 \ln X_{tt} \quad X_{tt} > 10 \quad (125)$$

The parameter X_{tt} , as defined by equation (84), is a function of flow quality, specific volume, and the viscosities of the liquid and vapor, and is not sensitive to the mass flux which in turn affects the slip ratio.

Experimental data of Staub and Zuber [48] for evaporating refrigerant 22 indicate that void fraction increases with increased mass flow rate. Comparison of their data with the void fraction predicted by the Martinelli-Nelson method showed a discrepancy which may lead to underestimation of the two-phase flow mean density by as much as 300 percent. (Note that a small difference in void fraction results in a large error in mean density prediction as density of the vapor and liquid are vastly different.) The results of Staub and Zuber also indicate that approximate agreement with the Martinelli-Nelson could be obtained at high mass fluxes, more than five times higher than those observed in usual heat pump evaporators.

In discussing the accuracy of a refrigerant mass prediction method, it should be noted that it also depends upon an accurate measurement of the internal volume of the system. Internal volume can be easily calculated for straight pipes. For valves, bends, etc., it can only be approximated.

Mass inventory of a heat pump system, as explained in Section 5.1, is intended to be used for iteration of refrigerant superheat at the compressor can inlet. The discussions above indicated that mass inventory calculations may have errors from two different sources. These sources are: inaccurate density prediction in the two-phase region, and inaccurate internal volume measurement. However, effect of these inaccuracies can be diminished since the program does not really need prediction of the absolute value of refrigerant mass in the system, but rather requires sensitivity in relative mass predictions in a heat pump with changes of operating conditions. This requirement

should be satisfied by the Lockhart-Martinelli method and very precise knowledge of the internal volume should not be that important. Since the refrigerant vapor superheat at the compressor can inlet is known at the cooling design operating mode at conditions (usually: outdoor temperature 95°F, indoor dry bulb temperature 80°F, and wet bulb 67°F) and does not have to be iterated for solution, refrigerant mass in the system can be calculated at this operating condition. This calculated mass of refrigerant may then be used as input data for refrigerant vapor superheat/quality iteration at other outdoor or indoor air conditions at which heat pump simulation results are required.

6. MODEL VERIFICATION

A verification of the computer model has been done by comparison of computer predictions with performance data obtained from laboratory tests conducted in NBS environmental chambers. In this stage of model development, laboratory data were available on one heat pump tested at two cooling and two heating standard rating operating conditions [50].

The sequence for running laboratory tests was as follows:

1. DoE cooling test A, $T_{\text{out}} = 95^{\circ}\text{F}$
indoor conditions: 80°F DB, 67°F WB
refrigerant charge adjusted to obtain 10° superheat at evaporator outlet
2. DoE cooling test B, $T_{\text{out}} = 82^{\circ}\text{F}$
indoor conditions: 80°F , 67°F WB
3. DoE high temperature heating test
outdoor conditions: 47°F DB, 43°F WB
indoor temperature: 70°F
4. DoE low temperature heating test
outdoor conditions: 17°F DB, 15°F WB
indoor temperature: 70°F

The laboratory data of DoE cooling Test A were used first to obtain compressor and four-way valve parameters as explained in Appendix E. Then computer model runs were started with simulation of performance at DoE cooling test A conditions with the appropriate imposed superheat at compressor can inlet that resulted in 10°F superheat at evaporator inlet as observed during laboratory test at this operating condition. Along with the performance results, refrigerant charge at the imposed superheat was obtained. This charge was then included in a heat pump data file to be used for iteration of superheat/quality of vapor at compressor can inlet, and composition of the circulating refrigerant at other operating conditions.

The laboratory and simulation results are given in Table 3 for the cooling mode and in Table 4 for heating operation.

During cooling operation no liquid was collected in the accumulator and the composition of the mixture did not change. In the heating mode refrigerant liquid was collected and a shift of composition of the circulating mixture occurred.

For both the cooling and heating modes, agreement between the results is very good, in fact, better than could be expected taking into account the number of simplifications used in the model, even the model is complex. It is not implied that this type of agreement can be obtained for every heat pump. It is believed that the accuracy of prediction stems partially from the effective cancellation of different simulation approximations, and this may not necessarily occur in simulation of every heat pump. However, the model

proved to provide physically consistent predictions including evaluation of refrigerant superheat/quality at the compressor can inlet and the circulating composition. Validation of this model with test data at other heat pumps charged with different mixtures, when available, would further increase confidence in the model.

Since two-phase heat transfer and pressure drop of non-azeotropic mixtures have not been fully presented in the literature and appropriate correlations are not available, the evaporative and condensation heat transfer coefficients and respective pressure drops are calculated in the model by formulas that may provide approximate values of the parameters in question. To evaluate the impact of possible approximate prediction on the final results, a few simulation runs were performed in which the specific parameter calculated within the program was altered by a preset factor. Change of calculated system capacity and COP as compared to performance obtained with unaltered parameter describes sensitivity of the model on this particular parameter.

The following results were obtained. Increase of the evaporative heat transfer coefficient by 50% resulted in 2.5% increase of system capacity and 1.8% of system COP. Decrease of the evaporative heat transfer coefficient by 50% caused reduction in capacity and COP by 6.9% and 5.4%, respectively. Increase in the condensation heat transfer coefficient by 50% increased capacity by 0.6% and COP by 1.1%. The same reduction in the condensation heat transfer coefficient caused reduction of system capacity by 2%, and reduction of COP by 3.1%.

Similar comparison was done for the evaporative and condensation pressure drops. Sensitivity of the final performance predictions was much smaller. In either considered case a change in pressure drop by 50% did not affect system capacity or COP by more than 1%.

Comparison numbers given above were obtained at indoor and outdoor air conditions specified for DoE cooling Test A. These findings should still apply for other operating conditions. Somewhat different results could be obtained simulating different systems.

Table 4. Laboratory Test and Computer Simulation Results in the Cooling Mode

Outdoor Temperature °F	Data Source	Circulating Mixture Composition	Capacity Btu/h	Energy Consumption Rate Watt	EER Btu/(h · Watt)	Refrigerant Mass Flow Rate lb/h	Evaporator Outlet Dew Point Temperature °F	Superheat Leaving Evaporator °F	Compressor Discharge		Condenser Outlet	
									Pressure psia	Temperature °F	Pressure psia	Temperature °F
95°F	Simulation	.65	28050	4359	6.43	562	50.5	9.7	304	212	291	101.6
	Test	.65	27880	4322	6.45	568	51.8	9.7	297	208	292	111.0
82°F	Simulation	.65	30000	4047	7.41	557	48.6	13.2	263	195	249	88.0
	Test	.65	29670	4048	7.33	552	47.8	14.7	252	196	247	97.2

Indoor Air Conditions - 80°DB/67°W

Table 5. Laboratory Test and Computer Simulation Results in the Heating Mode.

Outdoor Temperature	Data Source	Circulating Mixture Composition Weight Fraction of R13B1	Capacity Btu/h	Energy Consumption Watt	COP	Evaporator Outlet Temperature		Compressor Discharge		Condenser Outlet	
						°F	psia	Pressure	Temperature	Pressure	Temperature
47°F DB	Simulation	.667	35180	3673	2.81	37.1	230	220	159	220	84.9
43°F WB	Test	.667	33800	3776	2.62	40.1	231	231	176	231	93.3
17°F DB	Simulation	.743	21910	3206	2.00	11.4	211	203	150	203	80.2
15°F WB	Test	.734	21303	3298	1.89	14.1	209	195	177	195	82.1

Indoor Air Temperature 70°F

7. SUMMARY AND CONCLUSIONS

This report describes a model of a heat pump equipped with a constant flow area expansion device and operating with a non-azeotropic binary mixture refrigerant.

The model is able to simulate performance of a heat pump at imposed operating conditions without restriction on the refrigerant thermodynamic state at any point of the system. The simulation power includes simulation of the circulating mixture composition shift resulting from incomplete evaporation and accumulation of liquid refrigerant in the accumulator.

The model has a modular structure; it consists of independent models of heat pump components linked together by appropriate logic iterating refrigerant thermodynamic states in thirteen key locations of a heat pump thermodynamic cycle. Heat pump system component models were developed with emphasis on description of processes by fundamental heat transfer and pressure drop equations, and equation of state relationships among material properties. In the compressor model several refrigerant locations were identified and processes taking place between them accounted for all significant heat and pressure losses. Evaporator and condenser models were developed on a tube-by-tube basis where performance of each coil tube is computed separately by considering the cross-flow heat transfer with the external air-stream and the appropriate heat and mass transfer relationship. A constant flow area expansion device model was formulated with the aid of the Fanno flow theory.

The model utilizes an equation of state which is applicable for both liquid and vapor phases. Unlike the usual practice of using separate pressure-saturation temperature and liquid density-saturation temperature correlations with P-V-T relationship, one P-V-T relationship and physical equilibrium criteria are used in this project in establishing saturation state parameters and properties.

Since two-phase heat transfer and pressure drop of non-azeotropic mixtures have not been fully presented in the literature and appropriate correlations for their evaluation are not available, the evaporative and condensation heat transfer coefficients and respective pressure drops are calculated in the model by formulas either derived from a single component flow experiments or at best from limited experiments of one mixture only. There is certainly a need of research in these areas and this model should be updated when our knowledge about these phenomena is advanced.

The model was validated by comparing its performance predictions with laboratory test results of one heat pump charged with R13B1/R152a mixture at two cooling and heating operating conditions. The data presented in this report, established our confidence in the model. The choice of R13B1/R152a for development and verification of this model does not mean recommendation of this mixture as an optimum and precludes other mixtures to be more suitable for heat pump application.

This report is intended for research and development engineers to provide a tool for evaluation of the potential of non-azeotropic binary mixtures in heat

pumps. The User's Manual of the developed program, HPBI, is included in the Appendix Section to enable readers to prepare input data for other heat pumps and to run the program even without detailed knowledge of the main section of this report. An example run and a listing of the program are also included.

HPBI program is applicable to the standard split residential system. It is, however, limited in the form presented here to the R13B1/R152a mixture due to a saturated refrigerant property spline which was incorporated in the program to decrease required computing time. Once a more efficient scheme is developed to use the equation of state directly, the spline will be removed along with present model limitations.

The current version of the model can be used to evaluate performance of a heat pump charged with R13B1/R152a mixture at different operating conditions and different mixture compositions. It is intended to be used, after slight modification, to predict the possible potentials of mixtures working in modified thermodynamic cycles. The computing time required by the program will vary with the size of the heat exchangers and the accuracy of estimated refrigerant parameters given as inputs. On the average, HPBI program run on Sperry 1100/82 computer requires 10 minutes of CPU time to converge with iteration of superheat, and about 40 minutes to converge iterating refrigerant quality at the compressor can inlet and the circulating mixture composition.

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APPENDIX A. CALCULATION OF PROPERTIES OF MOIST AIR

For the selection of moist air property equations, it was assumed that air dry bulb temperature, T , relative humidity, RH , and pressure would be known, and the properties that would be the output were: humidity ratio, specific heat at constant pressure, gas constant, dynamic viscosity, and thermal conductivity. Presented here are the psychometric equations in their fundamental form. They are derived assuming moist air to be a mixture of two independent perfect gases, so that the perfect gas equation of state and Dalton's rule can be applied. The transport properties, dynamic viscosity, and thermal conductivity of air are assumed to be negligibly affected by the moisture content. Correlations for these two properties were based on dry air data.

Relations of this appendix were applied in air properties subroutine AIRPR.

The humidity ratio is determined by [16]:

$$w = 0.622 \text{ PSAT} / (\text{PATM} - RH \cdot \text{PSAT}) \quad (\text{A1})$$

where PATM = atmospheric pressure
 PSAT = saturation pressure of water vapor at temperature T
 w = humidity ratio (lb water/lb dry air)
 RH = relative humidity (-)

The saturated water vapor pressure, PSAT (psi), is calculated by the polynomial approximation [2]:

For $T > 32^\circ\text{F}$

$$\text{PSAT} = \exp(19.504 - 10.431z - 0.2755z^2 + 0.03940z^3)$$

For $32 \leq T < 180^\circ\text{F}$

$$\text{PSAT} = \exp(13.4353 - 5.0988z - 1.6896z^2 + 0.17829z^3)$$

For $T \geq 180^\circ\text{F}$

$$\text{PSAT} = \exp(16.8255 - 14.213z + 7.5568z^2 - 4.01506z^3 + 0.17692z^4)$$

where $z = \frac{1000}{460 + T} \quad (\text{A2})$

The specific heat at constant pressure is [16]:

$$C_p = (C_{p_{\text{dry}}} + 0.444) / (1 + w) \quad (\text{A3})$$

where C_p = specific heat at constant pressure of moist air
 (Btu/(lb moist air \cdot F))
 $C_{p_{\text{dry}}}$ = specific heat at constant pressure of dry air (Btu/(lb \cdot F))
 w = humidity ratio (lb water/lb dry air)

The specific heat at constant pressure of dry air is approximated by the following polynomial [2]:

$$C_{p_{\text{dry}}} = 0.2478786 - 0.4204563 \cdot 10^{-4} \cdot TR + 0.567857 \cdot 10^{-7} \cdot TR^2 - 0.14936056 TR^3 \quad (\text{A4})$$

where $TR = T + 460^{\circ}F$ is the dry bulb air temperature on the Rankine scale
 The gas constant is [16]:

$$R = (53.34 + 85.76 \cdot w) / (1 + w) \quad (A5)$$

where $R = \text{moist air gas constant} \frac{\text{lbf} \cdot \text{ft}}{\text{lb} \cdot R}$

$w = \text{humidity ratio (lb water/lb dry air)}$

The dynamic viscosity and thermal conductivity values are obtained from the respective equations [2]:

$$\begin{aligned} \mu = & 5.5029 \cdot 10^{-3} + 8.7157 \cdot 10^{-5} TR - 2.9464 \cdot 10^{-8} TR^2 \\ & + 6.25 \cdot 10^{-12} TR^3 \end{aligned} \quad (A6)$$

$$\begin{aligned} k = & -2.853 \cdot 10^{-4} + 3.268 \cdot 10^{-5} TR - 8.253 \cdot 10^{-9} TR^2 \\ & 1.239 \cdot 10^{-12} TR^3 \end{aligned} \quad (A7)$$

where $k = \text{moist air thermal conductivity (Btu/(h} \cdot F \cdot \text{ft))}$
 $TR = 460 + T = \text{dry bulb air temperature on the Rankine scale (R)}$

APPENDIX B. CALCULATION OF WATER AND FROST PROPERTIES

The equations presented below are for calculation of the properties of water and frost deposited on a heat pump evaporator outer surface. These properties, for specific conditions, are either solely a function of temperature or can be assumed to be constant. The equations below are used in the water and frost properties subroutine, WATPR.

The following fourth degree polynomial expression is used for calculating water density, conductivity, dynamic viscosity and latent heat of condensation [2]:

$$PROP = \sum_{I=1}^5 A(I) \cdot T^I - 1 \quad (B1)$$

where $A(I)$ = five constants per calculated property
 $PROP$ = calculated property
 T = water temperature (F)

Constants $A(I)$ for each property are given in Table B1. The specific heat of water at constant pressure is assumed to be independent of temperature and to have a constant value of $C_p = 1 \frac{\text{Btu}}{\text{lb} \cdot \text{F}}$

The density of frost, as proposed in [51], is calculated by the following equation:

$$\rho_f = \exp(b1 + b2 \cdot (TW - TP)) \quad (B2)$$

where $b1 = 11.9521 + 0.02422 \text{ TPR} + 35.5498 \text{ WA} - 9.1742 \cdot 10^{-7} \text{ VA} + 3.1138 \cdot 10^{-9} \text{ VA} \cdot \text{TPR} - 0.03838$
 $b2 = 13.1606 - 0.02133 \text{ TPR} - 81.955 \text{ WA}/32.018 - \text{TP}$
 TP = tube temperature (F)
 $\text{TPR} = TP + 460$. tube temperature (R)
 TW = water (frost) temperature (F)
 WA = air humidity ratio
 VA = air velocity (ft/sec)
 ρ_f = density of frost (lb/ft³)

The frost conductivity, as proposed in [52], is calculated by the equation:

$$k_f = 0.012138 + 3.8909 \cdot 10^{-3} \cdot \rho_f + 5.1409 \cdot 10^{-6} \cdot \rho_f \quad (B3)$$

where k_f = frost conductivity (Btu/(h · F · ft))
 ρ_f = frost density (lb/ft³)

Frost heat of sublimation, h_{SUBL} , and frost specific heat C_f , are assumed to have the following constant values [52]:

$$h_{\text{SUBL}} = 1219.0 \text{ Btu/lb}$$

$$C_f = 0.46 \text{ Btu/(lb} \cdot \text{F)}$$

Table B1. Water Property Evaluation Constants which are Used in Equation (B1)

Calculated Property A(I)	Density lb/ft ³	Dynamic viscosity lb/(h · ft)	Thermal conductivity Btu/(h · F · ft)	Latent heat of condensation Btu/lb
A(1)	0.11647 E+03	0.79422 E+03	-0.27694	0.31514 E+04
A(2)	-0.40054	0.47589 E+01	0.45215 E-03	0.13714 E+02
A(3)	0.10815 E-02	0.10622 E-01	0.49008 E-05	0.35945 E-01
A(4)	0.12387 E-05	0.10416 E-04	0.88613 E-08	0.43525 E-04
A(5)	0.49002 E-09	0.37690 E-08	0.41387 E-11	0.19695 E-07

APPENDIX C. CALCULATION OF CRITICAL PRESSURE FOR TWO-PHASE FANNO FLOW OF A NON-AZEOTROPIC MIXTURE

In this appendix, the procedure is explained for calculation of critical pressure of a flow if mass flow rate and stagnation enthalpy are given. The procedure focuses on the fact that Fanno flow assumes maximum entropy at the flow critical pressure. Hence, the point of maximum entropy on the Fanno line is being sought and once determined, the choking pressure is found.

The entropy of two-phase flow is calculated by the equation:

$$s = s_L + x(s_V - s_L) \quad (C1)$$

where s = entropy
 x = quality
subscripts L and V refer to liquid and vapor phase, respectively,
being in equilibrium

The quality for Fanno flow can be found using the energy equation:

$$i_o = i + \frac{G^2}{2} \cdot v^2 \quad (C2)$$

where G = mass flux
 i = enthalpy
 i_o = stagnation enthalpy
 v = specific volume

Two-phase specific enthalpy and specific volume are:

$$i = i_L + x(i_V - i_L) \quad (C3)$$

$$v = v_L + x(v_V - v_L) \quad (C4)$$

Substituting and rearranging, the following quadratic equation can be obtained [23]:

$$x^2 + x \cdot b + c = 0 \quad (C5)$$

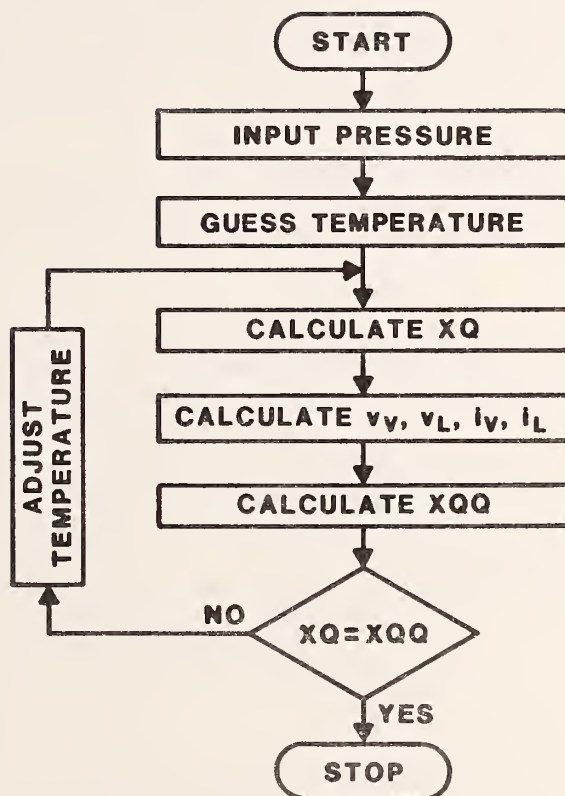
where
$$b = \frac{2(v_V - v_L) \cdot v_L \cdot G^2 + i_V - i_L}{G^2(v_V - v_L)^2}$$

$$c = \frac{2(i_L - i_o)}{G^2} + v_L^2 \frac{1}{(v_V - v_L)^2}$$

It should be noticed that saturated volumes of liquid and vapor and respective enthalpies are uniquely defined at a given pressure for a single component refrigerant. In such situation, refrigerant quality for a given Fanno flow can be calculated explicitly. When a non-azeotropic mixture is used, saturated liquid and vapor properties needed for evaluation of the quality are function of the quality thus an iterative procedure has to be used as shown

in the figure C1. Once the mixture quality (thus also temperature) at the Fanno flow at a given pressure is known, entropies of the saturated liquid and vapor are readily available and the flow entropy can be evaluated by equation C1.

The above described equations and procedures are applied in a double precision function, DFANNO. The Secant Method is used to provide the solution.



XQ - quality calculated by equation of state

XQQ - quality calculated by equation C5

Figure C1. Logic to evaluate the quality of the Fanno flow at a given pressure for a non-azeotropic mixture.

APPENDIX D. THE LOGIC FOR THE MAIN PROGRAM, BMAIN

The principle of the logic for the heat pump model has been explained in section 5.1. It is based on four balances:

- enthalpy balance
- pressure balance
- mass balance of non-azeotropic mixture
- mass balance of one of the mixture components

The enthalpy balance and pressure (mass flow rate) balance are interdependent and have to be found in a simultaneous iteration process. Their solution provides refrigerant properties in the system which are required for mass inventory calculations. The balances are performed in the manner as shown in figure D1, which presents the logic developed during this study and contained in the main program, BMAIN.

The objective of the logic is to iterate refrigerant states at key locations of a heat pump for given outdoor and indoor conditions. The addresses of the key locations used in the program BMAIN are consistent with those marked in figures 2 and D1:

- 1 - evaporator exit, suction tube inlet
- 2 - suction tube outlet, low pressure four-way valve inlet
- 3 - low pressure four-way valve exit, compressor can inlet
- 4 - inside compressor can
- 5 - compressor cylinder during suction stroke
- 6 - compressor cylinder during discharge
- 7 - discharge manifold
- 8 - compressor can exit, high pressure four-way valve inlet
- 9 - high pressure four-way valve exit, discharge tube inlet
- 10 - discharge tube outlet, condenser inlet
- 11 - condenser outlet, liquid line inlet
- 12 - liquid outlet, expansion device outlet
- 13 - evaporator inlet

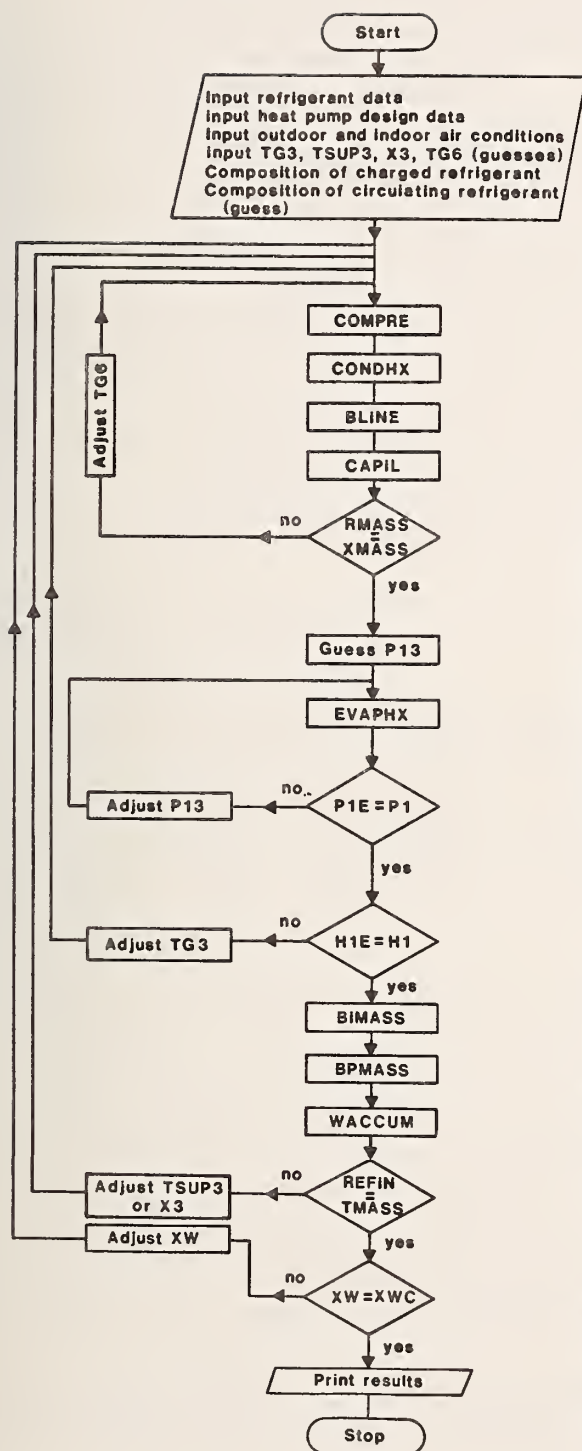
The thermodynamic processes shown in figure D1 are covered by five subroutines: COMPRE, CONDX, BLINE, CAPIL, and EVAPHX. The iteration process starts with an estimated refrigerant pressure, superheat or quality at the compressor can inlet, and an estimated compressor discharge pressure (points 3 and 6). From these data, the compressor simulation subroutine COMPRE computes the refrigerant mass flow rate through the compressor and refrigerant parameters from point 1 to 10. The condenser and liquid line subroutines CONDX and BLINE are called next to perform calculations which yield refrigerant states at point 11 and 12. Subsequently, the expansion device simulation program, CAPIL computes refrigerant mass flow rate through the flow restrictor. At this point, the mass flow rate through the compressor, RMASS, and mass flow rate through the expansion device, XMASS, are compared and mass flow balance is sought. This is done by resuming calculations by COMPRE, CONDX, BLINE, and CAPIL, holding constant refrigerant parameters at the compressor can inlet (point 3) and adjusting refrigerant saturation temperature at discharge (point 6) until an appropriate saturation temperature at point 6 is found at which RMASS and XMASS are equal.

The evaporator performance simulation routine, EVAPHX, was chosen in this logic to be run after the mass flow balance between the compressor and expansion device has been found. (EVAPHX requires the most computing time of all the subroutines.) Evaporator simulation requires input of refrigerant mass flow rate and refrigerant state at the evaporator inlet. The refrigerant enthalpy at the evaporator inlet, H13, is equal to that at point 11, H11, since the thermodynamic processes in the liquid line and flow restrictor are adiabatic. The refrigerant pressure at evaporator inlet, P13, is unknown and has to be guessed based on the known pressure at location 1 and the pressure drop across the evaporator (if known from the previous simulation of the evaporator). The output from EVAPHX is pressure and enthalpy at evaporator outlet, P1E and H1E, respectively. If the pressure P1E is not equal within imposed tolerance to P1, pressure P13 has to be adjusted and evaporator simulation repeated. After the condition of equal P1E and P1 pressures is satisfied, the refrigerant enthalpy at the evaporator outlet obtained from evaporator simulation, H1E, and enthalpy H1 are compared. If these enthalpies are not equal, the refrigerant saturation temperature at the compressor can inlet is adjusted and all calculations repeat.

Once an enthalpy balance and mass flow balance are established, two out of four refrigerant parameters estimated at the outset are temporarily obtained. To verify the third estimated parameter, i.e., the refrigerant vapor superheat at the compressor can inlet, the refrigerant mass inventory in the system is used. Each time the mass inventory is performed, it provides a result based on refrigerant states in the system found after solving enthalpy and pressure balances with the assumed vapor superheat (quality) at points 3 and refrigerant mixture composition. If the amount of refrigerant obtained from the mass inventory calculations is less than the refrigerant in the system, the superheat (quality) estimate has to be decreased and all calculations have to be repeated from the beginning.

After the refrigerant mixture mass inventory agrees with mass of charged refrigerant, the mass inventory of one of the mixture components is made. If, based on this inventory, calculated circulating mixture composition is not equal (within imposed tolerance) to circulating composition estimated for use for simulation calculations, new estimate for the circulating composition is computed and all the calculations have to be repeated from the outset. If the equality between the estimated and calculated compositions is obtained, the iteration process is ended and the results are printed.

In the course of the computing process, the main program gathers information about the heat pump components' performance, updated at each iteration loop, and applies them to anticipate changes of some parameters caused by a change in some state property. This allows for the iteration process to converge faster with each iteration loop. The Secant Method is used in the solution of the two most internal loops and the external loop of the main program. The Binary Search Method was applied to the mixture mass inventory loop due to highly non-linear characteristic of the accumulator.



Symbols:

H1 - refig. enthalpy at evaporator outlet calculated by COMPRE
 H1E - refig. enthalpy at evaporator outlet calculated by EVAPHX
 H12 - refig. enthalpy at point 12
 H13 - refig. enthalpy at point 13
 P1 - refig. pressure at evaporator outlet calculated by COMPRE
 P1E - refig. pressure at evaporator outlet calculated by EVAPHX
 REFIN - mass of refig. charged into a machine
 RMASS - refig. mass flow rate through a compressor
 TG3 - refig. dew point temp. at point 3
 TG6 - refig. dew point temp. at point 6
 TMASS - total mass of refig. calculated by inventory program
 TSUP3 - refig. superheat at point 3
 XMASS - refig. mass flow rate through an expansion device
 XW - circulating refrigerant mixture composition used for the calculations
 XWC - circulating refrigerant mixture composition resulting from the component refrigerant mass inventory
 X3 - refig. quality at point 3

BIMASS - calculates mass of refrigerant in a coil
 Input: refig. state at each coil tube end
 Output: mass of refig. in a coil
 BLINE - calculates pressure drop in a liquid line
 Input: RMASS and refig. state at point 11
 Output: refig. state at point 12
 BPMASS - Calculates mass of refig. in a tube
 Input: refig. state at tube ends
 Output: mass of refig. in a tube
 CAPIL - calculates performance of an expansion device
 Input: refig. state at point 12 and pressure at point 13
 Output: XMASS
 COMPRE - calculates performance of a compressor with a 4-way valve and tubing connecting compressor with both coils
 Input: refig. state at point 3 and pressure at point 6
 Output: RMASS and refig. state at points 1 through 10
 CONDHX - calculates performance of a condenser
 Input: RMASS and refig. state at point 10
 Output: refig. state at point 11
 EVAPHX - calculates performance of an evaporator
 Input: RMASS and refig. state at point 13
 Output: refig. state at point 1
 WACCUM - calculates mass of refrigerant in an accumulator
 Input: refig. state in an accumulator
 Output: mass of refig. in an accumulator and mean refig. composition

Figure D1. Overall logic of the program HPBI.

APPENDIX E. COMPRESSOR SIMULATION SUBROUTINE, COMPRE

Refer to figure 2 where the configuration of the main heat pump components is shown. It was decided to include in one subroutine called COMPRE all refrigerant processes from point 1 to point 10. The refrigerant path covered by this subroutine consists of:

- flow through a suction tube 1-2 and discharge tube 9-10
- flow through a reversing valve on suction side 2-3 and discharge side 8-9
- flow through a compressor 3-8

Input data to COMPRE is detailed in comment statements inserted in the program. Basically they consists of:

- connecting tubing design data
- compressor design and performance parameters
- four-way valve performance parameters
- refrigerant state at the compressor can inlet (point 3)
- refrigerant pressure after compressor (point 6)

The design input data are explained in Table H5. The compressor performance parameters include:

η_e = electric motor efficiency versus load
RPM = electric motor speed (RPM) versus load
 η_m = compressor mechanical efficiency
 C_e = compressor effective clearance
 η_p = compressor polytropic efficiency
CPC34, CPC45, CPC67, CPC78 = pressure drop parameters
CQC4C, CQCC0A, CQC45, CQC67, CQC78 = heat transfer parameters

One of the parameters, the compressor mechanical efficiency, η_m , is assumed here to be constant and equal to 0.96. The rest of the parameters can be calculated by means of subroutine COMPAR using compressor test data. The efficiency, η_e , and RPM as functions of mechanical load are both part of typical electric motor characteristic for its class (see figures 11 and 13). Coefficients describing these curves are required as an input to COMPAR. In the course of calculations, new coefficients are computed which retain the shape of the characteristic curves, although changes in absolute values of efficiency, η_e , and RPM do occur according to test data supplied. The rest of the parameters (compressor effective clearance, compressor polytropic efficiency, five heat transfer parameters, and four pressure drop parameters) are calculated by COMPAR using relations presented in section 5.2.2.

Subroutine COMPAR is incorporated in the general program HPBI and is called by the main program, BMAIN. It is executed by running the HPBI program with appropriate run control data input as explained in Appendix H.

Four-way valve performance parameters are the pressure drop parameter and heat transfer parameter. These parameters are calculated by subroutine VALVPA using four-way valve test data. Similar to the compressor case, VALVPA is a part of the program HPBI and can be accessed by executing it with appropriate run control input data (refer to Appendix H).

Organization of COMPRE may be followed by examining the program listing with inserted comment statements. Based on a known refrigerant state at the can inlet (point 3), refrigerant parameters at the suction valve (point 5) are estimated and since the compression pressure is given, the compression process can be computed. Using the refrigerant mass flow rate from this computation and applying equations given in section 5.2.2, enthalpy and pressure balances are conducted. If balance is not reached, the estimate of the refrigerant state at the suction valve, point 5, is adjusted. Calculations are repeated until balances are satisfied.

Once refrigerant states at compressor stations 3 to 8 and the mass flow rate are known, the properties for other refrigerant paths are computed. Simulation of the four-way valve and tubes are done by routines MVAL4 and PIPE, respectively.

The output from subroutine COMPRE is detailed in the program comment statements and consists of refrigerant thermodynamic properties for points 1 to 10, refrigerant mass flow rate, compressor RPM, energy consumption rate, electric motor efficiency, compression efficiency, and volumetric efficiency.

APPENDIX F. CONSTANT FLOW AREA EXPANSION DEVICE SUBROUTINE, CAPIL

The purpose of the capillary tube simulation subroutine is to calculate the refrigerant mass flow rate when the capillary dimensions, refrigerant inlet state, and pressure in the evaporator are given. Depending on the above input, there are several possible capillary tube operation modes. For example, the refrigerant can be either a single -or a two-phase state prior to the tube entrance. If there is a two-phase mixture at the inlet (a highly undesirable but possible condition under very low load operating conditions), two-phase flow will exist along the whole tube length.

For a subcooled liquid at the inlet, there are three flow alternatives:

- liquid flow only along the whole tube length
- liquid flow in the first portion of the tube and two-phase flow in the latter portion of the tube
- two-phase flow only along the whole tube length

Each of the above flow situations can exist with or without choking condition at the tube exit. The capillary tube simulation routine, CAPIL can distinguish among all of the above operational modes. During the course of the calculations, the refrigerant critical pressure is found by routine CHOK using a binary search iteration method. Then equations (62), (67), and (72) are used to compute the refrigerant mass flow rate. The density-pressure integral in equation (72) is evaluated by routine SIMP based on Simpson's rule. An iteration solution is required to obtain the final simulation result since choking pressure, friction factor, fractions of tube length with liquid and two-phase flow, and the velocity head used to correct enthalpy are functions of refrigerant mass flow rate which has to be found. The Secant Method is used in the iteration for the final CAPIL solution. The logic of the routine CAPIL is presented in figure F1. The input and output data are listed in comment statements in this routine.

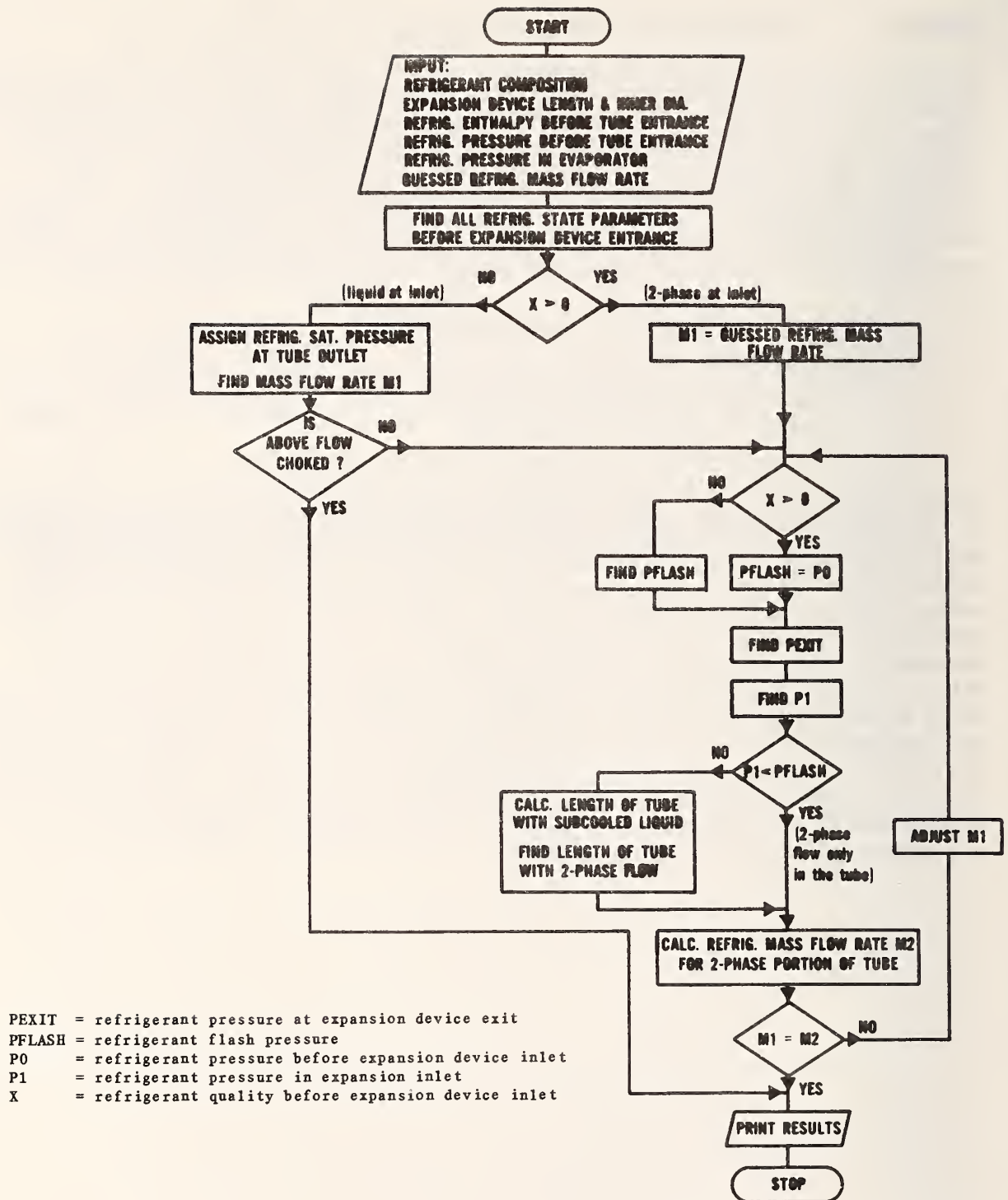


Figure F1. Logic of a constant flow area expansion device simulation program, CAPIL

APPENDIX G. EVAPORATOR AND CONDENSER SIMULATION SUBROUTINES, EVAPHX AND CONDHX

The tube-by-tube approach to coil simulation has been explained in section 5.4 where applicable heat, mass, and momentum transfer equations have also been presented. Though the evaporator and condenser simulation programs require somewhat different heat transfer relations, their logic is the same as shown in figure G1.

The input to coil simulation program are coil design data, refrigerant flow distribution sequence, and air and refrigerant inlet conditions. During the first bank of tubes' performance calculations, the air parameters upstream of each tube row are estimated. The program then computes the heat transfer rate and pressure drop tube-by-tube from the inlet to the exit of the coil. Mean refrigerant properties in each tube are used to calculate heat transfer rates and pressure drops applying equations described in section 5.4. An iterative procedure is used for each tube since only refrigerant conditions at tube inlet are known when the calculations begin, and initially the average refrigerant properties in a tube have to be assumed to be those at the inlet conditions. Upstream air temperature and humidity ratio may not be accurately known at the time tube performance is being calculated, so a second iterative loop is used to update air-side data after each loop. It should be realized that, while in the condenser case, update of air-side data means update of air temperature and humidity ratio; for the evaporator it also includes liquid (frost) layer thickness. When refrigerant enthalpy values at the coil exit obtained from two consecutive loops are within the imposed tolerance, the calculations of coil performance are completed and results printed.

Both the evaporator coil simulation model EVAPHX and condenser coil simulation model CONDHX comply with the logic and equations previously described. Input and output data for these programs are detailed in comment statements at the beginning of each subroutine. Other inserted comment statements along with the flow chart (figure G1) facilitate the understanding of both program's organization.

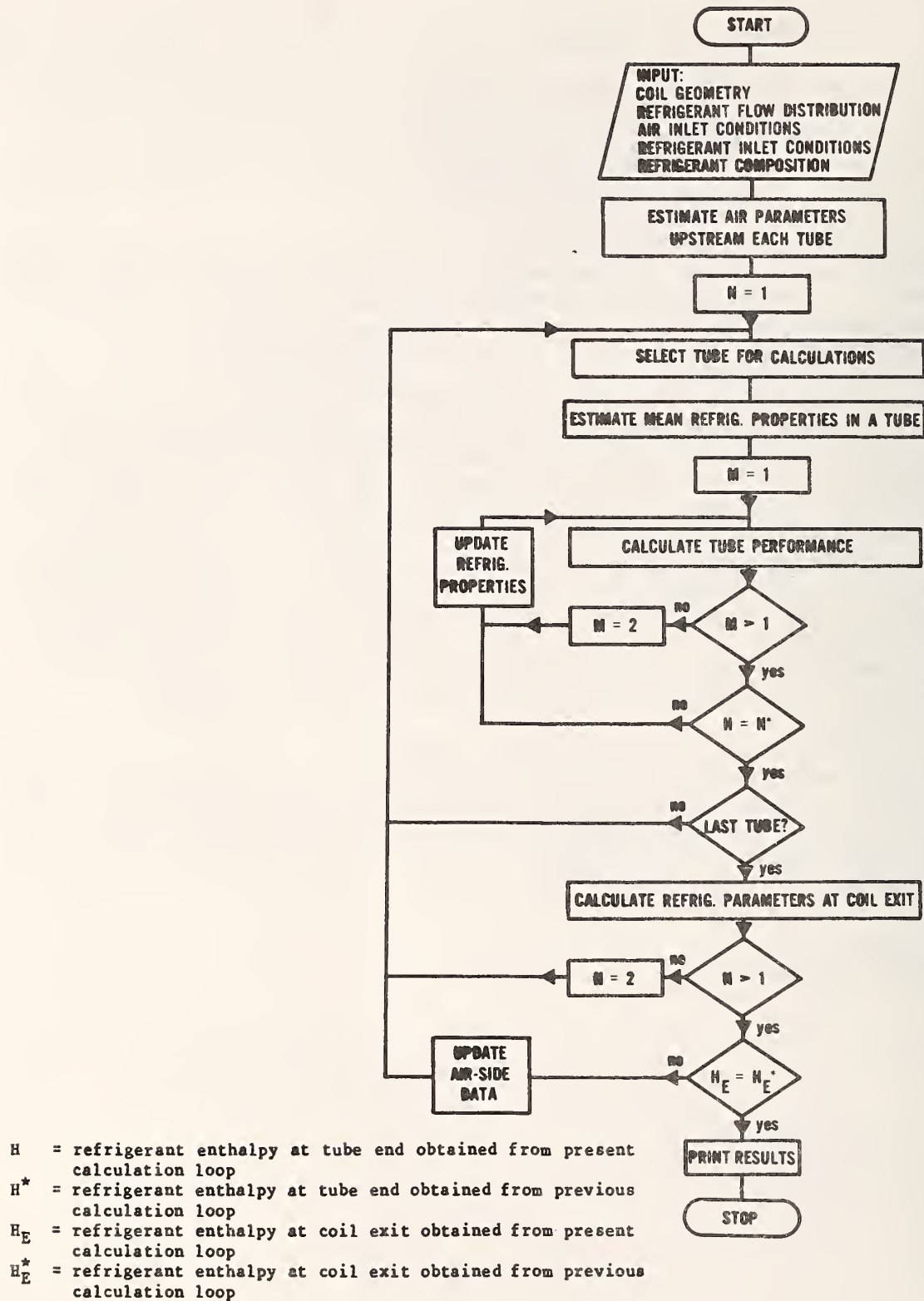


Figure G1. Flow chart for coil performance simulation programs.

APPENDIX H. PROGRAM USER'S GUIDE

H1. General Information

The simulation program of a heat pump charged with a non-azeotropic binary mixture, HPBI, is written in Fortran 77 and makes use of standard Fortran mathematical functions. It is built around the main program, BMAIN, and 65 subprograms for heat pump component simulation, and heat transfer, fluid mechanics, and fluid property calculation. Relations used in the heat pump component simulation subprograms, fluid mechanics, heat transfer and refrigerant property routines are described in the main text of this report. Equations used in moist air, water and frost property routines are described in Appendices A and B.

Tables H1, H2, and H3 list, by categories, all programs used in the heat pump model. Short descriptions of applications are stated in the tables for each program. A listing of all programs is given in Appendix J. A detailed description with specification of input and output parameters is given in the comment statements at the beginning of each program. Comment statements are also located within routines to facilitate understanding of the program organization. Organization of principal program modules is explained in Appendices D, E, F and G.

H2. Input and Output Data Coding

There are five categories of input values for the HPBI runs, namely:

- refrigerant constants
- heat pump data
- run controlling data
- indoor and outdoor air conditions
- refrigerant parameters

H2.1 Refrigerant Constants

Refrigerant constants for evaluation of thermodynamic properties are read into the program by subroutine BCONST from a disc file. File 7 is used in the program for input of these data. Table H4 presents a proper format of refrigerant data file. Comment statements inserted into subroutine BCONST help to identify specific constants.

Refrigerant constants for evaluation of transport properties are contained in function VISCON. Input of these data is not organized through reading of a data file since transport property constants may not have the same format for all mixtures. This may be due to different availability of data as well as different data possibly required by other kinds of mixing rules that could be more applicable for the new mixtures under consideration. Thus, if a new mixture is to be examined, function VISCON has to be redone from the source element.

H2.2 Heat Pump Data

Heat pump data are read into the program by the main program, BMAIN, from a disc file. File 8 is used in the program for input of heat pump data. Coding

of these data is described in Table H5 which includes Fortran symbols with their explanation. An example of a heat pump data file is given in Table H6. Heat pump data include information about each modeled component, i.e., compressor, condenser, expansion device, evaporator, accumulator, four-way valve, vapor line and liquid line. All components with the exception of the compressor and the four-way valve are described by design data only (dimensions, material properties, coil circuitry, etc.). The compressor and the four-way valve simulation also require performance parameters which are derived from test information using subroutines COMPAR and VALVPA, respectively, included in this report (see Appendix E). The subroutines are run by executing the HPBI program and providing appropriate run controlling data, as explained below in this appendix. Heat pump data also include the amount of refrigerant charge. This input is used in simulation runs in which iteration of refrigerant superheat (quality) at the compressor can inlet is required. For determining this input value refer to Chapter 5.6.

H2.3 Run Controlling Data, Indoor and Outdoor Air Conditions, Refrigerant Parameters and Output Data

This category of input data have to be contained in a runstream file or are read from a terminal if simulation run is executed in the interactive mode. They are clearly requested by the program and responses have to be given in a Fortran free format.

The sequence of requests depends on the response to the first request given by the program to determine which one of three possible tasks the user wants the program to perform, i.e.,

1. evaluation of compressor parameters
2. evaluation of four-way valve parameters
3. simulation of heat pump performance

Evaluation of compressor parameters and evaluation of four-way valve parameters has to be done once to generate performance parameters of these components needed as input for simulation of performance of a heat pump system.

Evaluation of Compressor Parameters

The following is the sequence of program requests and explanation of responses for evaluation of compressor parameters:

1. Request: COMPRESSOR PARAMETER (1), FOUR-WAY VALVE PARAMETERS (2) OR HEAT PUMP PERFORMANCE (3)
Response: 1
2. Request: DETERMINE COMPRESSOR PERFORMANCE PARAMETERS
ENTER: 0 FOR PARTIAL TEST DATA OR 1 FOR FULL TEST DATA =
Response: as explained in the request. Most likely user will not have detailed compressor test data. Assume then that response is 0

3. Request: ENTER: ELEFUL, ELEIPT, RPMCP, SWPVOL, RMASS, TOA =
Response: ELEFUL = compressor motor energy input rate at max. rated load (kW)
ELEIPT = compressor motor energy input rate at test conditions (kW)
RPMCP = compressor number of revolution per minute at test (1/min), enter 0 if not measured.
SWPVOL = total compressor displacement volume per revolution (in³)
RMASS = refrigerant mass flow rate at test conditions (lb/h)
TOA = ambient air temperature (F)
4. Request: WEIGHT COMPOSITION OF MIXTURE IN FRACTION OF MORE VOLATILE COMPONENT, XW =
Response: weight composition (decimal fraction)
5. Request: ENTER: T3, P3 =
Response: T3 = refrigerant temperature at compressor can inlet (F)
P3 = refrigerant pressure at compressor can inlet (psia)
6. Request: ENTER: T8, P8 =
Response: T8 = refrigerant temperature at compressor can inlet (F)
P8 = refrigerant pressure at compressor can inlet (psia)

At this point, the program will evaluate refrigerant state at key compressor locations and compressor parameters. Output symbols are explained in the comment statements in the beginning of subroutine COMPAR. Note, that evaluation of compressor parameters requires information on standard electric motor characteristic (motor efficiency vs. load and RPM vs. load) that should be contained in the mass storage data file in lines 2, 3, 4, 5 and 6 as explained in Table H5. This file should be assigned to number 8 for program execution.

Evaluation of Four-Way Valve Parameters

The following is the sequence of program requests and explanation of responses for evaluation of four-way valve parameters.

1. Request: COMPRESSOR PARAMETER (1), FOUR-WAY VALVE PARAMETERS (2) OR HEAT PUMP PERFORMANCE (3)
Response: 2
2. Request: EVALUATION OF FOUR-WAY VALVE PARAMETERS, ENTER: T2, P2, T3, P3 =
Response: T2 = refrigerant temperature at valve low pressure inlet (F)
P2 = refrigerant pressure at valve low pressure inlet (psia)
T3 = refrigerant temperature at valve low pressure outlet (F)
P3 = refrigerant pressure at valve low pressure outlet (psia)

3. Request: ENTER: T8, P8, T9, P9 =
Response: T8 = refrigerant temperature at valve high pressure inlet (F)
P8 = refrigerant pressure at valve high pressure inlet (psia)
T9 = refrigerant temperature at valve high pressure outlet (F)
P9 = refrigerant pressure at valve high pressure outlet (psia)
4. Request: ENTER: RMASS, XW =
Response: RMASS = refrigerant mass flow rate at test conditions (lb/h)
XW = refrigerant weight composition (decimal fraction of the more volatile component)

At this point, the program will evaluate the four-way valve pressure drop parameter, CPD, and heat transfer parameter, CQ. (All output symbols are explained in the comment statements in the beginning of subroutine VALVPA.) Parameters CPD and CQ are to be included in heat pump data file in line 10 as explained in Table H5.

Simulation of Heat Pump Performance

Simulation of heat pump performance is the ultimate purpose of the program HPBI. Once the heat pump data file is completed with the compressor and four-way valve performance parameters, simulation runs of heat pump performance can be conducted for a full range of operating conditions in the heating and cooling mode.

The following is the sequence of program requests and explanation of responses for the heat pump performance simulation run:

1. Request: COMPRESSOR PARAMETERS (1), FOUR-WAY VALVE PARAMETERS (2) OR HEAT PUMP PERFORMANCE (3)
Response: 3
2. Request: ANSWER 1 FOR YES OR 0 FOR NO
DO YOU WANT ANY INPUT DATA PRINTED? LPF =
Response: 0 for no input data print
or
1 for input data printout (request for specification of desired data will follow)
3. Request: OUTDOOR AND INDOOR AIR CONDITIONS, POA, TOA, RHOA, PRA, TRA, RHRA = ?
Response: POA = outside air pressure (psia)
TOA = outside air temperature (F)
RHOA = outside air relative humidity (decimal fraction)
PRA = indoor air pressure (psia)
TRA = indoor air temperature (F)
RHRA = indoor air relative humidity (decimal fraction)
4. Request: NSYS = 1 FOR HEATING, NSYS = 2 FOR COOLING, NSYS = ?
Response: as explained in the request

5. Request: IS ITERATION OF SUPERHEAT/QUALITY REQUESTED?
ITER = 0 FOR NO, ITER = 1 FOR YES, ITER = ?
Response: as explained in the request
6. Request: IS ITERATION OF CIRCULATING COMPOSITION REQUESTED?
ITERXW = 0 FOR NO, ITERXW = 1 FOR YES, ITERXW?
Response: as explained in the request
7. Request: COMPOSITION OF CHARGED REFRIGERANT =
(WEIGHT FRACTION OF MORE VOLATILE COMPONENT)
Response: as explained in the request (decimal fraction)
8. Request: COMPOSITION OF CIRCULATING REFRIGERANT =
(WEIGHT FRACTION OF MORE VOLATILE COMPONENT)
Response: as explained in the request (decimal fraction)
9. Request: REFRIGERANT STATE ESTIMATES: TG3, X3, TSUP3, TG6
Response: TG3 = dew point temperature of refrigerant at the
compressor can inlet (F)
X3 = refrigerant quality at the compressor can inlet
(decimal fraction)
TSUP3 = refrigerant superheat at the compressor can inlet
(F)
TG6 = dew point temperature of refrigerant vapor in
compressor cylinder after compression
(all above values are estimates, subject to iteration)

At this point, the program starts to iterate the refrigerant thermodynamic states at heat pump key locations following the logic explained in Appendix D. Once a solution is obtained the results are printed in the format shown in a printout of an example run (Appendix I).

The output data consist of refrigerant thermodynamic states at 13 key heat pump locations, identified on figure 4, along with results describing the heat pump performance. Symbols used for output data have the following meaning:

T,P,H,S,X	refrigerant temperature (F), pressure (psia), enthalpy (Btu/lb), entropy, (Btu/(lb · F), quality (decimal fraction)
COP	coefficient of performance (-)
ELUSE	total heat pump energy consumption rate (kW)
QLOAD	net capacity of the heat pump (Btu/h)
RMASS	refrigerant mass flow rate (lb/h)
TMASS	refrigerant charge (lb)
TG3	refrigerant dew point temperature at compressor can inlet

TG6 refrigerant dew point temperature at compressor delivery
 valve (F)

TSUP3 refrigerant superheat at compressor can inlet (F)

The last three output data TG3, TG6, and TSUP3 are final values iterated from TG3, TG6, and TSUP3 supplied as estimated input values.

It should be noted that the program prints the above results every time the enthalpy loop and pressure loop are closed, and proceeds with further calculations until the mass inventories of the mixture and one of mixture components are satisfied. The final results in the printout are the requested heat pump performance data at the imposed operating conditions.

Table H1. Refrigerant Property Single Precision Functions and Subroutines

NAME	PURPOSE
BUBPRE	Calculate bubble point pressure from a given composition and temperature.
BUBTEM	Calculate bubble point temperature from a given composition and pressure.
DEWPRE	Calculate dew point pressure from a given composition and temperature.
DEWTEM	Calculate dew point temperature from a given composition and pressure.
EBUBPR	Estimate bubble point pressure from a given composition and temperature.
EBUBTE	Estimate bubble point temperature from a given composition and pressure.
ENTROP	Calculate specific entropy of refrigerant being in a single phase from given composition, temperature and specific volume.
ENTRO2	Calculate specific entropy from given composition, temperature and pressure.
EQPAR	Calculate parameters for the equation of state.
ESVOL	Estimate specific volume of liquid or vapor from given composition, temperature and pressure
FGIBBS	Calculate Gibbs free energy
HCVCP	Calculate enthalpy, specific heat at constant volume and specific heat at constant pressure from given composition, temperature, and specific volume.
HPAR	Calculate parameters for calculation of specific enthalpy, specific heat at constant volume, specific heat at constant pressure and specific entropy.
HPIN	Calculate temperature from given composition, enthalpy and pressure.
HPPROP	Calculate thermodynamic and transport properties from given composition, pressure and enthalpy.
PXQIN	Calculate temperature of two-phase refrigerant from given composition, pressure and quality.
PXQIN2	Calculate temperature and enthalpy of two-phase refrigerant from given composition, pressure and quality.
QLITY	Calculate quality of R13B1/R152a mixture from given composition, temperature and pressure.
SATCOM	Calculate pressure, specific volume of liquid and specific volume of vapor of mixture pure components at saturation from given temperature.
SPIN	Calculate temperature from given composition, entropy, and pressure.
SATLIB	Estimate saturation pressure, specific volume of liquid and specific volume of vapor at given temperature for pure components.
TPPROP	Calculate specific enthalpy, specific volume and quality from given composition, temperature and pressure.
TXQIN	Calculate pressure of two phase refrigerant from given composition, temperature, and quality.

TXQIN2	Calculate pressure and specific enthalpy of two-phase refrigerant from given composition, temperature and quality.
VISCON	Calculate absolute viscosity and thermal conductivity of liquid or vapor and specific heat of saturated liquid of R13B1/R152a mixture from given composition and temperature.
VOLIT1	Calculate specific volume from given composition, temperature and pressure.

Table H2. Refrigerant Property Double Precision Function and Subroutines

NAME	PURPOSE
DBUBTE	as BUBTEM (Table H1)
DDEWTE	as DEWTEM (Table H1)
DENTRO	as ENTROP (Table H1)
DH	Calculate specific enthalpy from given composition, temperature and specific volume.
DQLITY	as QLITY (Table H1)
DVOL1	as VOLIT1 (Table H1)

Table H3. Functions and Subroutines for Heat Pump Component Simulation, Heat Transfer and Fluid Mechanics Calculations

NAME	PURPOSE
AIRHT	Calculate finned tube air-side heat transfer coefficient.
AIRPR	Calculate air properties.
BCONST	Read refrigerant constants for equation of state.
BIMASS	Calculate mass of refrigerant in a coil.
BLINE	Calculate frictional pressure drop in a liquid line.
BMAIN	Main program, solution logic contained.
BPMASS	Calculate mass of refrigerant in a tube.
BSIMP	Integrate numerically using Simpson's 1/3 Rule.
CAPIL	Simulate performance of a constant flow area expansion device.
CHOKE	Calculate the thermodynamic critical pressure for a non-azeotropic mixture in two-phase Fanno flow.
COMPAR	Evaluate compressor parameters.
COMPRE	Simulate compressor performance.
CONDX	Simulate condenser performance.
DDENFA	Calculate refrigerant density for two-phase Fanno flow (double precision).
DFANNO	Calculate refrigerant entropy for two-phase Fanno flow (double precision).
DPDYN1	Calculate dynamic pressure drop for a single-phase flow in a tube.
DPDYN2	Calculate dynamic pressure drop for a two-phase flow in a tube.
EVAPHX	Simulate evaporator performance.
EVDP	Calculate frictional evaporation pressure drop.
FINEFF	Calculate fin efficiency.
FEELIQ	Calculate Lockhart-Martinelli correction factor for two-phase pressure drop.
HTCCON	Calculate condensation heat transfer coefficient.
HTCEV	Calculate evaporation heat transfer coefficient.
HXCODE	Determine refrigerant and air flow distribution in a tube.
MVAL4	Simulate four-way valve performance.
OVLHTC	Calculate overall heat transfer coefficient for a dry finned tube.
OVLWET	Calculate overall heat transfer coefficient for a wet finned tube.
PFLASH	Calculate flashing pressure and temperature for a Fanno flow.
PIPE	Simulate flow through a tube.
SPHDP1	Calculate frictional single-phase pressure drop in a tube.
SPHTC	Calculate single-phase heat transfer coefficient in a tube.
VALVPA	Calculate four-way valve parameters.
WACCUM	Calculate mass of refrigerant in accumulator.
WATPR	Calculate water properties.

Table H4. Data File Containing Constants for Evaluation of Thermodynamic Properties of R13B1/R152a Mixture

1	25.4145, -0.063368, 4.140051E-05
2	0.1353977, -1.50409E-04, -1.354434E-07
3	27.392729669, -0.0594211671, 3.317695607E-5
4	0.1239878, -1.445514007E-4, -1.90223806E-8
5	0.1466, -2.241E-04
6	148.93, 66.05
7	340.15, 386.65
8	10.0522804, -2204.5632, 9636.5313
9	-0.05060051, 1.1455764E-03, -2.56392871E-06
10	0.2749422, -1.702569E-03, 3.71008313E-06
11	10.6410518, -2642.8994, 460.87585
12	0.0218192958, 5.416820778E-04, -1.24731336E-06
13	0.1023688715, -4.0752759E-04, 1.0409447E-06

Table H5. Heat Pump Input Data Code to Program HPBI

All input data are in FORTRAN free field input format with data values on the same line separated by commas.

Line 1: ATITLE
 ATITLE = title, maximum 80 characters

Line 2: EMETA(I), I = 1,5
 EMETA(I) = compressor motor efficiency in fraction at fraction of full load specified by EMOPT(I), I = 1,5 (-)

Line 3: EMETA(I), I = 6,11
 EMETA(I) = compressor motor efficiency in fraction at fraction of full load specified by EMOPT(I), I = 6,11 (-)

Line 4: EMOPT(I), I = 1,5
 EMOPT(I) = compressor motor full load fraction (decimal fraction)

Line 5: EMOPT(I), I = 6,11
 EMOPT(I) = compressor motor full load fraction (decimal fraction)

Line 6: EMRPM(I), I = 1,6
 EMRPM(I) = coefficient for compressor motor RPM calculations (-)

Line 7: ELEFUL, SWPVOL, ETALPY, CLREFF
 ELEFUL = compressor motor energy input rate at max. rated load (kW)
 SWPVOL = compressor displacement volume per revolution (in³)
 ETALPY = compressor polytropic efficiency (-)
 CLREFF = compressor clearance volume as fraction of displacement volume (-)

Line 8: CPC34, CPC45, CPC67, CPC78
 CPC34 = pressure drop parameter at compressor can inlet
 ((1bf · h²)/(1b · in² · ft³))
 CPC45 = pressure drop parameter at compressor suction valve
 ((1bf · h²)/(1b · in² · ft³))
 CPC67 = pressure drop parameter at compressor discharge valve
 ((1bf · h²)/(1b · in² · ft³))
 CPC78 = pressure drop parameter at compressor can exit
 ((1bf · h^{2.2})/(1b · in² · ft^{2.8}))

Line 9: CQC4C, CQCCOA, CQC45, CQC67, CQC78
 CQC4C = parameter for compressor can wall - refrigerant vapor heat transfer (ft^{0.2})
 CQCCOA = parameter for compressor can - ambient air heat transfer (Btu/h · F^{1.333})
 CQC45 = suction valve heat transfer parameter (ft^{0.2})
 CQC67 = discharge valve heat transfer parameter (ft^{0.2})
 CQC78 = heat transfer parameter at can exit (ft^{0.2})

Line 10: CQ, CPDR, VCAN, REFIN
 CQ = parameter for 4-way valve heat transfer (ft^{0.2})
 CPDR = pressure drop parameter for a 4-way valve (1bf·h²/(1b·in²·ft³))
 VCAN = volume of compressor can filled by liquid (ft³)
 REFIN = refrigerant charge (lb)

Line 11: AHGT, DACC, DHOLE(1), DHOLE(2), DTUBE, HDIS
 AHGT = distance between accumulator top and oil return hole (ft)
 DACC = inner diameter of accumulator (ft)
 DHOLE(1) = diameter of oil return hole (ft)
 DHOLE(2) = diameter of upper hole in accumulator tube (ft)

DTUBE = diameter of accumulator tube (ft)
HDIS = vertical distance between holes in accumulator tube (ft)

Line 12: NDEP(1), NROW(1)
NDEP(1) = number of indoor coil tube depth rows (-)
NROW(1) = number of tubes per indoor coil depth row (-)

Line 13: DI(1), DO(1), DT(1), RPCH(1), DPCH(1), WIDTH(1)
DI(1) = inner diameter of indoor coil tubes (in)
DO(1) = outer diameter of indoor coil tubes (in)
DT(1) = indoor coil fin tip diameter (in) (refer to figure 21)
RPCH(1) = pitch between tubes of the same depth row in indoor coil (in)
DPCH(1) = pitch for indoor coil tube depth rows (in)
WIDTP9H(1) = indoor coil width (equal tube length) (in)

Line 14: FPCH(1), FTK(1), FMK(1), TMK(1), AMAS(1)
FPCH(1) = indoor coil fin pitch (in)
FTK(1) = indoor coil fin thickness (in)
FMK(1) = indoor coil fin material thermal conductivity (Btu/(ft · h · F))
TMK(1) = indoor coil tube material thermal conductivity (Btu/ft · h · F)
AMAS(1) = air mass flow rate through indoor coil (lb/h)

Line 15: CONST(1), CPOW(1), ANGLE(1)
CONST(1) = constant for air side heat transfer correlation for indoor coil equal to 0.134 (-)
CPOW(1) = constant for air side heat transfer correlation for indoor coil equal to 0.681 (-)
ANGLE(1) = angle between indoor coil face and air streamlines (rad)

Line 16: EIDFAN
EIDFAN = indoor fan energy input rate (kW)

Line 17: NREPTI
NREPTI = number of repeating sections in indoor coil (-)

Line 18: NTUBE(1,I) I = 1,5
NTUBE(1,1) = number of tubes in first row in each section of indoor coil (-)
NTUBE(1,2) = number of tubes in second row in each section of indoor coil (-)
NTUBE(1,3) = number of tubes in third row in each section of indoor coil (-)
NTUBE(1,4) = number of tubes in fourth row in each section of indoor coil (-)
NTUBE(1,5) = number of tubes in fifth row in each section of indoor coil (-)

Line 19: IFROM(1,I), I = 1,10
IFROM(1,1) = number of tube of indoor coil from which tube 1 receives refrigerant when indoor coil works as evaporator (-)
IFROM(1,2) = number of tube of indoor coil from which tube 2 receives refrigerant when indoor coil works as evaporator (-)
IFROM(1,3) =
IFROM(1,9) =
IFROM(1,10) = number of tube of indoor coil from which tube 10 receives refrigerant when indoor coil works as evaporator (-)

Line 20: IFROM(1,I), I = 11,20
 IFROM(1,I) = number of tube of indoor coil from which tube I receives refrigerant when indoor coil works as evaporator (-)

Line 21: IFROM(1,I), I = 21,30
Line 22: IFROM(1,I), I = 31,40
Line 23: IFROM(1,I), I = 41,50
Line 24: IFROM(1,I), I = 51,60
Line 25: IFROM(1,I), I = 61,70
Line 26: IFROM(1,I), I = 71,80
Line 27: IFROM(1,I), I = 81,90
Line 28: IFROM(1,I), I = 91,100
Line 29: IFROM(1,I), I = 101,110
Line 30: IFROM(1,I), I = 111,120
Line 31: IFROM(1,I), I = 121,130
 IFROM(1,I) = number of tube of indoor coil from which tube I receives refrigerant when indoor coil works as evaporator (-)

Line 32: NDEP(2), NROW(2)
 NDEP(2) = number of outdoor coil tube row depths (-)
 NROW(2) = number of tubes per outdoor coil depth row (-)

Line 33: DI(2), DO(2), DT(2), RPCH(2), DPCH(2), WIDTH(2)
 DI(2) = inner diameter of outdoor coil tubes (in)
 DO(2) = outer diameter of outdoor coil tubes (in)
 DT(2) = outdoor coil fin tip diameter (in) (refer to figure 21)
 RPCH(2) = pitch between tubes of the same depth in outdoor coil (in)
 DPCH(2) = pitch between tube depth rows for outdoor coil (in)
 WIDTH(2) = outdoor coil width (equal tube length) (in)

Line 34: FPCH(2), FTK(2), FMK(2), TMK(2), AMAS(2)
 FPCH(2) = outdoor coil fin pitch (in)
 FTK(2) = outdoor coil fin thickness (in)
 FMK(2) = outdoor coil fin material thermal conductivity (Btu/(ft · h · F))
 TMK(2) = outdoor coil tube material thermal conductivity (Btu/(ft · h · F))
 AMAS(2) = air mass flow rate through outdoor coil (lb/h)

Line 35: CONST(2), CPOW(2), ANGLE(2)
 CONST(2) = constant for air side heat transfer correlation for outdoor coil equal to 0.134 (-)
 CPOW(2) = constant for air side heat transfer correlation for outdoor coil equal to 0.681 (-)
 ANGLE(2) = angle between outdoor coil face and air streamlines (rad)

Line 36: EIDFAN
 EIDFAN = outdoor fan energy input rate (kW)

Line 37: NREPTO
 NREPTO = number of repeating sections in outdoor coil (-)

Line 38: NTUBE(2,I), I = 1,5
 NTUBE(2,1) = number of tubes in first row in each section of outdoor coil (-)
 NTUBE(2,2) = number of tubes in second row in each section of outdoor coil (-)
 NTUBE(2,3) = number of tubes in third row in each section of outdoor coil (-)

NTUBE(2,4) = number of tubes in fourth row in each section of outdoor coil (-)

NTUBE(2,5) = number of tubes in fifth row in each section of outdoor coil (-)

Line 39: IFROM(2,I), I = 1,10

IFROM(2,1) = number of tube of outdoor coil from which tube 1 receives refrigerant when outdoor coil works as evaporator (-)

IFROM(2,2) = number of tube of outdoor coil from which tube 2 receives refrigerant when outdoor coil works as evaporator (-)

IFROM(2,3) =

.

.

.

IFROM(2,9) =

IFROM(2,10) = number of tube of outdoor coil from which tube 10 receives refrigerant when outdoor coil works as evaporator (-)

Line 40: IFROM(2,I), I = 11,20

IFROM(2,I) = number of tube of outdoor coil from which tube I receives refrigerant when outdoor coil works as evaporator (-)

Line 41: IFROM(2,I), I = 21,30

Line 42: IFROM(2,I), I = 31,40

Line 43: IFROM(2,I), I = 41,50

Line 44: IFROM(2,I), I = 51,60

Line 45: IFROM(2,I), I = 61,70

Line 46: IFROM(2,I), I = 71,80

Line 47: IFROM(2,I), I = 81,90

Line 48: IFROM(2,I), I = 91,100

Line 49: IFROM(2,I), I = 101,110

Line 50: IFROM(2,I), I = 111,120

Line 51: IFROM(2,I), I = 121,130

IFROM(2,I), = number of tube of outdoor coil from which tube I receives refrigerant when outdoor coil works as evaporator (-)

Line 52: CAPID1, CAPL1, NCPL1, CAPID2, CAPL2, NCPL2

CAPID1 = inner diameter of cooling operation expansion device (in)

CAPL1 = length of cooling operation expansion device (in)

NCPL1 = number of cooling operation expansion devices (-)

CAPID2 = inner diameter of heating operation expansion device (in)

CAPL2 = length of heating operation expansion device (in)

NCPL2 = number of heating operation expansion devices (-)

Line 53: YL, YD, YK1, YD1, YK2, YD2

YL = length of compressor-outdoor coil tubing (in)

YD, inner diameter of compressor-outdoor coil tubing (in)

YK1 = thermal conductivity of compressor-outdoor coil tubing material (Btu/(ft · h · F))

YD1 = outer diameter of compressor-outdoor coil tubing (in)

YK2 = thermal conductivity of compressor-outdoor coil tubing insulation (Btu/(ft · h · F))

YD2 = outer diameter of compressor-outdoor coil tubing insulation (Btu/(ft · h · F))

Line 54: RL, RD, RK1, RD1, RK2, RD2
 RL = length of compressor-indoor coil tubing (in)
 RD = inner diameter of compressor-indoor coil tubing (in)
 RK1 = thermal conductivity of compressor-indoor coil tubing
 material (Btu/(ft · h · F))
 RD1 = outdoor diameter of compressor-indoor coil tubing (in)
 RK2 = thermal conductivity of compressor-indoor coil tubing
 insulation (Btu/(ft · h · F))
 RD2 = outer diameter of compressor-outdoor coil tubing insulation
 (Btu/(ft · h · F))

Line 55: RYL, RYD
 RYL = liquid line length (in)
 RYD = liquid line diameter (in)

Note on coil input data:

The coil depth is in the direction perpendicular to a coil surface facing the incoming air. Lines 19-31 and 39-51: tubes are numbered consecutively from the first tube in the first row (facing incoming air) to the last tube in the last row. Enter 0 (zero), if considered tube receives refrigerant from coil inlet port; enter 999, if the tube is nonexistent.

Table H6. Example of a Heat Pump Data File

```

1  ***SYST4 WITH ACCUM., 13B1/152A 6-22-84***
2  3.9,3.64,0.706,0.06315
3  0.2076,0.3490,0.5665,0.6803,0.7238
4  0.7523,0.7712,0.7821,0.7900,0.7929,0.7940
5  0.05,0.1,0.2,0.3,0.4
6  0.5,0.6,0.7,0.8,0.9,1.0
7  3578.,-36.01,-39.0558,22.2747,-40.9712,14.7620
8  5.266E-7,1.504E-5,3.603E-4,1.173E-4
9  10.16,5.279,0.2060,0.2027,2.893
10 0.968,4.47E-5,0.104,5.082
11 0.92,0.42,0.0029,0.0033,0.0625,0.27
12 4,36
13 0.311,0.375,1.056,1.,0.875,18.
14 0.068,0.008,118.,223.,5000.
15 0.134,0.681,1.13
16 0.485
17 2
18 18,18,18,18,0
19 2,3,4,5,6,7,69,9,10,11
20 12,13,14,33,0,15,16,17,1,19
21 20,21,22,23,24,8,26,27,28,29
22 30,31,51,35,53,18,38,39,40,41
23 42,61,25,45,46,47,48,49,68,32
24 52,34,71,36,37,55,56,57,58,59
25 43,44,62,63,64,65,66,50,70,52
26 72,54,999,999,999,999,999,999,999,999
27 999,999,999,999,999,999,999,999,999,999
28 999,999,999,999,999,999,999,999,999,999
29 999,999,999,999,999,999,999,999,999,999
30 999,999,999,999,999,999,999,999,999,999
31 999,999,999,999,999,999,999,999,999,999
32 3,20
33 0.311,0.375,1.338,1.25,1.125,40.
34 0.075,0.008,118.,223.,10700.
35 0.134,0.681,1.52
36 0.380
37 1
38 20,20,20,0,0
39 21,1,2,3,6,7,8,48,0,9
40 12,10,33,13,14,15,18,19,20,40
41 41,43,44,44,24,46,28,49,50,11
42 11,52,34,55,56,54,36,58,59,60
43 42,22,23,45,27,25,26,47,29,30
44 31,51,32,53,35,57,37,37,38,39
45 999,999,999,999,999,999,999,999,999,999
46 999,999,999,999,999,999,999,999,999,999
47 999,999,999,999,999,999,999,999,999,999
48 999,999,999,999,999,999,999,999,999,999
49 999,999,999,999,999,999,999,999,999,999
50 999,999,999,999,999,999,999,999,999,999
51 999,999,999,999,999,999,999,999,999,999
52 0.08202,0.5,1,0.09,0.5,1
53 192.,0.68,223.,0.75,0.05,1.5
54 36.,0.68,223.,0.75,0.,0.
55 312.,0.249

```


APPENDIX I. EXAMPLE OF RUN OF THE PROGRAM HPBI

The following is a computer printout for a HPBI run in which performance of a heat pump in the heating mode was simulated. Input for this run was as follows:

- refrigerant data - as shown in Table H4
- heat pump data - as shown in Table H6
- run controlling data, operating conditions, estimated refrigerant parameters - as shown on the printout (lines 4-36).

The results of this run are included in Table 5.

The solution was iterated in two loops iterating composition of the circulating refrigerant (lines 39-2592 and 2593-4762). The second loop required eight internal loops in which the total refrigerant mass conservation was sought (TMASS = REFIN). The run required 33 minutes of CPU on a Sperry 1100/82 computer.


```

WEBER*PRT47(1)
1 @XQT Q$Q$Q$*DOMANSKI.HPB1
2
3 COMPRESSOR PARAMETERS(1), FOUR WAY VALVE PARAMETERS(2)
4 OR HEAT PUMP PERFORMANCE(3), IPTP=
5 3
6
7 ANSWER 1 FOR YES OR 0 FOR NO
8 DO YOU WANT ANY INPUT DATA PRINTED ? LPR=
9 0
10
11 OUTDOOR & INDOOR AIR CONDITIONS
12 POA, TOA, RHCA, FRA, TRA, RHRA=?
13 14.7, 47., 0.73, 14.7, 70., 0.56
14
15 NSYS=1 FOR HEATING, NSYS=2 FOR COOLING MODE NSYS=?
16 1
17
18 IS ITERATION OF SUPERHEAT/QUALITY REQUESTED ?
19 ITER=0 FOR NO, ITER=1 FOR YES, ITER=
20 1
21
22 IS ITERATION OF MIXTURE COMPOSITION REQUESTED ?
23 ITERXW=0 FOR NO, ITERXW=1 FOR YES, ITERXW=
24 1
25
26 COMPOSITION OF CHARGED REFRIGERANT =
27 (WEIGHT FRACTION OF MORE VOLATILE COMPONENT)
28 .65
29
30 COMPOSITION OF CIRCULATING REFRIGERANT =
31 (WEIGHT FRACTION OF MORE VOLATILE COMPONENT)
32 .65
33
34 REFRIGERANT STATE GUESSES: TG3,X3,TSUP3,TG6=
35 33.,1.,0.,110.
36
37
38 INPUT DATA TO COMPR:
39
40
41 P3 T3 H3 XG3 TG6 TRA TGA
42 6.697+001 3.300+001 8.929+001 1.000+000 1.100+002 7.000+001 4.700+001
43
44 COMPRESSOR ITERATION:
45
46 EI ETAE ETAC ETAV ETAV CPRPM RMASS
47 2.663+000 7.552+001 9.498+001 8.520+001 3.550+003 4.827+002
48
49 T P H X
50 1 3.828+001 7.458+001 8.778+001 9.771+001
51 2 3.765+001 7.872+001 8.773+001 9.773+001
52 3 3.300+001 6.697+001 8.929+001 1.000+000
53 4 6.995+001 6.698+001 9.536+001 1.000+000
54 5 7.348+001 6.407+001 9.606+001 1.000+000
55 6 1.808+002 2.343+002 1.097+002 1.000+000
56 7 1.738+002 2.118+002 1.090+002 1.000+000
57 8 1.553+002 2.107+002 1.055+002 1.000+000

```

9 1.475+002 2.081+002 1.040+002 1.000+000
 10 1.458+002 2.030+002 1.038+002 1.000+000

INPUT DATA TO CONDHX:

T	P	TAIR	RH	RMASS
1.468+002	2.080+002	7.000+001	5.600-001	4.827+002
H2, H2PH2=	40.91	.01		
H2, H2PH2=	41.12	-.21		
H2, H2PH2=	41.38	-.26		
H2, H2PH2=	41.52	-.13		
H2, H2PH2=	41.59	-.07		
H2, H2PH2=	41.52	-.03		
H2, H2PH2=	41.63	-.01		

CONDENSER ITERATION:

T	P	H	X
1.468+002	2.080+002	1.038+002	1.000+000
8.412+001	1.962+002	4.163+001	1.369-001

LIQUID LINE:

INPUT - PIN = 1.962+002 TIN = 8.412+001 XIN = 1.369-001 HIN = 4.163+001
 OUTPUT - POUT = 1.779+002 TOUT = 7.737+001 XOUT = 1.602-001

EXPANSION DEVICE:

INPUT - P12 = 1.779+002 H12 = 4.163+001 P13 = 8.258+001
 OUTPUT - POUT = 1.221+002 XMASS = 3.186+002

INPUT DATA TO COMPR:

P3	T3	H3	XQ3	TG6	TRA	TGA
6.697+001	3.300+001	8.929+001	1.000+000	1.125+002	7.000+001	4.700+001

COMPRESSOR ITERATION:

E1	ETAE	ETAC	ETAV	CPRPN	RMASS
2.719+000	7.576-001	9.483-001	8.580-001	3.549+003	4.756+002

I	T	P	H	X
1	3.820+001	7.448+001	8.772+001	9.766-001
2	3.758+001	7.563+001	9.767+001	9.768-001
3	3.300+001	6.697+001	8.923+001	1.000+000
4	7.069+001	6.688+001	9.548+001	1.000+000
5	7.437+001	6.410+001	9.620+001	1.000+000
6	1.849+002	2.426+002	1.103+002	1.000+000
7	1.781+002	2.212+002	1.096+002	1.000+000
8	1.597+002	2.202+002	1.059+002	1.000+000
9	1.511+002	2.177+002	1.043+002	1.000+000
10	1.503+002	2.176+002	1.042+002	1.000+000

INPUT DATA TO CONDHX:

T	P	TAIR	RH	RMASS
1.503+002	2.176+002	7.000+001	5.600-001	4.795+002
H2, H2PH2=	35.34	-.62		


```

174 T      P      H      X
175 1.511+002 2.196+002 1.642+002 1.000+000
176 8.528+001 2.116+002 3.541+001 .000
177
178
179
180 LIQUID LINE:
181 INPUT - PIN = 2.116+002 TIN = 8.528+001 XIN = .000
182 OUTPUT - POUT = 2.061+002 TOUT = 8.525+001 XOUT = .000
183
184
185
186 EXPANSION DEVICE:
187 INPUT - P12 = 2.061+002 H12 = 3.541+001 P13 = 8.246+001
188 OUTPUT - POUT = 1.593+002 XMASS = 4.802+002
189
190 INPUT DATA TO EVAPHX:
191
192 T      P      X      TAIR      RH      RMASS
193 2.834+001 8.246+001 2.548+001 4.700+001 7.300+001 4.790+002
194 H2= 92.6812
195 H2= 92.6334
196 QT, QS= 2.739+004 1.894+004
197
198 EVAPORATOR ITERATION:
199
200 T      P      H      X      TSUP
201 2.834+001 8.246+001 3.541+001 2.548+001
202 4.672+001 4.255+001 9.259+001 1.000+000 3.659+001
203
204 INPUT DATA TO EVAPHX:
205
206 T      P      X      TAIR      RH      RMASS
207 4.742+001 1.144+002 1.799+001 4.700+001 7.300+001 4.790+002
208 H2= 37.7119
209 H2= 37.0321
210 H2= 36.8217
211 QT, QS= 6.995+002 6.994+002
212
213 EVAPORATOR ITERATION:
214
215 T      P      H      X      TSUP
216 4.742+001 1.144+002 3.541+001 1.799+001
217 4.645+001 1.119+002 3.637+001 2.110+001 .000
218
219 INPUT DATA TO EVAPHX:
220
221 T      P      X      TAIR      RH      RMASS
222 3.770+001 9.715+001 2.190+001 4.700+001 7.300+001 4.790+002
223 H2= 85.2274
224 H2= 77.3103
225 H2= 75.9624
226 QT, QS= 1.942+004 1.854+004
227
228 EVAPORATOR ITERATION:
229
230 T      P      H      X      TSUP
231 3.770+001 9.715+001 3.541+001 2.190+001
232 3.915+001 8.021+001 7.596+001 8.362+001 .000

```


232	INPUT DATA TO EVAPHX:									
233	T	P	X	TAIR	RH	RMASS				
234	3.580+001	9.403+001	2.254-001	4.700+001	7.300-001	4.790+002				
235	P2= 91.0432									
236	H2= 89.3153									
237	H2= 87.1500									
238	H2= 87.0712									
239	H2= 86.9629									
240	QT, QS= 2.470+001 2.147+004									
241										
242										
243	EVAPORATOR ITERATION:									
244	T	P	H	X	TSUP					
245	3.590+001	9.403+001	3.541+001	2.264-001						
246	3.691+001	7.293+001	8.597+001	9.697-001	.000					
247	INPUT DATA TO EVAPHX:									
248	T	P	X	TAIR	RH	RMASS				
249	3.620+001	9.469+001	2.248-001	4.700+001	7.300-001	4.790+002				
250	H2= 90.6913									
251	H2= 87.0221									
252	H2= 84.6521									
253	H2= 84.5449									
254	QT, QS= 2.353+004 2.092+004									
255										
256	EVAPORATOR ITERATION:									
257	T	P	H	X	TSUP					
258	3.620+001	9.469+001	3.541+001	2.248-001						
259	3.752+001	7.459+001	8.454+001	9.412-001	.000					
260	P1, P1E, P1EP1, DENT2= 74.4634 74.5945 .1311 -3.1760									
261	INPUT DATA TO COMPR:									
262	P3	T3	H3	X03	T06	TRA	T0A			
263	6.447+001	3.100+001	8.909+001	1.000+000	1.120+002	7.000+001	4.700+001			
264	COMPRESSOR ITERATION:									
265	EI	ETAE	ETAC	ETAV	CFRPM	RMASS				
266	2.679+000	7.559-001	9.477-001	6.596-001	3.549+003	4.594+002				
267	I	T	P	H	X					
268	1	3.615+001	7.161+001	8.780+001	9.800-001					
269	2	3.554+001	7.080+001	8.774+001	9.799-001					
270	3	3.103+001	6.447+001	8.909+001	1.000+000					
271	4	7.043+001	6.436+001	9.555+001	1.000+000					
272	5	7.453+001	6.174+001	9.629+001	1.000+000					
273	6	1.870+002	2.410+002	1.108+002	1.000+000					
274	7	1.804+002	2.214+002	1.101+002	1.000+000					
275	8	1.615+002	2.204+002	1.053+002	1.000+000					
276	9	1.543+002	2.181+002	1.049+002	1.000+000					
277	10	1.534+002	2.180+002	1.048+002	1.000+000					
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T	1.534+002	P	2.180+002	TAIR	7.000+001	RH	5.600-001	RMASS	4.504+002
H2, H2PH2=	34.10								
H2, H2PH2=	34.16								
H2, H2PH2=	34.61								
H2, H2PH2=	34.76								
H2, H2PH2=	34.82								
H2, H2PH2=	34.82								

CONDENSER ITERATION:

T	1.534+002	P	2.180+002	H	1.048+002	X	1.000+000
8.300+001	2.110+002		3.482+001				

LIQUID LINE:

INPUT -	PIN = 2.110+002	TIN = 8.300+001	XIN = .000
OUTPUT -	POUT = 2.039+002	TOUT = 8.293+001	XOUT = .000

HIN = 3.482+001

EXPANSION DEVICE:

INPUT -	P12 = 2.039+002	H12 = 3.482+001	P13 = 9.183+001
OUTPUT -	POUT = 1.612+002	XMASS = 4.940+002	

INPUT DATA TO COMPR:

P3	T3	H3	XQ3	TG3	TRA	TOA
6.447+001	3.100+001	8.909+001	1.000+000	1.114+002	7.000+001	4.700+001

COMPRESSOR ITERATION:

E1	ETA1	ETAC	ETAV	CPRM	RMASS
2.665+000	7.553-001	9.481-001	8.547-001	3.550+003	4.592+002

I	T	P	H	X
1	3.617+001	7.164+001	8.782+001	3.801-001
2	3.555+001	7.083+001	8.775+001	9.800-001
3	3.100+001	6.447+001	8.909+001	1.000+000
4	7.023+001	6.438+001	9.551+001	1.000+000
5	7.410+001	6.174+001	9.625+001	1.000+000
6	1.853+002	2.388+002	1.106+002	1.000+000
7	1.793+002	2.189+002	1.093+002	1.000+000
8	1.605+002	2.160+002	1.062+002	1.000+000
9	1.534+002	2.156+002	1.048+002	1.000+000
10	1.525+002	2.156+002	1.047+002	1.000+000

INPUT DATA TO CONDHX:

T	1.525+002	P	2.156+002	TAIR	7.000+001	RH	5.600-001	RMASS	4.592+002
H2, H2PH2=	34.79								
H2, H2PH2=	34.74								
H2, H2PH2=	35.19								
H2, H2PH2=	35.46								
H2, H2PH2=	35.57								
H2, H2PH2=	35.58								

348 H2, H2PH2= 35.58 .00
 349
 350 CONDENSER ITERATION:
 351
 352 T P H X
 353 1.525+002 2.156+002 1.047+002 1.000+000
 354 8.591+001 2.077+002 3.558+001 .000
 355
 356
 357 LIQUID LINE:
 358 INPUT - PIN = 2.077+002 TIN = 8.591+001 XIN = .000 HIN = 3.558+001
 359 OUTPUT - POUT = 2.023+002 TOUT = 8.504+001 XOUT = 4.725-003
 360
 361
 362 EXPANSION DEVICE:
 363 INPUT - P12 = 2.025+002 H12 = 3.552+001 P13 = 9.186+001
 364 OUTPUT - POUT = 1.556+002 XPASS = 4.644+002
 365
 366 INPUT DATA TO COMPRESSOR:
 367
 368 P3 T3 H3 XQ3 TGS TRA TGA
 369 6.447+001 3.100+001 8.909+001 1.000+000 1.113+002 7.000+001 4.700+001
 370
 371 COMPRESSOR ITERATION:
 372
 373 E1 ETAE ETAC ETAV CPRFM RMASS
 374 2.662+000 7.532-001 9.481-001 8.549-001 3.550+003 4.593+002
 375
 376 I T P H X
 377 1 3.617+001 7.164+001 8.782+001 9.801-001
 378 2 3.556+001 7.083+001 8.775+001 9.900-001
 379 3 3.100+001 6.447+001 8.909+001 1.000+000
 380 4 7.025+001 6.438+001 9.551+001 1.000+000
 381 5 7.405+001 6.173+001 9.624+001 1.000+000
 382 6 1.857+002 2.394+002 1.106+002 1.000+000
 383 7 1.791+002 2.185+002 1.099+002 1.000+000
 384 8 1.603+002 2.175+002 1.052+002 1.000+000
 385 9 1.532+002 2.152+002 1.048+002 1.000+000
 386 10 1.523+002 2.151+002 1.047+002 1.000+000
 387
 388 INPUT DATA TO CONDENSER:
 389
 390 T P TAIR RH RMASS
 391 1.523+002 2.151+002 7.000+001 5.600-001 4.593+002
 392 H2, H2PH2= 35.09 -.44
 393 H2, H2PH2= 35.04 .05
 394 H2, H2PH2= 35.47 -.43
 395 H2, H2PH2= 35.75 -.28
 396 H2, H2PH2= 35.83 -.08
 397 H2, H2PH2= 35.84 -.01
 398
 399 CONDENSER ITERATION:
 400
 401 T P H X
 402 1.523+002 2.151+002 1.047+002 1.000+000
 403 8.671+001 2.072+002 3.584+001 9.987-004
 404
 405

LIQUID LINE:
 INPUT - PIN = 2.072+002 TIN = 8.671+001 XIN = 9.937-004 HIN = 3.584+001
 OUTPUT - POUT = 2.016+002 TOUT = 3.477+001 XOUT = 1.179-002

EXPANSION DEVICE:
 INPUT - P12 = 2.016+002 H12 = 3.584+001 P13 = 9.186+001
 OUTPUT - POUT = 1.537+002 XMASS = 4.558+002

INPUT DATA TO EVAPHX:
 T 3.454+001 9.186+001 2.372-001 4.700+001 7.300-001 4.592+002
 H2= 91.4649
 H2= 91.2135
 H2= 90.9388
 H2= 90.8369
 QT, QS= 2.530+004 2.185+004

EVAPORATOR ITERATION:
 T P X H X TSUP
 3.454+001 9.186+001 3.574+001 2.372-001
 4.301+001 6.872+001 9.084+001 1.000+000 8.535+000

INPUT DATA TO EVAPHX:
 T P X H X TSUP
 3.532+001 9.311+001 2.342-001 4.700+001 7.300-001 4.592+002
 H2= 91.2239
 H2= 90.4051
 H2= 89.6060
 H2= 89.0485
 H2= 88.9463
 QT, QS= 2.445+004 2.132+004

EVAPORATOR ITERATION:
 T P X H X TSUP
 3.532+001 9.311+001 3.573+001 2.342-001
 3.687+001 7.223+001 8.897+001 9.921-001 .000

INPUT DATA TO EVAPHX:
 T P X H X TSUP
 3.518+001 9.290+001 2.347-001 4.700+001 7.300-001 4.592+002
 H2= 91.2820
 H2= 90.6518
 H2= 89.8340
 H2= 89.6312
 H2= 89.4731
 QT, QS= 2.468+004 2.146+004

EVAPORATOR ITERATION:
 T P X H X TSUP
 3.518+001 9.290+001 3.573+001 2.347-001
 3.661+001 7.172+001 8.848+001 9.980-001 .000

464 P1,P1E,P1EP1,DENT2= 71.6394 71.7189 .0795 1.6604
 465
 466 INPUT DATA TO COMPR:
 467
 468 P3 T3 H3 XQ3 T96 TRA TOA
 469 6.532+001 3.169+001 8.916+001 1.000+000 1.119+002 7.000+001 4.700+001
 470
 471 COMPRESSOR ITERATION:
 472
 473 EI ETAE ETAC ETAV CPRPM RMASS
 474 2.687+000 7.563+001 9.481+001 8.556+001 3.549+003 4.650+002
 475
 476 I T P H X
 477 1 3.699+001 7.260+001 8.789+001 9.800+001
 478 2 3.623+001 7.178+001 8.783+001 9.801+001
 479 3 3.169+001 6.532+001 8.916+001 1.000+000
 480 4 7.045+001 6.523+001 9.551+001 1.000+000
 481 5 7.423+001 6.254+001 9.624+001 1.000+000
 482 6 1.858+002 2.403+002 1.105+002 1.000+000
 483 7 1.791+002 2.202+002 1.093+002 1.000+000
 484 8 1.604+002 2.192+002 1.061+002 1.000+000
 485 9 1.533+002 2.168+002 1.048+002 1.000+000
 486 10 1.525+002 2.168+002 1.045+002 1.000+000
 487
 488 INPUT DATA TO CONDHX:
 489
 490 T P TAIR RH RMASS
 491 1.525+002 2.168+002 7.000+001 5.600+001 4.560+002
 492 H2,H2PH2= 34.50 .12
 493 H2,H2PH2= 35.23 -.73
 494 H2,H2PH2= 34.95 .28
 495 H2,H2PH2= 35.30 -.36
 496 H2,H2PH2= 35.45 -.15
 497 H2,H2PH2= 35.53 -.07
 498 H2,H2PH2= 35.52 .01
 499
 500 CONDENSER ITERATION:
 501
 502 T P H X
 503 1.525+002 2.168+002 1.046+002 1.000+000
 504 8.569+001 2.028+002 3.552+001 .000
 505
 506
 507 LIQUID LINE:
 508 INPUT - PIN = 2.088+002 TIN = 8.569+001 XIN = .000
 509 OUTPUT - POUT = 2.035+002 TOUT = 6.537+001 XOUT = 1.694+003
 510
 511
 512
 513 EXPANSION DEVICE:
 514 INPUT - P12 = 2.035+002 H12 = 3.552+001 P13 = 9.386+001
 515 OUTPUT - POUT = 1.566+002 X12 = 4.690+002
 516
 517 INPUT DATA TO COMPR:
 518
 519 P3 T3 H3 XQ3 T96 TRA TOA
 520 6.532+001 3.169+001 8.916+001 1.000+000 1.118+002 7.000+001 4.700+001
 521 COMPRESSOR ITERATION:

EI	ETAE	ETAC	ETAV	CPRPM	RMASS
2.696+000	7.562+001	9.481+001	8.557+001	3.549+003	4.660+002

I	T	P	H	X
1	3.689+001	7.260+001	8.789+001	9.801+001
2	3.620+001	7.178+001	8.783+001	9.801+001
3	3.169+001	6.532+001	8.916+001	1.009+000
4	7.044+001	6.523+001	9.551+001	1.000+000
5	7.422+001	6.754+001	9.624+001	1.000+000
6	1.857+002	2.404+002	1.105+002	1.000+000
7	1.791+002	2.201+002	1.096+002	1.000+000
8	1.604+002	2.191+002	1.061+002	1.000+000
9	1.532+002	2.167+002	1.048+002	1.000+000
10	1.524+002	2.166+002	1.046+002	1.000+000

INPUT DATA TO CONDIX:

T	P	TAIR	RH	RMASS
1.524+002	2.166+002	7.000+001	5.600+001	4.360+002

H2, H2PH2=	34.60	.12
H2, H2PH2=	34.57	-.06
H2, H2PH2=	35.18	-.52
H2, H2PH2=	35.50	-.32
H2, H2PH2=	35.61	-.11
H2, H2PH2=	35.61	.00
H2, H2PH2=	35.61	.00

CONDENSER ITERATION:

T	P	H	X
1.524+002	2.166+002	1.046+002	1.000+000
8.604+001	2.033+002	3.561+001	.000

LIQUID LINE:

INPUT -	PIN = 2.025+002	TIN = 8.604+001	XIN = .000
OUTPUT -	POUT = 2.033+002	TOUT = 8.531+001	XOUT = 3.958+003

HIN = 3.561+001

EXPANSION DEVICE:

INPUT -	P12 = 2.033+002	H12 = 3.561+001	P13 = 9.386+001
OUTPUT -	POUT = 1.560+002	XMASS = 4.666+002	

INPUT DATA TO EVAPHX:

T	P	X	TAIR	RH	RMASS
3.575+001	9.386+001	2.303+001	4.700+001	7.300+001	4.660+002

H2=	91.0376
H2=	89.3069
H2=	87.2842
H2=	86.8236
QT, QS=	2.369+004 2.114+004

EVAPORATOR ITERATION:

T	P	H	X	TSUP
3.575+001	9.386+001	3.561+001	2.303+001	

```

580 3.735+001 7.356+001 8.689+001 9.632-001 .000
581
582 INPUT DATA TO EVAPHX:
583
584 T P X TAIR RH RMASS
585 3.554+001 9.352+001 2.311-001 4.700+001 7.300-001 4.650+002
586 H2= 91.1463
587 H2= 89.8963
588 H2= 88.5308
589 H2= 88.0712
590 QT,OS= 2.445+004 2.152+004
591
592 EVAPURATOR ITERATION:
593
594 T P H X TSUP
595 3.554+001 9.352-001 3.561+001 2.311-001
596 3.701+001 7.270+001 8.600+001 9.821-001 .000
597 P1,P1E,P1EP1,DENT2= 72.5026 72.6990 .0965 .1938
598 INTERM,INTER,P13= 0 0 156.00 93.52
599 COMPOSITION OF REFRIG. IN ACCUMULATOR = .650
600 (WEIGHT FRACTION OF MORE VOLATILE COMPONENT)
601
602 REFRIG. IN ACCUMULATOR = .181 LB
603
604 REFRIG. IN INDOOR COIL = 2.311 LB
605
606 REFRIG. IN OUTDOOR COIL= .848 LB
607
608 TMASS = 4.434+000 REFIN = 4.968+000
609
610 *****
611 **SYST4 WITH ACCUM., 13B1/152A 6-22-84***
612
613 TOA RHQA TRA RHRA
614 4.700+001 7.300-001 7.000+001 5.600-001
615
616 CFMIND CFMOUT
617 1.118+003 2.264+003
618
619
620
621
622
623
624
625
626
627
628
629
630
631
632
633
634
635
636
637

```

RESULTS:

I	T	P	H	S	X
1	3.689+001	7.260+001	8.789+001	1.921-001	9.801-001
2	3.623+001	7.178+001	8.785+001	1.921-001	9.801-001
3	3.169+001	6.532+001	8.916+001	1.955-001	1.000+000
4	7.044+001	6.523+001	9.551+001	2.039-001	1.000+000
5	7.422+001	6.254+001	9.624+001	2.110-001	1.000+000
6	1.857+002	2.404+002	1.105+002	2.122-001	1.000+000
7	1.701+002	2.201+002	1.098+002	2.126-001	1.000+000
8	1.604+002	2.191+002	1.061+002	2.068-001	1.000+000
9	1.533+002	2.167+002	1.048+002	2.043-001	1.000+000
10	1.524+002	2.166+002	1.046+002	2.045-001	1.000+000
11	8.604+001	2.036+002	3.561+001	8.132-002	.000
12	8.531+001	2.033+002	3.561+001	8.135-002	3.958-003
13	3.554+001	9.352+001	3.561+001	8.335-002	2.311-001

TG3 TSUP3 TG5 RMASS TMASS

3.169+001 .000 , 1.118+002 4.660+002 4.434+000

QLGAD ELUSE COP

3.381+004 3.551+000 2.790+000

REFRIG. COMPOSITION = .650
(WEIGHT FRACTION OF MORE VOLATILE COMPONENT)

INPUT DATA TO COMP:

F3 T3 H3 XQ3 TG6 TRA TGA
6.657+001 3.226+001 8.740+001 9.800-001 1.138+002 7.000+001 4.700+001

COMPRESSOR ITERATION:

EI ETAE ETAC ETAV CPRPM RMASS
2.749+000 7.589-001 9.461-001 8.563-001 3.543+003 4.823+002

I T P H X
1 3.756+001 7.407+001 8.622+001 9.504-001
2 3.692+001 7.320+001 8.617+001 9.506-001
3 3.226+001 6.657+001 8.740+001 9.800-001
4 6.457+001 6.648+001 9.448+001 1.000+000
5 6.829+001 6.369+001 9.521+001 1.000+000
6 1.812+002 2.471+002 1.094+002 1.000+000
7 1.744+002 2.263+002 1.087+002 1.000+000
8 1.550+002 2.253+002 1.050+002 1.000+000
9 1.492+002 2.229+002 1.037+002 1.000+000
10 1.485+002 2.228+002 1.036+002 1.000+000

INPUT DATA TO CONDHX:

T P TAIR RH RMASS
1.485+002 2.228+002 7.000+001 5.600-001 4.323+002

H2, H2PH2= 33.96 -.42
H2, H2PH2= 33.90 .06
H2, H2PH2= 34.44 -.54
H2, H2PH2= 34.20 .24
H2, H2PH2= 34.47 -.27
H2, H2PH2= 34.53 -.06
H2, H2PH2= 34.58 -.05
H2, H2PH2= 34.55 .01

CONDENSER ITERATION:

T P H X
1.485+002 2.228+002 1.036+002 1.000+000
8.205+001 2.152+002 3.456+001 .000

LIQUID LINE:

INPUT - PIN = 2.152+002 TIN = 8.205+001 XIN = .000
OUTPUT - POUT = 2.095+002 TOUT = 8.231+001 XOUT = .000

HIN = 3.456+001


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696 EXPANSION DEVICE:
697 INPUT - P12 =2.096+002 H12 =3.456+001 P13 =9.499+001
698 OUTPUT - POUT =1.668+002 XMASS =5.173+002
699
700 INPUT DATA TO COMP:
701
702 P3 T3 H3 X03 TG6 TRA T0A
703 6.657+001 3.226+001 8.740+001 9.809+001 1.134+002 7.009+001 4.700+001
704
705 COMPRESSOR ITERATION:
706
707 EI ETAE ETAC ETAV CPRPM RMASS
708 2.738+000 7.584+001 9.463+001 8.571+001 3.543+003 4.828+002
709
710 I T P H X
711 1 3.757+001 7.409+001 8.623+001 9.605+001
712 2 3.693+001 7.321+001 8.618+001 9.607+001
713 3 3.226+001 6.957+001 8.740+001 9.809+001
714 4 6.443+001 6.640+001 9.446+001 1.000+000
715 5 6.812+001 6.369+001 9.518+001 1.000+000
716 6 1.804+002 2.456+002 1.093+002 1.000+000
717 7 1.736+002 2.246+002 1.086+002 1.000+000
718 8 1.551+002 2.236+002 1.049+002 1.000+000
719 9 1.486+002 2.211+002 1.037+002 1.000+000
720 10 1.478+002 2.210+002 1.035+002 1.000+000
721
722 INPUT DATA TO CONDX:
723
724 T P TAIR RH RMASS
725 1.478+002 2.210+002 7.000+001 5.600+001 4.828+002
726 H2,H2PH2= 34.11 -.04
727 H2,H2PH2= 34.26 -.25
728 H2,H2PH2= 34.99 -.64
729 H2,H2PH2= 34.73 .27
730 H2,H2PH2= 34.97 -.24
731 H2,H2PH2= 35.06 -.09
732 H2,H2PH2= 35.11 -.05
733 H2,H2PH2= 35.10 .00
734
735 CONDENSER ITERATION:
736
737 T P H X
738 1.478+002 2.210+002 1.035+002 1.000+000
739 8.411+001 2.128+002 3.510+001 .000
740
741 LIQUID LINE:
742 INPUT - PIN = 2.128+002 TIN = 8.411+001 XIN = .000
743 OUTPUT - POUT = 2.072+002 TOUT = 8.408+001 XOUT = .000
744
745 HIN = 3.510+001
746
747 EXPANSION DEVICE:
748 INPUT - P12 =2.072+002 H12 =3.510+001 P13 =9.501+001
749 OUTPUT - POUT =1.612+002 XMASS =4.916+002
750
751 INPUT DATA TO EVAPHX:
752
753 T P X TAIR RH RMASS

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754 3.636+001 9.501+001 2.220+001 4.700+001 7.300+001 4.830+002
755 H2= 90.4425
756 H2= 85.8278
757 H2= 83.4042
758 H2= 83.3173
759 QT, QS= 2.321+004 2.072+004
760
761 EVAPORATOR ITERATION:
762
763 T P H X TSUP
764 3.636+001 9.501+001 3.528+001 2.220+001
765 3.762+001 7.516+001 8.334+001 9.272+001 .000
766
767 INPUT DATA TO EVAPHX:
768
769 T P X TAIR RH RMASS
770 3.613+001 9.463+001 2.229+001 4.700+001 7.300+001 4.830+002
771 H2= 90.7213
772 H2= 87.2462
773 H2= 84.9296
774 QT, QS= 2.397+004 2.152+004
775
776 EVAPORATOR ITERATION:
777
778 T P H X TSUP
779 3.613+001 9.463+001 3.528+001 2.229+001
780 3.732+001 7.419+001 8.491+001 9.457+001 .000
781 P1, PIE1, DENT2= 74.0893 74.1922 .1030 -1.3240
782
783 INPUT DATA TO COMPR:
784
785 P3 T3 H3 XQ3 TG6 TRA TGA
786 6.588+001 3.172+001 8.734+001 9.800+001 1.128+002 7.000+001 4.700+001
787
788 COMPRESSOR ITERATION:
789
790 EI ETAE ETAC ETAV CRRPM RMASS
791 2.717+000 7.575+001 9.465+001 8.565+001 3.549+003 4.775+002
792
793 I T P H X
794 1 3.706+001 7.331+001 8.617+001 9.505+001
795 2 3.636+001 7.244+001 8.611+001 9.607+001
796 3 3.172+001 6.569+001 8.734+001 9.800+001
797 4 6.422+001 6.579+001 9.445+001 1.000+000
798 5 6.794+001 6.502+001 9.518+001 1.000+000
799 6 1.802+002 2.435+002 1.093+002 1.000+000
800 7 1.734+002 2.228+002 1.066+002 1.000+000
801 8 1.548+002 2.217+002 1.049+002 1.000+000
802 9 1.483+002 2.193+002 1.037+002 1.000+000
803 10 1.475+002 2.192+002 1.035+002 1.000+000
804
805 INPUT DATA TO CONDHX:
806
807 T P TAIR RH RMASS
808 1.475+002 2.192+002 7.000+001 5.600+001 4.775+002
809 H2, H2PH2= 34.21 .03
810 H2, H2PH2= 34.42 -.21
811 H2, H2PH2= 35.16 -.74

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812 H2,H2PH2= 35.42 - .27
813 H2,H2PH2= 34.83 .59
814 H2,H2PH2= 35.01 - .18
815 H2,H2PH2= 35.14 - .12
816 H2,H2PH2= 35.21 - .07
817 H2,H2PH2= 35.20 .01
818
819 CONDENSER ITERATION:
820
821 T P H X
822 1.475+002 2.192+002 1.035+002 1.000+000
823 8.445+001 2.108+002 3.520+001 .000
824
825
826 LIQUID LINE:
827 INPUT - PIN = 2.108+002 TIN = 8.445+001 XIN = .000
828 OUTPUT - POUT = 2.053+002 TOUT = 8.444+001 XOUT = .000
829
830
831 EXPANSION DEVICE:
832 INPUT - P12 = 2.053+002 H12 = 3.520+001 P13 = 9.385+001
833 OUTPUT - POUT = 1.594+002 XMASS = 4.826+002
834
835 INPUT DATA TO COMPR:
836
837 P3 T3 H3 XQ3 TG6 TRA TGA
838 6.588+001 3.172+001 8.734+001 9.800+001 1.127+002 7.000+001 4.700+001
839
840 COMPRESSOR ITERATION:
841
842 EI ETAE ETAC ETAV CPRPM RMASS
843 2 714+000 7.574+001 9.465+001 8.568+001 3.549+003 4.776+002
844
845 I T P H X
846 1 3.700+001 7.332+001 8.617+001 9.606+001
847 2 3.606+001 7.245+001 8.611+001 9.607+001
848 3 3.172+001 6.588+001 8.734+001 9.800+001
849 4 6.420+001 6.573+001 9.445+001 1.000+000
850 5 6.791+001 6.303+001 9.517+001 1.000+000
851 6 1.801+002 2.432+002 1.093+002 1.000+000
852 7 1.732+002 2.224+002 1.086+002 1.000+000
853 8 1.547+002 2.214+002 1.049+002 1.000+000
854 9 1.481+002 2.190+002 1.037+002 1.000+000
855 10 1.474+002 2.108+002 1.035+002 1.000+000
856
857 INPUT DATA TO CONDHX:
858
859 T P TAIR RH RMASS
860 1.474+002 2.189+002 7.000+001 5.600+001 4.776+002
861 H2,H2PH2= 34.40 - .02
862 H2,H2PH2= 34.61 - .21
863 H2,H2PH2= 35.33 - .71
864 H2,H2PH2= 34.98 .35
865 H2,H2PH2= 35.22 - .24
866 H2,H2PH2= 35.35 - .13
867 H2,H2PH2= 35.42 - .08
868 H2,H2PH2= 35.42 .01
869

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CONDENSER ITERATION:

T	P	H	X
1.474+002	2.189+002	1.035+002	1.000+000
8.529+001	2.124+002	3.542+001	.000

LIQUID LINE:

INPUT -	PiN = 2.104+002	TiN = 8.529+001	XiN = .000
OUTPUT -	POUT = 2.049+002	TOUT = 8.538+001	XOUT = .000

HIN = 3.542+001

EXPANSION DEVICE:

INPUT -	P12 = 2.049+002	H12 = 3.542+001	P13 = 9.385+001
OUTPUT -	POUT = 1.576+002 <td>XMASS = 4.751+002 <td></td> </td>	XMASS = 4.751+002 <td></td>	

INPUT DATA TO EVAPHX:

T	P	X	TAIR	RH	RMASS
3.567+001	9.385+001	2.258-001	4.700+001	7.300-001	4.776+002

H2= 91.0917

H2= 89.5983

H2= 87.5974

H2= 87.4747

H2= 87.3638

QT, QS= 2.484+004 2.154+004

EVAPORATOR ITERATION:

T	P	H	X	TSUP
3.567+001	9.385+001	3.535+001	2.258-001	
3.683+001	7.269+001	8.737+001	9.742-001	.000

INPUT DATA TO EVAPHX:

T	P	X	TAIR	RH	RMASS
3.581+001	9.403+001	2.252-001	4.700+001	7.300-001	4.776+002

H2= 90.9892

H2= 89.0026

H2= 86.7588

H2= 86.6626

H2= 86.5777

QT, QS= 2.447+004 2.135+004

EVAPORATOR ITERATION:

T	P	H	X	TSUP
3.581+001	9.408+001	3.534+001	2.252-001	
3.706+001	7.326+001	8.659+001	9.652-001	.000
P1, P1E, P1EP1, PENT2=	73.3157	73.2571	-0.0586	.4132
INTERM, INTERE, POUT, P13=	1	0	157.55	94.08

INPUT DATA TO COMP:

P3	T3	H3	XG3	TG6	TRA	TOA
6.588+001	3.172+001	8.734+001	9.800-001	1.127+002	7.000+001	4.700+001

COMPRESSOR ITERATION:

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928 EI      ETAE      ETAC      ETAV      CPRPM      RMASS
929 2.715+000 7.575-001 5.465-001 8.567-001 3.549+003 4.776+002
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      T      P      H      X
1 3.700+001 7.331+001 8.617+001 9.606-001
2 3.635+001 7.245+001 8.611+001 9.607-001
3 3.172+001 5.588+001 8.734+001 9.800-001
4 6.420+001 6.579+001 9.445+001 1.000+000
5 6.791+001 6.303+001 9.518+001 1.000+000
6 1.301+002 2.433+002 1.083+002 1.000+000
7 1.733+002 2.225+002 1.086+002 1.000+000
8 1.547+002 2.215+002 1.042+002 1.000+000
9 1.482+002 2.191+002 1.037+002 1.000+000
10 1.474+002 2.190+002 1.035+002 1.000+000

      INPUT DATA TO CONDHX:

      T      P      TAIR      RH      RMASS
1.474+002 2.190+002 7.000+001 5.600-001 4.776+002
H2,H2PH2= 34.33 .04
H2,H2PH2= 34.55 -.22
H2,H2PH2= 35.27 -.72
H2,H2PH2= 34.91 .36
H2,H2PH2= 35.15 -.23
H2,H2PH2= 35.27 -.13
H2,H2PH2= 35.35 -.08
H2,H2PH2= 35.34 .01

      CONDENSER ITERATION:

      T      P      H      X
1.474+002 2.190+002 1.035+002 1.000+000
8.502+001 2.105+002 3.534+001 .000

      LIQUID LINE:
      INPUT - PIN = 2.105+002 TIN = 8.502+001 XIN = .000
      OUTPUT - POUT= 2.050+002 TOUT= 8.500+001 XOUT= .000
      HIN = 3.534+001

      EXPANSION DEVICE:
      INPUT - P12 =2.050+002 H12 =3.534+001 P13 =9.407+001
      DENFA DOES NOT CONVERGE, DIFFXQ2= -8.01178-006
      OUTPUT - POUT =1.534+002 XMASS =4.730+002

      INPUT DATA TO EVAPHX:

      T      P      X      TAIR      RH      RMASS
3.581+001 9.408+001 2.252-001 4.700+001 7.300-001 4.776+002
H2= 90.9594
H2= 89.0039
H2= 86.7603
H2= 86.6543
H2= 86.5792
QT,QS= 2.447+004 2.135+004

      EVAPORATOR ITERATION:

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986 T      P      H      X      TSUP
987 3.581+001 9.408+001 3.534+001 2.252-001
988 3.706+001 7.326+001 8.659+001 9.052-001 .000
989 P1,PIE,PIEP1,DENT2= 73.3148 73.2557 -.0591 .4153
990 INTERM,INTERE,FOUT,P13= 0 0 158.40 94.08
991 WACCUM DOES NOT CONVERGE, MAX.ERROR = 2.982-002 (LB)
992 COMPOSITION OF REFRIG. IN ACCUMULATOR = .333
993 (WEIGHT FRACTION OF MORE VOLATILE COMPONENT)
994 REFRIG. IN ACCUMULATOR = 2.740 LB
995 REFRIG. IN INDOOR COIL = 2.361 LB
996 REFRIG. IN OUTDOOR COIL = .885 L3
997
998 TMASS = 7.110+000 REFIN = 4.968+000
999
1000 *****
1001 **SYST4 WITH ACCUM., 1381/152A 6-22-84****
1002
1003 TOA      RHQA      TRA      RHRA
1004 4.700+001 7.300-001 7.000+001 5.600-001
1005 CFMIND      CFMCUT
1006 1.118+003 2.284+003

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RESULTS:

I	T	P	H	S	X
1	3.700+001	7.331+001	8.617+001	1.884-001	9.606-001
2	3.636+001	7.245+001	8.611+001	1.885-001	9.607-001
3	3.172+001	6.588+001	8.734+001	1.926-001	9.800-001
4	6.420+001	6.575+001	9.445+001	2.050-001	1.000+000
5	6.791+001	6.303+001	9.518+001	2.089-001	1.000+000
6	1.801+002	2.433+002	1.093+002	2.101-001	1.000+000
7	1.733+002	2.225+002	1.086+002	2.104-001	1.000+000
8	1.547+002	2.215+002	1.049+002	2.046-001	1.000+000
9	1.482+002	2.191+002	1.037+002	2.023-001	1.000+000
10	1.474+002	2.190+002	1.035+002	2.025-001	1.000+000
11	8.502+001	2.105+002	3.534+001	8.082-002	.000
12	8.500+001	2.050+002	3.534+001	8.095-002	.000
13	3.581+001	9.403+001	3.534+001	8.278-002	2.252-001

```

TG3      TSUP3      TG6      RMASS      TMASS
3.214+001 .000      1.127+002 4.776+002 7.110+000

QLOAD      ELUSE      COP
3.422+004 3.560+000 2.801+000

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REFRIG. COMPOSITION = .650
(WEIGHT FRACTION OF MORE VOLATILE COMPONENT)

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INPUT DATA TO CONF:

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1044 P3 T3 H3 XG3 TGG TRA TGA
1045 6.543+001 3.157+001 8.823+001 9.900+001 1.120+002 7.000+001 4.700+001
1046
1047 COMPRESSOR ITERATION:
1048
1049 EI ETAE ETAC ETAV CPRPM RMASS
1050 2.694+000 7.566+001 9.474+001 8.562+001 3.549+003 4.705+002
1051
1052 I T P H X
1053 1 3.691+001 7.277+001 8.702+001 9.704+001
1054 2 3.618+001 7.195+001 8.696+001 9.704+001
1055 3 3.157+001 6.543+001 8.823+001 9.900+001
1056 4 6.723+001 6.534+001 9.493+001 1.000+000
1057 5 7.102+001 6.262+001 9.571+001 1.000+000
1058 6 1.828+002 2.410+002 1.099+002 1.000+000
1059 7 1.760+002 2.205+002 1.092+002 1.000+000
1060 8 1.574+002 2.197+002 1.035+002 1.000+000
1061 9 1.506+002 2.170+002 1.042+002 1.000+000
1062 10 1.488+002 2.170+002 1.041+002 1.000+000
1063
1064 INPUT DATA TO CONDHX:
1065
1066 T P TAIR RH RMASS
1067 1.498+002 2.170+002 7.000+001 5.600+001 4.705+002
1068 H2,H2PH2= 34.65 .08
1069 H2,H2PH2= 34.75 -.10
1070 H2,H2PH2= 35.33 -.58
1071 H2,H2PH2= 35.65 -.32
1072 H2,H2PH2= 35.76 -.11
1073 H2,H2PH2= 35.76 .00
1074 H2,H2PH2= 35.76 .00
1075
1076 CONDENSER ITERATION:
1077
1078 T P H X
1079 1.498+002 2.170+002 1.041+002 1.000+000
1080 8.660+001 2.080+002 3.576+001 .000
1081
1082 LIQUID LINE:
1083 INPUT - PIN = 2.088+002 TIN = 8.660+001 XIN = .000
1084 OUTPUT - POUT = 2.033+002 TOUT = 8.533+001 XOUT = 6.996+003
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1102      I   T       P       H       X
1103      1   3.681+001  7.277+001  8.702+001  9.703-001
1104      2   3.618+001  7.192+001  8.696+001  9.704-001
1105      3   3.157+001  6.543+001  8.823+001  9.900-001
1106      4   6.731+001  6.534+001  9.498+001  1.000+000
1107      5   7.100+001  6.263+001  9.571+001  1.000+000
1108      6   1.820+002  2.413+002  1.099+002  1.000+000
1109      7   1.752+002  2.202+002  1.092+002  1.000+000
1110      8   1.575+002  2.198+002  1.055+002  1.000+000
1111      9   1.507+002  2.174+002  1.043+002  1.000+000
1112      10  1.439+002  2.173+002  1.041+002  1.000+000
1113
1114      INPUT DATA TO CONDHX:
1115
1116
1117      T       P       TAIR       RH       RMASS
1118      1.499+002  2.173+002  7.030+001  5.600-001  4.704+002
1119      H2,H2PH2=  34.43      .07
1120      H2,H2PH2=  35.21      -.78
1121      H2,H2PH2=  35.21      -.01
1122      H2,H2PH2=  35.29      -.07
1123      H2,H2PH2=  35.46      -.17
1124      H2,H2PH2=  35.55      -.09
1125      H2,H2PH2=  35.56      -.01
1126
1127      CONDENSER ITERATION:
1128
1129      T       P       H       X
1130      1.499+002  2.173+002  1.041+002  1.000+000
1131      8.584+001  2.092+002  3.556+001  .000
1132
1133      LIQUID LINE:
1134      INPUT -   PIN = 2.092+002   TIN = 8.584+001   XIN = .000
1135      OUTPUT -   POUT= 2.038+002   TOUT= 8.549+001   XOUT= 1.859-003
1136
1137
1138
1139      EXPANSION DEVICE:
1140      INPUT -   P12 =2.038+002   H12 =3.556+001   P13 =9.353+001
1141      OUTPUT -   POUT =1.571+002   XMASS =4.701+002
1142
1143      INPUT DATA TO EVAPHX:
1144
1145      T       P       X       TAIR       RH       RMASS
1146      3.553+001  9.353+001  2.302-001  4.700+001  7.300-001  4.704+002
1147      H2= 91.1647
1148      H2= 89.3796
1149      H2= 88.6931
1150      H2= 88.3154
1151      Q1,QS=  2.480+004  2.173+004
1152
1153      EVAPORATOR ITERATION:
1154
1155      T       P       H       X       TSUP
1156      3.553+001  9.353+001  3.553+001  2.302-001
1157      3.679+001  7.234+001  8.828+001  9.846-001  .000
1158
1159      INPUT DATA TO EVAPHX:

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HIN = 3.556+001

1218 INPUT -
1219 OUTPUT -
1220
1221

EXPANSION DEVICE:

1222 INPUT - P12 = 2.044+002 H12 = 3.546+001 P13 = 9.414+001
1223 OUTPUT - POUT = 1.572+002 XMASS = 4.727+002

1224 INPUT DATA TO EVAPHX:

1225
1226
1227
1228 T P X TAIR RH RMAS
1229 3.588+001 9.414+001 2.270-001 4.700+001 7.300-001 4.736+002
1230 H2= 90.9402
1231 H2= 88.7216
1232 H2= 86.4685
1233 H2= 86.2656
1234 QT, QS= 2.405+004 2.122+004

1235 EVAPORATOR ITERATION:

1236
1237
1238 T P H X TSUP
1239 3.588+001 9.414+001 3.546+001 2.270-001
1240 3.726+001 7.368+001 8.623+001 9.609-001 .000

1241 INPUT DATA TO EVAPHX:

1242
1243
1244 T P X TAIR RH RMAS
1245 3.578+001 9.390+001 2.274-001 4.700+001 7.300-001 4.736+002
1246 H2= 91.0247
1247 H2= 89.2156
1248 H2= 87.0729
1249 H2= 86.6713
1250 H2= 86.8123
1251 QT, QS= 2.433+004 2.127+004

1252 EVAPORATOR ITERATION:

1253
1254
1255 T P H X TSUP
1256 3.578+001 9.390+001 3.546+001 2.274-001
1257 3.712+001 7.327+001 8.682+001 9.678-001 .000
1258 P1, P1E, P1E1, DENT2= 73.2274 73.2701 .0427 -.2317
1259 INTERM, INTERE, POUT, P13= 0 0 157.22 93.98
1260 COMPOSITION OF REFRIG. IN ACCUMULATOR = .365
1261 (WEIGHT FRACTION OF MORE VOLATILE COMPONENT)

1262 REFRIG. IN ACCUMULATOR = 1.083 LB

1263 REFRIG. IN INDOOR COIL = 2.340 LB

1264 REFRIG. IN OUTDOOR COIL = .878 LB

1265 TMASS = 5.419+000 REFIN = 4.968+000

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SYST4 WITH ACCUM., 13B1/152A 6-22-84*

TOA RHOA TRA RHRA
4.700+001 7.300-001 7.000+001 5.600-001

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CFMIN D CFMOUT
1.118+003 2.234+003

RESULTS:

I	T	P	H	S	X
1	3.713+001	7.323+001	8.705+001	1.902-001	9.703-001
2	3.652+001	7.238+001	8.699+001	1.903-001	9.704-001
3	3.180+001	6.564+001	8.827+001	1.945-001	9.900-001
4	6.740+001	6.575+001	9.498+001	2.078-001	1.000+000
5	7.114+001	6.301+001	9.571+001	2.099-001	1.000+000
6	1.829+002	2.422+002	1.099+002	2.111-001	1.000+000
7	1.761+002	2.215+002	1.092+002	2.115-001	1.000+000
8	1.576+002	2.205+002	1.055+002	2.057-001	1.000+000
9	1.508+002	2.181+002	1.042+002	2.038-001	1.000+000
10	1.500+002	2.180+002	1.041+002	2.035-001	1.000+000
11	8.545+001	2.099+002	3.546+001	8.104-002	.000
12	6.543+001	2.044+002	3.546+001	8.106-002	.000
13	3.578+001	9.398+001	3.546+001	8.302-002	2.274-001

TG3	TSUP3	TG6	RMASS	TMASS
3.210+001	.000	1.124+002	4.736+002	5.419+000

QLOAD	ELUSE	COP
3.415+004	3.572+000	2.802+000

REFRIG. COMPOSITION = .650
(WEIGHT FRACTION OF MORE VOLATILE COMPONENT)

INPUT DATA TO COMPR:

P3	T3	H3	XQ3	TG6	TRA	TOA
6.582+001	3.199+001	8.873+001	9.950-001	1.123+002	7.000+001	4.700+001

COMPRESSOR ITERATION:

EI	ETA E	ETAC	ETAV	C:RPM	RMASS
2.702+000	7.569-001	9.477-001	8.563-001	3.549+003	4.717+002

I	T	P	H	X
1	3.723+001	7.319+001	8.749+001	9.752-001
2	3.661+001	7.236+001	8.743+001	9.753-001
3	3.199+001	6.582+001	8.873+001	9.950-001
4	6.899+001	6.573+001	9.524+001	1.000+000
5	7.273+001	6.301+001	9.597+001	1.000+000
6	1.843+002	2.419+002	1.102+002	1.000+000
7	1.775+002	2.213+002	1.095+002	1.000+000
8	1.590+002	2.203+002	1.059+002	1.000+000
9	1.520+002	2.179+002	1.045+002	1.000+000
10	1.512+002	2.178+002	1.043+002	1.000+000

INPUT DATA TO CONDHX:

1334	T	1.512+002	P	2.178+002	TAIR	RH	RMASS
1335	H2, H2PH2=	34.33			7.000+001	5.600-001	4.717+002
1336	H2, H2PI2=	35.12					
1337	H2, H2PH2=	34.88					
1338	H2, H2PH2=	35.21					
1339	H2, H2PH2=	35.36					
1340	H2, H2PH2=	35.44					
1341	H2, H2PH2=	35.44					
1342	H2, H2PH2=	35.44					
1343							
1344							
1345							
1346							
1347	T	1.512+002	P	2.173+002	H	X	
1348	8.538+001	2.097+002			1.043+002	1.000+000	
1349					3.544+001	.000	
1350							
1351							
1352							
1353							
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1356							
1357							
1358							
1359							
1360							
1361							
1362							
1363	T	3.575+001	P	9.394+001	TAIR	RH	RMASS
1364	H2=	91.0172			2.271-001	7.300-001	4.717+002
1365	H2=	69.1768					
1366	H2=	87.0406					
1367	H2=	86.8044					
1368	QT, QS=	2.421+004			2.132+004		
1369							
1370							
1371							
1372							
1373	T	3.575+001	P	9.394+001	H	X	TSUP
1374	3.720+001	7.339+001			3.544+001	2.271-001	
1375	P1, P1E, P1EP1, DENT2=	73.1945			8.878+001	9.669-001	.000
1376					73.3925	.1981	-.7351
1377							
1378							
1379							
1380	P3	T3			H3	XQ3	TG6
1381	6.545+001	3.168+001			8.870+001	9.950-001	1.120+002
1382							TRA
1383							TGA
1384							
1385							
1386							
1387							
1388							
1389							
1390							
1391							

CONDENSER ITERATION:

LIQUID LINE:

EXPANSION DEVICE:

INPUT DATA TO EVAFHX:

EVAPORATOR ITERATION:

COMPRESSOR ITERATION:

HIN = 3.544+001

1392 5 7.266+001 6.265+001 9.599+001 1.000+000
1393 6 1.843+002 2.410+002 1.102+002 1.000+000
1394 7 1.776+002 2.206+002 1.095+002 1.000+000
1395 8 1.589+002 2.196+002 1.058+002 1.000+000
1396 9 1.520+002 2.172+002 1.045+002 1.000+000
1397 10 1.512+002 2.171+002 1.044+002 1.000+000
1398
1399 INPUT DATA TO CONDHX:
1400
1401 T P TAIR RH RMASS
1402 1.512+002 2.171+002 7.000+001 5.600-001 4.687+002
1403 H2,H2PH2= 34.46 .09
1404 H2,H2PH2= 35.21 -.75
1405 H2,H2PH2= 34.96 .25
1406 H2,H2PH2= 35.32 -.35
1407 H2,H2PH2= 35.47 -.15
1408 H2,H2PH2= 35.54 -.08
1409 H2,H2PH2= 35.54 .01
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1411 CONDENSER ITERATION:
1412
1413 T P H X
1414 1.512+002 2.171+002 1.044+002 1.000+000
1415 8.575+001 2.091+002 3.554+001 .000
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1419 LIQUID LINE:
1420 INPUT - PIN = 2.091+002 TIN = 8.575+001 XIN = .000
1421 OUTPUT - POUT = 2.037+002 TOUT = 8.546+001 XOUT = 1.539-003
1422
1423
1424 EXPANSION DEVICE:
1425 INPUT - P12 = 2.037+002 H12 = 3.554+001 P13 = 9.351+001
1426 OUTPUT - POUT = 1.566+002 XMASS = 4.695+002
1427
1428 INPUT DATA TO EVAPHX:
1429 T P X TAIR RH RMASS
1430 3.551+001 9.351+001 2.298-001 4.700+001 7.300-001 4.587+002
1431 H2= 91.1552
1432 H2= 89.9336
1433 H2= 88.6044
1434 H2= 88.2019
1435 QT, QS= 2.458+004 2.165+004
1436
1437 EVAPORATOR ITERATION:
1438
1439 T P H X TSUP
1440 3.551+001 9.351+001 3.554+001 2.298-001
1441 3.688+001 7.249+001 8.818+001 9.834-001 .000
1442
1443 INPUT DATA TO EVAPHX:
1444
1445 T P X TAIR RH RMASS
1446 3.557+001 9.361+001 2.296-001 4.700+001 7.300-001 4.687+002
1447 H2= 91.1332
1448 H2= 89.8058
1449 H2= 88.2652

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H2= 87.8343
QT,CS= 2.453+004 2.156+004

EVAPORATOR ITERATION:

T P H X TSUP
3.557+001 9.361+001 3.554+001 2.296-001
3.698+001 7.273+001 8.788+001 9.798-001 .000
P1,P1E,P1CP1,DENT2= 72.7637 72.7340 -.0297
INTERM,INTERE,ROUT,P13= 0 0 156.59 93.61
COMPOSITION OF REFRIG. IN ACCUMULATOR = .625
(WEIGHT FRACTION OF MORE VOLATILE COMPONENT) .4190

REFRIG. IN ACCUMULATOR = .196 LB

REFRIG. IN INDOOR COIL = 2.324 LB

REFRIG. IN OUTDOOR COIL = .854 LB

TMASS = 4.482+000 REFIN = 4.968+000

SYST4 WITH ACCUM., 13B1/152A 6-22-84*

TOA RHQA TRA RHRA
4.700+001 7.300-001 7.000+001 5.600-001

CFMIND CFMOUT
1.118+003 2.284+003

RESULTS:

I	T	P	H	S	X
1	3.691+001	7.276+001	8.746+001	1.912-001	9.752-001
2	3.629+001	7.193+001	8.740+001	1.912-001	9.753-001
3	3.168+001	6.545+001	8.870+001	1.955-001	9.950-001
4	6.890+001	6.536+001	9.524+001	2.084-001	1.000+000
5	7.266+001	6.263+001	9.598+001	2.105-001	1.000+000
6	1.843+002	2.410+002	1.102+002	2.117-001	1.000+000
7	1.776+002	2.206+002	1.095+002	2.120-001	1.000+000
8	1.589+002	2.196+002	1.059+002	2.062-001	1.000+000
9	1.520+002	2.172+002	1.045+002	2.043-001	1.000+000
10	1.512+002	2.171+002	1.044+002	2.040-001	1.000+000
11	8.575+001	2.091+002	3.554+001	8.118-002	.000
12	8.546+001	2.037+002	3.554+001	8.121-002	1.539-003
13	3.557+001	9.361+001	3.554+001	8.319-002	2.296-001

TG3 TSUP3 TG6 RMASS TMASS
3.179+001 .000 1.120+002 4.687+002 4.482+000

QLOAD ELUSE COP
3.391+004 3.557+000 2.794+000

REFRIG. COMPOSITION = .650
(WEIGHT FRACTION OF MORE VOLATILE COMPONENT)

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INPUT DATA TO COMP:

P3 T3 H3 XQ3 TG6 TRA TGA
6.565+001 3.179+001 8.848+001 9.925-001 1.122+002 7.000+001 4.700+001

COMPRESSOR ITERATION:

EI ETAE ETAC ETAV CPRPM PMASS
2.699+000 7.568-001 9.475-001 8.562-001 3.549+003 4.712+002

I	T	P	H	X
1	3.703+001	7.300+001	8.726+001	9.728-001
2	3.641+001	7.216+001	8.720+001	9.729-001
3	3.179+001	6.565+001	8.849+001	9.925-001
4	6.814+001	6.556+001	9.511+001	1.000+000
5	7.189+001	6.224+001	9.584+001	1.000+000
6	1.833+002	2.416+002	1.100+002	1.000+000
7	1.768+002	2.211+002	1.093+002	1.000+000
8	1.592+002	2.201+002	1.057+002	1.000+000
9	1.513+002	2.177+002	1.044+002	1.000+000
10	1.505+002	2.176+002	1.042+002	1.000+000

INPUT DATA TO CONDHX:

T	P	TAIR	RH	RMASS
1.505+002	2.176+002	7.000+001	5.600-001	4.712+002
H2,H2PH2=	34.37	.06		
H2,H2PH2=	35.16	-.79		
H2,H2PH2=	34.92	.23		
H2,H2PH2=	35.26	-.34		
H2,H2PH2=	35.41	-.15		
H2,H2PH2=	35.49	-.08		
H2,H2PH2=	35.49	.00		

CONDENSER ITERATION:

T	P	H	X
1.505+002	2.176+002	1.042+002	1.000+000
8.557+001	2.095+002	2.549+001	.000

LIQUID LINE:

INPUT - PIN = 2.095+002 TIN = 8.557+001 XIN = .000
O:INPUT - POUT = 2.041+002 TOUT = 8.555+001 XOUT = .000
HIN = 3.549+001

EXPANSION DEVICE:

INPUT - P12 = 2.041+002 H12 = 3.549+001 P13 = 9.385+001
OUTPUT - POUT = 1.571+002 XMASS = 4.712+002

INPUT DATA TO EVAPHX:

T	P	X	TAIR	RH	RMASS
3.571+001	9.385+001	2.222-001	4.700+001	7.300-001	4.712+002
H2= 91.0547					
H2= 89.4687					

H2= 87.4577
 H2= 87.1759
 QT, QS= 2.433+004 2.139+004
 EVAPORATOR ITERATION:
 T P H X TSUP
 3.571+001 3.385+001 3.549+001 2.282-001
 3.711+001 7.315+001 8.713+001 9.712-001 .000
 P1,PIE,P1EP1,DENT2= 73.0043 73.1545 .1503
 INTERM,INTER,P13= 0 0 157.13 93.85
 COMPOSITION OF REFRIG. IN ACCUMULATOR = .615
 (WEIGHT FRACTION OF MORE VOLATILE COMPONENT)
 REFRIG. IN ACCUMULATOR = .204 LB
 REFRIG. IN INDOOR COIL = 2.333 LB
 REFRIG. IN OUTDOOR COIL = .872 LB
 TMASS = 4.525+000 REFIN = 4.968+000

 SYST4 WITH ACCUM., 1301/152A 6-22-94*

- .1317

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RESULTS:
 I T P H S X
 1 3.703+001 7.300+001 8.726+001 1.907-001 9.728-001
 2 3.641+001 7.216+001 8.720+001 1.908-001 9.729-001
 3 3.179+001 6.565+001 8.848+001 1.950-001 9.925-001
 4 5.814+001 6.555+001 9.511+001 2.081-001 1.000+000
 5 7.189+001 6.284+001 9.584+001 2.102-001 1.000+000
 6 1.836+002 2.415+002 1.100+002 2.114-001 1.000+000
 7 1.768+002 2.211+002 1.093+002 2.117-001 1.000+000
 8 1.582+002 2.201+002 1.057+002 2.059-001 1.000+000
 9 1.513+002 2.177+002 1.044+002 2.040-001 1.000+000
 10 1.505+002 2.176+002 1.042+002 2.038-001 1.000+000
 11 8.557+001 2.095+002 3.549+001 8.109-002 .000
 12 8.555+001 2.041+002 3.549+001 8.112-002 .000
 13 3.571+001 9.385+001 3.549+001 8.308-002 2.282-001
 TG3 TSUP3 TG6 RMASS TMASS
 3.195+001 .000 1.122+002 4.712+002 4.523+000
 QLOAD ELUSE COP
 3.404+001 3.564+000 2.799+000
 REFRIG. COMPOSITION = .650
 (WEIGHT FRACTION OF MORE VOLATILE COMPONENT)

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INPUT DATA TO COMPR:

P3	T3	H3	XG3	TG5	TRA	TGA
6.777+001	3.344+001	8.854+001	9.912-001	1.141+002	7.000+001	4.700+001

COMPRESSOR ITERATION:

E1	ETA	ETAC	ETAV	CPRPM	RMASS
2.765+000	7.596-001	9.471-001	8.577-001	3.543+003	4.831+002

I	T	P	H	X
1	3.876+001	7.540+001	8.732+001	9.714-001
2	3.813+001	7.453+001	8.727+001	9.717-001
3	3.344+001	6.777+001	8.854+001	9.912-001
4	6.837+001	6.767+001	9.506+001	1.000+000
5	7.205+001	6.485+001	9.579+001	1.000+000
6	1.838+002	2.479+002	1.099+002	1.000+000
7	1.769+002	2.265+002	1.092+002	1.000+000
8	1.585+002	2.254+002	1.055+002	1.000+000
9	1.518+002	2.229+002	1.043+002	1.000+000
10	1.510+002	2.228+002	1.041+002	1.000+000

INPUT DATA TO CONDHX:

T	P	TAIR	RH	RMASS
1.510+002	2.228+002	7.000+001	5.600-001	4.881+002
H2,H2PH2=	33.92	-26		
H2,H2PH2=	34.60	-68		
H2,H2PH2=	34.59	.01		
H2,H2PH2=	34.78	-20		
H2,H2PH2=	34.88	-10		
H2,H2PH2=	34.95	-06		
H2,H2PH2=	34.95	.00		

CONDENSER ITERATION:

T	P	H	X
1.510+002	2.228+002	1.041+002	1.000+000
8.350+001	2.151+002	3.495+001	.000

LIQUID LINE:

INPUT	-	PIN = 2.151+002	TIN = 8.350+001	XIN = .000
OUTPUT	-	POUT = 2.093+002	TOUT = 8.348+001	XOUT = .000

HIN = 3.495+001

EXPANSION DEVICE:

INPUT	-	P12 = 2.093+002	H12 = 3.495+001	P13 = 9.625+001
OUTPUT	-	POUT = 1.643+002	XMASS = 5.049+002	

INPUT DATA TO COMPR:

P3	T3	H3	XG3	TG6	TRA	TGA
6.777+001	3.344+001	8.854+001	9.912-001	1.139+002	7.000+001	4.700+001

COMPRESSOR ITERATION:

EI	ETA E	ETAC	ETAV	CPRPM	RMASS
2.760+000	7.594-001	9.472-001	8.580-001	3.543+003	4.850+002
I	T	P	H	X	
1	3.876+001	7.541+001	8.732+001	9.715-001	
2	3.813+001	7.454+001	8.723+001	9.717-001	
3	3.344+001	6.777+001	8.854+001	9.912-001	
4	6.830+001	6.767+001	9.505+001	1.000+000	
5	7.197+001	6.485+001	9.577+001	1.000+000	
6	1.834+002	2.472+002	1.090+002	1.000+000	
7	1.765+002	2.257+002	1.091+002	1.000+000	
8	1.582+002	2.247+002	1.055+002	1.000+000	
9	1.515+002	2.221+002	1.042+002	1.000+000	
10	1.507+002	2.221+002	1.041+002	1.000+000	

INPUT DATA TO CONDHX:

T	P	TAIR	RH	RMASS
1.507+002	2.221+002	7.000+001	5.600-001	4.883+002
H2, H2PH2=	34.20	- .11		
H2, H2PH2=	34.44	- .25		
H2, H2PH2=	34.58	- .14		
H2, H2PH2=	34.82	- .23		
H2, H2PH2=	34.98	- .17		
H2, H2PH2=	35.02	- .04		
H2, H2PH2=	35.03	- .01		

CONDENSER ITERATION:

T	P	H	X
1.507+002	2.221+002	1.041+002	1.000+000
8.383+001	2.140+002	3.503+001	.000

LIQUID LINE:

INPUT -	PIN = 2.140+002	TIN = 8.383+001	XIN = .000
OUTPUT -	POUT = 2.080+002	TOUT = 8.380+001	XOUT = .000

HIN = 3.503+001

EXPANSION DEVICE:

INPUT -	P12 = 2.080+002	H12 = 3.503+001	P13 = 9.626+001
OUTPUT -	POUT = 1.631+002	XMASS = 4.383+002	

INPUT DATA TO EVAPHX:

T	P	X	TAIR	RH	RMASS
3.709+001	9.626+001	2.167-001	4.700+001	7.300-001	4.887+002
H2=	80.2819				
H2=	80.3385				
H2=	78.9247				
QT, QG=	2.140+004	1.974+004			

EVAPORATOR ITERATION:

T	P	H	X	TSUP
3.709+001	9.626+001	3.515+001	2.167-001	

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1740 3.831+001 7.777+001 7.894+001 8.742-001 .000
1741
1742 INPUT DATA TO EVAPHX:
1743
1744 T P X TAIR RH RMASS
1745 3.658+001 9.542+001 2.187-001 4.700+001 7.300-001 4.887+002
1746 H2= 89.9611
1747 H2= 84.2842
1748 H2= 81.9561
1749 H2= 81.8844
1750 QT, QS= 2.286+004 2.053+004
1751
1752 EVAPORATOR ITERATION:
1753
1754 T P H X TSUP
1755 3.658+001 9.542+001 3.515+001 2.187-001
1756 3.773+001 7.581+001 8.193+001 9.107-001 .000
1757
1758 INPUT DATA TO EVAPHX:
1759
1760 T P X TAIR RH RMASS
1761 3.647+001 9.525+001 2.191-001 4.700+001 7.300-001 4.897+002
1762 H2= 90.1903
1763 H2= 85.0249
1764 H2= 82.5664
1765 QT, QS= 2.319+004 2.091+004
1766
1767 EVAPORATOR ITERATION:
1768
1769 T P H X TSUP
1770 3.647+001 9.525+001 3.515+001 2.191-001
1771 3.760+001 7.539+001 8.260+001 9.187-001 .000
1772 P1, P1E, P1EPI, DENT2= 75.4106 75.3918 -.0188 -4.7252
1773
1774 INPUT DATA TO COMPRESSOR:
1775
1776 P3 T3 H3 X33 TGS TRA TGA
1777 6.531+001 3.149+001 8.834+001 9.912-001 1.119+002 7.000+001 4.700+001
1778
1779 COMPRESSOR ITERATION:
1780
1781 EI ETAE ETAC ETAV CPRPN RMASS
1782 2.690+000 7.564-001 9.475-001 8.561-001 3.540+003 4.690+002
1783
1784 I T P H X
1785 1 3.672+001 7.262+001 8.712+001 9.716-001
1786 2 3.610+001 7.178+001 8.706+001 9.716-001
1787 3 3.149+001 6.531+001 8.834+001 9.912-001
1788 4 6.765+001 6.521+001 9.564+001 1.000+000
1789 5 7.140+001 6.251+001 9.577+001 1.000+000
1790 6 1.831+002 2.406+002 1.100+002 1.000+000
1791 7 1.764+002 2.202+002 1.093+002 1.000+000
1792 8 1.578+002 2.192+002 1.056+002 1.000+000
1793 9 1.500+002 2.168+002 1.043+002 1.000+000
1794 10 1.501+002 2.167+002 1.042+002 1.000+000
1795
1796 INPUT DATA TO CONDHX:
1797

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1798	T	1.501+002	P	2.167+002	TAIR	7.000+001	RH	5.600-00	RMASS	4.690+002
1799	H2,H2PH2=	34.86								
1800	H2,H2PH2=	34.89								
1801	H2,H2PH2=	35.39								
1802	H2,H2PH2=	35.66								
1803	H2,H2PH2=	35.78								
1804	H2,H2PH2=	35.79								
1805	H2,H2PH2=	35.79								
1806	H2,H2PH2=	35.79								
1807	CONDENSER ITERATION:									
1808	T	1.501+002	P	2.167+002	H	1.042+002	X	1.000+000		
1809		8.671+001		2.036+002		3.579+001		.000		
1810	LIQUID LINE:									
1811	INPUT -	PIN = 2.086+002		TIN = 8.671+001		XIN = .000		HIN = 3.579+001		
1812	OUTPUT -	POUT = 2.030+002		TCUT = 8.525+001		XOUT = 8.115-003				
1813	EXPANSION DEVICE:									
1814	INPUT -	P12 = 2.030+002		H12 = 3.579+001		P13 = 9.246+001				
1815	OUTPUT -	POUT = 1.553+002		XMASS = 4.615+002						
1816	INPUT DATA TO COMPR:									
1817	P3	T3	H3	XQ3	TG6	TRA	TOA			
1818	6.531+001	3.149+001	8.834+001	9.912-00	1.121+002	7.000+001	4.700+001			
1819	COMPRESSOR ITERATION:									
1820	E1	ETA	ETAE	ETAC	ETAV	CPRPM	RMASS			
1821	2.695+000	7.566-001	9.474-001	8.557-001	3.519+003	4.688+002				
1822	I	T	P	H	X					
1823	1	3.672+001	7.261+001	8.711+001	9.715-001					
1824	2	3.609+001	7.177+001	8.705+001	9.716-001					
1825	3	3.149+001	6.531+001	8.834+001	9.912-001					
1826	4	6.771+001	6.521+001	9.505+001	1.000+000					
1827	5	7.147+001	6.251+001	9.579+001	1.000+000					
1828	6	1.835+002	2.413+002	1.100+002	1.000+000					
1829	7	1.768+002	2.210+002	1.093+002	1.000+000					
1830	8	1.581+002	2.200+002	1.056+002	1.000+000					
1831	9	1.512+002	2.176+002	1.043+002	1.000+000					
1832	10	1.504+002	2.175+002	1.042+002	1.000+000					
1833	INPUT DATA TO CONDHX:									
1834	T	1.504+002	P	2.175+002	TAIR	7.000+001	RH	5.600-001	RMASS	4.603+002
1835	H2,H2PH2=	34.72								
1836	H2,H2PH2=	34.80								
1837	H2,H2PH2=	34.85								
1838	H2,H2PH2=	34.96								
1839	H2,H2PH2=	35.10								
1840	H2,H2PH2=	35.27								


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1856 H2, H2PH2= 35.28 -.01
1857
1858 CONDENSER ITERATION:
1859
1860 T P H X
1861 1.504+002 2.175+002 1.042+002 1.000+000
1862 8.477+001 2.096+002 3.528+001 .000
1863
1864 LIQUID LINE:
1865 INPUT - PIN = 2.036+002 TIN = 8.477+001 XIN = .000
1866 OUTPUT - POUT = 2.042+002 TOUT = 8.475+001 XOUT = .000
1867
1868 HIN = 3.528+001
1869
1870 EXPANSION DEVICE:
1871 INPUT - P12 = 2.042+002 H12 = 3.528+001 P13 = 9.245+001
1872 DDENFA DOES NOT CONVERGE, DIFFNO2 = -5.66260-006
1873 OUTPUT - POUT = 1.580+002 XMASS = 4.766+002
1874
1875 INPUT DATA TO COMPR:
1876
1877 P3 T3 H3 XQ3 TGS TRA TGA
1878 6.531+001 3.149+001 8.834+001 9.912+001 1.120+002 7.000+001 4.700+001
1879
1880 COMPRESSOR ITERATION:
1881
1882 EI ETAE ETAC ETAV CPRPM RMASS
1883 2.692+000 7.565+001 9.475+001 8.559+001 3.549+003 4.639+002
1884
1885 I T P H X
1886 1 3.672+001 7.262+001 8.712+001 9.716+001
1887 2 3.610+001 7.178+001 8.705+001 9.716+001
1888 3 3.149+001 6.531+001 8.834+001 9.912+001
1889 4 6.768+001 6.521+001 9.505+001 1.000+000
1890 5 7.144+001 6.251+001 9.578+001 1.000+000
1891 6 1.833+002 2.410+002 1.100+002 1.000+000
1892 7 1.766+002 2.203+002 1.093+002 1.000+000
1893 8 1.579+002 2.195+002 1.056+002 1.000+000
1894 9 1.511+002 2.172+002 1.043+002 1.000+000
1895 10 1.503+002 2.171+002 1.042+002 1.000+000
1896
1897 INPUT DATA TO CONDHX:
1898
1899 T P TAIR RH RMASS
1900 1.503+002 2.171+002 7.000+001 5.600+001 4.689+002
1901 H2, H2PH2= 34.44 .08
1902 H2, H2PH2= 35.20 .76
1903 H2, H2PH2= 34.97 .23
1904 H2, H2PH2= 35.32 .35
1905 H2, H2PH2= 35.47 .15
1906 H2, H2PH2= 35.55 .08
1907 H2, H2PH2= 35.54 .01
1908
1909 CONDENSER ITERATION:
1910
1911 T P H X
1912 1.503+002 2.171+002 1.042+002 1.000+000
1913 8.577+001 2.091+002 3.554+001 .000

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LIQUID LINE:
INPUT - PIN = 2.091+002 TIN = 8.577+001 XIN = .000 HIN = 3.554+001
OUTPUT - POUT = 2.037+002 TOUT = 8.544+001 XOUT = 1.710-003

EXPANSION DEVICE:
INPUT - P12 = 2.037+002 H12 = 3.554+001 P13 = 3.245+001
OUTPUT - POUT = 1.562+002 XMASS = 4.692+002

INPUT DATA TO EVAPHX:
T 3.486+001 P 9.245+001 X 2.325-001 TAIR 4.700+001 RH 7.300-001 RMASS 4.689+002
H2= 91.4095
H2= 91.0235
H2= 90.6844
H2= 90.4082
H2= 90.3057
QT, QS= 2.567+004 2.190+004

EVAPORATOR ITERATION:
T 3.486+001 P 9.245+001 H 3.554+001 X 2.325-001 TSUP
3.988+001 6.970+001 9.028+001 1.000+000 4.700+000

INPUT DATA TO EVAPHX:
T 3.561+001 P 9.367+001 X 2.295-001 TAIR 4.700+001 RH 7.300-001 RMASS 4.689+002
H2= 91.1227
H2= 89.7332
H2= 88.0675
H2= 87.7171
QT, QS= 2.441+004 2.148+004

EVAPORATOR ITERATION:
T 3.561+001 P 9.367+001 H 3.554+001 X 2.295-001 TSUP
3.703+001 7.287+001 8.766+001 9.773-001 .000

INPUT DATA TO EVAPHX:
T 3.555+001 P 9.357+001 X 2.298-001 TAIR 4.700+001 RH 7.300-001 RMASS 4.689+002
H2= 91.1433
H2= 89.6541
H2= 88.4274
H2= 88.0450
QT, QS= 2.461+004 2.161+004

EVAPORATOR ITERATION:
T 3.555+001 P 9.357+001 H 3.554+001 X 2.298-001 TSUP

1972 3.693+001 7.261+001 8.803+001 9.815+001 .000
1973 P1,P1E,P1EP1,DENT2= 72.6171 72.6117 -.0054 .9032
1974
1975 INPUT DATA TO COMPR:
1976
1977 P3 T3 H3 XQ3 TG6 TRA TGA
1978 6.570+001 3.181+001 8.837+001 9.912+001 1.123+002 7.000+001 4.700+001
1979
1980 COMPRESSOR ITERATION:
1981
1982 EI ETAE ETAC ETAV ETAV CPRPM RMASS
1983 2.702+000 7.569+001 9.474+001 8.563+001 3.549+003 4.720+002
1984
1985 I T P H X
1986 1 3.705+001 7.306+001 8.715+001 9.716+001
1987 2 3.643+001 7.222+001 8.709+001 9.716+001
1988 3 3.181+001 6.570+001 8.837+001 9.912+001
1989 4 6.776+001 6.561+001 9.505+001 1.000+000
1990 5 7.150+001 6.238+001 9.578+001 1.000+000
1991 6 1.832+002 2.418+002 1.100+002 1.000+000
1992 7 1.763+002 2.212+002 1.093+002 1.000+000
1993 8 1.579+002 2.202+002 1.056+002 1.000+000
1994 9 1.511+002 2.170+002 1.043+002 1.000+000
1995 10 1.503+002 2.177+002 1.041+002 1.000+000
1996
1997 INPUT DATA TO CONDHX:
1998
1999 T P TAIR RH RMASS
2000 1.503+002 2.177+002 7.000+001 5.600+001 4.720+002
2001 H2,H2PH2= 34.34 .06
2002 H2,H2FH2= 35.14 -.80
2003 H2,H2PH2= 34.92 .23
2004 H2,H2PH2= 35.25 -.34
2005 H2,H2PH2= 35.41 -.15
2006 H2,H2PH2= 35.49 -.08
2007 H2,H2PH2= 35.48 .01
2008
2009 CONDENSER ITERATION:
2010
2011 T P H X
2012 1.503+002 2.177+002 1.041+002 1.000+000
2013 8.554+001 2.096+002 3.548+001
2014
2015 LIQUID LINE:
2016 INPUT - PIN = 2.096+002 TIN = 8.554+001 XIN = .000
2017 OUTPUT - POUT = 2.042+002 TOUT = 8.552+001 XOUT = .000
2018
2019
2020
2021 EXPANSION DEVICE:
2022 INPUT - P12 = 2.042+002 H12 = 3.548+001 P13 = 9.402+001
2023 OUTPUT - POUT = 1.563+002 XMASS = 4.714+002
2024
2025 INPUT DATA TO EVAPHX:
2026
2027 T P X TAIR RH RMASS
2028 3.581+001 9.402+001 2.277+001 4.700+001 7.300+001 4.720+002
2029 H2= 90.9949

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H2= 89.0392
H2= 86.8653
H2= 86.6291
QT,QS= 2.413+004 2.127+004

EVAPORATOR ITERATION:

T      P      H      X      TS'IP
3.581+001 9.402+001 3.543+001 2.277-001
3.723+001 7.350+001 8.659+001 9.650-001 .000

INPUT DATA TO EVAPHX:

T      P      X      TAIR      RH      RMASS
3.570+001 9.385+001 2.281-001 4.700+001 7.300-001 4.720+002
H2= 91.0722
H2= 89.5154
H2= 87.5181
H2= 87.2596
H2= 87.1924
QT,QS= 2.441+004 2.131+004

EVAPORATOR ITERATION:

T      P      H      X      TSUP
3.570+001 9.385+001 3.548+001 2.281-001
3.706+001 7.306+001 8.720+001 9.721-001 .000
P1,PIE,P1EP1,DENT2= 73.0641 73.0637 -.0004 .0495
INTERM,INTER,FOUT,P13= 0 0 156.78 93.85
COMPOSITION OF REFRIG IN ACCUMULATOR = .570
(WEIGHT FRACTION OF MORE VOLATILE COMPONENT)

REFRIG. IN ACCUMULATOR = .240 LB
REFRIG. IN INDOOR COIL = 2.336 LB
REFRIG. IN OUTDOOR COIL = .871 LB

TMASS = 4.563+000 REFIN = 4.963+000

*****
**SYST4 WITH ACCUM., 13B1/152A F-22-34***

TGA      RHGA      TRA      RHRA
4.700+001 7.200-001 7.000+001 5.600-001

CFMIND      CFMOUT
1.118+003 2.294+003

RESULTS:

I      T      P      H      S      X
1 3.705+001 7.306+001 8.715+001 1.905-001 9.716-001
2 3.643+001 7.222+001 8.709+001 1.905-001 9.716-001
3 3.181+001 6.570+001 8.937+001 1.947-001 9.912-001
4 6.776+001 6.561+001 9.505+001 2.079-001 1.000+000
5 7.150+001 6.288+001 9.578+001 2.101-001 1.000+000

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2088	6	1.832+002	2.418+002	1.100+002	2.113-001	1.000+000
2089	7	1.765+002	2.212+002	1.093+002	2.116-001	1.000+000
2090	8	1.579+002	2.202+002	1.056+002	2.058-001	1.000+000
2091	9	1.511+002	2.178+002	1.043+002	2.039-001	1.000+000
2092	10	1.503+002	2.177+002	1.041+002	2.037-001	1.000+000
2093	11	8.551+001	2.096+002	3.548+001	8.106+002	.000
2094	12	8.552+001	2.042+002	3.548+001	8.111+002	.000
2095	13	8.570+001	9.585+001	3.548+001	8.307-002	2.281-001
2096						
2097	TG3	TSUP3	TG6	RMAS5	TMAS5	
2098	3.199+001	.000	1.123+002	4.720+002	4.563+000	
2099	QLGAD	ELUSE	COF			
2100	3.407+004	3.567+000	2.799+000			
2101						
2102						
2103						
2104						
2105						
2106						
2107						
2108						
2109						
2110						
2111	P3	T3	H3	XQ3	TG6	TRA
2112	6.622+001	3.221+001	8.836+001	9.906-001	1.128+002	7.000+001
2113						
2114						
2115						
2116	EI	ETA5	ETAC	ETAV	CPRPM	RMAS5
2117	2.721+000	7.577-001	9.472-001	8.565-001	3.549+003	4.762+002
2118						
2119	I	T	P	H	X	
2120	1	3.747+001	7.366+001	8.714+001	9.709-001	
2121	2	3.685+001	7.280+001	8.709+001	9.710-001	
2122	3	3.221+001	6.622+001	8.855+001	9.906-001	
2123	4	6.776+001	6.613+001	9.502+001	1.000+000	
2124	5	7.149+001	5.338+001	9.575+001	1.000+000	
2125	6	1.834+002	2.437+002	1.099+002	1.000+000	
2126	7	1.766+002	2.230+002	1.092+002	1.000+000	
2127	8	1.580+002	2.219+002	1.056+002	1.000+000	
2128	9	1.512+002	2.195+002	1.043+002	1.000+000	
2129	10	1.504+002	2.194+002	1.041+002	1.000+000	
2130						
2131						
2132						
2133	T	P	TAIR	RIH	RMAS5	
2134	1.504+002	2.194+002	7.000+001	5.600-001	4.762+002	
2135	H2, F2PH2=	34.17	.00			
2136	H2, H2PH2=	34.90	-.72			
2137	H2, H2PH2=	34.95	-.05			
2138	H2, H2PH2=	35.07	-.12			
2139	H2, H2PH2=	35.20	-.12			
2140	H2, H2PH2=	35.26	-.07			
2141	H2, H2PH2=	35.27	-.01			
2142						
2143						
2144						
2145						

REFRIG. COMPOSITION = .650
(WEIGHT FRACTION OF MORE VOLATILE COMPONENT)

INPUT DATA TO COMPRESSOR ITERATION:

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2203

1.504+002 2.194+002 1.041+002 1.000+000
8.475+001 2.116+002 3.527+001 .000

LIQUID LINE:
INPUT - PIN = 2.116+002 TIN = 8.475+001 XIN = .000
OUTPUT - POUT = 2.061+002 TOUT = 8.472+001 XOUT = .000
HIN = 3.527+001

EXPANSION DEVICE:
INPUT - P12 = 2.061+002 H12 = 3.527+001 P13 = 9.444+001
OUTPUT - POUT = 1.597+002 XMASS = 4.835+002

INPUT DATA TO COMPRESSOR:

P3 T3 T6 XQ3 T96 TRA TGA
6.622+001 3.221+001 8.835+001 9.905+001 1.127+002 7.000+001 4.700+001

COMPRESSOR ITERATION:

EI ETAE ETAC ETAV CPRPM RMASS
2.719+000 7.576+001 9.473+001 8.567+001 3.549+003 4.763+002

I	T	P	H	X
1	3.748+001	7.366+001	8.714+001	9.709+001
2	3.685+001	7.281+001	9.708+001	9.710+001
3	3.221+001	6.622+001	8.836+001	9.906+001
4	6.773+001	6.610+001	9.502+001	1.000+000
5	7.145+001	6.038+001	9.575+001	1.000+000
6	1.832+002	2.434+002	1.099+002	1.000+000
7	1.764+002	2.226+002	1.092+002	1.000+000
8	1.579+002	2.215+002	1.055+002	1.000+000
9	1.511+002	2.191+002	1.043+002	1.000+000
10	1.503+002	2.190+002	1.041+002	1.000+000

INPUT DATA TO CONDHX:

T P TAIR RH RMASS
1.503+002 2.190+002 7.000+001 5.600+001 4.763+002
H2,H2PH2= 34.38 .01
H2,H2PH2= 34.56 -.18
H2,H2PH2= 35.32 -.76
H2,H2PH2= 35.35 -.03
H2,H2PH2= 35.42 -.08
H2,H2PH2= 35.45 -.02
H2,H2PH2= 34.62 .63
H2,H2PH2= 34.98 -.16
H2,H2PH2= 35.08 -.11

CONDHX DOES NOT CONVERGE, MAX. ERROR= -1.059+001 (BTU/LP)

CONDENSER ITERATION:

T P H X
1.503+002 2.190+002 1.041+002 1.000+000
8.401+001 2.110+002 3.500+001 .000

LIQUID LINE:

HIN = 3.508+001

XIN = .000
XOUT = .000

TIN = 8.401+001
TOUT = 8.399+001

INPUT - PIN = 2.110+002
OUTPUT - POUT = 2.055+002

EXPANSION DEVICE:

INPUT - P12 = 2.055+002 H12 = 3.508+001 P13 = 9.444+001
OUTPUT - POUT = 1.600+002 XMASS = 4.861+002

INPUT DATA TO EVAPHX:

T	P	X	TAIR	RH	RMASS
3.615+001	9.444+001	2.322-001	4.700+00	7.300-001	4.753+002
H2 = 90.6330					
H2 = 88.4190					
H2 = 86.1198					
H2 = 85.9054					
QT, QS = 2.333+004	2.110+004				

EVAPORATOR ITERATION:

T	P	H	X	TSUP
3.615+001	9.444+001	3.581+001	2.322-001	
3.733+001	7.386+001	8.589+001	9.559-00	.000

INPUT DATA TO EVAPHX:

T	P	X	TAIR	RH	RMASS
3.610+001	9.436+001	2.324-001	4.700+001	7.300-001	4.753+002
H2 = 90.9397					
H2 = 88.6861					
H2 = 86.4198					
H2 = 86.2107					
QT, QS = 2.397+004	2.118+004				

EVAPORATOR ITERATION:

T	P	H	X	TSUP
3.610+001	9.436+001	3.581+001	2.324-001	
3.725+001	7.366+001	8.618+001	9.603-001	.000
P1, P1E, F1EPI, DENT2 =	73.6589	73.6508		.0019
				- .9602

INPUT DATA TO COMP:

P3	T3	H3	XQ3	T96	TRA	T9A
6.581+001	3.168+001	8.832+001	9.906-001	1.129+002	7.000+001	4.700+001

COMPRESSOR ITERATION:

E1	ETA E	ETAC	ETAV	CPRPM	RMASS
2.717+000	7.575-001	9.471-001	6.556-001	3.540+003	4.725+002

I	T	P	H	X
1	3.712+001	7.317+001	8.710+001	9.709-001
2	3.649+001	7.282+001	8.704+001	9.710-001
3	3.188+001	6.581+001	8.832+001	9.706-001
4	6.774+001	6.571+001	9.504+001	1.000+000
5	7.151+001	6.299+001	9.577+001	1.000+000
6	1.839+002	2.408+002	1.100+002	1.000+000

7 1.772+002 2.234+002 1.093+002 1.000+000
 8 1.535+002 2.224+002 1.056+002 1.000+000
 9 1.517+002 2.200+002 1.043+002 1.000+000
 10 1.503+002 2.199+002 1.042+002 1.000+000
 INPUT DATA TO CONDHX:
 T P TAIR RH RMASS
 1.508+002 2.199+002 7.000+001 5.600-001 4.725+002
 H2, H2PH2= 34.04 .00
 H2, H2PH2= 34.22 -.18
 H2, H2PH2= 34.86 -.64
 H2, H2PH2= 34.93 -.08
 H2, H2PH2= 34.45 .48
 H2, H2PH2= 34.63 -.18
 H2, H2PH2= 34.74 -.11
 H2, H2PH2= 34.78 -.04
 H2, H2PH2= 34.77 .01
 CONDENSER ITERATION:
 T P H X
 1.508+002 2.139+002 1.042+002 1.000+000
 8.280+001 2.123+002 3.477+001 .000
 LIQUID LINE:
 INPUT - PIN = 2.123+002 TIN = 8.280+001 XIN = .000
 OUTPUT - POUT = 2.039+002 TOUT = 8.273+001 XOUT = .000
 HIN = 3.477+001
 EXPANSION DEVICE:
 INPUT - P12 = 2.069+002 H12 = 3.477+001 P13 = 9.387+001
 OUTPUT - POUT = 1.626+002 XMASS = 4.999+002
 INPUT DATA TO COMPRESSOR:
 P3 T3 H3 X03 T66 TRA TOA
 6.581+001 3.183+001 8.832+001 9.906-001 1.140+002 7.000+001 4.700+001
 COMPRESSOR ITERATION:
 EI ETAE ETAC ETAV ETAV CPRPM RMASS
 2.742+000 7.536-001 9.464-001 8.539-001 3.548+003 4.711+002
 I T P H X
 1 3.708+001 7.313+001 8.708+001 9.707-001
 2 3.645+001 7.229+001 8.702+001 9.708-001
 3 3.183+001 6.581+001 8.832+001 9.905-001
 4 6.868+001 6.571+001 9.309+001 1.000+000
 5 7.191+001 6.300+001 9.584+001 1.000+000
 6 1.857+002 2.475+002 1.103+002 1.000+000
 7 1.791+002 2.275+002 1.036+002 1.000+000
 8 1.602+002 2.265+002 1.058+002 1.000+000
 9 1.533+002 2.242+002 1.045+002 1.000+000
 10 1.524+002 2.241+002 1.043+002 1.000+000
 INPUT DATA TO CONDHX:
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T      P      TAIR      RH      RMASS
1.524+002  2.241+002  7.000+001  5.600-001  4.711+002
H2,H2PH2= 33.54      -.28
H2,H2PH2= 33.62      -.08
H2,H2PH2= 33.87      -.26
H2,H2PH2= 33.73      .14
H2,H2PH2= 33.89      -.15
H2,H2PH2= 33.95      -.06
H2,H2PH2= 33.94      .01

CONDENSER ITERATION:
T      P      H      X
1.524+002  2.241+002  1.043+002  1.000+000
7.962+001  2.176+002  3.394+001  .000

LIQUID LINE:
INPUT - PIN = 2.176+002  TIN = 7.962+001  XIN = .000
OUTPUT - POUT= 2.122+002  TOUT= 7.959+001  XOUT= .000
HIN = 3.394+001

EXPANSION DEVICE:
INPUT - P12 =2.122+002  H12 =3.394+001  P13 =9.383+001
OUTPUT - POUT =1.820+002  XMASS =5.922+002

INPUT DATA TO COMPR:
P3      T3      H3      XQ3      TQ6      TRA      TOA
6.581+001  3.188+001  0.832+001  9.906-001  1.125+002  7.000+001  4.700+001

COMPRESSOR ITERATION:
EI      ETAE      ETAC      ETAV      CPRPM      RMASS
2.710+000  7.572-001  9.473-001  8.561-001  3.549+003  4.729+002

I      T      P      H      X
1  3.713+001  7.318+001  8.710+001  9.709-001
2  3.650+001  7.233+001  8.704+001  9.710-001
3  3.188+001  6.581+001  8.832+001  9.906-001
4  6.755+001  6.571+001  9.502+001  1.000+000
5  7.139+001  6.296+001  9.575+001  1.000-000
6  1.834+002  2.227+002  1.100+002  1.000+000
7  1.766+002  2.222+002  1.093+002  1.000+000
8  1.580+002  2.211+002  1.056+002  1.000+000
9  1.512+002  2.187+002  1.043+002  1.000+000
10 1.504+002  2.187+002  1.041+002  1.000+000

INPUT DATA TO CONDHX:
T      P      TAIR      RH      RMASS
1.504+002  2.187+002  7.000+001  5.600-001  4.729+002
H2,H2PH2= 34.31      .03
H2,H2PH2= 34.47      -.16
H2,H2PH2= 35.22      -.75
H2,H2PH2= 35.26      -.04
H2,H2PH2= 35.33      -.07

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2378 H2,H2PH2= 35.35 -.02
2379 H2,H2PH2= 35.37 -.02
2380
2381 CONDENSER ITERATION:
2382
2383 T P H X
2384 1.504+002 2.187+002 1.041+002 1.000+000
2385 8.513+001 2.108+002 3.537+001 .000
2386
2387 LIQUID LINE:
2388 INPUT - PIN = 2.108+002 TIN = 3.513+001 XIN = .000
2389 OUTPUT - POUT = 2.054+002 TOUT = 3.510+001 XOUT = .000
2390
2391 HIN = 3.537+001
2392
2393 EXPANSION DEVICE:
2394 INPUT - P12 =2.054+002 H12 =3.537+001 P13 =9.388+001
2395 DDENFA DOES NOT CONVERGE, DIFXQ2= -9.61527-006
2396 OUTPUT - POUT =1.588+002 XMASS =4.789+002
2397
2398 INPUT DATA TO EVAPHX:
2399
2400 T P X X TAIR RH RMASS
2401 3.572+001 9.388+001 2.274-001 4.700+001 7.000-001 4.730+002
2402 H2= 91.0375
2403 H2= 89.4282
2404 H2= 87.3639
2405 H2= 87.1429
2406 H2= 87.0711
2407 QT, QS= 2.442+004 2.132+004
2408
2409 EVAPORATOR ITERATION:
2410
2411 T P H X TSUP
2412 3.572+001 9.388+001 3.544+001 2.274-001
2413 3.707+001 7.311+001 3.708+001 9.707-001 .000
2414 P1,P1E,P1EP1,DENT2= 73.1819 73.1082 -.0737 -.0226
2415 INTERM,ITERE,POUT,P13= 1 0 158.83 93.88
2416
2417 INPUT DATA TO COMPR:
2418
2419 P3 T3 H3 XQ3 T96 TRA TGA
2420 6.581+001 3.188+001 8.832+001 9.005-001 1.125+002 7.000+001 4.700+001
2421
2422 COMPRESSOR ITERATION:
2423
2424 EI ETAE ETAC ETAV ETAV CPRPM RMASS
2425 2.708+000 7.572-001 9.473-001 8.563-001 3.543+003 4.730+002
2426
2427 I T P H X
2428 1 3.713+001 7.318+001 8.710+001 9.709-001
2429 2 3.650+001 7.234+001 8.704+001 9.710-001
2430 3 3.183+001 6.581+001 8.832+001 9.305-001
2431 4 6.763+001 6.571+001 9.502+001 1.000+000
2432 5 7.137+001 6.298+001 9.575+001 1.000+000
2433 6 1.833+002 2.425+002 1.100+002 1.000+000
2434 7 1.765+002 2.219+002 1.093+002 1.000+000
2435 8 1.579+002 2.209+002 1.056+002 1.000+000

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2436 9 1.511+002 2.184+002 1.043+002 1.000+000
2437 10 1.503+002 2.184+002 1.041+002 1.000+000
2438
2439 INPUT DATA TO CONDHX:
2440
2441 T P TAIR RH RMASS
2442 1.503+002 2.184+002 7.000+001 5.600-001 4.730+002
2443 H2, H2PH2= 34.60 -.54
2444 H2, H2PH2= 34.71 -.10
2445 H2, H2PH2= 34.75 -.04
2446 H2, H2PH2= 34.86 -.11
2447 H2, H2PH2= 35.08 -.21
2448 H2, H2PH2= 35.17 -.09
2449 H2, H2PH2= 35.18 -.01
2450
2451 CONDENSER ITERATION:
2452
2453 T P H X
2454 1.503+002 2.184+002 1.041+002 1.000+000
2455 8.439+001 2.104+002 3.518+001 .000
2456
2457 LIQUID LINE:
2458
2459 INPUT - PIN = 2.104+002 TIN = 8.439+001 XIN = .000
2460 OUTPUT - POUT = 2.049+002 TOUT = 8.437+001 XOUT = .000
2461
2462
2463
2464 EXPANSION DEVICE:
2465 INPUT - P12 = 2.049+002 H12 = 3.518+001 P13 = 9.389+001
2466 OUTPUT - POUT = 1.509+002 XMASS = 4.814+002
2467
2468 INPUT DATA TO COMPR:
2469
2470 P3 T3 H3 X03 TG6 TRA TOA
2471 6.581+001 3.188+001 8.932+001 9.906-001 1.124+002 7.000+001 4.700+001
2472
2473 COMPRESSOR ITERATION:
2474
2475 EI ETAE ETAC ETAV ETAV CPRPM RMASS
2476 2.706+000 7.571-001 9.474-001 8.564-001 3.549+003 4.731+002
2477
2478 1 T P H X
2479 1 3.713+001 7.319+001 8.710+001 9.709-001
2480 2 3.650+001 7.234+001 8.705+001 9.710-001
2481 3 3.189+001 6.531+001 8.632+001 9.306-001
2482 4 6.759+001 5.571+001 9.501+001 1.000-000
2483 5 7.133+001 6.293+001 9.574+001 1.000+000
2484 6 1.831+002 2.421+002 1.099+002 1.000+000
2485 7 1.763+002 2.215+002 1.092+002 1.000+000
2486 8 1.577+002 2.203+002 1.055+002 1.000+000
2487 9 1.509+002 2.187+002 1.043+002 1.000+000
2488 10 1.501+002 2.180+002 1.041+002 1.000+000
2489
2490 INPUT DATA TO CONDHX:
2491
2492 T P TAIR RH RMASS
2493 1.501+002 2.180+002 7.000+001 5.600-001 4.731+002
H2, H2PH2= 34.82 -.50

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2494 H2,H2PH2= 34.92 -.10
2495 H2,H2PH2= 34.77 .16
2496 H2,H2PH2= 35.16 -.39
2497 H2,H2PH2= 35.36 -.21
2498 H2,H2PH2= 35.44 -.07
2499 H2,H2PH2= 35.43 .01
2500
2501 CONDENSER ITERATION:
2502
2503 T P H X
2504 1.501+002 2.180+002 1.041+002 1.000+000
2505 8.534+001 2.093+002 3.543+001 .000
2506
2507 LIQUID LINE:
2508 INPUT - PIN = 2.099+002 TIN = 8.534+001 XIN = .000
2509 OUTPUT - POUT = 2.044+002 TOUT = 8.532+001 XOUT = .000
2510
2511 HIN = 3.543+001
2512
2513 EXPANSION DEVICE:
2514 INPUT - P12 = 2.044+002 H12 = 3.543+001 P13 = 9.289+001
2515 OUTPUT - POUT = 1.576+002 XMASS = 4.737+002
2516
2517 INPUT DATA TO EVAPHX:
2518
2519 T P X TAIR RH FMASS
2520 3.572+001 9.389+001 2.271-001 4.700+001 7.300-001 4.731+002
2521 H2= 91.0529
2522 P2= 89.3513
2523 H2= 87.3205
2524 H2= 87.1003
2525 H2= 87.0362
2526 QT,QS= 2.442+004 2.132+004
2527
2528 EVAPORATOR ITERATION:
2529
2530 T P H X TSUP
2531 3.572+001 9.389+001 3.543+001 2.271-001
2532 3.707+001 7.313+001 8.704+001 9.703-001 .000
2533 P1,P1E,P1EP1,DENT2= 73.1884 73.1278 -.0507 -.0520
2534 INTERN, INTERE, POUT, P13= 0 0 157.59 93.09
2535 COMPOSITION OF REFRIG. IN ACCUMULATOR = .404
2536 (WEIGHT FRACTION OF MORE VOLATILE COMPONENT)
2537
2538 REFRIG. IN ACCUMULATOR = .649 LB
2539
2540 REFRIG. IN INDOOR COIL = 2.314 LB
2541
2542 REFRIG. IN OUTDOOR COIL = .875 LB
2543
2544 TMASS = 4.924+000 REFIN = 4.968+000
2545
2546 *****
2547 **SYST4 WITH ACCUM., 1381/152A 6-22-84***
2548
2549 TQA RHQA TRA RHRA
2550 4.700+001 7.300-001 7.000+001 5.600-001
2551

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2552 CFMIND CFMOUT
2553 1.118+003 2.284+003
2554
2555
2556 RESULTS:
2557
2558 I T P H S X
2559 1 3.713+001 7.313+001 8.710+001 1.903-001 9.709-001
2560 2 3.650+001 7.234+001 8.705+001 1.904-001 9.710-001
2561 3 3.188+001 6.581+001 8.832+001 1.946-001 9.905-001
2562 4 6.759+001 6.571+001 9.501+001 2.073-001 1.009+000
2563 5 7.133+001 6.298+001 9.574+001 2.100-001 1.000+000
2564 6 1.831+002 2.421+002 1.099+002 2.112-001 1.000+000
2565 7 1.763+002 2.215+002 1.092+002 2.115-001 1.000+000
2566 8 1.577+002 2.205+002 1.055+002 2.057-001 1.000+000
2567 9 1.509+002 2.183+002 1.043+002 2.033-001 1.003+000
2568 10 1.501+002 2.180+002 1.041+002 2.036-001 1.000+000
2569 11 8.534+001 2.099+002 3.543+001 8.093-002 .000
2570 12 8.532+001 2.044+002 3.543+001 8.101-002 .000
2571 13 3.572+001 9.339+001 3.543+001 8.296-002 2.271-001
2572
2573 TG3 TSUP3 TG6 RMASS THASS
2574 3.208+001 .000 1.124+002 4.731+002 4.934+000
2575
2576 QLOAD ELUSE COP
2577 3.415+004 3.571+000 2.802+000
2578
2579 REFRIG. COMPOSITION = .650
2580 (WEIGHT FRACTION OF MORE VOLATILE COMPONENT)
2581
2582
2583
2584
2585
2586 FOR THE AMOUNT OF REFRIGERANT STORED IN THE ACCUMULATOR
2587 CIRCULATING COMPOSITION SHOULD BE .687
2588
2589 NEW CALCULATED COMPOSITION FOR THE NEXT LOOP IS .668
2590 (WEIGHT FRACTION OF MORE VOLATILE COMPONENT)
2591
2592 *****
2593 INPUT DATA TO COMPR:
2594
2595 P3 T3 H3 X03 TG6 TRA TGA
2596 6.773+001 3.187+001 8.642+001 9.906-001 1.124+002 7.000+001 4.700+001
2597
2598 COMPRESSOR ITERATION:
2599
2600 EI ETAE ETAC ETAV X03 CPRPM RMASS
2601 2.763+000 7.595-001 9.470-001 8.577-001 3.548+003 4.934+002
2602
2603 I T P H X
2604 1 3.722+001 7.547+001 8.524+001 9.708-001
2605 2 3.658+001 7.459+001 8.518+001 9.709-001
2606 3 3.187+001 6.773+001 8.642+001 9.906-001
2607 4 6.722+001 6.764+001 9.295+001 1.000+000
2608 5 7.087+001 6.477+001 9.366+001 1.000+000
2609

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6 1.825+002 2.478+002 1.075+002 1.000+000
 7 1.757+002 2.253+002 1.068+002 1.000+000
 8 1.573+002 2.249+002 1.033+002 1.000+000
 9 1.506+002 2.223+002 1.020+002 1.000+000
 10 1.408+002 2.222+002 1.019+002 1.000+000

INPUT DATA TO CONDHX:

T	P	TAIR	RH	RMASS
1.498+002	2.222+002	7.000+001	5.600+001	4.965+002
H2,H2PH2=	34.64	- .09		
H2,H2PH2=	35.39	- .75		
H2,H2PH2=	35.80	- .41		
H2,H2PH2=	35.95	- .15		
H2,H2PH2=	36.02	- .07		
H2,H2PH2=	36.07	- .05		
H2,H2PH2=	36.10	- .03		
H2,H2PH2=	36.11	- .01		

CONDENSER ITERATION:

T	P	H	X
1.498+002	2.222+002	1.019+002	1.000+000
8.773+001	2.130+002	3.611+001	1.170+002

LIQUID LINE:

INPUT - PIN = 2.130+002 TIN = 8.773+001 XIN = 1.170+002
 OUTPUT - POUT = 2.058+002 TOUT = 8.524+001 XOUT = 2.552+002

EXPANSION DEVICE:

INPUT - P12 = 2.058+002 H12 = 3.611+001 P13 = 9.617+001
 OUTPUT - POUT = 1.549+002 XMASS = 4.529+002

INPUT DATA TO COMPR:

P3	T3	H3	XG3	TG6	TRA	TGA
6.773+001	3.187+001	8.542+001	9.905+001	1.129+002	7.000+001	4.700+001

COMPRESSOR ITERATION:

EI	ETA	ETAC	ETAV	CFRPM	RMASS
2.775+000	7.600+001	9.467+001	8.568+001	3.548+003	4.558+002

I	T	P	H	X
1	3.720+001	7.545+001	8.523+001	9.707+001
2	3.657+001	7.457+001	8.518+001	9.709+001
3	3.187+001	6.773+001	8.642+001	9.905+001
4	6.733+001	6.764+001	9.298+001	1.000+000
5	7.107+001	6.277+001	9.370+001	1.000+000
6	1.831+002	2.496+002	1.076+002	1.000+000
7	1.766+002	2.280+002	1.070+002	1.000+000
8	1.582+002	2.269+002	1.033+002	1.000+000
9	1.514+002	2.244+002	1.021+002	1.000+000
10	1.506+002	2.243+002	1.019+002	1.000+000

INPUT DATA TO CONDHX:

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2725

T	P	TAIR	RH	RMASS
1.506+002	2.243+002	7.000+001	5.600-001	4.959+002
H2, H2PH2=	34.14	- .10		
H2, H2PH2=	34.41	- .27		
H2, H2PH2=	34.94	- .53		
H2, H2PH2=	34.59	.35		
H2, H2PH2=	34.81	- .22		
H2, H2PH2=	34.92	- .11		
H2, H2PH2=	34.99	- .07		
H2, H2PH2=	34.93	.01		

CONDENSER ITERATION:

T	P	H	X
1.506+002	2.243+002	1.019+002	1.000+000
8.545+001	2.153+002	3.498+001	.000

LIQUID LINE:

INPUT -	PIN = 2.158+002	TIN = 8.545+001	XIN = .000
OUTPUT -	POUT = 2.039+002	TOUT = 8.543+001	XOUT = .000

HIN = 3.498+001

EXPANSION DEVICE:

INPUT -	P12 = 2.099+002	H12 = 3.498+001	P13 = 9.615+001
OUTPUT -	POUT = 1.627+002	XMASS = 4.909+002	

INPUT DATA TO COMP:

P3	T3	H3	XQ3	TG6	TRA	TGA
6.773+001	3.187+001	8.642+001	9.906-001	1.130+002	7.000+001	4.700+001

COMPRESSOR ITERATION:

E1	ETA	ETAC	ETAV	CPRPM	RMASS
2.777+000	7.601-001	9.466-001	8.567-001	3.543+003	4.957+002

I	T	P	H	X
1	3.720+001	7.545+001	8.523+001	9.707-001
2	3.657+001	7.457+001	8.517+001	9.708-001
3	3.187+001	6.773+001	8.542+001	9.906-001
4	6.740+001	6.764+001	9.208+001	1.000+000
5	7.109+001	6.477+001	9.370+001	1.000+000
6	1.835+002	2.498+002	1.076+002	1.000+000
7	1.767+002	2.282+002	1.070+002	1.000+000
8	1.582+002	2.272+002	1.034+002	1.000+000
9	1.515+002	2.245+002	1.021+002	1.000+000
10	1.507+002	2.245+002	1.020+002	1.000+000

INPUT DATA TO CONDHX:

T	P	TAIR	RH	RMASS
1.507+002	2.246+002	7.000+001	5.600-001	4.957+002
H2, H2PH2=	34.01	- .11		
H2, H2PH2=	34.27	- .26		
H2, H2PH2=	34.82	- .55		
H2, H2PH2=	35.06	- .24		

H2,H2PH2= 35.11 -.05
 H2,H2PH2= 35.12 -.01

CONDENSER ITERATION:

T	P	H	X
1.507+002	2.246+002	1.020+002	1.000+000
8.602+001	2.162+002	3.512+001	.000

LIQUID LINE:
 INPUT - PIN = 2.162+002 TIN = 8.602+001 XIN = .000
 OUTPUT - POUT = 2.104+002 TOUT = 8.599+001 XOUT = .000

HIN = 3.512+001

EXPANSION DEVICE:
 INPUT - P12 = 2.104+002 H12 = 3.512+001 P13 = 9.615+001
 OUTPUT - POUT = 1.524+002 XMASS = 4.883+002

INPUT DATA TO EVAPHX:

T	P	X	TAIR	RH	RMASS
3.576+001	9.615+001	2.212+001	4.700+001	7.300+001	4.050+002
H2= 89.0996					
H2= 87.5662					
H2= 85.5736					
QT,OS= 2.522+004	2.238+004				

EVAPORATOR ITERATION:

T	P	H	X	TSUP
3.576+001	9.615+001	3.467+001	2.212+001	
3.685+001	7.485+001	8.552+001	9.745+001	.000

INPUT DATA TO EVAPHX:

T	P	X	TAIR	RH	RMASS
3.591+001	9.639+001	2.203+001	4.700+001	7.300+001	4.050+002
H2= 88.9841					
H2= 86.9319					
H2= 84.7277					
QT,OS= 2.482+004	2.210+004				

EVAPORATOR ITERATION:

T	P	H	X	TSUP
3.591+001	9.639+001	3.467+001	2.206+001	
3.707+001	7.546+001	8.471+001	9.648+001	.000
P1,P1E,P1EP1,DENT2=	75.4489	75.4489	.0068	
INTERM,INTERE,POUT,P13=	1	0	162.35	96.39

INPUT DATA TO COMPR:

P3	T3	H3	X03	TG6	TRA	T0A
6.773+001	3.187+001	8.642+001	9.906+001	1.128+002	7.000+001	4.700+001

COMPRESSOR ITERATION:

2784	ETAE	7.599-001	ETAC	8.458-001	ETAV	CPRPM	RMASS
2785	2.772+000				8.571-001	3.548+003	4.960+002
2786							
2787	I	T	P	H	X		
2788	1	3.721+001	7.546+001	8.523+001	9.708-001		
2789	2	3.657+001	7.457+001	8.519+001	9.709-001		
2790	3	3.187+001	6.773+001	8.542+001	9.906-001		
2791	4	6.734+001	6.764+001	9.297+001	1.000+000		
2792	5	7.101+001	6.477+001	9.369+001	1.000+000		
2793	6	1.832+002	2.491+002	1.076+002	1.000+000		
2794	7	1.763+002	2.274+002	1.069+002	1.000+000		
2795	8	1.579+002	2.263+002	1.033+002	1.000+000		
2796	9	1.512+002	2.238+002	1.021+002	1.000+000		
2797	10	1.504+002	2.237+002	1.019+002	1.000+000		
2798							
2799	INPUT DATA TO CONDHX:						
2800							
2801	T	P	TAIR	RH	RMASS		
2802	1.504+002	2.237+002	7.000+001	5.600-001	4.960+002		
2803	H2, H2PH2=	34.09	- .10				
2804	H2, H2PH2=	34.97	- .88				
2805	H2, H2PH2=	34.77	.20				
2806	H2, H2PH2=	35.10	- .33				
2807	H2, H2PH2=	35.25	- .15				
2808	H2, H2PH2=	35.33	- .08				
2809	H2, H2PH2=	35.32	.00				
2810							
2811	CONDENSER ITERATION:						
2812							
2813	T	P	H	X			
2814	1.504+002	2.237+002	1.019+002	1.000+000			
2815	8.677+001	2.150+002	3.532+001	.000			
2816							
2817							
2818	LIQUID LINE:						
2819	INPUT -	PIN = 2.150+002	TIN = 8.677+001	XIN = .000			
2820	OUTPUT -	POUT = 2.091+002	TOUT = 8.626+001	XOUT = 2.741-003			
2821							
2822							
2823	EXPANSION DEVICE:						
2824	INPUT -	P12 = 2.091+002	H12 = 3.532+001	P13 = 3.640+001			
2825	OUTPUT -	POUT = 1.503+002	XMASS = 4.795+002				
2826							
2827	INPUT DATA TO COMP:						
2828							
2829	P3	T3	H3	XG3	TG6	TRA	TGA
2830	6.773+001	3.187+001	8.542+001	9.906-001	1.123+002	7.000+001	4.700+001
2831	COMPRESSOR ITERATION:						
2832							
2833	EI	ETAE	ETAC	ETAV	CPRPM	RMASS	
2834	2.760+000	7.594-001	9.471-001	6.578-001	3.548+003	4.960+002	
2835							
2836	I	T	P	H	X		
2837	1	3.722+001	7.548+001	8.524+001	9.708-001		
2838	2	3.659+001	7.459+001	8.519+001	9.710-001		
2839	3	3.187+001	6.773+001	8.642+001	9.906-001		
2840	4	6.720+001	6.764+001	9.295+001	1.000+000		
2841							

2842	5	7.084+001	6.476+001	9.366+001	1.000+000
2843	6	1.823+002	2.474+002	1.075+002	1.000+000
2844	7	1.755+002	2.255+002	1.068+002	1.000+000
2845	8	1.572+002	2.244+002	1.032+002	1.000+000
2846	9	1.524+002	2.218+002	1.020+002	1.000+000
2847	10	1.497+002	2.218+002	1.019+002	1.000+000
2848	INPUT DATA TO CONDHX:				
2849					
2850					
2851	T	P	TAIR	RH	RMASS
2852	H2,H2PH2=	34.99	7.000+001	5.600-001	4.966+002
2853	H2,H2PH2=	35.62	- .08		
2854	H2,H2PH2=	35.95	- .33		
2855	H2,H2PH2=	36.13	- .18		
2856	H2,H2PH2=	36.23	- .09		
2857	H2,H2PH2=	36.29	- .06		
2858	H2,H2PH2=	36.32	- .03		
2859	H2,H2PH2=	36.33	- .02		
2860	CONDENSER ITERATION:				
2861					
2862					
2863					
2864	T	P	H	X	
2865	1.497+002	2.218+002	1.019+002	1.000+000	
2866	8.756+001	2.124+002	3.633+001	1.752-002	
2867	LIQUID LINE:				
2868					
2869	INPUT -	PIN = 2.124+002	TIN = 8.756+001	XIN = 1.752-002	HIN = 3.633+001
2870	OUTPUT -	POUT = 2.047+002	TOUT = 8.491+001	XOUT = 3.220-002	
2871	EXPANSION DEVICE:				
2872	INPUT -	P12 = 2.047+002	H12 = 3.633+001	F13 = 9.541+001	
2873	OUTPUT -	POUT = 1.536+002	XMASS = 4.455+002		
2874	INPUT DATA TO COMPR:				
2875					
2876	P3	T3	H3	XQ3	TG6
2877	6.773+001	3.187+001	8.642+001	9.506-001	1.130+002
2878	COMPRESSOR ITERATION:				
2879					
2880	EI	ETAE	ETAC	ETAV	CP3PM
2881	2.777+000	7.501-001	9.466-001	8.567-001	3.548+003
2882					RMASS
2883					4.957+002
2884	I	T	P	H	X
2885	1	3.720+001	7.545+001	8.523+001	9.707-001
2886	2	3.656+001	7.457+001	8.517+001	9.703-001
2887	3	3.187+001	6.773+001	8.642+001	9.906-001
2888	4	6.741+001	6.764+001	9.299+001	1.000+000
2889	5	7.109+001	6.477+001	9.370+001	1.000+000
2890	6	1.835+002	2.499+002	1.077+002	1.000+000
2891	7	1.767+002	2.283+002	1.070+002	1.000+000
2892	8	1.583+002	2.273+002	1.034+002	1.000+000
2893	9	1.515+002	2.247+002	1.021+002	1.000+000
2894	10	1.507+002	2.247+002	1.020+002	1.000+000
2895					
2896					
2897					
2898					
2899					

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2900 INPUT DATA TO CONDHX:
2901
2902 T P TAIR RH RMASS
2903 1.507+002 2.247+002 7.000+001 5.600-001 4.957+002
2904 H2,H2PH2= 33.97 -.11
2905 H2,H2PH2= 34.23 -.26
2906 H2,H2PH2= 34.79 -.55
2907 H2,H2PH2= 35.03 -.24
2908 H2,H2PH2= 35.08 -.05
2909 H2,H2PH2= 35.09 -.01
2910
2911 CONDENSER ITERATION:
2912
2913 T P H X
2914 1.507+002 2.247+002 1.020+002 1.000+000
2915 8.533+001 2.163+002 3.509+001 .000
2916
2917
2918 LIQUID LINE:
2919 INPUT - PIN = 2.163+002 TIN = 8.538+001 XIN = .000
2920 OUTPUT - POUT= 2.105+002 TOUT= 8.534+001 XOUT= .000
2921
2922 HIN = 3.509+001
2923
2924 EXPANSION DEVICE:
2925 INPUT - P12 =2.105+002 H12 =3.509+001 P13 =9.639+001
2926 DDENFA DOES NOT CONVERGE, DIFXQ2= -8.00648-006
2927 OUTPUT - POUT =1.626+002 XMASS =4.904+002
2928
2929 INPUT DATA TO COMP:
2930
2931 P3 T3 H3 XQ3 TQ5 TRA TGA
2932 6.773+001 3.187+001 8.642+001 9.906-001 1.131+002 7.000+001 4.700+001
2933
2934 COMPRESSOR ITERATION:
2935
2936 EI ETAE ETAC ETAV CPRPM RMASS
2937 2.779+000 7.602-001 9.466-001 8.566-001 3.543+003 4.050+002
2938
2939 I T P H X
2940 1 3.720+001 7.544+001 8.523+001 9.707-001
2941 2 3.656+001 7.436+001 8.517+001 9.706-001
2942 3 3.187+001 6.773+001 8.642+001 9.906-001
2943 4 6.744+001 6.764+001 9.299+001 1.000+000
2944 5 7.113+001 6.478+001 9.371+001 1.000+000
2945 6 1.837+002 2.502+002 1.077+002 1.000+000
2946 7 1.769+002 2.286+002 1.070+002 1.000+000
2947 8 1.584+002 2.276+002 1.034+002 1.000+000
2948 9 1.516+002 2.251+002 1.021+002 1.000+000
2949 10 1.509+002 2.250+002 1.020+002 1.000+000
2950
2951 INPUT DATA TO CONDHX:
2952
2953 T P TAIR RH RMASS
2954 1.509+002 2.250+002 7.000+001 5.600-001 4.956+002
2955 H2,H2PH2= 33.80 -.12
2956 H2,H2PH2= 34.57 -.77
2957 H2,H2PH2= 34.47 .10
2958 H2,H2PH2= 34.78 -.31

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2958 H2,H2PH2= 34.88 -.10
2959 H2,H2PH2= 34.94 -.06
2960 H2,H2PH2= 34.94 .00
2961
2962 CONDENSER ITERATION:
2963
2964 T P H X
2965 1.509+002 2.250+002 1.020+002 1.000+000
2966 8.530+001 2.167+002 3.494+001 .000
2967
2968
2969 LIQUID LINE:
2970 INPUT - PIN = 2.167+002 TIN = 8.530+001 XIN = .000
2971 OUTPUT - POUT = 2.109+002 TOUT = 8.527+001 XOUT = .000
2972
2973
2974 EXPANSION DEVICE:
2975 INPUT - P12 = 2.109+002 H12 = 3.494+001 P13 = 9.633+001
2976 OUTPUT - POUT = 1.638+002 XMASS = 4.954+002
2977
2978 INPUT DATA TO EVAPHX:
2979
2980 T P X X TAIR RH RMAS
2981 3.597+001 9.638+001 2.253-001 4.700+001 7.300-001 4.950+002
2982 H2= 89.0563
2983 H2= 87.3074
2984 H2= 85.2126
2985 H2= 85.1421
2986 H2= 85.0411
2987 QT,QS= 2.483+004 2.153+004
2988
2989 EVAPORATOR ITERATION:
2990
2991 T P H X TSUP
2992 3.597+001 9.638+001 3.494+001 2.253-001
2993 3.698+001 7.520+001 8.505+001 9.689-001 .000
2994
2995 INPUT DATA TO EVAPHX:
2996
2997 T P X X TAIR RH RMAS
2998 3.603+001 9.648+001 2.251-001 4.700+001 7.300-001 4.950+002
2999 H2= 89.0079
3000 H2= 87.0411
3001 H2= 84.8757
3002 H2= 84.6077
3003 H2= 84.7199
3004 QT,QS= 2.458+004 2.145+004
3005
3006 EVAPORATOR ITERATION:
3007
3008 T P H X TSUP
3009 3.603+001 9.648+001 3.494+001 2.251-001
3010 3.707+001 7.544+001 8.473+001 9.651-001 .000
3011 P1,P1E,P1CP1,DENT2= 75.4447 75.4423 -.0024
3012 INTERM,INTERE,FOUT,P13= 0 0 163.83 96.48
3013 COMPOSITION OF REFRIG. IN ACCUMULATOR = .414
3014 (WEIGHT FRACTION OF MORE VOLATILE COMPONENT)
3015

HIN = 3.494+001

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REFRIG. IN ACCUMULATOR = .796 LB
REFRIG. IN INDOOR COIL = 2.492 LB
REFRIG. IN OUTDOOR COIL = .915 LB
TMASS = 5.350+000 REFIN = 4.969+000

SYST4 WITH ACCUM., 13E1/152A 6-22-81*

TOA RHQA TRA RHRA
4.700+001 7.300+001 7.000+001 5.600+001
CFMIND CFMOUT
1.118+003 2.284+003

RESULTS:

I	T	P	H	S	X
1	3.720+001	7.544+001	8.323+001	1.862+001	9.707+001
2	3.656+001	7.456+001	8.517+001	1.863+001	9.703+001
3	3.187+001	6.773+001	8.642+001	1.905+001	9.906+001
4	6.744+001	6.764+001	9.259+001	2.035+001	1.000+000
5	7.113+001	6.478+001	9.371+001	2.056+001	1.000+000
6	1.837+002	2.502+002	1.077+002	2.067+001	1.000+000
7	1.769+002	2.286+002	1.070+002	2.071+001	1.000+000
8	1.534+002	2.276+002	1.034+002	2.014+001	1.000+000
9	1.516+002	2.251+002	1.021+002	1.996+001	1.000+000
10	1.509+002	2.250+002	1.020+002	1.993+001	1.000+000
11	8.530+001	2.167+002	3.494+001	7.987+002	.000
12	8.527+001	2.109+002	3.494+001	7.989+002	.000
13	3.603+001	9.648+001	3.494+001	8.179+002	2.251+001

TG3	TSUP3	TG6	RMASS	TMASS
3.208+001	.000	1.131+002	4.955+002	5.350+000

QLOAD	ELUSE	COP
3.487+004	3.644+000	2.804+000

REFRIG. COMPOSITION = .668
(WEIGHT FRACTION OF MORE VOLATILE COMPONENT)

INPUT DATA TO COMP:

P3	T3	H3	XQ3	TG6	TRA	TOA
6.713+001	3.151+001	8.723+001	1.000+000	1.121+002	7.000+001	4.700+001

COMPRESSOR ITERATION:

EI	ETAE	ETAC	ETAV	CPRPM	RMASS
2.750+000	7.539+001	9.475+001	8.562+001	3.548+003	4.877+002

I	T	P	H	X
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1 3.689+001 7.474+001 8.599+001 9.799-001
2 3.627+001 7.383+001 8.593+001 9.799-001
3 3.161+001 6.713+001 8.723+001 1.000+000
4 7.018+001 6.704+001 9.346+001 1.000+000
5 7.389+001 6.422+001 9.418+001 1.000+000
6 1.858+002 2.470+002 1.032+002 1.000+000
7 1.791+002 2.257+002 1.075+002 1.000+000
8 1.606+002 2.246+002 1.039+002 1.000+000
9 1.535+002 2.221+002 1.026+002 1.000+000
10 1.527+002 2.221+002 1.024+002 1.000+000

INPUT DATA TO CONDHX:

T	P	TAIR	RH	RMASS
1.527+002	2.221+002	7.000+001	5.600-001	4.877+002
H2, H2PH2=	34.66	- .68		
H2, H2PH2=	34.60	.06		
H2, H2PH2=	35.22	-.62		
H2, H2PH2=	35.47	-.25		
H2, H2PH2=	35.57	-.10		
H2, H2PH2=	35.57	.00		
H2, H2PH2=	35.57	.00		

CONDENSER ITERATION:

T	P	H	X
1.527+002	2.221+002	1.024+002	1.000+000
8.772+001	2.135+002	3.557+001	.000

LIQUID LINE:

INPUT - PIN = 2.135+002 TIN = 8.772+001 XIN = .000
OUTPUT - POUT = 2.074+002 TOUT = 8.571+001 XOUT = 1.127-002
HIN = 3.557+001

EXPANSION DEVICE:

INPUT - P12 = 2.074+002 H12 = 3.557+001 P13 = 9.577+001
OUTPUT - POUT = 1.582+002 XMASS = 4.691+002

INPUT DATA TO COMP:

P3	T3	H3	X03	T06	TRA	T0A
6.713+001	3.161+001	8.723+001	1.000+000	1.124+002	7.000+001	4.700+001

COMPRESSOR ITERATION:

E1	ETAE	ETAC	ETAV	CP3PM	RM/SS
2.756+000	7.592-001	9.473-001	8.559-001	3.548+003	4.873+002

I	T	P	H	X
1	3.688+001	7.473+001	8.599+001	9.798-001
2	3.626+001	7.387+001	8.593+001	9.799-001
3	3.161+001	6.713+001	8.723+001	1.000+000
4	7.026+001	6.704+001	9.348+001	1.000+000
5	7.389+001	6.423+001	9.420+001	1.000+000
6	1.863+002	2.480+002	1.083+002	1.000+000
7	1.796+002	2.268+002	1.076+002	1.000+000
8	1.610+002	2.258+002	1.039+002	1.000+000

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3132 9 1.540+002 2.233+002 1.026+002 1.000+000
3133 10 1.531+002 2.232+002 1.025+002 1.000+000
3134
3135 INPUT DATA TO CONDHX:
3136
3137 T P TAIR RH RMASS
3138 1.531+002 2.232+002 7.000+001 5.600+001 4.973+002
3139 H2, H2PH2= 34.30 -.58
3140 H2, H2PH2= 34.33 -.03
3141 H2, H2PH2= 34.32 -.59
3142 H2, H2PH2= 34.49 .43
3143 H2, H2PH2= 34.71 -.22
3144 H2, H2PH2= 34.82 -.11
3145 H2, H2PH2= 34.89 -.07
3146 H2, H2PH2= 34.88 .01
3147
3148 CONDENSER ITERATION:
3149
3150 T P H X
3151 1.531+002 2.232+002 1.025+002 1.000+000
3152 8.504+001 2.149+002 3.488+001 .000
3153
3154 LIQUID LINE:
3155 INPUT - PIN = 2.149+002 TIN = 8.504+001 XIN = .000
3156 OUTPUT - POUT = 2.093+002 TOUT = 8.502+001 XOUT = .000
3157
3158 HIN = 3.488+001
3159
3160 EXPANSION DEVICE:
3161 INPUT - P12 = 2.093+002 H12 = 3.488+001 P13 = 9.576+001
3162 OUTPUT - POUT = 1.620+002 XMASS = 4.903+002
3163
3164 INPUT DATA TO CONFR:
3165
3166 P3 T3 H3 XQ3 T66 TRA TGA
3167 6.713+001 3.161+001 8.723+001 1.000+000 1.124+002 7.030+001 4.700+001
3168
3169 COMPRESSOR ITERATION:
3170
3171 EI ETAE ETAC ETAV ETAV CPRPM RMASS
3172 2.755+000 7.592+001 9.474+001 3.553+001 3.548+003 4.874+002
3173
3174 I T P H X
3175 1 3.688+001 7.473+001 3.599+001 9.798+001
3176 2 3.626+001 7.389+001 3.593+001 9.799+001
3177 3 3.161+001 6.713+001 8.723+001 1.000+000
3178 4 7.025+001 6.704+001 9.347+001 1.000+000
3179 5 7.393+001 6.423+001 9.419+001 1.000+000
3180 6 1.862+002 2.478+002 1.083+002 1.000+000
3181 7 1.795+002 2.267+002 1.076+002 1.000+000
3182 8 1.609+002 2.256+002 1.039+002 1.000+000
3183 9 1.539+002 2.231+002 1.026+002 1.000+000
3184 10 1.531+002 2.230+002 1.025+002 1.000+000
3185
3186 INPUT DATA TO CONDHX:
3187
3188 T P TAIR RH RMASS
3189 1.531+002 2.230+002 7.000+001 5.600+001 4.874+002

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3190 H2, H2PH2= 34.40 - .56
 3191 H2, H2PH2= 34.42 - .03
 3192 H2, H2PH2= 35.00 - .58
 3193 H2, H2PH2= 34.60 - .41
 3194 H2, H2PH2= 34.83 - .23
 3195 H2, H2PH2= 34.93 - .11
 3196 H2, H2PH2= 34.99 - .06
 3197 H2, H2PH2= 34.98 .01
 3198
 3199 CONDENSER ITERATION:
 3200
 3201 T P H X
 3202 1.531+002 2.230+002 1.025+002 1.000+000
 3203 8.546+001 2.147+002 3.498+001 .000
 3204
 3205
 3206 LIQUID LINE:
 3207 INPUT - PIN = 2.147+002 TIN = 8.546+001 XIN = .000
 3208 OUTPUT - POUT = 2.091+002 TOUT = 8.544+001 XOUT = .000
 3209
 3210
 3211 EXPANSION DEVICE:
 3212 INPUT - P12 = 2.091+002 H12 = 3.498+001 P13 = 9.576+001
 3213 DDENFA DOES NOT CONVERGE, DIFFX02 = -3.49636-006
 3214 OUTPUT - POUT = 1.612+002 XMASS = 4.867+002
 3215
 3216 INPUT DATA TO EVAPHX:
 3217
 3218 T P X X TAIR RH RMASS
 3219 3.560+001 9.576+001 2.276-001 4.700+001 7.300-001 4.874+002
 3220 H2= 89.2099
 3221 H2= 88.1069
 3222 H2= 86.9356
 3223 H2= 85.5808
 3224 QT, QS= 2.513+004 2.190+004
 3225
 3226 EVAPORATOR ITERATION:
 3227
 3228 T P H X TSUP
 3229 3.560+001 9.576+001 3.497+001 2.276-001
 3230 3.676+001 7.436+001 8.655+001 9.864-001 .000
 3231
 3232 INPUT DATA TO EVAPHX:
 3233
 3234 T P X X TAIR RH RMASS
 3235 3.569+001 9.591+001 2.273-001 4.700+001 7.300-001 4.874+002
 3236 H2= 89.1770
 3237 H2= 87.9079
 3238 H2= 86.4816
 3239 H2= 86.1747
 3240 QT, QS= 2.493+004 2.177+004
 3241
 3242 EVAPORATOR ITERATION:
 3243
 3244 T P H X TSUP
 3245 3.569+001 9.591+001 3.499+001 2.273-001
 3246 3.691+001 7.472+001 8.614+001 9.815-001 .000
 3247 P1, P1E, P1EP1, DENT2= 74.7282 74.7190 -.0091 .1507

3249 INTER, INTERE, POUT, F13= 0 0 161.18 95.91
3249 COMPOSITION OF REFRIG. IN ACCUMULATOR = .668
3250 (WEIGHT FRACTION OF MORE VOLATILE COMPONENT)
3251

3252 REFRIG. IN ACCUMULATOR = .190 LB
3253

3254 REFRIG. IN INDOOR COIL = 2.392 LB
3255

3256 REFRIG. IN OUTDOOR COIL = .882 LB
3257

3258 TMASS = 4.602+000 REFIN = 4.968+000
3259

3260 *****
3261 **SYST4 WITH ACCUM., 13B1/152A 6-22-84***
3262

3263 TOA RHQA TRA RHRA
3264 4.700+001 7.300+001 7.000+001 5.600+001
3265

3266 CFMIND CFMOUT
3267 1.118+003 2.284+003
3268

RESULTS:

I	T	P	H	S	X
1	3.688+001	7.473+001	8.599+001	1.879+001	9.798+001
2	3.626+001	7.388+001	8.573+001	1.880+001	9.799+001
3	3.161+001	6.713+001	8.723+001	1.923+001	1.000+000
4	7.025+001	6.704+001	9.347+001	2.045+001	1.000+000
5	7.358+001	6.423+001	9.419+001	2.066+001	1.000+000
6	1.062+002	2.473+002	1.083+002	2.078+001	1.000+000
7	1.795+002	2.267+002	1.076+002	2.082+001	1.000+000
8	1.609+002	2.256+002	1.039+002	2.025+001	1.000+000
9	1.539+002	2.231+002	1.026+002	2.005+001	1.000+000
10	1.531+002	2.230+002	1.025+002	2.003+001	1.000+000
11	3.546+001	2.147+002	3.498+001	7.936+002	.000
12	8.514+001	2.091+002	3.496+001	7.938+002	.000
13	3.569+001	3.591+001	3.498+001	8.191+002	2.273+001

TG3	TSUP3	TG6	RMASS	TMASS
3.161+001	.000	1.124+002	4.874+002	4.602+000

QLOAD	FLUSE	COP
3.455+004	3.620+000	2.795+000

3290 REFRIG. COMPOSITION = .668
3291 (WEIGHT FRACTION OF MORE VOLATILE COMPONENT)
3292

INPUT DATA TO COMPR:

P3	T3	H3	XG3	TG6	TRA	TOA
6.743+001	3.173+001	8.683+001	9.953+001	1.127+002	7.000+001	4.700+001

COMPRESSOR ITERATION:

3306	EI	ETAE	ETAC	ETAV	CPRPM	RMASS
3307	2	767+000	7.597-001	9.470-001	8.562-001	3.548+003
3308						4.914+002
3309	I	T	P	H	X	
3310	1	3.703+001	7.508+001	8.551+001	9.753-001	
3311	2	3.641+001	7.421+001	8.555+001	9.754-001	
3312	3	3.173+001	6.743+001	8.683+001	9.953-001	
3313	4	6.884+001	6.733+001	9.323+001	1.000+000	
3314	5	7.255+001	6.449+001	9.395+001	1.000+000	
3315	6	1.850+002	2.430+002	1.080+002	1.000+000	
3316	7	1.782+002	2.273+002	1.073+002	1.000+000	
3317	8	1.597+002	2.266+002	1.037+002	1.000+000	
3318	9	1.528+002	2.241+002	1.024+002	1.000+000	
3319	10	1.570+002	2.240+002	1.022+002	1.000+000	
3320						
3321						
3322						
3323	T	P	TAIR	RH	RMASS	
3324	1	520+002	2.240+002	7.000+001	5.600-001	4.914+002
3325	H2, H2PH2=	34.03	- .07			
3326	H2, H2PH2=	34.28	- .25			
3327	H2, H2PH2=	34.78	- .51			
3328	H2, H2PH2=	35.03	- .25			
3329	H2, H2PH2=	35.08	- .05			
3330	H2, H2PH2=	35.09	- .01			
3331						
3332						
3333						
3334	T	P	H	X		
3335	1	520+002	2.240+002	1.022+002	1.000+000	
3336	8	589+001	2.158+002	3.503+001	.000	
3337						
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3357						
3358	I	T	P	H	X	
3359	1	3.703+001	7.508+001	8.551+001	9.753-001	
3360	2	3.641+001	7.421+001	8.555+001	9.754-001	
3361	3	3.173+001	6.743+001	8.683+001	9.953-001	
3362	4	6.884+001	6.733+001	9.323+001	1.000+000	
3363	5	7.254+001	6.450+001	9.395+001	1.000+000	

INPUT DATA TO CONDHX:

LIQUID LINE:

INPUT - PIN = 2.158+002 TIN = 8.589+001 XIN = .000

OUTPUT - POUT = 2.100+002 TOUT = 8.586+001 XOUT = .000

HIN = 3.509+001

EXPANSION DEVICE:

INPUT - P12 = 2.100+002 H12 = 3.509+001 P13 = 9.626+001

OUTPUT - POUT = 1.518+002 XMASS = 4.878+002

INPUT DATA TO COMP:

P3 T3 H3 X03 X06 T06 TRA T0A

6.743+001 3.173+001 6.683+001 9.353-001 1.128+002 7.000+001 4.700+001

COMPRESSOR ITERATION:

EI ETAE ETAC ETAV CPRPM RMASS

2.769+000 7.597-001 9.469-001 8.561-001 3.548+003 4.914+002

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3364 6 1.850+002 2.491+002 1.080+002 1.000+000
3365 7 1.783+002 2.278+002 1.073+002 1.000+000
3366 8 1.597+002 2.268+002 1.037+002 1.000+000
3367 9 1.529+002 2.242+002 1.024+002 1.000+000
3368 10 1.520+002 2.242+002 1.022+002 1.000+000
3369
3370 INPUT DATA TO CONDIX:
3371
3372 T P TAIR RH RMASS
3373 1.520+002 2.242+002 7.000+001 5.500-001 4.914+002
3374 H2,H2PH2= 33.94 -.07
3375 H2,H2PH2= 34.18 -.24
3376 H2,H2PH2= 34.70 -.52
3377 H2,H2PH2= 34.94 -.24
3378 H2,H2PH2= 34.99 -.05
3379 H2,H2PH2= 35.01 -.01
3380
3381 CONDENSER ITERATION:
3382
3383 T P H X
3384 1.520+002 2.242+002 1.022+002 1.000+000
3385 8.556+001 2.160+002 3.501+001 .000
3386
3387 LIQUID LINE:
3388 INPUT - PIN = 2.160+002 TIN = 8.556+001 XIN = .000
3389 OUTPUT - POUT = 2.102+002 TOUT = 8.552+001 XOUT = .000
3390
3391 HIN = 3.501+001
3392
3393 EXPANSION DEVICE:
3394 INPUT - P12 = 2.102+002 H12 = 3.501+001 P13 = 9.626+001
3395 OUTPUT - POUT = 1.628+002 XMASS = 4.915+002
3396
3397 INPUT DATA TO EVAPHX:
3398
3399 T P X TAIR RH RMASS
3400 3.591+001 9.626+001 2.268-001 4.700+001 7.300-001 4.914+002
3401 H2= 89.0907
3402 H2= 87.5102
3403 H2= 85.5433
3404 H2= 85.3536
3405 H2= 85.2817
3406 QT, QS= 2.471+004 2.146+004
3407
3408 EVAPORATOR ITERATION:
3409
3410 T P H X TSUP
3411 3.591+001 9.626+001 3.501+001 2.268-001
3412 3.704+001 7.520+001 8.529+001 9.716-001 .000
3413 P1,P1E,P1EP1,DEHT2= 75.0783 75.2007 .1225 -.3170
3414 INTERM,INFERE,POUT,P13= 0 0 162.77 96.26
3415 COMPOSITION OF REFRIG. IN ACCUMULATOR = .645
3416 (WEIGHT FRACTION OF MORE VOLATILE COMPONENT)
3417
3418 REFRIG. IN ACCUMULATOR = .205 LB
3419
3420 REFRIG. IN INDOOR COIL = 2.475 LB
3421

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REFRIG. IN OUTDOOR COIL= .903 LB

TMASS = 4.726+000 REFIN = 4.968+000

SYST4 WITH ACCUM., 13B1/152A 6-22-81*

TOA RHQA TPA RHRA
4.700+001 7.300+001 7.000+001 5.600+001

CFMIND CFMOUT
1.118+003 2.204+003

RESULTS:

I	T	P	H	S	X
1	3.703+001	7.503+001	8.561+001	1.871+001	9.753+001
2	3.641+001	7.421+001	8.555+001	1.872+001	9.754+001
3	3.173+001	6.743+001	8.633+001	1.914+001	9.953+001
4	6.883+001	6.733+001	9.323+001	2.040+001	1.000+000
5	7.254+001	6.450+001	9.395+001	2.031+001	1.000+000
6	1.850+002	2.491+002	1.080+002	2.073+001	1.000+000
7	1.783+002	2.278+002	1.073+002	2.076+001	1.000+000
8	1.597+002	2.265+002	1.037+002	2.019+001	1.000+000
9	1.528+002	2.242+002	1.024+002	2.001+001	1.000+000
10	1.520+002	2.242+002	1.022+002	1.998+001	1.000+000
11	8.556+001	2.160+002	3.501+001	9.000+002	.000
12	8.552+001	2.102+002	3.501+001	8.002+002	.000
13	3.591+001	9.626+001	3.501+001	8.194+002	2.203+001

TG3 TSUP3 TG6 RMASS TMASS
3.184+001 .000 1.123+002 4.914+002 4.726+000

QLOAD ELUSE COP
3.460+004 3.634+000 2.797+000

REFRIG. COMPOSITION = .668
(WEIGHT FRACTION OF MORE VOLATILE COMPONENT)

INPUT DATA TO COMPR:

P3 T3 H3 X03 TG6 TFA TOA
6.800+001 3.213+001 8.603+001 9.930+001 1.135+002 7.000+001 4.700+001

COMPRESSOR ITERATION:

E1 ETAE ETAC ETAV CPRPM RMASS
2.792+000 7.607+001 9.465+001 8.563+001 3.548+003 4.965+002

I	T	P	H	X
1	3.745+001	7.573+001	8.545+001	9.729+001
2	3.682+001	7.485+001	8.539+001	9.731+001
3	3.213+001	6.800+001	8.603+001	9.930+001
4	6.834+001	6.791+001	9.313+001	1.000+000

3480 5 7.204+001 6.504+001 9.365+001 1.000+000
3481 6 1.849+002 2.517+002 1.079+002 1.000+000
3482 7 1.781+002 2.302+002 1.072+002 1.000+000
3483 8 1.506+002 2.232+002 1.035+002 1.000+000
3484 9 1.527+002 2.266+002 1.023+002 1.000+000
3485 10 1.519+002 2.266+002 1.021+002 1.000+000
3486
3487 INPUT DATA TO CONDHX:
3488
3489 T P TAIR RH RMASS
3490 1.519+002 2.256+002 7.000+001 5.600+001 4.965+002
3491 H2,H2FH2= 33.58 - .25
3492 H2,H2PH2= 33.83 - .25
3493 H2,H2PH2= 34.28 - .45
3494 H2,H2FH2= 34.53 - .25
3495 H2,H2FH2= 34.57 - .04
3496 H2,H2FH2= 34.57 .00
3497
3498 CONDENSER ITERATION:
3499
3500 T P H X
3501 1.519+002 2.256+002 1.021+002 1.000+000
3502 8.385+001 2.187+002 3.457+001 .000
3503
3504
3505 LIQUID LINE:
3506 INPUT - PIN = 2.187+002 TIN = 8.385+001 XIN = .000
3507 OUTPUT - POUT = 2.129+002 TOUT = 8.383+001 XOUT = .000
3508
3509
3510 EXPANSION DEVICE:
3511 INPUT - P12 = 2.129+002 H12 = 3.457+001 P13 = 9.691+001
3512 OUTPUT - POUT = 1.675+002 XMASS = 5.131+002
3513
3514 INPUT DATA TO COMPR:
3515
3516 P3 T3 H3 XQ3 TQ6 TRA TOA
3517 6.800+001 3.213+001 8.663+001 9.930+001 1.133+002 7.000+001 4.700+001
3518
3519 COMPRESSOR ITERATION:
3520
3521 E1 ETAE ETAC ETAV ETAV CPRPM RMASS
3522 2.787+000 7.605+001 9.467+001 8.566+001 3.545+003 4.000+002
3523
3524 I T P H X
3525 1 3.746+001 7.574+001 8.545+001 9.730+001
3526 2 3.633+001 7.486+001 8.540+001 9.731+001
3527 3 3.213+001 6.800+001 8.666+001 9.930+001
3528 4 6.828+001 6.791+001 9.312+001 1.000+000
3529 5 7.196+001 6.504+001 9.323+001 1.000+000
3530 6 1.845+002 2.503+002 1.079+002 1.000+000
3531 7 1.777+002 2.294+002 1.071+002 1.000+000
3532 8 1.592+002 2.283+002 1.035+002 1.000+000
3533 9 1.524+002 2.257+002 1.022+002 1.000+000
3534 10 1.515+002 2.257+002 1.021+002 1.000+000
3535
3536 INPUT DATA TO CONDHX:
3537

3538	T	1.516+002	P	2.257+002	TAIR	RH	RMASS	
3539	H2, H2PH2=	34.03		- .58		5.600+001	4.938+002	
3540	H2, H2PH2=	34.25		- .22				
3541	H2, H2PH2=	34.19		.06				
3542	H2, H2PH2=	34.54		- .35				
3543	H2, H2PH2=	34.70		- .16				
3544	H2, H2PH2=	34.74		- .04				
3545	H2, H2PH2=	34.74		.00				
3546	H2, H2PH2=	34.74						
3547								
3548	CONDENSER ITERATION:							
3549								
3550	T	1.516+002	P	2.257+002	H	X		
3551	8.452+001	2.175+002	1.021+002	1.000+000				
3552			3.474+001	.000				
3553								
3554								
3555	LIQUID LINE:							
3556	INPUT -	PIN = 2.175+002		TIN = 8.452+001	XIN = .000			
3557	OUTPUT -	POUT = 2.117+002		TEUT = 8.418+001	XOUT = .000			
3558								HIN = 3.474+001
3559								
3560	EXPANSION DEVICE:							
3561	INPUT -	P12 = 2.117+002		H12 = 3.474+001	P13 = 3.602+001			
3562	OUTPUT -	POUT = 1.656+002		X1ASS = 5.038+002				
3563								
3564	INPUT DATA TO EVAPHX:							
3565								
3566	T	3.627+001	P	9.692+001	2.227+001	TAIR	RH	RMASS
3567	H2=	83.7629			4.700+001	7.300+001	4.970+002	
3568	H2=	85.5537						
3569	H2=	83.2565						
3570	QT, QS=	2.403+004	2.155+004					
3571								
3572	EVAPORATOR ITERATION:							
3573								
3574								
3575	T	3.627+001	P	9.692+001	H	X	TSUP	
3576	3.741+001	7.649+001	3.486+001	2.227+001				
3577			8.322+001	9.469+001			.000	
3578	INPUT DATA TO EVAPHX:							
3579								
3580								
3581	T	3.610+001	P	9.662+001	2.234+001	TAIR	RH	RMASS
3582	H2=	83.9210			4.700+001	7.300+001	4.970+002	
3583	H2=	86.5753						
3584	H2=	84.3140						
3585	QT, QS=	2.456+004	2.193+004					
3586								
3587	EVAPORATOR ITERATION:							
3588								
3589								
3590	T	3.610+001	P	9.662+001	H	X	TSUP	
3591	3.717+001	7.574+001	3.486+001	2.234+001				
3592			8.429+001	9.598+001			.000	
3593	P1, P1E, P1EP1, DENT2=	75.7395	75.7413				.0017	-1.1637
3594	INPUT DATA TO COMPR:							
3595								

3596 P3 T3 H3 XQ3 TG6 TRA TOA
3597 6.743+001 3.173+001 8.662+001 9.930+001 1.128+002 7.000+001 4.700+001
3598
3599 COMPRESSOR ITERATION:
3600
3601 EI ETAE ETAC ETAV ETAV CPRPM RMASS
3602 2.770+000 7.596+001 9.460+001 8.563+001 3.543+003 4.928+002
3603
3604
3605 I T P H X
3606 1 3.704+001 7.516+001 8.541+001 9.730+001
3607 2 3.641+001 7.428+001 8.505+001 9.731+001
3608 3 3.173+001 6.729+001 8.562+001 9.330+001
3609 4 6.810+001 6.739+001 9.311+001 1.000+000
3610 5 7.180+001 6.434+001 9.383+001 1.000+000
3611 6 1.843+002 2.493+002 1.078+002 1.000+000
3612 7 1.775+002 2.279+002 1.071+002 1.000+000
3613 8 1.590+002 2.268+002 1.035+002 1.000+000
3614 9 1.522+002 2.243+002 1.023+002 1.000+000
3615 10 1.514+002 2.242+002 1.021+002 1.000+000
3616
3617 INPUT DATA TO CONDHX:
3618
3619 T TAIR RH RMASS
3620 1.514+002 2.242+002 7.000+001 5.600+001 4.928+002
3621 H2,H2PH2= 33.99 -.08
3622 H2,H2PH2= 34.24 -.25
3623 H2,H2PH2= 34.77 -.53
3624 H2,H2PH2= 35.01 -.24
3625 H2,H2PH2= 35.07 -.05
3626 H2,H2PH2= 35.08 -.01
3627
3628 CONDENSER ITERATION:
3629
3630 T P H X
3631 1.514+002 2.242+002 1.021+002 1.000+000
3632 8.583+001 2.190+002 3.508+001 .000
3633
3634
3635 LIQUID LINE:
3636 INPUT - PIN = 2.160+002 TIN = 3.533+001 XIN = .000
3637 OUTPUT - POUT = 2.102+002 TOUT = 3.530+001 XOUT = .000
3638
3639
3640 EXPANSION DEVICE:
3641 INPUT - P12 = 2.102+002 H12 = 3.508+001 P13 = 9.604+001
3642 DDENFA DOES NOT CONVERGE, DIFXQ2= -5.18214-006
3643 OUTPUT - POUT = 1.628+002 XFMSS = 4.997+002
3644
3645 INPUT DATA TO COMPR:
3646
3647 P3 T3 H3 XQ3 TG6 TRA TOA
3648 6.749+001 3.173+001 8.662+001 9.930+001 1.129+002 7.000+001 4.700+001
3649
3650 COMPRESSOR ITERATION:
3651
3652 EI ETAE ETAC ETAV ETAV CPRPM RMASS
3653 2.772+000 7.599+001 9.467+001 8.562+001 3.543+003 4.927+002

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3654
3655      I   T       P       H       X
3656      1   3.704+001  7.515+001  8.541+001  9.730+001
3657      2   3.641+001  7.428+001  8.525+001  9.731+001
3658      3   3.173+001  6.748+001  8.662+001  9.930+001
3659      4   6.813+001  6.739+001  9.311+001  1.000+000
3660      5   7.183+001  6.455+001  9.383+001  1.000+000
3661      6   1.844+002  2.495+002  1.078+002  1.000+000
3662      7   1.776+002  2.281+002  1.072+002  1.000+000
3663      8   1.591+002  2.271+002  1.035+002  1.000+000
3664      9   1.523+002  2.246+002  1.023+002  1.000+000
3665     10   1.515+002  2.245+002  1.021+002  1.000+000
3666
3667      INPUT DATA TO CONDHX:
3668
3669      T       P       TAIR       RH       RMASS
3670      1.515+002  2.245+002  7.000+001  5.600+001  4.927+002
3671      H2,H2PH2= 33.95      -.09
3672      H2,H2PH2= 34.09      -.25
3673      H2,H2PH2= 34.65      -.55
3674      H2,H2PH2= 34.68      -.24
3675      H2,H2PH2= 34.93      -.05
3676      H2,H2PH2= 34.94      -.01
3677
3678      CONDENSER ITERATION:
3679
3680      T       P       H       X
3681      1.515+002  2.245+002  1.021+002  1.000+000
3682      8.530+001  2.163+002  3.494+001  .000
3683
3684
3685      LIQUID LINE:
3686      INPUT -   PIN = 2.163+002      TIN = 8.530+001      XIN = .000
3687      OUTPUT -  POUT = 2.105+002     TOUT = 8.527+001     XOUT = .000
3688
3689
3690
3691      EXPANSION DEVICE:
3692      INPUT -   P12 = 2.105+002      H12 = 3.494+001      P13 = 9.604+001
3693      OUTPUT -  POUT = 1.635+002     XMASS = 1.942+002
3694
3695      INPUT DATA TO EVAPHX:
3696
3697      T       P       X       TAIR       RH       RMASS
3698      3.576+001  9.604+001  2.262+001  4.700+001  7.300+001  4.927+002
3699      H2= 89.1718
3700      H2= 87.8456
3701      H2= 86.2828
3702      H2= 86.0935
3703      H2= 86.0277
3704      QT,QS= 2.516+004  2.170+004
3705
3706      EVAPORATOR ITERATION:
3707
3708      T       P       H       X       TSUP
3709      3.576+001  9.604+001  3.494+001  2.262+001  .000
3710      3.679+001  7.453+001  8.601+001  9.902+001  .000
3711
3712      INPUT DATA TO EVAPHX:

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3712 T 3.589+001 9.626+001 2.257-001 4.700+001 7.300-001 4.927+002
3713 P 3.589+001 9.626+001 2.257-001 4.700+001 7.300-001 4.927+002
3714 H2= 89.0000
3715 H2= 87.4873
3716 H2= 85.5003
3717 H2= 85.3471
3718 H2= 85.2536
3719 QT, QS= 2.479+004 2.150+004
3720
3721 EVAPORATOR ITERATION:
3722
3723
3724 T P H X TSUP
3725 3.509+001 9.626+001 3.494+001 2.257-001
3726 3.700+001 7.515+001 8.527+001 9.714-001 .000
3727 P1,PIE,P1EPI,DENT2= 75.1531 75.1549 .0019 -.1412
3728 INTERM,INTER,FOOT,P13= 0 0 163.49 96.26
3729 COMPOSITION OF REFRIG. IN ACCUMULATOR = .637
3730 (WEIGHT FRACTION OF MORE VOLATILE COMPONENT)
3731
3732 REFRIG. IN ACCUMULATOR = .212 LB
3733
3734 REFRIG. IN INDOOR COIL = 2.487 LB
3735
3736 REFRIG. IN OUTDOOR COIL = .905 LB
3737
3738 TMASS = 4.749+000 REFIN = 4.938+000
3739
3740 *****
3741 **SYST4 WITH ACCUM., 13B1/152A G-22-34***
3742
3743 TQA RHOA TRA RHRA
3744 4.700+001 7.300-001 7.000+001 5.600-001
3745
3746 CFMIND CFMOUT
3747 1.118+003 2.231+003
3748
3749
3750
3751
3752 RESULTS:
3753 I T P H X S
3754 1 3.704+001 7.515+001 8.541+001 1.867-001 9.730-001
3755 2 3.641+001 7.420+001 8.505+001 1.868-001 9.731-001
3756 3 3.173+001 6.749+001 8.662+001 1.903-001 9.930-001
3757 4 6.813+001 6.739+001 9.311+001 2.038-001 1.000+000
3758 5 7.183+001 6.455+001 9.383+001 2.059-001 1.000+000
3759 6 1.844+002 2.455+002 1.070+002 2.070-001 1.000+000
3760 7 1.776+002 2.201+002 1.072+002 2.074-001 1.000+000
3761 8 1.591+002 2.271+002 1.035+002 2.017-001 1.000+000
3762 9 1.523+002 2.246+002 1.023+002 1.998-001 1.000+000
3763 10 1.515+002 2.245+002 1.021+002 1.996-001 1.000+000
3764 11 8.530+001 2.163+002 3.494+001 7.987-002 .000
3765 12 8.527+001 2.165+002 3.491+001 7.989-002 .000
3766 13 3.589+001 9.626+001 3.491+001 8.130-002 2.257-001
3767
3768 TGA TSUP3 TGA TMASS
3769 3.180+001 .000 1.129+002 4.927+002 4.749+000

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LOAD ELUSE COP
3.475+004 3.637+000 2.800+000

REFRIG. COMPOSITION = .668
(WEIGHT FRACTION OF MORE VOLATILE COMPONENT)

INPUT DATA TO COMPR:

P3	T3	H3	X03	TG6	TRA	TOA
6.805+001	3.214+001	8.655+001	9.913+001	1.139+002	7.000+001	4.700+001

COMPRESSOR ITERATION:

EI	ETAE	ETAC	ETAV	CFRPM	RMASS
2.801+000	7.611+001	9.462+001	8.558+001	3.547+003	4.969+002

I	T	P	H	X
1	3.745+001	7.577+001	8.534+001	9.717+001
2	3.682+001	7.489+001	8.529+001	9.719+001
3	3.214+001	6.805+001	8.655+001	9.918+001
4	6.807+001	6.795+001	9.308+001	1.000+000
5	7.178+001	6.508+001	9.380+001	1.000+000
6	1.851+002	2.530+002	1.079+002	1.000+000
7	1.783+002	2.316+002	1.072+002	1.000+000
8	1.597+002	2.306+002	1.035+002	1.000+000
9	1.529+002	2.280+002	1.023+002	1.000+000
10	1.521+002	2.230+002	1.021+002	1.000+000

INPUT DATA TO CONDHX:

T	P	TAIR	RH	RMASS
1.521+002	2.280+002	7.000+001	5.600+001	4.969+002
H2, H2PH2=	33.61	-.28		
H2, H2PH2=	33.54	.07		
H2, H2PH2=	33.99	-.45		
H2, H2PH2=	34.20	-.20		
H2, H2PH2=	33.85	.34		
H2, H2PH2=	33.98	-.13		
H2, H2PH2=	34.06	-.08		
H2, H2PH2=	34.11	-.05		
H2, H2PH2=	34.09	.02		

CONDENSER ITERATION:

T	P	H	X
1.521+002	2.290+002	1.021+002	1.000+000
8.202+001	2.204+002	3.409+001	.000

LIQUID LINE:

INPUT -	PIN = 2.204+002	TIN = 8.202+001	XIN = .000
OUTPUT -	POUT = 2.146+002	TOUT = 8.198+001	XOUT = .000

HIN = 3.409+001

EXPANSION DEVICE:

3828 INPUT - P12 =2.146+002 H12 =3.409+001 P13 =9.688+001
3829 OUTPUT - POUT =1.712+002 XMASS =5.341+002
3830
3831 INPUT DATA TO COMPR:
3832
3833 P3 T3 H3 XQ3 TG6 TPA TQA
3834 6.805+001 3.214+001 8.655+001 9.913+00 1.130+002 7.000+001 4.700+001
3835
3836 COMPRESSOR ITERATION:
3837
3838 EI ETAE ETAC ETAV CPRFH RMAS
3839 2.781+000 7.603+001 9.468+001 8.572+001 3.548+003 4.980+002
3840
3841 I T P H X
3842 1 3.748+001 7.581+001 8.536+001 9.719+001
3843 2 3.685+001 7.482+001 8.530+001 9.720+001
3844 3 3.214+001 6.805+001 8.635+001 9.913+001
3845 4 6.782+001 6.795+001 9.304+001 1.000+000
3846 5 7.148+001 6.507+001 9.375+001 1.000+000
3847 6 1.836+002 2.500+002 1.077+002 1.000+000
3848 7 1.768+002 2.282+002 1.070+002 1.000+000
3849 8 1.584+002 2.271+002 1.034+002 1.000+000
3850 9 1.516+002 2.246+002 1.021+002 1.000+000
3851 10 1.508+002 2.245+002 1.020+002 1.000+000
3852
3853 INPUT DATA TO CONDHY:
3854
3855 T P TAIR RH RMAS
3856 1.508+002 2.245+002 7.000+001 5.600+00 4.980+002
3857 H2,H2PH2= 34.37 -.67
3858 H2,H2PH2= 34.59 -.23
3859 H2,H2PH2= 35.06 -.47
3860 H2,H2PH2= 34.67 .39
3861 H2,H2PH2= 34.91 -.24
3862 H2,H2PH2= 35.03 -.12
3863 H2,H2PH2= 35.03 -.07
3864 H2,H2PH2= 35.08 .01
3865
3866 CONDENSER ITERATION:
3867
3868 T P H X
3869 1.508+002 2.245+002 1.020+002 1.000+000
3870 8.584+001 2.159+002 3.508+001 .000
3871
3872
3873 LIQUID LINE:
3874 INPUT - PIN = 2.159+002 TIN = 8.584+001 XIN = .000
3875 OUTPUT - POUT = 2.100+002 TOUT = 8.582+001 XOUT = .000
3876
3877
3878 EXPANSION DEVICE:
3879 INPUT - P12 =2.100+002 H12 =3.503+001 P13 =9.691+001
3880 OUTPUT - POUT =1.619+002 XMASS =4.880+002
3881
3882 INPUT DATA TO EVAPHX:
3883
3884 T P X TAIR RH RMAS
3885 3.627+001 9.691+001 2.230+001 4.700+001 7.300+001 4.977+002

3686 H2= 88.7753
3687 H2= 85.6629
3688 H2= 83.3127
3689 QT, QS= 2.412+004 2.161+004
3690
3691 EVAPORATOR ITERATION:
3692
3693 T P H X TSUP
3694 3.627+001 9.691+001 3.487+001 2.230+001
3695 3.737+001 7.639+001 8.333+001 9.483+001 .000
3696
3697 INPUT DATA TO EVAPHX:
3698
3699 T P X TAIR RH RMAS3
3700 3.613+001 9.668+001 2.235+001 4.700+001 7.300+001 4.977+002
3701 H2= 88.3993
3702 H2= 86.4500
3703 H2= 84.1672
3704 QT, QS= 2.453+004 2.190+004
3705
3706 EVAPORATOR ITERATION:
3707
3708 T P H X TSUP
3709 3.613+001 9.668+001 3.487+001 2.235+001
3710 3.718+001 7.581+001 8.415+001 9.582+001 .000
3711 P1, P1E, P1EP1, DENT2= 75.8052 75.8075 .0023 -1.2055
3712
3713 INPUT DATA TO COMPRESSOR:
3714
3715 P3 T3 H3 XQ3 T66 TRA TGA
3716 6.751+001 3.172+001 8.651+001 9.918+001 1.129+002 7.000+001 4.700+001
3717
3718 COMPRESSOR ITERATION:
3719
3720 EI ETAE ETAC ETAV CPRPM PMASS
3721 2.772+000 7.599+001 9.467+001 8.564+001 3.548+003 4.934+002
3722
3723 I T P H X
3724 1 3.704+001 7.519+001 8.531+001 9.719+001
3725 2 3.640+001 7.431+001 8.525+001 9.720+001
3726 3 3.172+001 6.751+001 8.651+001 9.918+001
3727 4 6.774+001 6.741+001 9.305+001 1.000+000
3728 5 7.144+001 6.456+001 9.377+001 1.000+000
3729 6 1.840+002 2.495+002 1.078+002 1.000+000
3730 7 1.772+002 2.281+002 1.071+002 1.000+000
3731 8 1.587+002 2.270+002 1.034+002 1.000+000
3732 9 1.519+002 2.245+002 1.022+002 1.000+000
3733 10 1.511+002 2.244+002 1.020+002 1.000+000
3734
3735 INPUT DATA TO CONDCHX:
3736
3737 T P TAIR RH RMAS3
3738 1.511+002 2.244+002 7.000+001 5.600+001 4.934+002
3739 H2, H2PH2= 33.93 -.09
3740 H2, H2PH2= 34.18 -.25
3741 H2, H2PH2= 34.73 -.55
3742 H2, H2PH2= 34.97 -.24
3743 H2, H2PH2= 35.02 -.05


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3944 H2,H2PH2= 35.03 -.01
3945
3946 CONDENSER ITERATION:
3947
3948 T P H X
3949 1.511+002 2.244+002 1.020+002 1.000+000
3950 8.565+001 2.162+002 3.503+001 .000
3951
3952
3953 LIQUID LINE:
3954 INPUT - PIN = 2.162+002 TIN = 8.565+001 XIN = .000
3955 OUTPUT - POUT = 2.104+002 TOUT = 8.561+001 XOUT = .000
3956
3957
3958 EXPANSION DEVICE:
3959 INPUT - P12 = 2.104+002 H12 = 3.503+001 P13 = 9.605+001
3960 OUTPUT - POUT = 1.630+002 XMASS = 4.915+002
3961
3962 INPUT DATA TO EVAPHX:
3963
3964 T P X TAIR RH RMASS
3965 3.530+001 9.606+001 2.277+001 4.700+001 7.300+001 4.934+002
3966 H2= 89.1784
3967 H2= 87.8766
3968 H2= 86.3823
3969 H2= 86.1836
3970 H2= 86.0991
3971 QT, QS= 2.520+004 2.172+004
3972
3973 EVAPORATOR ITERATION:
3974
3975 T P H X TSUP
3976 3.580+001 9.606+001 3.503+001 2.277+001
3977 3.674+001 7.449+001 8.610+001 9.813+001 .000
3978
3979 INPUT DATA TO EVAPHX:
3980
3981 T P X TAIR RH RMASS
3982 3.596+001 9.634+001 2.270+001 4.700+001 7.300+001 4.934+002
3983 H2= 89.0790
3984 H2= 87.4263
3985 H2= 85.4180
3986 H2= 85.2754
3987 H2= 85.1920
3988 QT, QS= 2.475+004 2.148+004
3989
3990 EVAPORATOR ITERATION:
3991
3992 T P H X TSUP
3993 3.596+001 9.634+001 3.503+001 2.270+001
3994 3.701+001 7.519+001 8.520+001 9.706+001 .000
3995 P1,P1E,P1EP1,DENT2= 75.1867 75.1882 .0014 -.1117
3996 INTERM,INTEC,POUT,P13= 0 0 163.04 96.34
3997 COMPOSITION OF REFRIG. IN ACCUMULATOR = .631
3998 (WEIGHT FRACTION OF MORE VOLATILE COMPONENT)
3999
4000 REFRIG. IN ACCUMULATOR = .216 LB
4001

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REFRIG. IN INDOOR COIL = 2.475 LB
 REFRIG. IN OUTDOOR COIL= .905 LB
 TMASS = 4.740+000 REFIN = 4.963+000

 SYSTEM WITH ACCUM., 13P1/152A 6-22-84*

TOA RHQA TRA RHRA
 4.700+001 7.500-001 7.000+001 5.600-001
 CFMIND CFMOUT
 1.118+003 2.284+003

RESULTS:

I	T	P	H	S	X
1	3.704+001	7.519+001	8.531+001	1.865-001	9.719-001
2	3.610+001	7.431+001	8.525+001	1.865-001	9.720-001
3	3.172+001	6.751+001	8.651+001	1.907-001	9.318-001
4	6.774+001	6.741+001	9.305+001	2.036-001	1.000+000
5	7.144+001	6.456+001	9.377+001	2.057-001	1.000+000
6	1.840+002	2.495+002	1.078+002	2.069-001	1.000+000
7	1.772+002	2.281+002	1.071+002	2.073-001	1.000+000
8	1.587+002	2.270+002	1.034+002	2.016-001	1.000+000
9	1.519+002	2.245+002	1.022+002	1.997-001	1.000+000
10	1.511+002	2.244+002	1.020+002	1.995-001	1.000+000
11	8.565+001	2.162+002	3.503+001	8.004-002	.000
12	8.561+001	2.104+002	3.503+001	8.006-002	.000
13	3.596+001	9.634+001	3.503+001	8.198-002	2.273-001

TG3	TSUP3	TG6	RMASS	TMASS
3.190+001	.000	1.129+002	4.931+002	4.742+000

QLOAD	ELUSE	COP
3.472+004	3.637+000	2.797+000

REFRIG. COMPOSITION = .668
 (WEIGHT FRACTION OF MORE VOLATILE COMPONENT)

INPUT DATA TO COMPR:

P3	T3	H3	XG3	TG5	TRA	TGA
6.686+001	3.120+001	8.641+001	3.912-001	1.132+002	7.000+001	4.700+001

COMPRESSOR ITERATION:

EI	ETAE	ETAC	ETAV	CPRPM	RMASS
2.774+000	7.592-001	9.462-001	8.545-001	3.543+003	4.874+002

I	T	P	H	X
1	3.647+001	7.442+001	8.519+001	9.712-001
2	3.584+001	7.356+001	8.513+001	9.712-001

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3 3.120+001 6.566+001 8.541+001 9.912-001
4 6.764+001 6.677+001 9.306+001 1.000+000
5 7.141+001 6.396+001 9.379+001 1.000+000
6 1.852+002 2.506+002 1.090+002 1.000+000
7 1.734+002 2.299+002 1.073+002 1.000+000
8 1.597+002 2.268+002 1.036+002 1.000+000
9 1.529+002 2.264+002 1.023+002 1.000+000
10 1.520+002 2.263+002 1.022+002 1.000+000

INPUT DATA TO CONDHX:

T	P	TAIR	RH	RMASS
1.520+002	2.263+002	7.000+001	5.600-001	4.874+002
H2, H2PH2=	33.58	- .34		
H2, H2PH2=	33.57	.01		
H2, H2PH2=	33.99	- .42		
H2, H2PH2=	34.18	- .19		
H2, H2PH2=	33.83	.35		
H2, H2PH2=	33.97	- .13		
H2, H2PH2=	34.05	- .08		
H2, H2PH2=	34.09	- .04		
H2, H2PH2=	34.08	.02		

CONDENSER ITERATION:

T	P	H	X
1.520+002	2.263+002	1.022+002	1.000+000
8.194+001	2.190+002	3.408+001	.000

LIQUID LINE:

INPUT -	PIN = 2.190+002	TIN = 8.194+001	XIN = .000
OUTPUT -	POUT = 2.134+002	TOUT = 8.191+001	XOUT = .000

HIN = 3.408+001

EXPANSION DEVICE:

INPUT -	P12 = 2.134+002	H12 = 3.408+001	P13 = 9.557+001
OUTPUT -	POUT = 1.699+002	XMASS = 5.295+002	

INPUT DATA TO COMPRESSOR:

P3	T3	H3	XQ3	TG6	TRA	TGA
6.686+001	3.120+001	8.641+001	9.912-001	1.124+002	7.000+001	4.700+001

COMPRESSOR ITERATION:

E1	ETAE	ETAC	ETAV	CPRFM	RMASS
2.754+000	7.591-001	9.467-001	8.558-001	3.548+003	4.081+002

I	T	P	H	X
1	3.640+001	7.445+001	8.521+001	9.713-001
2	3.586+001	7.358+001	8.515+001	9.714-001
3	3.120+001	6.585+001	8.541+001	9.912-001
4	6.740+001	6.677+001	9.302+001	1.000+000
5	7.113+001	6.395+001	9.374+001	1.000+000
6	1.838+002	2.480+002	1.078+002	1.000+000
7	1.770+002	2.269+002	1.071+002	1.000+000
8	1.585+002	2.258+002	1.035+002	1.000+000

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4118 9 1.517+002 2.233+002 1.022+002 1.000+000
4119 10 1.509+002 2.233+002 1.020+002 1.000+000
4120
4121 INPUT DATA TO CONDHX:
4122
4123 T P TAIR RH RMASS
4124 1.509+002 2.233+002 7.000+001 5.600-001 4.804+002
4125 H2,H2PH2= 34.24 -.62
4126 H2,H2PH2= 34.41 -.18
4127 H2,H2PH2= 34.89 -.47
4128 H2,H2PH2= 34.47 .42
4129 H2,H2PH2= 34.70 -.23
4130 H2,H2PH2= 34.82 -.12
4131 H2,H2PH2= 34.88 -.07
4132 H2,H2PH2= 34.87 .01
4133
4134 CONDENSER ITERATION:
4135
4136 T P H X
4137 1.509+002 2.233+002 1.020+002 1.000+000
4138 8.503+001 2.150+002 3.437+001 .000
4139
4140 LIQUID LINE:
4141 INPUT - PIN = 2.150+002 TIN = 8.503+001 XIN = .000
4142 OUTPUT - POUT= 2.093+002 TOUT= 8.501+001 XOUT= .000
4143
4144 HIN = 3.487+001
4145
4146 EXPANSION DEVICE:
4147 INPUT - P12 =2.093+002 H12 =3.487+001 P13 =9.560+001
4148 OUTPUT - POUT =1.624+002 XMASS =4.910+002
4149
4150 INPUT DATA TO EVAPHX:
4151
4152 T P X TAIR RH RMASS
4153 3.549+001 9.560+001 2.270-001 4.700+001 7.300-001 4.884+002
4154 H2= 89.2513
4155 H2= 88.3391
4156 H2= 87.3552
4157 H2= 86.9869
4158 QT,QS= 2.542+004 2.207+004
4159
4160 EVAPORATOR ITERATION:
4161
4162 T P H X TSUP
4163 3.549+001 9.560+001 3.493+001 2.270-001
4164 3.655+001 7.393+001 8.690+001 9.914-001 .000
4165
4166 INPUT DATA TO EVAPHX:
4167
4168 T P X TAIR RH RMASS
4169 3.562+001 9.581+001 2.265-001 4.700+001 7.300-001 4.884+002
4170 H2= 89.1969
4171 H2= 88.0232
4172 H2= 86.7563
4173 H2= 86.4446
4174 QT,QS= 2.514+004 2.190+004
4175

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4176 EVAPORATOR ITERATION:

4177 T P H X TSUP

4178 3.562+001 9.581+001 3.493+001 2.265-001

4179 3.678+001 7.445+001 8.640+001 9.846-001 .000

4180 P1,PIE,P1EP1,DENT2= 74.4542 74.4459 -.0083 1.1886

4181

4182 INPUT DATA TO COMP:

4183 P3 T3 H3 XQ3 TG6 TRA TGA

4184 6.739+001 3.161+001 8.645+001 9.912-001 1.127+002 7.000+001 4.700+001

4185

4186 COMPRESSOR ITERATION:

4187 EI ETAE ETAC ETAV ETPRM RMASS

4188 2.766+000 7.596-001 9.467-001 8.564-001 3.548+003 4.927+002

4189

4190 I T P H X

4191 1 3.693+001 7.505+001 8.525+001 9.713-001

4192 2 3.630+001 7.418+001 8.519+001 9.714-001

4193 3 3.161+001 6.739+001 8.645+001 9.912-001

4194 4 6.751+001 6.729+001 9.302+001 1.000-000

4195 5 7.121+001 6.445+001 9.373+001 1.000+000

4196 6 1.837+002 2.489+002 1.077+002 1.000-000

4197 7 1.769+002 2.275+002 1.070+002 1.000+000

4198 8 1.584+002 2.265+002 1.034+002 1.000+000

4199 9 1.516+002 2.240+002 1.022+002 1.000+000

4200 10 1.508+002 2.239+002 1.020+002 1.000+000

4201

4202 INPUT DATA TO CONDHX:

4203 T P TAIR RH RMASS

4204 1.508+002 2.239+002 7.000+001 5.600-001 4.927+002

4205 H2,H2PH2= 34.12 -.08

4206 H2,H2PH2= 34.38 -.25

4207 H2,H2PH2= 34.90 -.52

4208 H2,H2PH2= 34.55 .35

4209 H2,H2PH2= 34.76 -.21

4210 H2,H2PH2= 34.87 -.11

4211 H2,H2PH2= 34.94 -.07

4212 H2,H2PH2= 34.93 .01

4213

4214 CONDENSER ITERATION:

4215 T P H X

4216 1.508+002 2.239+002 1.020+002 1.000+000

4217 8.525+001 2.155+002 3.493+001 .000

4218

4219 LIQUID LINE:

4220 INPUT - PIN = 2.155+002 TIN = 8.525+001 XIN = .000

4221 OUTPUT - POUT = 2.097+002 TOUT = 8.523+001 XOUT = .000

4222

4223

4224

4225

4226

4227

4228

4229

4230 EXPANSION DEVICE:

4231 INPUT - P12 = 2.097+002 H12 = 3.493+001 P13 = 9.641+001

4232 OUTPUT - POUT = 1.626+002 XMASS = 4.911+002

4233

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4234 INPUT DATA TO EVAPHX:
4235
4236 T P X TAIR RH RMAS
4237 3.598+001 9.641+001 2.252-001 4.700+001 7.300-001 4.927+002
4238 H2= 89.0122
4239 H2= 87.0705
4240 H2= 84.9381
4241 H2= 84.8139
4242 H2= 84.7435
4243 QT, QS= 2.455+004 2.138+004
4244
4245 EVAPORATOR ITERATION:
4246
4247 T P H X TSUP
4248 3.598+001 9.641+001 3.493+001 2.252-001
4249 3.715+001 7.554+001 8.475+001 9.653-001 .000
4250
4251 INPUT DATA TO EVAPHX:
4252
4253 T P X TAIR RH RMAS
4254 3.587+001 9.622+001 2.256-001 4.700+001 7.300-001 4.927+002
4255 H2= 89.0816
4256 H2= 87.5769
4257 H2= 85.6249
4258 H2= 85.4721
4259 H2= 85.3758
4260 QT, QS= 2.486+004 2.154+004
4261
4262 EVAPORATOR ITERATION:
4263
4264 T P H X TSUP
4265 3.587+001 9.622+001 3.493+001 2.256-001
4266 3.697+001 7.506+001 8.538+001 9.728-001 .000
4267 P1, P1E, P1EP1, DENT2= 75.0545 75.0616 .0071 .1320
4268 INTERM, INTERE, POUT, P13= 0 0 162.61 96.22
4269 COMPOSITION OF REFRIG. IN ACCUMULATOR = .508
4270 (WEIGHT FRACTION OF MORE VOLATILE COMPONENT)
4271
4272 REFRIG. IN ACCUMULATOR = .368 LB
4273
4274 REFRIG. IN INDOOR COIL = 2.407 LB
4275
4276 REFRIG. IN OUTDOOR COIL = .903 LB
4277
4278 TMASS = 4.822+000 REFIN = 4.968+000
4279
4280 *****
4281 **SYST4 WITH ACCUM , 13B1/152A 6-22-84***
4282
4283 TOA RHOA TRA RHRA
4284 4.700+001 7.300-001 7.000+001 5.500-001
4285 CFMIND CFMOUT
4286 1.118+003 2.284+003
4287
4288
4289
4290
4291 RESULTS:

```

4292	I	T	P	H	S	X
4293	1	3.693+001	7.505+001	8.525+001	1.864-001	9.713-001
4294	2	3.630+001	7.418+001	8.519+001	1.865-001	9.714-001
4295	3	3.161+001	6.739+001	8.645+001	1.906-001	9.912-001
4296	4	6.751+001	6.729+001	9.302+001	2.036-001	1.000+000
4297	5	7.121+001	6.445+001	9.373+001	2.057-001	1.000+000
4298	6	1.837+002	2.489+002	1.077+002	2.069-001	1.000+000
4299	7	1.769+002	2.275+002	1.070+002	2.072-001	1.000+000
4300	8	1.584+002	2.265+002	1.034+002	2.016-001	1.000+000
4301	9	1.516+002	2.240+002	1.022+002	1.997-001	1.000+000
4302	10	1.508+002	2.239+002	1.020+002	1.994-001	1.000+000
4303	11	8.525+001	2.155+002	3.493+001	7.985-002	.000
4304	12	8.523+001	2.097+002	3.493+001	7.988-002	.000
4305	13	3.587+001	9.622+001	3.493+001	8.179-002	2.256-001
4306	TG3	TSUP3	TG6	RM3SS	TM3SS	
4307	3.181+001	.000	1.127+002	4.927+002	4.822+000	
4308	QLQAD	ELUSE	COP			
4309	3.470+004	3.631+000	2.800+000			
4310	REFRIG. COMPOSITION = .668					
4311	(WEIGHT FRACTION OF MORE VOLATILE COMPONENT)					
4312						
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INPUT DATA TO COMP:

P3	T3	H3	XG3	TG6	TRA	TGA
6.717+001	3.144+001	8.641+001	9.309-001	1.124+002	7.000+001	4.700+001

COMPRESSOR ITERATION:

EI	ETAE	ETAC	ETAV	CPRPM	RMASS
2.758+000	7.593-001	9.468-001	8.564-001	3.548+003	4.912+002

INPUT DATA TO CONDHX:

T	P	TAIR	RH	RMASS
1.505+002	2.230+002	7.000+001	5.600-001	4.912+002
H2,H2PH2=	34.11	-.06		
H2,H2PH2=	34.96	-.85		
H2,H2PH2=	34.75	.21		
H2,H2PH2=	35.08	-.34		
H2,H2PH2=	35.24	-.15		

```

4350 H2,H2PH2= 35.31 -.08
4351 H2,H2PH2= 35.31 .00
4352
4353 CONDENSER ITERATION:
4354
4355 T P H X
4356 1.505+002 2.230+002 1.020+002 1.000+000
4357 8.670+001 2.144+002 3.531+001 .000
4358
4359
4360 LIQUID LINE:
4361 INPUT - PIN = 2.144+002 TIN = 8.670+001 XIN = .000
4362 OUTPUT - POUT= 2.086+002 TOUT= 8.607+001 XOUT= 3.445-003
4363
4364
4365 EXPANSION DEVICE:
4366 INPUT - P12 =2.086+002 H12 =3.531+001 P13 =9.598+001
4367 OUTPUT - POUT =1.600+002 XMASS =4.785+002
4368
4369 INPUT DATA TO COMPR:
4370
4371 P3 T3 H3 XQ3 TG6 TRA TGA
4372 6.717+001 3.144+001 8.641+001 9.909-001 1.127+002 7.000+001 4.700+001
4373
4374 COMPRESSOR ITERATION:
4375
4376 EI ETAE ETAC ETAV ETAV X
4377 2.763+000 7.595-001 9.466-001 8.560-00 3.548+003 4.909+002
4378
4379 I T P H X
4380 1 3.674+001 7.481+001 8.521+001 9.710-001
4381 2 3.611+001 7.393+001 8.515+001 9.711-001
4382 3 3.144+001 6.717+001 8.641+001 9.909-001
4383 4 6.740+001 6.708+001 9.301+001 1.000+000
4384 5 7.111+001 6.424+001 9.372+001 1.000+000
4385 6 1.838+002 2.408+002 1.077+002 1.000+000
4386 7 1.770+002 2.276+002 1.070+002 1.000+000
4387 8 1.585+002 2.265+002 1.034+002 1.000+000
4388 9 1.517+002 2.240+002 1.022+002 1.000+000
4389 10 1.509+002 2.240+002 1.020+002 1.000+000
4390
4391 INPUT DATA TO CONDHX:
4392
4393 T P TAIR TARR RH RMASS
4394 1.509+002 2.240+002 7.000+001 5.600-001 4.909+002
4395 H2,H2PH2= 33.95 -.08
4396 H2,H2PH2= 34.19 -.24
4397 H2,H2PH2= 34.74 -.55
4398 H2,H2PH2= 34.98 -.24
4399 H2,H2PH2= 35.03 -.05
4400 H2,H2PH2= 35.04 -.01
4401
4402 CONDENSER ITERATION:
4403
4404 T P H X
4405 1.509+002 2.240+002 1.020+002 1.000+000
4406 8.569+001 2.158+002 3.504+001 .000
4407

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LIQUID LINE:
 INPUT - PIN = 2.158+002 TIN = 8.563+001 XIN = .000 HIN = 3.504+001
 OUTPUT - POUT = 2.100+002 TOUT = 8.565+001 XOUT = .000

EXPANSION DEVICE:
 INPUT - P12 = 2.100+002 H12 = 3.504+001 P13 = 9.597+001
 DOENFA DOES NOT CONVERGE, DIFFX2 = -9.38367-006
 OUTPUT - POUT = 1.626+002 XMASS = 4.900+002

INPUT DATA TO EVAPHX:
 T P X TAIR RH RMASS
 3.574+001 9.597+001 2.281-001 4.700+001 7.300-001 4.909+002
 H2 = 89.1871
 H2 = 87.9454
 H2 = 86.5718
 H2 = 86.3075
 H2 = 86.2429
 QT, QS = 2.513+004 2.169+004

EVAPORATOR ITERATION:
 T P H X TSUP
 3.574+001 9.597+001 3.504+001 2.281-001
 3.677+001 7.448+001 8.624+001 9.828-001 .000

INPUT DATA TO EVAPHX:
 T P X TAIR RH RMASS
 3.582+001 9.610+001 2.270-001 4.700+001 7.300-001 4.909+002
 H2 = 89.1628
 H2 = 87.7873
 H2 = 86.1445
 H2 = 85.9260
 H2 = 85.8617
 QT, QS = 2.494+004 2.158+004

EVAPORATOR ITERATION:
 T P H X TSUP
 3.582+001 9.610+001 3.504+001 2.278-001
 3.690+001 7.480+001 8.584+001 9.781-001 .000
 P1, P1E, P1EP1, DENT2 = 74.8051 74.8045 -.0006 .6326

INPUT DATA TO COMPR:
 P3 T3 H3 XQ3 TG6 TRA TGA
 6.745+001 3.166+001 8.643+001 9.909-001 1.129+002 7.000+001 4.700+001

COMPRESSOR ITERATION:
 EI ETAE ETAC ETAV CPRPM RMASS
 2.770+000 7.598-001 9.466-001 8.563-001 3.548+003 4.932+002

I T P H X
 1 3.697+001 7.512+001 8.523+001 9.710-001

4466 2 3.634+001 7.425+001 8.517+001 9.711-001
4467 3 3.166+001 6.745+001 8.643+001 9.909-001
4468 4 6.746+001 6.736+001 9.300+001 1.000+000
4469 5 7.116+001 6.451+001 9.372+001 1.000+000
4470 6 1.837+002 2.494+002 1.077+002 1.000+000
4471 7 1.769+002 2.280+002 1.070+002 1.000+000
4472 8 1.584+002 2.270+002 1.034+002 1.000+000
4473 9 1.517+002 2.245+002 1.021+002 1.000+000
4474 10 1.509+002 2.244+002 1.020+002 1.000+000
4475
4476 INPUT DATA TO CONDHX:
4477
4478 T P TAIR RH RMASS
4479 1.509+002 2.244+002 7.000+001 5.600-001 4.932+002
4480 H2, H2PH2= 33.91 -.09
4481 H2, H2PH2= 34.16 -.25
4482 H2, H2PH2= 34.71 -.56
4483 H2, H2PH2= 34.95 -.24
4484 H2, H2PH2= 35.00 -.05
4485 H2, H2PH2= 35.01 -.01
4486
4487 CONDENSER ITERATION:
4488
4489 T P H X
4490 1.509+002 2.244+002 1.020+002 1.000+000
4491 8.559+001 2.162+002 3.501+001 .000
4492
4493 LIQUID LINE:
4494 INPUT - PIN = 2.162+002 TIN = 8.559+001 XIN = .000
4495 OUTPUT - POUT = 2.104+002 TOUT = 8.555+001 XOUT = .000
4496
4497 EXPANSION DEVICE:
4498 INPUT - P12 = 2.104+002 H12 = 3.501+001 P13 = 9.642+001
4499 OUTPUT - POUT = 1.632+002 XMASS = 4.919+002
4500
4501 INPUT DATA TO EVAPHX:
4502
4503 T P X TAIR RH RMASS
4504 3.601+001 9.642+001 2.266-001 4.700+001 7.300-001 4.932+002
4505 H2= 89.0363
4506 H2= 87.1973
4507 H2= 85.0951
4508 H2= 84.9684
4509 H2= 84.8924
4510 QT, QS= 2.461+004 2.141+004
4511
4512 EVAPORATOR ITERATION:
4513
4514 T P H X TSUP
4515 3.601+001 9.642+001 3.501+001 2.266-001
4516 3.710+001 7.542+001 8.490+001 9.670-001 .000
4517
4518 INPUT DATA TO EVAPHX:
4519
4520 T P X TAIR RH RMASS
4521 3.594+001 9.630+001 2.268-001 4.700+001 7.300-001 4.932+002
4522
4523

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4524 H2= 89.0900
4525 H2= 87.5004
4526 H2= 85.5169
4527 H2= 85.3627
4528 H2= 85.2751
4529 QT, QS= 2.479+004 2.150+004
4530
4531 EVAPORATOR ITERATION:
4532
4533 T P H X TSUP
4534 3.594+001 9.630+001 3.501+001 2.268-00'
4535 3.699+001 7.513+001 8.528+001 9.716-00' .000
4536 P1,PIE,PIEP,DENT2= 75.1244 75.1281 .0037 .0523
4537 INTERM,INTERE,POUT,P13= 0 0 163.24 96.30
4538 COMPOSITION OF REFRIG. IN ACCUMULATOR = .445
4539 (WEIGHT FRACTION OF MORE VOLATILE COMPONENT)
4540
4541 REFRIG. IN ACCUMULATOR = .571 LB
4542
4543 REFRIG. IN INDOOR COIL = 2.477 LB
4544
4545 REFRIG. IN OUTDOOR COIL= .904 LB
4546
4547 TMASS = 5.097+000 REFIN = 4.968+000
4548
4549 *****
4550 **SYST4 WITH ACCUM., 13B1/152A 6-22-84***
4551
4552 TGA RHGA TRA RHRA
4553 4.700+001 7.300-001 7.000+001 5.600-001
4554
4555 CFMIND CFMOUT
4556 1.118+003 2.284+003
4557
4558
4559
4560
4561 RESULTS:
4562 I T P H S X
4563 1 3.697+001 7.512+001 8.523+001 1.863-001 9.710-001
4564 2 3.634+001 7.425+001 8.517+001 1.864-001 9.711-001
4565 3 3.166+001 6.745+001 8.643+001 1.906-001 9.909-001
4566 4 6.746+001 6.735+001 9.300+001 2.036-001 1.000+000
4567 5 7.116+001 6.451+001 9.372+001 2.057-001 1.000+000
4568 6 1.837+002 2.494+002 1.077+002 2.068-001 1.000+000
4569 7 1.769+002 2.280+002 1.070+002 2.072-001 1.000+000
4570 8 1.584+002 2.270+002 1.034+002 2.015-001 1.000+000
4571 9 1.517+002 2.245+002 1.021+002 1.996-001 1.000+000
4572 10 1.509+002 2.244+002 1.020+002 1.994-001 1.000+000
4573 11 8.559+001 2.162+002 3.501+001 8.001-002 .000
4574 12 8.555+001 2.104+002 3.501+001 8.003-002 .000
4575 13 3.594+001 9.630+001 3.501+001 8.195-002 2.268-001
4576
4577 TG3 TSUP3 TG6 RMASS TMASS
4578 3.186+001 .000 1.129+002 4.932+002 5.097+000
4579
4580 QLOAD ELUSE COP
4581 3.469+004 3.635+000 2.796+000

```

REFRIG. COMPOSITION = .668
(WEIGHT FRACTION OF MORE VOLATILE COMPONENT)

INPUT DATA TO COMPR:

P3 T3 H3 XG3 TG6 TRA TGA
6.742+001 3.164+001 8.644+001 9.911-001 1.128+002 7.000+001 4.700+001

COMPRESSOR ITERATION:

E1 ETAE ETAC ETAV CPRPM RMASS
2.769+000 7.597-001 9.467-001 8.564-001 3.548+003 4.930+002

I	T	P	H	X
1	3.695+001	7.509+001	8.524+001	9.711-001
2	3.632+001	7.421+001	8.518+001	9.712-001
3	3.164+001	6.742+001	8.644+001	9.911-001
4	6.749+001	6.733+001	9.501+001	1.000+000
5	7.119+001	6.448+001	9.373+001	1.000+000
6	1.837+002	2.492+002	1.077+002	1.000+000
7	1.769+002	2.278+002	1.070+002	1.000+000
8	1.584+002	2.267+002	1.034+002	1.000+000
9	1.516+002	2.242+002	1.022+002	1.000+000
10	1.508+002	2.242+002	1.020+002	1.000+000

INPUT DATA TO CONDHX:

T	P	TAIR	RH	RMASS
1.508+002	2.242+002	7.000+001	5.600-00*	4.930+002
H2, H2PH2=	34.01	-.09		
H2, H2PH2=	34.26	-.25		
H2, H2PH2=	34.80	-.54		
H2, H2PH2=	35.04	-.24		
H2, H2PH2=	35.10	-.05		
H2, H2PH2=	35.11	-.01		

CONDENSER ITERATION:

T	P	H	X
1.508+002	2.242+002	1.020+002	1.000+000
8.593+001	2.159+002	3.511+001	.000

LIQUID LINE:

INPUT - PIN = 2.159+002 TIN = 8.593+001 XIN = .000
OUTPUT - POUT = 2.101+002 TOUT = 8.591+001 XOUT = .000

EXPANSION DEVICE:

INPUT - P12 = 2.101+002 H12 = 3.511+001 P13 = 9.627+001
OUTPUT - POUT = 1.618+002 XMASS = 4.878+002

INPUT DATA TO COMPR:

P3 T3 H3 XG3 TG6 TRA TGA
HIN = 3.511+001

4640 6.742+001 3.164+001 8.644+001 9.911-001 1.129+002 7.000+001 4.700+001
4641
4642 COMPRESSOR ITERATION:
4643
4644 E1 ETAE ETAC ETAV CPRPM RMASS
4645 2.771+000 7.598-001 9.466-001 8.562-001 3.548+003 4.923+002
4646
4647 I T P H X
4648 1 3.695+001 7.509+001 8.524+001 9.711-001
4649 2 3.632+001 7.421+001 8.518+001 9.712-001
4650 3 3.164+001 6.742+001 8.644+001 9.911-001
4651 4 6.752+001 6.733+001 9.302+001 1.000+000
4652 5 7.122+001 6.448+001 9.373+001 1.000+000
4653 6 1.839+002 2.495+002 1.077+002 1.000+000
4654 7 1.771+002 2.282+002 1.070+002 1.000+000
4655 8 1.586+002 2.271+002 1.034+002 1.000+000
4656 9 1.518+002 2.246+002 1.022+002 1.000+000
4657 10 1.510+002 2.246+002 1.020+002 1.000+000
4658
4659 INPUT DATA TO CONDHX:
4660
4661 T P TAIR RH RMASS
4662 1.510+002 2.246+002 7.000+001 5.600-001 4.928+002
4663 H2, H2PH2= 33.81 -.10
4664 H2, H2PH2= 34.57 -.76
4665 H2, H2PH2= 34.46 .11
4666 H2, H2PH2= 34.77 -.32
4667 H2, H2PH2= 34.87 -.10
4668 H2, H2PH2= 34.93 -.06
4669 H2, H2PH2= 34.93 .00
4670
4671 CONDENSER ITERATION:
4672
4673 T P H X
4674 1.510+002 2.246+002 1.020+002 1.000+000
4675 8.525+001 2.164+002 3.493+001 .000
4676
4677
4678 LIQUID LINE:
4679 INPUT - PIN = 2.164+002 TIN = 8.525+001 XIN = .000
4680 OUTPUT - POUT = 2.106+002 TOUT = 8.522+001 XOUT = .000
4681
4682
4683 EXPANSION DEVICE:
4684 INPUT - P12 = 2.106+002 H12 = 3.493+001 P13 = 9.626+001
4685 OUTPUT - POUT = 1.637+002 XMASS = 4.946+002
4686
4687 INPUT DATA TO EVAPHX:
4688
4689 T P X TAIR RH RMASS
4690 3.589+001 9.626+001 2.254-001 4.700+001 7.300-001 4.928+002
4691 H2= 89.0827
4692 H2= 87.4651
4693 H2= 85.4580
4694 H2= 85.3255
4695 H2= 85.2310
4696 QT, QS= 2.480+004 2.151+004
4697

HIN = 3.493+001

EVAPORATOR ITERATION:

```

T      P      H      X      TSUP
3.589+001  9.626+001  3.493+001  2.254-001
3.700+001  7.517+001  8.524+001  9.711-001      .000
PI,PIE,PIEP1,DENT2= 75.0878 75.1672      .0795
INTERM,INTERE,POUT,P13= 0 0 163.71 96.26
COMPOSITION OF REFRIG. IN ACCUMULATOR = .470
(WEIGHT FRACTION OF MORE VOLATILE COMPONENT)

```

REFRIG. IN ACCUMULATOR = .469 LB

REFRIG. IN INDOOR COIL = 2.490 LB

REFRIG. IN OUTDOOR COIL = .905 LB

TMASS = 5.009+000 REFIN = 4.968+000

 SYST4 WITH ACCUM., 13B1/152A 6-22-84

```

TOA      RHGA      TRA      RHRA
4.700+001  7.300-001  7.000+001  5.600-001

```

```

CFMIND      CFMOUT
1.118+003  2.284+003

```

RESULTS:

I	T	P	H	S	X
1	3.695+001	7.509+001	8.524+001	1.863-001	9.711-001
2	3.632+001	7.421+001	8.518+001	1.864-001	9.712-001
3	3.164+001	6.742+001	8.644+001	1.906-001	9.911-001
4	6.752+001	6.733+001	9.302+001	2.036-001	1.000+000
5	7.122+001	6.448+001	9.373+001	2.057-001	1.000+000
6	1.839+002	2.495+002	1.077+002	2.069-001	1.000+000
7	1.771+002	2.282+002	1.070+002	2.072-001	1.000+000
8	1.586+002	2.271+002	1.034+002	2.015-001	1.000+000
9	1.518+002	2.246+002	1.022+002	1.997-001	1.000+000
10	1.510+002	2.246+002	1.020+002	1.994-001	1.000+000
11	8.525+001	2.164+002	3.493+001	7.985-002	.000
12	8.522+001	2.106+002	3.493+001	7.987-002	.000
13	3.589+001	9.626+001	3.493+001	8.178-002	2.254-001

```

TG3      TSUP3      TG6      RMASS      TMASS
3.183+001      .000      1.129+002  4.928+002  5.009+000

```

```

QLOAD      ELUSE      COP
3.472+004  3.636+000  2.798+000

```

REFRIG. COMPOSITION = .668
 (WEIGHT FRACTION OF MORE VOLATILE COMPONENT)

FOR THE AMOUNT OF REFRIGERANT STORED IN THE ACCUMULATOR
CIRCULATING COMPOSITION SHOULD BE .669

I=999 FOR DISCONTINUATION
999

4756
4757
4758
4759
4760
4761
4762
END PRT

EOFF

APPENDIX J. LISTING OF THE PROGRAM, HPBI

The following is a complete listing of the program HPBI. All source elements of the program are listed in alphabetical order. For fast reference, the reader can review Tables H1, H2 and H3, which contain name lists of all functions and subroutines with a short statement of their purpose.


```

***** AIRHT *****
@ELT, L DD.AIRHT
ELT 0R1 S7401C 07/21/94 15:54:51 (0)
  10. 00 FUNCTION AIRHT(AMASS,ACP,AMU,AK,DO,DT,NROW,WIDTH,
  20. 00 & RPCH,FPCH,FTK,CONST,CPOW,ANGLE)
  30. 00 C
  40. 00 C**** PURPOSE:
  50. 00 C TO CALCULATE AIR-SIDE FINNED-TUBE
  60. 00 C HEAT TRANSFER COEFFICIENT
  70. 00 C
  80. 00 C**** INPUT DATA:
  90. 00 C ACP - AIR SPEC. HEAT AT CONSTANT PRESSURE (BTU/LBM*F)
 100. 00 C AK - AIR THERMAL CONDUCTIVITY (BTU/H*FT*F)
 110. 00 C AMASS - AIR MASS FLOW RATE (LBM/H)
 120. 00 C AMU - AIR DYNAMIC VISCOSITY (LBM/FT*H)
 130. 00 C ANGLE - ANGLE BETWEEN COIL FACE & AIR STREAMLINES (RAD)
 140. 00 C CONST - COEFFICIENT FOR NUSSELT NUMBER (-)
 150. 00 C CPOW - POWER FOR REYNOLDS NUMBER (-)
 160. 00 C DO - TUBE OUTSIDE DIAMETER (FT)
 170. 00 C DT - FIN TIP DIAMETER (FT)
 180. 00 C FPCH - FIN PITCH (FT)
 190. 00 C FTK - FIN THICKNESS (FT)
 200. 00 C NROW - NUMBER OF TUBES PER ROW (-)
 210. 00 C RPCH - TUBE ROW PITCH (FT)
 220. 00 C WIDTH - HEAT EXCHANGER WIDTH (FT)
 230. 00 C
 240. 00 C**** OUTPUT DATA:
 250. 00 C AIRHT - AIR-SIDE HEAT TRANSFER COEFFICIENT (BTU/H*F*FT**2)
 260. 00 C
 270. 00 C APR=AMU*ACP/AK
 280. 00 C FS=FPCH-FTK
 290. 00 C FL=0.5*(DT-DO)
 300. 00 C HGT=NROW*RPCH
 310. 00 C TA=DO*WIDTH*NROW
 320. 00 C FN=WIDTH/FPCH
 330. 00 C FA=(RPCH-DO)*FTK*FN
 340. 00 C FA=NROW*FA
 350. 00 C EA=(WIDTH*HGT-TA-FA)*SIN(ANGLE)
 360. 00 C G=AMASS/EA
 370. 00 C RE=O*DO/AMU
 380. 00 C ANU=CONST*RE*CPOW*APR**0.333*(FS/FL)**0.2*(FS/FTK)**0.1134
 390. 00 C AIRHT=ANU*AK/DO
 400. 00 C RETURN
 410. 00 C END

```

END ELT. ERRORS: NONE. TIME: 0.098 SEC. IMAGE COUNT: 41

@HDS,P ***** AIRPR ***** .L,0

***** AIRPR *****

```

@ELT, L DD, AIRPR
ELT 8R1 S7401C 07/21/84 15:54:51 (0)
10. 00 SUBROUTINE AIRPR(I, T, PATH, RH, W, CP, R, AM, AK)
20. 00 C
30. 00 C**** PURPOSE:
40. 00 C TO CALCULATE AIR PROPERTIES
50. 00 C
60. 00 C**** INPUT DATA:
70. 00 C I = 1 IF RELATIVE HUMIDITY IS GIVEN (-)
80. 00 C = 2 IF HUMIDITY RATIO IS GIVEN (-)
90. 00 C T - AIR TEMPERATURE (-)
100. 00 C PATH - AIR PRESSURE (PSIA)
110. 00 C RH - RELATIVE HUMIDITY IN FRACTION (IF I=1) (-)
120. 00 C W - HUMIDITY RATIO (IF I=2) (LBM H2O/LBM DRY AIR)
130. 00 C
140. 00 C**** OUTPUT DATA:
150. 00 C AK - AIR THERMAL CONDUCTIVITY (BTU/H*F*FT)
160. 00 C AM - AIR DYNAMIC VISCOSITY (LBM/FT*H)
170. 00 C CP - AIR SPEC. HEAT AT CONSTANT PRESSURE (BTU/LBM*F)
180. 00 C R - GAS CONSTANT OF AIR (LBF*FT/LBM*R)
190. 00 C RH - RELATIVE HUMIDITY IN FRACTION (FOR I=2) (-)
200. 00 C W - HUMIDITY RATIO (FOR I=1) (LBM H2O/LBM DRY AIR)
210. 00 C
220. 00 C
230. 00 C
240. 00 IF (I.EQ.1)GOTO100
250. 00 P=W*PATH/(C.622+W)
260. 00 100 CONTINUE
270. 00 TR=T+460.
280. 00 Z=1000./TR
290. 00 IF (TR.GE.492.)GOTO110
300. 00 PSAT=EXP(0.03940*Z**3-0.2755*Z*Z-10.431*Z+19.509)
310. 00 GOTO30
320. 00 10 IF (TR.GE.672.)GOTO20
330. 00 PSAT=EXP(0.17829*Z*Z*Z-1.6893*Z*Z-5.0988*Z+13.4353)
340. 00 GOTO30
350. 00 20 PSAT=EXP(0.71692*Z**4-4.01503*Z**3+7.5368*Z*Z-14.2131*Z+16.8255)
360. 00 30 IF (I.EQ.1)GOTO110
370. 00 RH=P/PSAT
380. 00 110 CONTINUE
390. 00 IF (RH.GE.0.00001)GOTO40
400. 00 W=0.
410. 00 GOTO50
420. 00 40 PW=RH*PSAT
430. 00 W=0.622*PW/(PATM-PW)
440. 00 50 CONTINUE
450. 00 CP=0.2478786-0.4204563E-04*TR+0.5767657F-07*TR**2
460. 00 & -0.1493056E-10*TR**3
470. 00 CP=(CP+0.414*W)/(1.+W)
480. 00 R=(53.34+85.76*W)/(1.+W)
490. 00 AM=5.5029E-03+8.7157E-05*TR-2.9464E-08*TR**2
500. 00 & +6.250E-12*TR**3
510. 00 AK=-2.853E-04+2.268E-05*TR-8.253E-09*TR*TR
520. 00 & +1.239E-12*TR**3
530. 00 RETURN
END

```

END ELT. ERRORS: NONE. TIME: 0.093 SEC. IMAGE COUNT: 53

```

***** BCONST *****
@ELT,L DD,BCONST
ELT 8R1 S7401C 07/21/84 15:54.52 (0)
10. 00 SUBROUTINE BCONST
20. 00 C
30. 00 C
40. 00 C**** PURPOSE:
50. 00 C      TO READ CONSTANTS FOR PURE COMPONENTS OF MIXTURE.
60. 00 C
70. 00
80. 00 COMMON/RDATA1/A3,A4,A5,A6,A7,A8,B3,B4,B5,
90. 00 & B6,B7,B8,F0,F1
100. 00 COMMON/FDATA2/WMOL1,WMOL2,TC1,TC2
110. 00 COMMON/ESDATA/AA(2,3),BB(2,3),CC(2,3)
120. 00 C
130. 00 C**** READ EQ. OF STATE CONSTANTS
140. 00 C      MORE VOLATILE COMPONENT
150. 00 C      A0,A1,A2,B0,B1,B2
160. 00 C      READ(7,777)A3,A4,A5
170. 00 C      READ(7,777)B3,B4,B5
180. 00 C
190. 00 C      LESS VOLATILE COMPONENT
200. 00 C      A0,A1,A2,B0,B1,B2
210. 00 C      READ(7,777)A6,A7,A8
220. 00 C      READ(7,777)B5,B7,B8
230. 00 C
240. 00 C      INTERACTION COEFFICIENTS; C,D
250. 00 C      READ(7,777)F1,F0
260. 00 C
270. 00 C*****
280. 00 C      READ MOLECULAR WEIGHTS
290. 00 C      MORE & LESS VOLATILE COMPONENT
300. 00 C      READ(7,777)WMOL1,WMOL2
310. 00 C
320. 00 C      READ CRITICAL TEMPERATURES
330. 00 C      MORE & LESS VOLATILE COMPONENT
340. 00 C      READ(7,777)TC1,TC2
350. 00 C*****
360. 00 C      READ CONSTANTS FOR ESTIMATING EQUATIONS
370. 00 C      PSAT=EXP(AA1+AA2/T+AA3/T**2)
380. 00 C      VAPOR=T*(B31+B2*T+D53*T**2)/P
390. 00 C      LIQUID=CC1+CC2*T+CC3*T**2
400. 00 C      MORE VOLATILE COMPONENT
410. 00 C      READ(7,777)AA(1,1),AA(1,2),AA(1,3)
420. 00 C      READ(7,777)BB(1,1),BB(1,2),BB(1,3)
430. 00 C      READ(7,777)CC(1,1),CC(1,2),CC(1,3)
440. 00 C
450. 00 C      LESS VOLATILE COMPONENT
460. 00 C      READ(7,777)AA(2,1),AA(2,2),AA(2,3)
470. 00 C      READ(7,777)BB(2,1),BB(2,2),BB(2,3)
480. 00 C      READ(7,777)CC(2,1),CC(2,2),CC(2,3)
490. 00 C
500. 00 C      777 FORMAT(
510. 00 C      RETURN
520. 00 C      END
530. 00

```

END ELT. ERRORS: NONE. TIME: 0.101 SEC. IMAGE COUNT: 53

***** BIMASS *****

@ELT, L DD.BIMASS

ELT 8R1 S7401C 07/21/84 15:54:52 (0)

FUNCTION BIMASS(N)

10. 00

20. 00

30. 00

40. 00

50. 00

60. 00

70. 00

80. 00

90. 00

100. 00

110. 00

120. 00

130. 00

140. 00

150. 00

160. 00

170. 00

180. 00

190. 00

200. 00

210. 00

220. 00

230. 00

240. 00

250. 00

260. 00

270. 00

280. 00

290. 00

300. 00

310. 00

320. 00

330. 00

340. 00

350. 00

360. 00

370. 00

380. 00

390. 00

400. 00

410. 00

420. 00

430. 00

440. 00

450. 00

460. 00

470. 00

480. 00

490. 00

500. 00

510. 00

520. 00

530. 00

540. 00

550. 00

560. 00

570. 00

580. 00

C**** PURPOSE:

C TO COMPUTE MASS OF A NON-AZEOTROPIC REFRIGERANT

C IN A COIL

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C

- INNER DIAMETER OF COIL TUBES (FT)

= 1 FOR AN INDOOR COIL (-)

= 2 FOR AN OUTDOOR COIL (-)

- NUMBER OF REPEATING SECTIONS IN COIL (-)

- NUMBER OF TUBES PER COIL SECTION (-)

- REFRIG. TEMP. AT TUBE 1 END OF HIGHER REFRIG. ENTHALPY (F)

- REFRIG. TEMP. AT TUBE 1 END OF LOWER REFRIG. ENTHALPY (F)

- SPEC. VOL. OF REFRIG. VAPOR IN TUBE 1 END OF HIGHER REFRIG. ENTHALPY (FT**3/LB)

- SPEC. VOL. OF REFRIG. VAPOR IN TUBE 1 END OF LOWER REFRIG. ENTHALPY (FT**3/LB)

- SPEC. VOL. OF REFRIG. LIQUID IN TUBE 1 END OF HIGHER REFRIG. ENTHALPY (FT**3/LB)

- SPEC. VOL. OF REFRIG. LIQUID IN TUBE 1 END OF LOWER REFRIG. ENTHALPY (FT**3/LB)

- REFRIG. QUALITY AT TUBE K END OF HIGHER REFRIG. ENTHALPY (-)

- REFRIG. QUALITY AT TUBE K END OF LOWER REFRIG. ENTHALPY (-)

- FRACTION OF TUBE K WITH SUPERHEATED VAPOR (WHEN 2-PHASE FLOW IS IN REST OF TUBE) (-)

OR

- FRACTION OF TUBE K WITH 2-PHASE FLOW (WHEN SUB-COOLED LIQUID IS IN REST OF TUBE) (-)

- LOCKHART-MARTINELLI PARAMETER FOR REFRIG. IN TUBE K (-)

- WEIGHT COMPOSITION (FRACTION OF LESS VOLATILE COMPONENT)

- LENGTH OF COIL TUBES (FT)

- MASS OF REFRIGERANT IN COIL (LB)

- MASS OF REFRIGERANT IN COIL (LB)

- MASS OF REFRIGERANT IN COIL (LB)

- MASS OF REFRIGERANT IN COIL (LB)

- MASS OF REFRIGERANT IN COIL (LB)

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- MASS OF REFRIGERANT IN COIL (LB)

- MASS OF REFRIGERANT IN COIL (LB)

C**** SUPPROGRAMS CALLED BY BIMASS:

C RUBTEM, NEWTEM, VISCON

C

C

C

C

C

C

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C

C

C

C

C

C

C

C

C

C

C

C

C COMTON/RDATA3/XW, XH, WH

C COILION/HFIX/NDEP(2), NROM(2), DI(2), DO(2), DT(2), RPCH(2), DPCH(2),

C &WIDTH(2), FICH(2), FTK(2), FMK(2), NFA(2), ANGLE(2),

C &CONFST(2), CPOM(2), NTUB(2,5), IFROM(2,130), NSECT(2), NTPS(2)

C COILION/MASS/TRM(2,2,130), PRM(2,2,130), XRM(2,2,130),

C & VLM(2,2,130), VGH(2,2,130), XTUBE(2,130), XTT(2,130)

C

C

C

C

C

C

C

C

C

C

C

C AREA=3.14159*DI(N)*DI(N)/4.

C VTUBE=AREA*WIDTH(N)

C VBEND=AREA*20.0*DI(N)

C BIMASS=0.

C

C

C

C

C

C

C

C DO 100 1=1, NTPS(N)

C IF (XRM(N,2,1).GT..9999) THEN

C CH=VTUBE/(.5*(VGH(N,1,1)+VGH(N,2,1)))

C CH=CH+VBEND/VGH(N,2,1)

C GOTO 210

C @SUPERHEATED VAPOR ONLY


```

***** BMASS *****
590.
600.
610.
620.
630.
640.
650.
660.
670.
680.
690.
700.
710.
720.
730.
740.
750.
760.
770.
780.
790.
800.
810.
820.
830.
840.
850.
860.
870.
880.
890.
900.
910.
920.
930.
940.
950.
960.
970.
980.
990.
1000.
1010.
1020.
1030.
1040.
1050.
1060.
1070.
1080.
1090.
1100.
1110.
1120.
1130.
1140.
1150.
1160.

C
END IF
IF(XRM(N,1,1).GT.0.9999)THEN @SUPERHEATED VAPOR AND 2-PHASE
VSA1=VGM(N,1,1)+XTUBE(N,1)*(VGM(N,2,1)-VGM(N,1,1))
TM1=VTUBE*XTUBE(N,1)/(0.5*(VSA1+VGM(N,1,1)))
PDEW=PRM(N,1,1)+XTUBE(N,1)*(PRM(N,2,1)-PRM(N,1,1))
PA=PDEW/14.6959
TKDEW=(TMM(N,2,1)+460.)/1.8
CALL DEWTEH(1,PA,XM,TKDEW,XL)
TDEW=TKDEW*1.8-459.67
XRAV=0.5*(1.+XRM(N,2,1))
TRAV=.5*(TDEW+TRM(N,2,1))
VISL=VISCON(1,TRAV,XW)
VISV=VISCON(3,TRAV,XW)
XTTT=(1.-XRAV)/XRAV**9*(VISL/VISV)**.1
XTTT=XTTT*(VLM(N,2,1)/VGM(N,2,1))**.5
IF(XTTT.LT.10.)VOID=(1.+XTTT**.8)**(-.378)
IF(XTTT.GE.10.)VOID=.823-.157*ALOG(XTTT)
TM2=VTUBE*(1.-XTUBE(N,1))*VOID/(0.5*(VSA1+VGM(N,2,1)))
TM3=VTUBE*(1.-XTUBE(N,1))*(1.-VOID)/VLM(N,2,1)
TM4=VBEND*((1.-VOID)/VLM(N,2,1)+VOID/VGM(N,2,1))
CM=TM1+TM2+TM3+TM4
GOTO 210
END IF

C
IF(XRM(N,2,1).GT.0.9)THEN
XRAV=.5*(XRM(N,1,1)+XRM(N,2,1))
TRAV=.5*(TRM(N,1,1)+TRM(N,2,1))
VISL=VISCON(1,TRAV,XW)
VISV=VISCON(3,TRAV,XW)
XTTT=((1.-XRAV)/XRAV)**9*(VISL/VISV)**.1
XTTT=XTTT*(VLM(N,2,1)/VGM(N,2,1))**.5
IF(XTTT.LT.10.)VOID=(1.+XTTT**.8)**(-.378)
IF(XTTT.GE.10.)VOID=.823-.157*ALOG(XTTT)
TM1=VTUBE*VOID/(0.5*(VGM(N,1,1)+VGM(N,2,1)))
TM2=VTUBE*(1.-VOID)/(0.5*(VLM(N,2,1)+VLM(N,1,1)))
TM3=VBEND*(VOID/VGM(N,2,1)+(1.-VOID)/VLM(N,2,1))
CM=TM1+TM2+TM3
GOTO 210
END IF

C
IF(XRM(N,2,1).GT.1)THEN
IF(XTT(N,1).LT.10.)VOID=(1.+XTT(N,1)**.8)**(-.378)
IF(XTT(N,1).GE.10.)VOID=.823-.157*ALOG(XTT(N,1))
TM1=VTUBE*VOID/(0.5*(VGM(N,1,1)+VGM(N,2,1)))
TM2=VTUBE*(1.-VOID)/(0.5*(VLM(N,2,1)+VLM(N,1,1)))
TM3=VBEND*(VOID/VGM(N,2,1)+(1.-VOID)/VLM(N,2,1))
CM=TM1+TM2+TM3
GOTO 210
END IF

C
IF(XRM(N,2,1).GT.0.0001)THEN
XRAV=.5*(XRM(N,1,1)+XRM(N,2,1))
TRAV=.5*(TRM(N,1,1)+TRM(N,2,1))
VISL=VISCON(1,TRAV,XW)
VISV=VISCON(3,TRAV,XW)
XTTT=((1.-XRAV)/XRAV)**9*(VISL/VISV)**.1
XTTT=XTTT*(VLM(N,2,1)/VGM(N,2,1))**.5

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***** BIMASS *****

DATE 072184

```

1170. 00 IF(XTTT.LT.10.)VOID=(1.+XTTT*.8)**(-.378)
1180. 00 IF(XTTT.GE.10.)VOID=.823-.157*ALOG(XTTT)
1190. 00 TM1=VTUBE*VOID/(0.5*(VGM(N,1,1)+VGM(N,2,1)))
1200. 00 TM2=VTUBE*(1.-VOID)/(5*(VLM(N,2,1)+VLM(N,1,1)))
1210. 00 TM3=VBEND*(VOID/VGM(N,2,1)+(1.-VOID)/VLM(N,2,1))
1220. 00 CM=TM1+TM2+TM3
1230. 00 GOTO 210
1240. 00 END IF
1250. 00
1260. 00 C
1270. 00 IF(XRM(N,1,1).GT.0.0001)THEN @2-PHASE AND SUBCOOLED LIQUID
1280. 00 XRAV=0.5*XRM(N,1,1)
1290. 00 PA=PRM(N,1,1)+XTUBE(N,1)*(PRM(N,2,1)-PRM(N,1,1))
1300. 00 PA=PA/14.6959
1310. 00 CALL BUBTET(0,PA,XN,TK,XV)
1320. 00 TRAV=.5*(TK*1.8-459.67+TRM(N,1,1))
1330. 00 VISL=VISCON(1,TRAV,XV)
1340. 00 VISV=VISCON(3,TRAV,XV)
1350. 00 XTTT=((1.-XRAV)/XRAV)**.9*(VISL/VISV)**.1
1360. 00 XTTT=XTTT*(VLM(N,1,1)/VGM(N,1,1))**.5
1370. 00 IF(XTTT.LT.10.)VOID=(1.+XTTT*.8)**(-.378)
1380. 00 IF(XTTT.GE.10.)VOID=.823-.157*ALOG(XTTT)
1390. 00 TM1=XTUBE(N,1)*VOID*VTUBE/VGM(N,1,1)
1400. 00 TM2=XTUBE(N,1)*(1.-VOID)*VTUBE/VLM(N,1,1)
1410. 00 TM3=((1.-XTUBE(N,1))*VTUBE*VBEND)/VLM(N,2,1)
1420. 00 CM=TM1+TM2+TM3
1430. 00 GOTO 210
1440. 00 END IF
1450. 00 C
1460. 00 CM=VTUBE/(.5*(VLM(N,1,1)+VLM(N,2,1)))
1470. 00 CM=CM*VBEND/VLM(N,2,1)
210 CONTINUE
1490. 00 BIMASS=BIMASS+CM
100 CONTINUE
1500. 00 BIMASS=BIMASS*NSECT(N)
1510. 00 IF(N.EQ.1)THEN
1520. 00 WRITE(6,220)BIMASS
220 FORMAT(2X,'REFRIG. IN INDOOR COIL =',F8.3,' LB')
ELSE
1540. 00 WRITE(6,222)BIMASS
222 FORMAT(2X,'REFRIG. IN OUTDOOR COIL =',F8.3,' LB')
1550. 00 END IF
1560. 00 RETURN
1570. 00
1580. 00
1590. 00

```

END FLT. ERRORS: NONE. TIME: 0.237 SEC. IMAGE COUNT: 157

@HDG,P ***** BISYS1 ***** .L,0

```

***** BLINE *****
@ELT,L DD,BLINE
ELT 8R1 S74Q1C 07/21/84 15:54:53 (0)
10. 00 SUBROUTINE BLINE(D,TL,RMASS)
20. 00 C
30. 00 C**** PURPOSE:
40. 00 C TO CALCULATE FRICTIONAL PRESSURE DROP
50. 00 C OF A NON-AZEOTROPIC MIXTURE IN A LIQUID LINE
60. 00 C
70. 00 C**** INPUT DATA:
80. 00 C D - INNER LIQUID LINE DIAMETER (FT)
90. 00 C H11 - REFRIG. ENTHALPY AT INLET (BTU/LBM)
100. 00 C P11 - REFRIG. PRESSURE AT INLET (PSIA)
110. 00 C RMASS - REFRIG. MASS FLOW RATE (LBM/H)
120. 00 C TL - LIQUID LINE LENGTH (FT)
130. 00 C T11 - REFRIG. TEMPERATURE AT INLET (F)
140. 00 C XM - MOLAR COMPOSITION (FRACTION OF LESS VOLATILE COMPONENT)
150. 00 C XW - WEIGHT COMPOSITION (FRACTION OF LESS VOLATILE COMPONENT)
160. 00 C X11 - REFRIG. QUALITY AT INLET (-)
170. 00 C WM - MIXTURE MOLECULAR WEIGHT (G/MOL)
180. 00 C
190. 00 C**** OUTPUT DATA:
200. 00 C H12 - REFRIG. ENTHALPY AT OUTLET (BTU/LBM)
210. 00 C P12 - REFRIG. PRESSURE AT OUTLET (PSIA)
220. 00 C T12 - REFRIG. TEMPERATURE AT OUTLET (F)
230. 00 C X12 - REFRIG. QUALITY AT OUTLET (-)
240. 00 C
250. 00 C**** SUBPROGRAMS CALLED BY BLINE:
260. 00 C
270. 00 C COMMON/BDATA2/W1,W2,TC1,TC2
280. 00 C COMMON/BDATA3/X2,XM,WM
290. 00 C COMMON/COND11/P11,T11,H11,X11
300. 00 C COMMON/COND12/P12,T12,H12,X12
310. 00 C DATA NO,N1/O,1/
320. 00 C
330. 00 C WRITE(6,100)P11,T11,X11,H11
340. 00 C H12=H11
350. 00 C IF(X11.GT.999)THEN
360. 00 C X12=X11
370. 00 C P12=P11
380. 00 C T12=T11
390. 00 C WRITE(6,99)
400. 00 C FORMAT(' ERROR IN CALLING BLINE, X11=0.')
410. 00 C GOTO 50
420. 00 C END IF
430. 00 C TLL=TL
440. 00 C TK11=(T11+459.67)/1.8
450. 00 C PA11=P11/14.6959
460. 00 C ACCUR=0.003
470. 00 C IF(X11.GT.0.)GOTO 10
480. 00 C
490. 00 C**** SUBCOOLED LIQUID AT ENTRANCE
500. 00 C CALL VOLIT(0,TK11,PA11,X11,VL)
510. 00 C VL=VL*16.01845/MM
520. 00 C VISL=VISCON(1,T11,XW)
530. 00 C PD1=SPHDF1(RMASS,TL,D,VL,VISL)
540. 00 C P12=P11-PD1
550. 00 C PA12=P12/14.6959
560. 00 C TK12=TK11

```

***** BLINE *****

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570. CALL HPIN(N1,H12,PA12,XM,ACCUR,TK12,X12,XL,XV,VL,VV)
580. T12=TK12*1.8-459.67
590. IF(X12.LT.0.0001)GOTO 50
600.
610. C
620. C
630. PRESSURE DROPS BELOW SAT. PRESSURE
640. CALL BUBPRE(NO,TK11,XM,PARI,XVIN)
650. XLIN=XM
660. CALL VOLIT1(NO,TK11,PARI,XLIN,VLIN)
670. CALL VOLIT1(N1,TK11,PARI,XVIN,VVIN)
680. TRI=T11
690. PRI=P/RI*14.6959
700. XRI=0.
710. TLL=TL*(PRI-P12)/PD1
720. GOTO 20
730. C
740. C***** 2-PHASE AT INLET
750. 10 CALL GLITY(TK11,PA11,XM,XQ11,XVIN,XLIN)
760. CALL VOLIT1(NO,TK11,PA11,XLIN,VLIN)
770. CALL VOLIT1(N1,TK11,PA11,XVIN,VVIN)
780. XRI=X11
790. TRI=T11
800. PRI=P11
810. PRE=PRI
820. 20 X12=XRI+0.02
830. T12=TRI
840. TK12=(T12+459.67)/1.8
850. XLE=XLIN
860. XVE=XVIN
870. VLE=VLIN
880. VVE=VVIN
890. DO 40 I=1,15
900. XRAV=0.5*(XRI+X12)
910. TRAV=0.5*(TRI+T12)
920. VL=0.5*(VLIN+VLE)
930. VV=0.5*(VVIN+VVE)
940. XL=0.5*(XLIN+XLE)
950. XV=0.5*(XVIN+XVE)
960. W1L=W1*(1.-XL)+12*XL
970. VLW=VL*16.04816/WML
980. XLV=XL/(W1/W2*(1.-XL)+XL)
990. WMV=W1*(1.-XV)+W2*XV
1000. VVW=VV*16.04816/WMV
1010. VMIX=VLW*(1.-XRAV)+VVW*XRAV
1020. VISL=VISCON(1,TRAV,XLV)
1030. PD1=SFIDP1(RMASS,TLL,D,VMIX,VISL)
1040. P12=PRI-PD1
1050. IF(P12.LE.0.)THEN
1060. WRITE(6,98)
1070. X12=X11
1080. T12=T11
1090. P12=P11
1100. GOTO 50
1110. END IF
1120.
1130. C
1140. PA12=P12/14.6959
      CALL HPIN(N1,H12,PA12,XM,ACCUR,TK12,X12,XLE,XVE,VLE,VVE)
      T12=TK12*1.8-459.67

```


***** BLINE *****

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1150. 00      DELP=PRE-P12
1160. 00      IF (ABS(DELP).LT.0.001)GOTO 50
1170. 00      PRE=P12
1180. 00      40 CONTINUE
1190. 00      WRITE(5,105)DELP
1200. 00      50 WRITE(6,110)P12,T12,X12
1210. 00      100 FORMAT(// ' LIQUID LINE: ',/
1220. 00      @ ' INPUT - PIN = ',1PE10.3,3X,'TIN = ',1PE10.3,3X,
1230. 00      @ ' XIN = ',1PE10.3,3X,'HIN = ',1PE10.3)
1240. 00      105 FORMAT(' BLINE DOES NOT CONVERGE, DELP= ',1PE14.5)
1250. 00      110 FORMAT(' OUTPUT - POUT= ',1PE10.3,3X,'TCUT= ',1PE10.3,3X,
1260. 00      @ 'XOUT= ',1PE10.3)
1270. 00      RETURN
1280. 00      END

```

END ELT. ERRORS: NONE. TIME: 0.195 SEC. IMAGE COUNT: 128

@HDG,P ***** BMAIN ***** .L,0

***** BMAIN *****

@ELT,L DD.BMAIN

ELT 0R1 S74Q1C 07/21/84 15:54:54 (0)

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10. C
20. C ***** THIS IS THE MAIN PROGRAM OF ** HPBI **
30. C PROGRAM FOR SIMULATION OF VAPOR COMPRESSION CYCLE
40. C WITH CONSTANT FLOW AREA EXPANSION DEVICE,
50. C WORKING WITH NON-AZEOTROPIC, BINARY REFRIGERANT.
60. C
70. C ***** THIS PROGRAM HAS BEEN DEVELOPED AT NATIONAL BUREAU OF STANDARDS,
80. C WASHINGTON, D.C.,
90. C UNDER THE CONTRACT FROM ELECTRIC POWER RESEARCH INSTITUTE,
100. C PALO ALTO, CA.
110. C FEBRUARY, 1984.
120. C
130. C
140. C ***** THIS PROGRAM PROVIDES LOGIC AND ITERATION PROCEDURES
150. C FOR EVALUATION OF PERFORMANCE OF VAPOR COMPRESSION CYCLE
160. C WITH CONSTANT FLOW AREA EXPANSION DEVICE
170. C
180. C ***** THIS PROGRAM PROVIDES ALSO LOGIC FOR EVALUATION
190. C OF PERFORMANCE PARAMETERS OF RECIPROCATING, HERMETIC COMPRESSOR.
200. C
210. C ***** INPUT DATA:
220. C [1] REFRIGERANT CONSTANTS - CONTAINED IN PROGRAM BOONST
230. C [2] HEAT PUMP DATA - READ FROM FILE 8
240. C REFER TO*****
250. C [3] RUN CONTROLLING DATA - READ FROM A TERMINAL
260. C [4] INDOOR/OUTDOOR AIR CONDITIONS DATA - READ FROM A TERMINAL
270. C [5] REFRIGERANT PARAMETERS (GUESS) - READ FROM A TERMINAL
280. C
290. C [2],[3],[4] & [5] ARE EXPLAINED IN ALPHABETIC ORDER BELOW
300. C AHGT - DISTANCE BETWEEN ACCUMULATOR TOP AND OIL RETURN HOLE (FT)
310. C AMAS(110) - AIR MASS FLOW RATE THROUGH COIL (LBM/H)
320. C ANGLE(110) - ANGLE BETWEEN COIL FACE & AIR STREAMLINES (RAD)
330. C CAPL1 - LENGTH OF COOLING OPERATION EXPANSION DEVICE (IN)
340. C CAPL2 - LENGTH OF HEATING OPERATION EXPANSION DEVICE (IN)
350. C CAPID1 - INNER DIA. OF COOLING OPERATION EXPANSION DEVICE (IN)
360. C CAPID2 - INNER DIA. OF HEATING OPERATION EXPANSION DEVICE (IN)
370. C CONST(110) - CONSTANT FOR AIR SIDE HEAT TRANSFER CORRELATION (-)
380. C CPOW(110) - CONSTANT FOR AIR SIDE HEAT TRANSFER CORRELATION (-)
390. C CLRCFF - COMPRESSOR CLEARANCE VOLUME AS FRACTION OF STROKE VOLUME (-)
400. C CPC34 - PRESSURE DROP PARAMETER AT COMPRESSOR CAN INLET
410. C (LBF*H**2/LB*IN**2*FT**3)
420. C CPC45 - PRESSURE DROP PARAMETER AT COMPRESSOR SUCTION VALVE
430. C (LBF*H**2/LB*IN**2*FT**3)
440. C CPC67 - PRESSURE DROP PARAMETER AT COMPRESSOR DISCHARGE VALVE
450. C (LBF*H**2/LB*IN**2*FT**3)
460. C CPC78 - PRESSURE DROP PARAMETER AT COMPRESSOR CAN EXIT
470. C (LBF*H**2/LB*IN**2*FT**3)
480. C CPDR - PRESSURE DROP PARAMETER FOR 4-WAY VALVE
490. C (LBF*H**2/LB*IN**2*FT**3)
500. C CQ - PARAMETER FOR 4-WAY VALVE HEAT TRANSFER (FT**2)
510. C CQCC0A - PARAMETER FOR COMPRESSOR CAN WALL-AMBIENT AIR HEAT TRANSFER
520. C CQC4C - PARAMETER FOR COMPRESSOR CAN WALL-REFRIG. VAPOR HEAT TRANSFER
530. C CQC45 - SUCTION VALVE HEAT TRANSFER PARAMETER (FT**2)
540. C CQC67 - DISCHARGE VALVE HEAT TRANSFER PARAMETER (FT**2)
550. C CQC78 - CAN EXIT HEAT TRANSFER PARAMETER (FT**2)
560. C DACC - ACCUMULATOR INNER DIAMETER (FT)

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(BTU/H*F**1.333
(FT**2)

(FT**2)

(FT**2)

(FT**2)

(FT**2)

(FT**2)

(FT**2)

(FT**2)

(FT**2)

(FT**2)

(FT**2)

***** BMAIN *****

DATE 072184

570.	C	00	DHOLE(1)	- DIA. OF A HOLE IN ACCUMULATOR TUBE (FT)
580.	C	00	I=1	- OIL RETURN HOLE
590.	C	00	I=2	- UPPER HOLE
600.	C	00	DI(110)	- INNER DIAMETER OF COIL TUBES (IN)
610.	C	00	DO(110)	- OUTER DIAMETER OF COIL TUBES (IN)
620.	C	00	DPCI(110)	- DEPTH PITCH FOR COIL TUBES (IN)
630.	C	00	DT(110)	- FIN TIP DIAMETER (IN)
640.	C	00	DTUBE	- INNER DIA. OF ACCUMULATOR TUBE (FT)
650.	C	00	ETOFAN	- INDOOR FAN ENERGY INPUT RATE (KW)
660.	C	00	ELEFUL	- COMPRESSOR MOTOR ENERGY INPUT RATE AT FULL LOAD (KW)
670.	C	00	EMETA(K)	- COMPRESSOR MOTOR EFFICIENCY IN FRACTION AT FRACTION OF FULL LOAD SPECIFIED BY EMOPT(K) (-)
680.	C	00	EMOPT(K)	- COMPRESSOR MOTOR FULL LOAD FRACTION (-)
690.	C	00	ENRPM(M)	- COEFFICIENT FOR COMPRESSOR MOTOR RPM CALCULATION (-)
700.	C	00	FOOFAN	- OUTDOOR FAN ENERGY INPUT RATE (KW)
710.	C	00	ETAPLY	- COMPRESSOR POLYTROPIC EFFICIENCY (-)
720.	C	00	FMK(110)	- FIN MATERIAL THERMAL CONDUCTIVITY (BTU/FT*H*F)
730.	C	00	FPCH(110)	- FIN PITCH (IN)
740.	C	00	FTK(110)	- FIN THICKNESS (IN)
750.	C	00	HDIS	- VERTICAL DISTANCE BETWEEN HOLES IN AN ACCUMULATOR (FT)
760.	C	00	IFROM(110,M)	- NUMBER OF TUBE FROM WHICH TUBE M RECEIVES REFRIG. WHEN COIL WORKS AS EVAPORATOR (-)
770.	C	00	I10	= 1 FOR INDOOR COIL (-)
780.	C	00	I20	= 2 FOR OUTDOOR COIL (-)
790.	C	00	IPTP	= 1 FOR EVALUATION OF COMPRESSION CYCLE PERFORMANCE (-)
800.	C	00	I2TP	= 2 FOR EVALUATION OF COMPRESSOR CYCLE PERFORMANCE (-)
810.	C	00	ITER	= 1 FOR ENTHALPY & REF. FLOW BALANCE (-)
820.	C	00	I2ER	= 2 FOR ABOVE + REFRIG. CHARGE BALANCE (-)
830.	C	00	LPR	= 0 FOR NO INPUT DATA PRINTOUT (-)
840.	C	00	NCPL1	= 1 FOR INPUT DATA PRINTOUT REQUEST (-)
850.	C	00	NCPL2	- NUMBER OF COOLING OPERATION EXPANSION DEVICES (-)
860.	C	00	NDECT(1)	- NUMBER OF HEATING OPERATION EXPANSION DEVICES (-)
870.	C	00	NDECT(2)	- NUMBER OF COIL TUBE ROW DEPTHS (-)
880.	C	00	NROW(110)	- NUMBER OF REPEATING SECTIONS IN INDOOR COIL (-)
890.	C	00	NTUB(110,1)	- NUMBER OF TUBES PER ROW (-)
900.	C	00	NSYS	- NUMBER OF TUBES IN ROW 1 FOR EACH SECTION OF COIL (-)
910.	C	00	POA	= 1 FOR HEATING OPERATION (-)
920.	C	00	PRA	= 2 FOR COOLING OPERATION (-)
930.	C	00	REFIN	- OUTDOOR AIR PRESSURE (PSIA)
940.	C	00	RD	- INDOOR AIR PRESSURE (PSIA)
950.	C	00	RD1	- REFRIGERANT CHARGE (REQUIRED FOR ITER=1 ONLY) (LBM)
960.	C	00	RD2	- INNER DIAMETER COMPRESSOR-INDOOR COIL TUBING (IN)
970.	C	00	RHOA	- OUTER DIAMETER COMPRESSOR-INDOOR COIL TUBING (IN)
980.	C	00	RHRA	- OUTER DIAMETER OF COMPRESSOR-INDOOR COIL TUBING INSULATION (IN)
990.	C	00	RK1	- INSULATION (BTU/FT*H*F)
1000.	C	00	RK2	- THERMAL CONDUCTIVITY OF COMPRESSOR-INDOOR COIL TUBING INSULATION (BTU/FT*H*F)
1010.	C	00	RL	- LENGTH OF COMPRESSOR-INDOOR COIL TUBING (IN)
1020.	C	00	RPCH(110)	- PITCH BETWEEN TUBES OF THE SAME DEPTH (IN)
1030.	C	00	RYL	- LIQUID LINE DIAMETER (IN)
1040.	C	00	RYL	- LIQUID LINE LENGTH (IN)
1050.	C	00	SWPVOL	- COMPRESSOR SWEPT VOLUME PER REVOLUTION (IN**3)
1060.	C	00	TMK(110)	- TUBE MATERIAL THERMAL CONDUCTIVITY (BTU/FT*H*F)
1070.	C	00		
1080.	C	00		
1090.	C	00		
1100.	C	00		
1110.	C	00		
1120.	C	00		
1130.	C	00		
1140.	C	00		


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1150. 00 C T63 - REFRIG. SAT. TEMP. AT COMPRESSOR CAN INLET (F) (GUESS)
1160. 00 C T66 - REFRIG. SAT. TEMP. AT COMPRESSOR DISCHARGE VALVE (F) (GUESS)
1170. 00 C T0A - OUTDOOR AIR TEMPERATURE (F)
1180. 00 C TRA - INDOOR AIR TEMPERATURE (F)
1190. 00 C TSUP3 - REFRIG. SUPERHEAT AT COMPRESSOR CAN INLET (F) (GUESS)
1200. 00 C VCAN - VOLUME OF COMPRESSOR CAN FILLED BY REFRIG. (FT**3)
1210. 00 C WIDTH(110) - COIL WIDTH (IN)
1220. 00 C X3 - REFRIG. QUALITY AT COMPRESSOR CAN INLET (-)
1230. 00 C YD - INNER DIAMETER COMPRESSOR-OUTDOOR COIL TUBING (IN)
1240. 00 C YD1 - OUTER DIAMETER OF COMPRESSOR-OUTDOOR COIL TUBING (IN)
1250. 00 C YD2 - OUTER DIAMETER OF COMPRESSOR-OUTDOOR COIL TUBING INSULATION (IN)
1260. 00 C YK1 - THERMAL CONDUCTIVITY OF COMPRESSOR-OUTDOOR COIL TUBING
1270. 00 C - MATERIAL (BTU/FT*H*F)
1280. 00 C YK2 - THERMAL CONDUCTIVITY OF COMPRESSOR-OUTDOOR COIL TUBING
1290. 00 C - INSULATION (BTU/FT*H*F)
1300. 00 C YL - LENGTH OF COMPRESSOR-OUTDOOR COIL TUBING (IN)
1310. 00 C
1320. 00 C ***** OUTPUT DATA:
1330. 00 C CFMIND - VOLUMETRIC AIR FLOW THROUGH INDOOR COIL (FT**3/MIN)
1340. 00 C CFMOUT - VOLUMETRIC AIR FLOW THROUGH OUTDOOR COIL (FT**3/MIN)
1350. 00 C COP - COEFFICIENT OF PERFORMANCE (-)
1360. 00 C ELUSE - TOTAL HEAT PUMP ENERGY CONSUMPTION (KW)
1370. 00 C RMASS - REFRIG. MASS FLOW RATE (LBM/H)
1380. 00 C T63 - REFRIG. SAT. TEMP. AT COMPRESSOR CAN INLET (F)
1390. 00 C T66 - REFRIG. SAT. TEMP. AT COMPRESSOR DISCHARGE VALVE (F)
1400. 00 C TMASS - REFRIGERANT CHARGE (LBM)
1410. 00 C TSUP3 - REFRIG. SUPERHEAT AT COMPRESSOR CAN INLET (F)
1420. 00 C QLOAD - NET CAPACITY OF A HEAT PUMP (BTU/H)
1430. 00 C REFRIG. PARAMETERS H,P,S,T,X FOR SYSTEM LOCATIONS FROM 1 TO 13
1440. 00 C WHERE:
1450. 00 C H - ENTHALPY (BTU/LBM)
1460. 00 C P - PRESSURE (PSIA)
1470. 00 C S - ENTROPY (BTU/LBM*F)
1480. 00 C T - TEMPERATURE (F)
1490. 00 C X - QUALITY (-)
1500. 00 C
1510. 00 C 1 - EVAPORATOR EXIT
1520. 00 C 2 - LOW PRESSURE 4-WAY VALVE INLET
1530. 00 C 3 - COMPRESSOR INLET
1540. 00 C 4 - INSIDE COMPRESSOR CAN
1550. 00 C 5 - CYLINDER AT SUCTION
1560. 00 C 6 - CYLINDER AT DISCHARGE
1570. 00 C 7 - DISCHARGE MANIFOLD
1580. 00 C 8 - COMPRESSOR CAN EXIT
1590. 00 C 9 - HIGH PRESSURE 4-WAY VALVE EXIT
1600. 00 C 10 - CONDENSER INLET
1610. 00 C 11 - CONDENSER OUTLET
1620. 00 C 12 - EXPANSION DEVICE INLET
1630. 00 C 13 - EVAPORATOR INLET
1640. 00 C
1650. 00 C ***** SUBPROGRAMS CALLED BY THIS MAIN PROGRAM:
1660. 00 C AIRPR,BIMASS,BLINE,BPMASS,CAPIL,COMPAR,COMPRE,CONDXH,DEMPRE,
1670. 00 C ENTR02,EVAPHX,HCVCP,PXQIN2,VALVPA,VOL1,T1,VACCUM
1680. 00 C
1690. 00 C
1700. 00 C COMMON/RDATA1/A3,A4,A5,A6,A7,A8,B3,B4,B5,B6,B7,B8,F0,F1
1710. 00 C COMMON/RDATA2/W1,W2,TC1,TC2
1720. 00 C COMMON/RDATA3/X1,XM,XM

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***** BMAIN *****

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1730. 00 COMMON/PARAM/A,B,C1,C2,D1,D2
1740. 00 COMMON/STORE1/A1,A2,B1,B2,SFG31
1750. 00 COMMON/ESDATA/AA(2,3),BB(2,3),CC(2,3)
1760. 00 COMMON/ACCURA/IACC,ITM
1770. 00
1780. 00 COMMON/COND1/P1,T1,H1,X1
1790. 00 COMMON/COND2/P2,T2,H2,X2
1800. 00 COMMON/COND3/P3,T3,H3,X3
1810. 00 COMMON/COND4/P4,T4,H4,X4,P5,T5,H5,X5,S5,P6,T6,H6,X6,
1820. 00 @ P7,T7,H7,X7
1830. 00 COMMON/COND5/P8,T8,H8,X8
1840. 00 COMMON/COND6/P9,T9,H9,X9
1850. 00 COMMON/COND10/P10,T10,H10,X10
1860. 00 COMMON/COND11/P11,T11,H11,X11
1870. 00 COMMON/COND12/P12,T12,H12,X12
1880. 00 COMMON/COND13/P13,T13,H13,X13
1890. 00 COMMON/COMP/CPC34,CPC40,CPC45,CPC45,CPC45,
1900. 00 & EMETAL(11),EMOPT(11),EMRPM(G),ELFFUL,SUPVOL,
1910. 00 & ETAPLY,CLREFF,CPC67,CPC67,CPC78,CPC78
1920. 00 COMMON/WAY4/CQ,CPDR
1930. 00 COMMON/HPTHX/NDEF(2),NROW(2),DI(2),DI(2),DT(2),RPCH(2),DPCH(2),
1940. 00 & WIDTH(2),FPCH(2),FTK(2),FMK(2),TIK(2),AMAS(2),ANGLE(2),
1950. 00 & CONST(2),CPOW(2),NTUB(2,5),IFROM(2,130),NJECT(2),NTPS(2)
1960. 00 COMMON/CFIN/CHX(2,8)
1970. 00 COMMON/MERG/MERGE(2,20,2),IMER(2),ISTART(2,20),IST(2),
1980. 00 & IDEPTH(2,130),FLOW(2,130),JFROM(2,130),KEEFD(2,130,3),
1990. 00 & KSTART(2,20),KST(2)
2000. 00 COMMON/MACS/TRM(2,2,130),PRM(2,2,130),XRM(2,2,130),
2010. 00 & VRM(2,2,130),VGM(2,2,30),XTUBE(2,130),XTT(2,130)
2020. 00 COMMON/RLINE/RL,RO,RK1,RO1,RK2,RD2
2030. 00 COMMON/YLINE/YL,YD,YK1,YD1,YK2,YD2
2040. 00 COMMON/ACCDIM/AHGT,DACC,DHGLE(2),DTUBE,HDIS
2050. 00 DIMENSION ATITLE(20),CEFI(8,3)
2060. 00 DATA X6,X7,X8,X9,X10/5*1./
2070. 00 DATA CEFF(1,1),CEFF(1,2),CEFF(1,3)/1,0,0./
2080. 00 *CEFF(2,1),CEFF(2,2),CEFF(2,3)/-.02392, -.13755, .20130E-01/
2090. 00 *CEFF(3,1),CEFF(3,2),CEFF(3,3)/.16105, .8189E-01, -.1144E-01/
2100. 00 *CEFF(4,1),CEFF(4,2),CEFF(4,3)/-.64375, -.5556E-01, -.28753E-01/
2110. 00 *CEFF(5,1),CEFF(5,2),CEFF(5,3)/.53481, .1804E-01, .42477E-01/
2120. 00 *CEFF(6,1),CEFF(6,2),CEFF(6,3)/-.19286, .36404E-03, -.20335E-01/
2130. 00 *CEFF(7,1),CEFF(7,2),CEFF(7,3)/.03454, .1056E-02, .40947E-02/
2140. 00 *CEFF(8,1),CEFF(8,2),CEFF(8,3)/-.0020972, .1241E-03, -.29673E-03/
2150. 00 DATA NO,N1/O,1/
2160. 00
2170. 00
2180. 00 ***** INPUT REFRIGERANT DATA *****
2190. 00 CALL ECONST
2200. 00
2210. 00
2220. 00 WRITE(6,894)
2230. 00 READ(5,777)IPTP
2240. 00 IF(IPTP.EQ.2)GOTO 340
2250. 00
2260. 00
2270. 00
2280. 00 ***** INPUT HEAT PUMP DATA *****
2290. 00 ***** INPUT MOTOR/COMPRESSOR DATA *****
2300. 00 READ(8,800)ATITLE
2310. 00 READ(8,777)(EMETA(1),I=1,5)
2320. 00 READ(8,777)(EMETA(1),I=6,11)
2330. 00 READ(8,777)(EMOPT(1),I=1,5)

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***** BMAIN *****

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2310. READ(8,777)(ENOPT(1),I=6,11)
2320. READ(8,777)(ENRPM(1),I=1,6)
2330. IF(IPTP.EQ.1)GOTO 301
2340. READ(8,777)FLEFUL,S/PCVL,ETAPLY,CLREFF
2350. READ(8,777)CPC34,CPC45,CPC67,CPC78
2360. READ(8,777)CQC45,CQC67A,CQC65,CQC67,CQC78
2370. READ(8,777)CQ,CQDR,VCAN,JEFIN
2380. READ(8,777)AHGT,DACC,DHOLE(1),DHOLE(2),DIUPE,HDIS
2390. C**** INPUT INDOOR COIL DATA
2400. READ(8,777)NDEP(1),NROW(1)
2410. READ(8,777)DI(1),DO(1),DT(1),RPCH(1),DPCCH(1),WIDTH(1)
2420. READ(8,777)FPCCH(1),FTK(1),FMK(1),TNK(1),AMPS(1)
2430. READ(8,777)CONST(1),CPOW(1),ANGLE(1)
2440. READ(8,777)EIDFAN
2450. READ(8,777)INSECT(1)
2460. READ(8,777)(NTUB(1),I=1,5)
2470. DO 12 I=1,13
2480. N=10*I
2490. M=N-9
2500. 12 READ(8,777)(IFROM(1,J),J=M,N)
2510. C**** INPUT OUTDOOR COIL DATA
2520. READ(8,777)NDEP(2),NROW(2)
2530. READ(8,777)DI(2),DO(2),DT(2),RPCH(2),DPCCH(2),WIDTH(2)
2540. READ(8,777)FPCCH(2),FTK(2),FMK(2),TNK(2),AMPS(2)
2550. READ(8,777)CONST(2),CPOW(2),ANGLE(2)
2560. READ(8,777)EIDFAN
2570. READ(8,777)INSECT(2)
2580. READ(8,777)(NTUB(2),I=1,5)
2590. DO 13 I=1,13
2600. N=10*I
2610. M=N-9
2620. 13 READ(8,777)(IFROM(2,J),J=M,N)
2630. C**** INPUT EXPANSION DEVICE DATA
2640. READ(8,777)CAPID1,CAPL1,HCP11,CAPID2,CAPL2,HCP12
2650. C**** INPUT CONNECTING TUDING DATA
2660. READ(8,777)YL,YD,YK1,YD1,YK2,YD2
2670. READ(8,777)RL,RD,RK1,RD1,RK2,RD2
2680. READ(8,777)RYL,RYD
2690. C
2700. C*****
2710. C**** EVALUATION OF PERFORMANCE
2720. C**** OF VAPOR COMPRESSION CYCLE
2730. C*****
2740. C**** WRITE INPUT DATA
2750. LPR1=1
2760. LPR2=1
2770. LPR3=1
2780. WRITE(6,851)
2790. READ(5,777)LPR
2800. IF(LPR.EQ.0)GOTO 32
2810. WRITE(6,852)
2820. READ(5,777)LPR1
2830. WRITE(6,853)
2840. READ(5,777)LPR2
2850. WRITE(6,854)
2860. READ(5,777)LPR3
2870. WRITE(6,855)
2880. READ(5,777)LPR4
2890.

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***** BMAIN *****
2900. 00
2910. 00
2920. 00
2930. 00
2940. 00
2950. 00
2960. 00
2970. 00
2980. 00
2990. 00
3000. 00
3010. 00
3020. 00
3030. 00
3040. 00
3050. 00
3060. 00
3070. 00
3080. 00
3090. 00
3100. 00
3110. 00
3120. 00
3130. 00
3140. 00
3150. 00
3160. 00
3170. 00
3180. 00
3190. 00
3200. 00
3210. 00
3220. 00
3230. 00
3240. 00
3250. 00
3260. 00
3270. 00
3280. 00
3290. 00
3300. 00
3310. 00
3320. 00
3330. 00
3340. 00
3350. 00
3360. 00
3370. 00
3380. 00
3390. 00
3400. 00
3410. 00
3420. 00
3430. 00
3440. 00
3450. 00
3460. 00
3470. 00

IF (LPR1.EQ.0) GOTO 22
WRITE(6,905)
WRITE(6,856) ELEFUL, SWFV2L, ETAPLY, CLREFF
WRITE(6,599)
WRITE(6,856) (EMETA(1), I=1, 5)
WRITE(6,856) (EMETA(1), I=6, 11)
WRITE(6,901)
WRITE(6,856) (EMOPT(1), I=1, 5)
WRITE(6,856) (EMOPT(1), I=6, 11)
WRITE(6,900)
WRITE(6,856) (EMRPM(1), I=1, 6)
WRITE(6,906)
WRITE(6,856) CPC34, CPC45, CPC67, CPC78
WRITE(6,907)
WRITE(6,856) CQC4C, CQCCOA, CQC45, CQC67, CQC78
WRITE(6,951)
WRITE(6,856) CO, CPDR, VCAN, REFIN
WRITE(6,952)
WRITE(6,856) AHGT, DACC, DHOLE(1), DHOLE(2), DTURE, HDIS
22 CONTINUE
IF (LPR2.EQ.0) GOTO 28
WRITE(6,911)
WRITE(6,912) NDEF(1), NROW(1)
WRITE(6,913)
WRITE(6,856) DI(1), DO(1), DT(1), RPCH(1), RPCH(1), WIDTH(1)
WRITE(6,914)
WRITE(6,856) FPCH(1), FTK(1), FMK(1), TMK(1), AMAS(1)
WRITE(6,918)
WRITE(6,856) CONST(1), CPQW(1), ANGLE(1)
WRITE(6,935) EIDFAN
WRITE(6,915) NSECT(1)
WRITE(6,916) (NTUB(1,1), I=1, 5)
DO 24 I=1, 13
N=10*I
M=N-9
24 WRITE(6,917) (IFROM(1,J), J=M,N)
WRITE(6,919)
WRITE(6,920) NDEF(2), NROW(2)
WRITE(6,921)
WRITE(6,856) DI(2), DO(2), DT(2), RPCH(2), DPCH(2), WIDTH(2)
WRITE(6,922)
WRITE(6,856) FPCH(2), FTK(2), FMK(2), TMK(2), AMAS(2)
WRITE(6,923)
WRITE(6,856) CONST(2), CPQW(2), ANGLE(2)
WRITE(6,936) EIDFAN
WRITE(6,915) NSECT(2)
WRITE(6,924) (NTUB(2,1), I=1, 5)
DO 26 I=1, 13
N=10*I
M=N-9
26 WRITE(6,917) (IFROM(2,J), J=M,N)
28 CONTINUE
IF (LPR3.EQ.0) GOTO 30
WRITE(6,946) CAP1D1, CAP1, NCPL1, CAP1D2, CAP12, NCPL2
30 CONTINUE
IF (LPR4.EQ.0) GOTO 32
WRITE(6,903)
WRITE(6,929)

```

```

3480. 00 WRITE(6,855)RL, RD, RK1, RD1, RK2, RD2
3490. 00 WRITE(6,909)
3500. 00 WRITE(6,910)
3510. 00 WRITE(6,956)YL, YD, YK1, YD1, YK2, YD2
3520. 00 WRITE(6,948)
3530. 00 WRITE(6,949)
3540. 00 WRITE(6,855)RYL, RYD
3550. 00
3560. 00 32 CONTINUE
3570. 00 C*****
3580. 00 C***** PREPARE DATA FOR CALCULATIONS
3590. 00 C***** INDOOR @ OUTDOOR COIL
3600. 00 DO 34 I=1,2
3610. 00 D1(I)=D1(I)/12.
3620. 00 DQ(I)=DQ(I)/12.
3630. 00 DT(I)=DT(I)/12.
3640. 00 RPCH(I)=RPCH(I)/12.
3650. 00 DPCH(I)=DPCH(I)/12.
3660. 00 WIDTH(I)=WIDTH(I)/12.
3670. 00 FPCH(I)=FPCH(I)/12.
3680. 00 34 FTK(I)=FTK(I)/12.
3690. 00 CALL HXCODE(1)
3700. 00 CALL HXCODE(2)
3710. 00 DO 40 I=1,2
3720. 00 CRR=DT(I)/DQ(I)
3730. 00 DO 40 J=1,8
3740. 00 CHX(I,J)=0.
3750. 00 DO 40 K=1,3
3760. 00 L=K-1
3770. 00 40 CHX(I,J)=CHX(I,J)*CEFF(J,K)*CRR**L
3780. 00 C***** EXPANSION DEVICE
3790. 00 CAPID1=CAPID1/12.
3800. 00 CAPL1=CAPL1/12.
3810. 00 CAPID2=CAPID2/12.
3820. 00 CAPL2=CAPL2/12.
3830. 00 C***** CONNECTING TUBING
3840. 00 RL=RL/12.
3850. 00 RD=RD/12.
3860. 00 RD1=RD1/12.
3870. 00 RD2=RD2/12.
3880. 00 YL=YL/12.
3890. 00 YD=YD/12.
3900. 00 YD1=YD1/12.
3910. 00 YD2=YD2/12.
3920. 00 RYL=RYL/12.
3930. 00 RYD=RYD/12.
3940. 00
3950. 00 C*****
3960. 00 50 WRITE(6,926)
3970. 00 READ(5,777)POA, TOA, RHOA, PRA, TRA, RHRA
3980. 00 WRITE(6,940)
3990. 00 READ(5,777)NSYS
4000. 00 IF(NSYS.EQ.1)THEN
4010. 00 CAPID=CAPID2
4020. 00 CAPL=CAPL2
4030. 00 NCPL=NCPL2
4040. 00 PAIRC=PRA
4050. 00 TAIRC=TRA
4060. 00 RHC=RHRA

```



```

***** BMAIN *****
4060. 00
4070. 00
4080. 00
4090. 00
4100. 00
4110. 00
4120. 00
4130. 00
4140. 00
4150. 00
4160. 00
4170. 00
4180. 00
4190. 00
4200. 00
4210. 00
4220. 00
4230. 00
4240. 00
4250. 00
4260. 00
4270. 00
4280. 00
4290. 00
4300. 00
4310. 00
4320. 00
4330. 00
4340. 00
4350. 00
4360. 00
4370. 00
4380. 00
4390. 00
4400. 00
4410. 00
4420. 00
4430. 00
4440. 00
4450. 00
4460. 00
4470. 00
4480. 00
4490. 00
4500. 00
4510. 00
4520. 00
4530. 00
4540. 00
4550. 00
4560. 00
4570. 00
4580. 00
4590. 00
4600. 00
4610. 00
4620. 00
4630. 00

```

```

      PAIRE=POA
      TAIRE=TOA
      RHC=RHOA
    ELSE
      CAPID=CAPID1
      NCPL=NCPL1
      PAIRE=TRA
      TAIRE=TRA
      RHE=RHRA
      PAIRC=POA
      TAIRC=TOA
      RHC=RHOA
    END IF
      WRITE(6,941)
      READ(5,777)ITER
      WRITE(6,954)
      READ(5,777)ITERXW
      IF(ITER.EQ.0)ITERXW=0
      WRITE(6,927)
      READ(5,777)XWORG
      WRITE(6,928)
      READ(5,777)XWW
      WRITE(6,908)
      READ(5,777)TG3,X3,TSUP3,TG6
      XW=1.-XWW
      CALL AIRPR(1,TRA,PRA,RHRA,W,CT,R,AM,AT)
      V=R*(459.67+TRA)/144./PRA
      CF11=ANAS(1)*V/60.
      CALL AIRPR(1,TOA,POA,RHOA,W,CT,R,AM,AT)
      V=R*(459.67+TOA)/144./POA
      CF12=ANAS(2)*V/60.
      IACC=0
      RTGG=0.
      PDR11G=8.
      FILTER=0.
      SPAD=0.
*****
***** MAIN ITERATION PROCESS *****
*****
C ***** START MIXTURE COMPOSITION LOOP
      DO 250 NXW=1,4
        X3MIN=0.8
        X3MAX=1.
        XM=XV/(W2/W1*(1.-XW)+XW)
        WM=W1*(1.-XM)+W2*XM
C *****
C ***** START REFRIGERANT MASS CONSERVATION LOOP
      DO 200 NMASS=1,12
C *****
C ***** START ENTHALPY LOOP
      DO 130 ITH=1,8
        TKG3=(TG3+459.67)/1.8
        CALL DEWPRE(NO,TKG3,XM,PA3,XL)
        P3=PA3*14.6959
        IF(X3.LT.1.)THEN
          CALL PXGIN2(XM,PA3,X3,TK3,XL,XV,VL,VV,H3)

```

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***** BMAIN *****
4640. 00 T3=TK3*1.8-459.67
4650. 00 ELSE
4660. 00 T3=T63+TSUP3
4670. 00 TK3=(T3+459.67)/1.8
4680. 00 CALL VCLIT1(N1,TK3,PA3,XM,V)
4690. 00 CALL HCVCP(N1,TK3,V,XI1,H3,CV,CP)
4700. 00 END IF
4710. 00
4720. 00 C
4730. 00 C**** START REFRIGERANT FLOW RATE LOOP
4740. 00 INTERM=1
4750. 00 DO 120 ICLAR=1,3
4760. 00 DO 112 ITM=1,8
4770. 00 IF (INTERM.EQ.0) GOTO 114
4780. 00 CALL COMPTE(NSYS,TG6,TRA,TOA,EI,RMASS)
4790. 00
4800. 00 C
4810. 00 CALL CONDIX(NSYS,RMASS,T10,P10,TAIRC,PAIRC,RHC,T11,H11,X11)
4820. 00
4830. 00 C
4840. 00 X12A=X12
4850. 00 CALL BLINE(RYD,RYL,RMASS)
4860. 00
4870. 00 C
4880. 00 P13=P1+PDR113
4890. 00 POUT=P13
4900. 00 XMASS=RMASS
4910. 00 CALL CAPIL(XW,XMASS,P12,H12,FOUT,CAPL,CAPID,NCPL)
4920. 00
4930. 00 C
4940. 00 DRMS2=RMASS-XMASS
4950. 00 INTERM=0
4960. 00 IF (ABS(DRMS2/RMASS).LT.0.005) GOTO 114
4970. 00 INTERM=1
4980. 00
4990. 00 C
5000. 00 IF (ITM.EQ.1) THEN
5010. 00 TG6X=TG6
5020. 00 IF (FILTER.NE.0.) THEN
5030. 00 TG6CH=-FILTER*DRMS2
5040. 00 IF (ABS(TG6CH).GT.8.) TG6CH=SIGN(8,TG6CH)
5050. 00 ELSE
5060. 00 TG6CH=SIGN(2.5,DRMS2)
5070. 00 END IF
5080. 00 TG6=TG6X+TG6CH
5090. 00 GOTO 108
5100. 00 END IF
5110. 00
5120. 00 C
5130. 00 FILTER=(TG6X-TG6)/(DRMS21-DRMS2)
5140. 00 DDR=DRMS2/(DRMS21-DRMS2)
5150. 00 IF ((X12+X12A).EQ.0.) OR (X12*X12A).NE.0.) THEN
5160. 00 IF (ABS(DRMS2/RMASS).GT.0.05) GOTO 105
5170. 00 TG6=TG6-DDR*(TG6X-TG6)
5180. 00 RMASS=RMASS-DDR*(RMASS1-RMASS)
5190. 00 H12=H12-DDR*(H121-H12)
5200. 00 IF (ICLAR.NE.1) GOTO 112
5210. 00 GOTO 114
5220. 00 END IF
5230. 00
5240. 00 C
5250. 00 105 TG6Y=TG6
5260. 00 TG6CH=-DDR*(TG6X-TG6)
5270. 00 IF (ABS(TG6CH).GT.8.) TG6CH=SIGN(8.,TG6CH)
5280. 00 TG6=TG6Y+TG6CH

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***** BMAIN *****

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5220. 00      TG6X=TG6Y
5230. 00      RMASS1=RMASS
5240. 00      H121=H12
5250. 00      DRMS21=DRMS2
5260. 00      CONTINUE @@@@@@@@@@@@@@@@@@
5270. 00      WRITE(6,113)DRMS2,TG6CH
5280. 00      113 FORMAT(// ' ITH LOOP DID NOT CONVERGE, DRMS2=',F6.2, ' TG6CH=',F6.2)
5290. 00      114 CONTINUE
5300. 00      H13=H12
5310. 00      TK13=(T1+459.67)/1.8
5320. 00
5330. 00      C***** ITERATE EVAPORATOR TO CLOSE ENTHALPY LOOP
5340. 00      DO 116 ITE=1,6
5350. 00      PA13=P13/14.6959
5360. 00      CALL HPIN(N1,H13,PA13,XM,0.003,TK13,X13,XL,XV,VL,VV)
5370. 00      T13=TK13*1.8-459.67
5380. 00      NSY=2/NSYS
5390. 00      CALL EVAPHX(NSY,RM'ASS,T13,P13,TAIRE,PAIRE,RHE,X13,T1E,P1E,H1E,X1E)
5400. 00      P1EP1=P1E-P1
5410. 00      INTERE=0
5420. 00      IF(ABS(P1EP1).LT.0.2)GOTO 118
5430. 00      INTERE=1
5440. 00      TV=P13
5450. 00      IF(ITE.EQ.1)THEN
5460. 00      P13=P13-P1EP1
5470. 00      IF(SPAD.NE.0.)P13=TV-P1EP1*SPAD
5480. 00      ELSE
5490. 00      SPAD=(P13-P13F)/(P1EP1-P1EP1F)
5500. 00      P13=P13-P1EP1*SPAD
5510. 00      H1EE=H1E-P1EP1*(H1E-H1F)/(P1EP1-P1EP1F)
5520. 00      END IF
5530. 00      P13F=TV
5540. 00      P1EP1F=P1EP1
5550. 00      H1F=H1E
5560. 00      116 CONTINUE
5570. 00      WRITE(6,117)
5580. 00      117 FORMAT(// ' ITH LOOP FAILED TO CONVERGE=' )
5590. 00      INTERE=1
5600. 00      H1E=H1EE
5610. 00
5620. 00      C
5630. 00      118 PDR113=P13-P1
5640. 00      DENT2=H1E-H1
5650. 00      PRINT 640,P1,P1E,P1EP1,DENT2
5660. 00      640 FORMAT(' P1,P1E,P1EP1,DENT2=',4F10.4)
5670. 00      IF(ABS(DENT2).GT.0.6)GOTO 123
5680. 00      PRINT 3020,INTERM,INTERE,POUT,P13
5690. 00      3020 FORMAT(' INTERM,INTERE,POUT,P13=',2I4,2F8.2)
5700. 00      IF(INTERM.NE.0)GOTO 120
5710. 00      IF(INTERE.NE.0)GOTO 120
5720. 00      IF(POUT-P13).LT.-0.5)GOTO 120
5730. 00      GOTO 140
5740. 00      120 CONTINUE
5750. 00      WRITE(6,642)
5760. 00      642 FORMAT(' LOOP 120 DID NOT CONVERGE, RUN TERMINATED')
5770. 00      GOTO 999
5780. 00      C
5790. 00      C***** ITERATE INPUT DATA FOR NEXT LOOP (ITH)
5800. 00      123 IF(ITH.EQ.1)THEN

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```

***** BMAIN *****
5800. 00 DENT1=DENT2
5810. 00 TG3A=TG3
5820. 00 TG6A=TG6
5830. 00 IF (RTG6.NE.0.) THEN
5840. 00 TG3=TG3-DENT2*RTG3
5850. 00 TG6=TG6-DENT2*RTG6
5860. 00 ELSE
5870. 00 TCH=SIGN(2.,DENT2)
5880. 00 TG3=TG3+TCH
5890. 00 TG6=TG6+TCH/2.
5900. 00 END IF
5910. 00 ELSE
5920. 00 RTG3=(TG3A-TG3)/(DENT1-DENT2)
5930. 00 RTG6=(TG6A-TG6)/(DENT1-DENT2)
5940. 00 IF (ABS(DENT1).LT.ABS(DENT2)) GO TO 124
5950. 00 DENT1=DENT2
5960. 00 TG3A=TG3
5970. 00 TG6A=TG6
5980. 00 124 TG3=TG3A-DENT1*RTG3
5990. 00 TG6=TG6A-DENT1*RTG6
6000. 00 END IF
6010. 00 130 CONTINUE
6020. 00 WRITE(6,133) DENT2
6030. 00 133 FORMAT(// ' ITH LOOP DID NOT CONVERGE, DENT2=',F8.3)
6040. 00 140 CONTINUE
6050. 00 C*****
6060. 00 C**** END OF ENTHALPY AND FLOW RATE LOOP
6070. 00 C**** DO REFRIGERANT MASS INVENTORY
6080. 00 PMASS1=BPMASS(RD,RL,P1,T1,P2,T2)
6090. 00 PMASS2=DPMASS(YD,YL,P9,T9,P10,T100)
6100. 00 CALL WACCUM(T3,P3,PMASS,PMASS3,XWA)
6110. 00 PM1.S4=BPMASS(1.128,VCAN,P4,T4,P4,T4)
6120. 00 PMASS5=BPMASS(RYD,RYL,P11,T11,P12,T12)
6130. 00 PMASS1=PMASS1+PMASS2+PMASS3+PMASS4+PMASS5
6140. 00 TMASS=PMASS1+PMASS(1)+BIMASS(2)
6150. 00 C**** PRINT RESULTS: MASS INVENTORY, REF. STATES, PERFORMANCE
6160. 00 WRITE(6,862) TMASS,REFIN
6170. 00 CALL ENTRO2(XV,T1,P1,S1,XQ)
6180. 00 CALL ENTRO2(XW,T2,P2,S2,XQ)
6190. 00 CALL ENTRO2(XV,T3,P3,S3,XQ)
6200. 00 CALL ENTRO2(XW,T4,P4,S4,XQ)
6210. 00 CALL ENTRO2(XV,T5,P5,S5,XQ)
6220. 00 CALL ENTRO2(XW,T6,P6,S6,XQ)
6230. 00 CALL ENTRO2(XV,T7,P7,S7,XQ)
6240. 00 CALL ENTRO2(XW,T8,P8,S8,XQ)
6250. 00 CALL ENTRO2(XV,T9,P9,S9,XQ)
6260. 00 CALL ENTRO2(XW,T10,P10,S10,XQ)
6270. 00 CALL ENTRO2(XV,T11,P11,S11,XQ)
6280. 00 CALL ENTRO2(XW,T12,P12,S12,XQ)
6290. 00 CALL ENTRO2(XV,T13,P13,S13,XQ)
6300. 00 WRITE(6,864)
6310. 00 WRITE(6,800) ATITLE
6320. 00 WRITE(6,934) TOA,PHOA,TRA,RHRA
6330. 00 WRITE(6,947) CFM1,CFM2
6340. 00 WRITE(6,931)
6350. 00 INN=1
6360. 00 WRITE(6,933) INN,T1,P1,H1,S1,X1
6370. 00 INN=2

```



```

***** BMAIN *****
6380. 00
6390. 00
6400 00
6410. 00
6420. 00
6430. 00
6440. 00
6450. 00
6460. 00
6470. 00
6480. 00
6490. 00
6500. 00
6510. 00
6520. 00
6530. 00
6540. 00
6550. 00
6560. 00
6570. 00
6580. 00
6590. 00
6600. 00
6610. 00
6620. 00
6630. 00
6640. 00
6650. 00
6660. 00
6670. 00
6680. 00
6690. 00
6700. 00
6710. 00
6720. 00
6730. 00
6740. 00
6750. 00
6760. 00
6770. 00
6780. 00
6790. 00
6800. 00
6810. 00
6820. 00
6830. 00
6840. 00
6850. 00
6860. 00
6870. 00
6880. 00
6890. 00
6900. 00
6910. 00
6920. 00
6930. 00
6940. 00
6950. 00

```

```

WRITE(6,933)INN,T2,P2,H2,S2,X2
INN=3
WRITE(6,933)INN,T3,P3,H3,S3,X3
INN=4
WRITE(6,933)INN,T4,P4,H4,S4,X4
INN=5
WRITE(6,933)INN,T5,P5,H5,S5,X5
INN=6
WRITE(6,933)INN,T6,P6,H6,S6,X6
INN=7
WRITE(6,933)INN,T7,P7,H7,S7,X7
INN=8
WRITE(6,933)INN,T8,P8,H8,S8,X8
INN=9
WRITE(6,933)INN,T9,P9,H9,S9,X9
INN=10
WRITE(6,933)INN,T10,P10,H10,S10,X10
INN=11
WRITE(6,933)INN,T11,P11,H11,S11,X11
INN=12
WRITE(6,933)INN,T12,P12,H12,S12,X12
INN=13
WRITE(6,933)INN,T13,P13,H13,S13,X13
QLOAD=RMASS*(H10-H11)+3412.66*EIDFAN
IF(NSYS.EQ.2)QLOAD=RMASS*(H1-H13)-3412.66*EIDFAN
ELUSE=E1+E2DFAN+EIDFAN
COP=QLOAD/(3412.66*ELUSE)
WRITE(6,942)TG3,TSUP3,TGG,RMASS,TMASS
WRITE(6,943)QLOAD,ELUSE,COP
WRITE(6,945)XMW
IF(ITER.EQ.0)GOTO 260

C
C*** CHECK ON REFRIG. CHARGE REQUESTED
DMASS=TMASS-REFIN
C*** REFRIG. BALANCE NOT OBTAINED
C*** ITERATE INPUT DATA FOR THE NEXT LOOP (DMASS)
IF(ABS(DMASS/REFIN).LT.0.015)GOTO 202
IF(DMASS.LE.1)GOTO 164
158 DMASS1=DMASS
TSUP31=TSUP3
X31=X3
TG31=TG3
TGG1=TGG
CC+++ X3MAX=X3
X3MIN=X3
TG3MAX=TG3
TG3MIN=TG3
TGGMIN=TGG
TGGMAX=TGG
DMASS1=DMASS1
DMIN=DMASS1
CC+++ IF(DMASS.GT.0.)GOTO 160
IF(TSUP3.GT.0.)THEN
TSUP3=TSUP31-4.
TSUP3=AMAX1(0.,TSUP3)
TG3=TG31+(TSUP31-TSUP3)/4.
@MOVE TOWARDS SMALLER TSUP3 OR X3

```

***** BMAIN *****

```

6960.    TG6=TG61+(TSUP31-TSUP3)/2.
6970.    ELSE
6980.        X3=X31-0.02
6990.        TG3=TG31+1.
7000.        TG6=TG61+2.
7010.        X3MAX=X31
7020.        DMAX=DMASS1
7030.        TG3MAX=TG31
7040.        TG6MAX=TG61
7050.    END IF
7060.    GOTO 190
7070.    160 IF(X3.LT.1.)THEN @MOVE TOWARDS HIGHER TSUP3 OF X3
7080.        X3MIN=X31
7090.        DMIN=DMASS1
7100.        TG3MIN=TG31
7110.        TG6MIN=TG61
7120.        X3=X31+0.02
7130.        X3=AMIN1(1.,X3)
7140.        TG3=TG31-(X3-X31)/0.02
7150.        TG6=TG61-(X3-X31)/0.01
7160.    ELSE
7170.        IF(NXW.NE.1)THEN
7180.            WRITE(6,956)
7190.            GOTO 202
7200.        END IF
7210.        TSUP3=TSUP31+4.
7220.        TG3=TG31-1.
7230.        TG6=TG61-2.
7240.    END IF
7250.    GOTO 196
7260.    C***** INTERPOLATION *****
7270.    164 DMAS2=DMASS
7280.        TSUP32=TSUP3
7290.        X32=X3
7300.        TG32=TG3
7310.        TG62=TG6
7320.        FACT=-DMASS2/(DMASS1-DMASS2)
7330.    C
7340.        IF(DMASS.GT.0.)GOTO 166
7350.        IF(TSUP3.GT.0.)THEN @MOVE TOWARDS SMALLER TSUP3 OR X3
7360.            TSUPCH=(TSUP31-TSUP32)*FACT
7370.            IF(TSUPCH.LT.-10.)THEN
7380.                TSUPCH=-10.
7390.                FACT=-10./(TSUP31-TSUP32)
7400.            END IF
7410.            TSUP3=TSUP3+TSUPCH
7420.            IF(TSUP3.LT.0.)THEN
7430.                TSUP3=0.
7440.                X3=1.
7450.                FACT=-TSUP32/(TSUP31-TSUP32)
7460.            END IF
7470.        ELSE
7480.            X3MAX=X3
7490.            TG3MAX=TG3
7500.            TG6MAX=TG6
7510.            DMAX=DMASS
7520.            IF(X3MIN.LT.X3MAX)THEN
7530.                X3=0.5*(X3MIN+X3MAX)

```

***** BMAIN *****

```

7540. 00      TG3=0.5*(TG3MIN+TG3MAX)
7550. 00      TG6=0.5*(TG6MIN+TG6MAX)
7560. 00      ELSE
7570. 00      X3=X3MAX-0.015
7580. 00      END IF
7590. 00      IF((ABS(DMASS1-DMASS2)/REFIN).LT.0.005)THEN
7600. 00      XCH=-0.01
7610. 00      IF(X3MIN.EQ..8)GOTO 165
7620. 00      X3=X3MIN+0.00199
7630. 00      X31=X3MIN
7640. 00      TG31=TG3MIN
7650. 00      TG51=TG5MIN
7660. 00      DMASS1=D1MIN
7670. 00      TG3=TG3MIN
7680. 00      TG6=TG6MIN
7690. 00      GOTO 198
7700. 00      END IF
7710. 00      XCH=(X31-X32)*FACT
7720. 00      IF(XCH.LT.-0.06)THEN
7730. 00      XCH=-0.06
7740. 00      FACT=-0.06/(X31-X32)
7750. 00      END IF
7760. 00      X3=X32+XCH
7770. 00      IF(X3.LT.X3MIN)THEN
7780. 00      FACT=-DMASS2/(DM1MIN-DMASS2)
7790. 00      X3=X32+(X3MIN-X32)*FACT
7800. 00      END IF
7810. 00      END IF
7820. 00      GOTO 170
7830. 00
7840. 00      C 166 IF(X3.LT.1.)THEN
7850. 00      X3MIN=X3
7860. 00      TG6MIN=TG6
7870. 00      TG3MIN=TG3
7880. 00      DMMIN=DMASS
7890. 00      IF(X3MAX.GT.X3MIN)THEN
7900. 00      X3=0.5*(X3MIN+X3MAX)
7910. 00      TG3=0.5*(TG3MIN+TG3MAX)
7920. 00      TG6=0.5*(TG6MIN+TG6MAX)
7930. 00      ELSE
7940. 00      X3=X3MIN+0.015
7950. 00      X3=AMIN1(1.,X3)
7960. 00      END IF
7970. 00      X3=X32+(X31-X32)*FACT
7980. 00      IF(X3.GT.1.)THEN
7990. 00      X3=1.0001
8000. 00      FACT=(1.-X32)/(X31-X32)
8010. 00      END IF
8020. 00      IF(X3.GT.X3MAX)THEN
8030. 00      FACT=-DMASS2/(DMMAX-DMASS2)
8040. 00      X3=X32+(X3MAX-X32)*FACT
8050. 00      END IF
8060. 00      ELSE
8070. 00      IF(NXW.NE.1)THEN
8080. 00      WRITE(6,956)
8090. 00      GOTO 202
8100. 00      END IF
8110. 00      IF(TSUP31.EQ.0..AND.TSUP32.EQ.0.)GOTO 158

```

REMOVE TOWARDS HIGHER TSUP3 OR X3

***** BMAIN *****

```

8120. 00 TSUPCH=(TSUP31-TSUP32)*FACT
8130. 00 IF (TSUPCH.GT.10.) THEN
8140. 00 TSUPCH=10.
8150. 00 FACT=10./(TSUP31-TSUP32)
8160. 00
8170. 00 END IF
8180. 00 TSUP3=TSUP3+TSUPCH
8190. 00 END IF
8200. 00 170 TG3=TG32*(TG31-TG32)*FACT
8210. 00 TG6=TG62*(TG61-TG62)*FACT
8220. 00 DMASS1=DMASS2
8230. 00 TSUP31=TSUP32
8240. 00 X31=X32
8250. 00 TG31=TG32
8260. 00 TG61=TG62
8270. 00 196 IF (X3.LT.0.80.OR.TSUP3.GT.55.) THEN
8280. 00 WRITE(6,944)
8290. 00 GOTO 202
8300. 00 198 END IF
8310. 00 C+ 198 IF ((X3MAX-X3MIN).LT..002) THEN
8320. 00 C+ WRITE(6,950)X3MAX,X3MIN
8330. 00 C+ GOTO 202
8340. 00 C+ END IF
8350. 00 200 CONTINUE
8360. 00 WRITE(6,201)DMASS
8370. 00 201 FORMAT('/// NMASS LOOP DID NOT CONVERGE, DMASS=',F9.3)
8380. 00 C**** END OF REFRIGERANT MASS CONSERVATION LOOP
8390. 00 C
8400. 00 C*****
8410. 00 202 IF (ITERXV.EQ.0) GOTO 250
8420. 00 XWCIR=(TMASS*(1.-XWORG)-PMASS3*XVA)/(TMASS-PMASS3)
8430. 00 XWCIRW=1.-XWCIR
8440. 00 PRINT 957,XWCIRW
8450. 00 XWDIF=XW-XWCIR
8460. 00 IF (ABS(XWDIF).LT.0.01) GOTO 250
8470. 00 IF (XW.EQ.1) THEN
8480. 00 XW1=XW
8490. 00 XW=XW-.5*(XW-XWCIR)
8500. 00 XWDIF1=XWDIF
8510. 00 ELSE
8520. 00 FOCI=(XW-XW1)/(XWDIF-XWDIF1)
8530. 00 XW0=XW-XWDIF*FOCI
8540. 00 XW1=XW
8550. 00 XW=XW0
8560. 00 XW=AMIN1(XW,.95)
8570. 00 XW=AMAX1(XW,.05)
8580. 00 IF (XW.EQ.XW1) GOTO 260
8590. 00 XWDIF1=XWDIF
8600. 00 END IF
8610. 00 XW=1.-XW
8620. 00 PRINT 958,XW
8630. 00 250 CONTINUE
8640. 00 C**** END OF MIXTURE COMPOSITION LOOP
8650. 00 C
8660. 00 WRITE(6,955)
8670. 00 260 CONTINUE
8680. 00 C**** END OF EVALUATION OF PERFORMANCE
8690. 00 C**** OF VAPOR COMPRESSION CYCLE
8690. 00 C**** WRITE(6,953)

```


***** BMAIN *****

```

9280. 00 914 FORMAT(/2X,'FPCH(1)',4X,'FTK(1)',5X,'FMK(1)',5X,
9290. 00 &'TMK(1)',5X,'AMAS(1)')
9300. 00 915 FORMAT(/2X,'NO. OF REPEATING SECTIONS:',I3)
9310. 00 916 FORMAT(/2X,'NTUB(1):',515//2X,'IFROM(1):')
9320. 00 917 FORMAT(10I5)
9330. 00 918 FORMAT(/2X,'CONST(1)',3X,'CPGW(1)',4X,'ANGLE(1)')
9340. 00 919 FORMAT(/2X,'OUTDOOR COIL DATA:')
9350. 00 920 FORMAT(/2X,'NDEF(2)',4X,'NROW(2)'/14,I11)
9360. 00 921 FORMAT(/2X,'DI(2)',6X,'DO(2)',6X,'DT(2)',6X,'RPCH(2)',4X,
9370. 00 &'DPCH(2)',4X,'WIDTH(2)')
9380. 00 922 FORMAT(/2X,'FPCH(2)',4X,'FTK(2)',5X,'FMK(2)',5X,
9390. 00 &'TMK(2)',5X,'AMAS(2)')
9400. 00 923 FORMAT(/2X,'CONST(2)',3X,'CPGW(2)',4X,'ANGLE(2)')
9410. 00 924 FORMAT(/2X,'NTUB(2):',515//2X,'IFROM(2):')
9420. 00 926 FORMAT(/2X,'OUTDOOR & INDOOR AIR CONDITIONS',
9430. 00 /2X,'POA,TOA,RHQA,PRA,TRA,RHQA=?')
9440. 00 927 FORMAT(/2X,'COMPOSITION OF CHARGED REFRIGERANT =',
9450. 00 1 /2X,'(WEIGHT FRACTION OF MORE VOLATILE COMPONENT)')
9460. 00 928 FORMAT(/2X,'COMPOSITION OF CIRCULATING REFRIGERANT =',
9470. 00 1 /2X,'(WEIGHT FRACTION OF MORE VOLATILE COMPONENT)')
9480. 00 929 FORMAT(/2X,'RL',7X,'RD',6X,'RK1',6X,
9490. 00 &'RD1',6X,'RK2',6X,'RD2')
9500. 00 931 FORMAT(/2X,'RESULTS:',/2X,'I',2X,'T',10X,'P',10X,'H',10X,
9510. 00 &'S',10X,'X')
9520. 00 933 FORMAT(13,5(1PE11.3))
9530. 00 934 FORMAT(/2X,'TOA',8X,'RHQA',7X,'TRA',8X,'RHEA'/4(1PE11.3))
9540. 00 935 FORMAT(/2X,'INDOOR FAN KW=',1PE11.3)
9550. 00 936 FORMAT(/2X,'OUTDOOR FAN KW=',1PE11.3)
9560. 00 940 FORMAT(/2X,'NSYS=1 FOR HEATING, NSYS=2 FOR COOLING MODE',
9570. 00 &4X,'NSYS=?')
9580. 00 941 FORMAT(/2X,'IS ITERATION OF SUPERHEAT/QUALITY REQUESTED ?',/
9590. 00 2X,'ITER=0 FOR NO, ITER=1 FOR YES, ITER=')
9600. 00 942 FORMAT(/2X,'TG3',8X,'TSU*3',6X,'TGG',8X,'RMAS',6X,'TMASS'/
9610. 00 &5(1PE11.3))
9620. 00 943 FORMAT(/2X,'GLOAD',6X,'ELUSE',6X,'COP'/3(1PE11.3))
9630. 00 944 FORMAT(' QUALITY OR SUPERHEAT LIMIT EXCEEDED')
9640. 00 945 FORMAT(/1X,' REFRIG. COMPOSITION =',F7.3,/
9650. 00 &' (WEIGHT FRACTION OF MORE VOLATILE COMPONENT)',///)
9660. 00 946 FORMAT(/2X,'EXP. DEVICE:',8X,'ID',5X,'LENGTH',#'/
9670. 00 @ ' AT INDOOR COIL',F10.4,F8.2,16,/
9680. 00 @ ' AT OUTDOOR COIL',F9.4,F8.2,16)
9690. 00 947 FORMAT(/2X,'CFMIND',5X,'CFMCUT'/2(1PE11.3))
9700. 00 948 FORMAT(/2X,'LIQUID LINE')
9710. 00 949 FORMAT(/2X,'RYL',8X,'RYD')
9720. 00 950 FORMAT(' QUALITY X3 IS WITHIN THE RANGE',F6.3,'-',F5.3)
9730. 00 951 FORMAT(/2X,'CQ',11X,'CPDR',7X,'VCAN',7X,'REFIN')
9740. 00 952 FORMAT(/2X,'AUGT',7X,'DACC',7X,'DHOLE(1)',8X,'DHOLE(2)',
9750. 00 @ 3X,'DTUBE',6X,'IDIS')
9760. 00 954 FORMAT(/2X,'IS ITERATION OF MIXTURE COMPOSITION REQUESTED ?',/
9770. 00 &2X,'ITERXW=0 FOR NO, ITERXW=1 FOR YES, ITERXW=')
9780. 00 955 FORMAT(/2X,' CONVERGENCE NOT OBTAINED IN COMPOSITION LOOP')
9790. 00 956 FORMAT(/2X,' GUESSED COMPOSITION TO RICH IN MORE VOLATILE',
9800. 00 &' COMPONENT')
9810. 00 957 FORMAT('/' FOR THE AMOUNT OF REFRIGERANT STORED IN THE ACCUMULA
9820. 00 &'TOR',/ ' CIRCULATING COMPOSITION SHOULD BE',F7.3)
9830. 00 958 FORMAT(/
9840. 00 &' NEW CALCULATED COMPOSITION FOR THE NEXT LOOP IS',F7.3,/
9550. 00 &' (WEIGHT FRACTION OF MORE VOLATILE COMPONENT)')//,

```

DATE 072184

```
***** BMAIN *****
9860.      00      & ' ***** ' )
9870.      00      999 STOP
9880.      00      END

END ELT.  ERRORS: NONE.  TIME:  1.146 SEC.  IMAGE COUNT: 987

@HDG,P  ***** BPMASS ***** .L,0
```

***** BPMASS *****

@ELT,L 00.BPMASS

ELT 8R1 S74Q1C 07/21/84 15:54:55 (0)

FUNCTION BPMASS(DI,DL,P1,T1,P2,T2)

```

10. 00 C
20. 00 C**** PURPOSE
30. 00 C TO COMPUTE MASS OF A NON-AZEOTROPIC REFRIGERANT
40. 00 C IN A TUBE
50. 00 C (HOMOGENEOUS FLOW ASSUMED)
60. 00 C
70. 00 C**** INPUT DATA:
80. 00 C DI - INNER DIAMETER OF TUBE (FT)
90. 00 C DL - LENGTH OF TUBE (FT)
100. 00 C P1 - REFRIG. INLET PRESSURE (PSIA)
110. 00 C P2 - REFRIG. OUTLET PRESSURE (PSIA)
120. 00 C T1 - REFRIG. INLET TEMPERATURE (F)
130. 00 C T2 - REFRIG. OUTLET TEMPERATURE (F)
140. 00 C XW - WEIGHT COMPOSITION (FRACTION OF LESS VOLATILE COMPONENT)
150. 00 C
160. 00 C**** OUTPUT DATA:
170. 00 C BPMASS - REFRIG. MASS IN TUBE (LB)
180. 00 C
190. 00 C**** SUBPROGRAMS CALLED BY BPMASS:
200. 00 C TPROP
210. 00 C
220. 00 C COMMON/RDATA3/XW,XM,WM
230. 00 C
240. 00 C AREA=3.14159*DI*DI/4.
250. 00 C VTUBE=AREA*DL
260. 00 C CALL TPROP(T1,P1,XW,XQ,H,V1)
270. 00 C CALL TPROP(T2,P2,XW,XQ,H,V2)
280. 00 C BPMASS=2.*VTUBE/(V1+V2)
290. 00 C RETURN
300. 00 C
310. 00 C END

```

END ELT. ERRORS: NONE. TIME: 0.075 SEC. IMAGE COUNT: 31

@HDG,P ***** BSIMP ***** .L,0

***** BSIMP *****

@ELT,L DD.BSIMP

ELT 8R1 S74QIC 07/21/84 15:54:55 (0)

FUNCTION BSIMP(FX,A1,A2,F,MAXIT,H1,GN,XM)

C-----

C NUMERICAL INTEGRATION USING SIMPSON-S 1/3 RULE

C-----

C INITIALIZATIONS

REAL*8 FX

C

S1=0.

PR=0.

X=A2

H=A1-A2

M=2

C LOOP TO COUNT THE MAXIMUM NUMBER OF ITERATIONS

DO 6 J=0,MAXIT

S=0.

C LOOP TO COUNT THE NUMBER OF FUNCTION EVALUATIONS

DO 1 I=1,M

C SUM THE FUNCTION EVALUATIONS

S=S+FX(XM,X,H1,GN)

C INCREMENT X

1 X=X+H

C OBTAIN NEW VALUE OF INTEGRATION

BSIMP=(2.*S+S1)*H/3.

IF(J-1) 2,4,3

C FIRST LOOP. SET M TO 1 AND DIVIDE THE FUNCTION EVALUATIONS

C BY 2

2 M=1

S1=-S/2.

GO TO 5

C CHECK ERROR CONTROL

3 PR=PR-BSIMP

PR=PR/BSIMP

C EVALUATION WITHIN ERROR LIMITS. FINISH

IF(ABS(PR).LT.E) GO TO 7

C SECOND ITERATION. HALVE H AND DOUBLE M

4 H=H*.5

M=2*M

C OBTAIN NEW LOWER FOR FUNCTION EVALUATIONS

5 X=A2+.5*H

C PREVIOUS VALUE OF INTEGRATION

PR=BSIMP

6 S1=S1+S

7 RETURN

END

END ELT. ERRORS: NONE. TIME: 0.096 SEC. IMAGE COUNT: 44

@HDG,P ***** BUBPRE ***** .L,0

```

***** BUBPRE *****
@ELT,L DD,BUBPRE
ELT 8R1 S74Q1C 07/21/84 15:54:56 (0)
10. 00 SUBROUTINE BUBPRE(IG,T,X,P,XV)
20. 00 C
30. 00 C
40. 00 C
50. 00 C
60. 00 C
70. 00 C
80. 00 C
90. 00 C
100. 00 C
110. 00 C
120. 00 C
130. 00 C
140. 00 C
150. 00 C
160. 00 C
170. 00 C
180. 00 C
190. 00 C
200. 00 C
210. 00 C
220. 00 C
230. 00 C
240. 00 C
250. 00 C
260. 00 C
270. 00 C
280. 00 C
290. 00 C
300. 00 C
310. 00 C
320. 00 C
330. 00 C
340. 00 C
350. 00 C
360. 00 C
370. 00 C
380. 00 C
390. 00 C
400. 00 C
410. 00 C
420. 00 C
430. 00 C
440. 00 C
450. 00 C
460. 00 C
470. 00 C
480. 00 C
490. 00 C
500. 00 C
510. 00 C
520. 00 C
530. 00 C
540. 00 C
550. 00 C
560. 00 C

*****
PURPOSE:
TO CALC. BUBBLE POINT PRESSURE OF BINARY MIXTURE
FROM GIVEN TEMPERATURE AND COMPOSITION

***** INPUT:
IG = 0, IF GUESS OF PRESSURE IS NOT GIVEN
= 1, IF GUESS OF PRESSURE IS GIVEN
T - TEMPERATURE (K)
P - GUESS OF PRESSURE, OPTIONAL (STD ATM)
X - MOLAR CONCENTRATION (FRACTION OF LESS VOLATILE COMPONENT)

***** OUTPUT:
P - BINARY MIXTURE PRESSURE AT BUBBLE POINT (STD ATM)
XV - MOLAR CONCENTRATION OF LESS VOLATILE COMPONENT
IN SAT. VAPOR IN EQUILIBRIUM WITH LIQUID (-)

***** SUBPROGRAMS CALLED BY BUBPRE:
EBUBFR,QLITY

DATA SLOPE/0.,TLAST/0./

IF(ABS(T-TLAST).GT.1.E-3)GOTO 10
IF(ABS(X-XLAST).GT.1.E-3)GOTO 10
P=PLAST
XV=XVLAST
RETURN

10 IF(IG.NE.1)P=EBUEPR(T,X)
DO 50 I=1,20
CALL QLITY(T,P,X,XQ,XV,XL)
P2=P
XDIF2=X-XL
IF(ABS(XDIF2).LT.0.00001)GOTO 100
IF(I.NE.1)GOTO 20

15 P1=P2
XDIF1=XDIF2
IF(SLOPE.NE.0.)THEN
DP=XDIF2*SLOPE
IF(ABS(DP).GT.1.)DP=SIGN(1.,DP)
P=P2-DP
GOTO 50
END IF
P=P2+0.5
IF(XDIF2.GT.0.)P=P2-0.5
GOTO 50

20 IF(XDIF1.EQ.XDIF2)GOTO 15
SLOPE=(P2-P1)/(XDIF2-XDIF1)
IF(ABS(XDIF1).LT.ARS(XDIF2))GOTO 30
P1=P2
XDIF1=XDIF2
DP=XDIF1*SLOPE
IF(ABS(DP).GT.1.)DP=SIGN(1.,DP)

30 DP=XDIF1*SLOPE
IF(ABS(DP).GT.1.)DP=SIGN(1.,DP)

```

***** BUBPRE *****

```
570.      00      P=P1-DP
580.      00      50 CONTINUE
590.      00      WRITE(6,600)XDIF2
600.      00      600 FORMAT(' ERROR 600 IN BUBPRE, XDIF2=',1PE12.4)
610.      00      100 TLAST=T
620.      00      XLAST=X
630.      00      PLAST=P
640.      00      XVLAST=XV
650.      00      RETURN
660.      00      END
```

END ELT. ERRORS: NONE. TIME: 0.128 SEC. IMAGE COUNT: 66

@HDG,P ***** BUBTEM ***** .L,0

247

***** BUBTEM *****

```
560. 00 IF(ABS(DT).GT.10.)DT=SIGN(10.,DT)
570. 00 T=T1-DT
580. 00 50 CONTINUE
590. 00 WRITE(6,600)IG,P,X,TIN,XDIF2,DT
600. 00 600 FORMAT(' ERROR 600 IN BUBTEM, IG,P,X,TIN,XDIF2,DT=',12,5F10.6)
610. 00 100 PLAST=P
620. 00 XLAST=X
630. 00 TLAST=T
640. 00 XVLAST=XV
650. 00 RETURN
660. 00 END
```

END ELT. ERRORS: NONE. TIME: 0.138 SEC. IMAGE COUNT: 67

@HGG,P ***** CAPIL ***** .L,0

***** CAPIL *****

```

@ELT,L DD,CAPIL
ELT 8R1 S7401C 07/21/84 15:54:56 (0)
SUBROUTINE CAPIL(XW,XMASS,P1,H1,P2,TL,D,NC)
10. 00 C
20. 00 C
30. 00 C
40. 00 C
50. 00 C
60. 00 C
70. 00 C
80. 00 C
90. 00 C
100. 00 C
110. 00 C
120. 00 C
130. 00 C
140. 00 C
150. 00 C
160. 00 C
170. 00 C
180. 00 C
190. 00 C
200. 00 C
210. 00 C
220. 00 C
230. 00 C
240. 00 C
250. 00 C
260. 00 C
270. 00 C
280. 00 C
290. 00 C
300. 00 C
310. 00 C
320. 00 C
330. 00 C
340. 00 C
350. 00 C
360. 00 C
370. 00 C
380. 00 C
390. 00 C
400. 00 C
410. 00 C
420. 00 C
430. 00 C
440. 00 C
450. 00 C
460. 00 C
470. 00 C
480. 00 C
490. 00 C
500. 00 C
510. 00 C
520. 00 C
530. 00 C
540. 00 C
550. 00 C

***** PURPOSE:
TO COMPUTE MASS FLOW RATE OF NON-AZEOTROPIC MIXTURE
THROUGH EXPANSION DEVICE OF CONSTANT FLOW AREA

***** INPUT DATA:
D - EXPANSION DEVICE DIAMETER (FT)
NC - NUMBER OF EXPANSION DEVICES (-)
H1 - REFRIG. ENTHALPY BEFORE EXP. DEVICE (BTU/LBM)
P1 - REFRIG. PRESSURE BEFORE EXP. DEVICE (PSIA)
P2 - EVAPORATOR INLET PRESSURE (PSIA)
TL - EXPANSION DEVICE LENGTH (FT)
XMASS - MASS FLOW RATE THROUGH NC EXP. DEVICES (GUESS) (LB/HR)
XW - COMPOSITION (WEIGHT FRACTION OF LESS VOLATILE COMPONENT)

***** OUTPUT DATA:
P2 - REFRIG. PRESSURE AT OUTLET OF EXP. DEVICE (PSIA)
XMASS - MASS FLOW RATE THROUGH NC EXP. DEVICES (LB/HR)

***** SUBPROGRAMS CALLED BY CAPIL:
BSIMP, CIOKE, DDEIFA, DFANNO, HFPROP, PFLASH, VISCON

COMMON/RDATA2/W1,W2,TC1,TC2
EXTERNAL DDENFA
REAL*8 DFANNO,S1,S2,DDENFA

NO=0
PRINT 44, P1,H1,P2
IF(P1.LE.P2)GOTO 50
***** @ERROR IN CALLING CAPIL
PRELIMINARY CALCULATIONS
PA1=P1/14.6959
XM=XW/(W2/W1*(1.-XW)+XW)
AREA=3.141593*D*D/4.
G=XMASS/AREA/3600./FLCAT(NC)
IF(3.LT.200.)G=200.
IF(TL.LT.0.2.AND.G.LT.1000.)G=1000.
GMIN=0.
GMAY=5.E4
***** FIND REFRIG. COND. BEFORE EXPANSION DEVICE
ACC=0.0005
CALL HFPROP(NO,H1,PA1,XM,ACC,TK1,XIN,XML,YMV,VLM,VVM,
@ VM,V,CP,CV,AM,AK)
TIN=TK1*1.8-459.67
IF(XIN.GT.0.)THEN
VL=V
FVL=TL
PFLA=P1
TFJA=TIN
GOTO 20
END IF
E=G
DO 4 LIQ=1,15
EE=E+E
EN=EE/(64.4*778.104)

```

```

***** CAPIL *****
560. 00 CALL FFLASH(XM,H1,EN,TIN,TFLA,PFLA,VL)
570. 00 POUT=AMAX1(PFLA,P2)
580. 00 ZF=VISCOS(1,TIN,XM)
590. 00 RE=3500.*E*D/ZF
600. 00 FF=16./RE
610. 00 IF(RE.GT.2000.)FF=0.045/RE*.0.2
620. 00 PDL=EE*FF*TL*VL*2./(32.2*144.*D)
630. 00 PDE=1.15*EE*VL/(64.4*144.)
640. 00 P1A=POUT+PDL+PDE
650. 00 P1P1A=P1-P1A
660. 00 IF(ABS(P1P1A).LT.0.02)GOTO 6
670. 00 IF(LIQ.EQ.1)GOTO 2
680. 00 EY=E-P1P1A*(EB-E)/(P1P1B-P1P1A)
690. 00 EB=E
700. 00 E=EY
710. 00 GOTO 3
720. 00 2 EB=E
730. 00 E=.8*E
740. 00 IF(P1P1A.GT.0.)E=1.5*E
750. 00 3 P1P1B=P1P1A
760. 00 4 CONTINUE
770. 00 PRINT 5
780. 00 5 FORMAT(' CAPIL*ERROR 5')
790. 00 C
800. 00 6 CONTINUE
810. 00 G2=E
820. 00 IF(POUT.EQ.P2)GOTO 1000
830. 00 S1=DFANNO(XM,PFLA,H1,EN)
840. 00 S2=DFANNO(XM,PFLA-C.1,H1,EN)
850. 00 IF(S2.LT.S1)GOTO 1000
860. 00 GMIN=G2
870. 00 G=1.05*GMIN
880. 00 IF((P1-POUT).LT.1.)G=20.*G
890. 00 C**** START THE LOOP
900. 00 20 DO 40 IA=1,25
910. 00 DO 30 II=1,10
920. 00 GG=G*G
930. 00 GG=GG/(64.4*778.104)
940. 00 C**** FIND FLASHING PRESSURE
950. 00 IF(XIN.EQ.0.)CALL PFLASH(XM,H1,GN,TIN,TFLA,PFLA,VL)
960. 00 C**** FIND PRESSURE AT EXIT
970. 00 POUT=CHKE(XM,PFLA,P2,H1,GN)
980. 00 IF(POUT.EQ.PFLA)GOTO 32
990. 00 C**** FIND PRESSURE IN ENTRANCE
1000. 00 PINN=0.
1010. 00 DO 35 IEN=1,5
1020. 00 PIN=P1-1.15*GG*VL/(64.4*144.)
1030. 00 IF(PIN.GE.PFLA)GOTO 22
1040. 00 IF(PIN.LT.POUT)GOTO 32
1050. 00 VL=1./DDENFA(XM,PIN,H1,GN)
1060. 00 IF(ABS(PIN-PINN).LT.0.01)GOTO 21
1070. 00 PINN=PIN
1080. 00 35 CONTINUE
1090. 00 21 P2PH=PIN
1100. 00 FVL=TL
1110. 00 GOTO 23
1120. 00 C**** CALC. LENGHT OF SUBCOOLED LIQUID
1130. 00 22 P2PH=PFLA

```

```

***** CAPIL *****
1140. 00 ZF=VISCEN(1,TFLA,XW)
1150. 00 RE=2600.*G*D/ZF
1160. 00 FF=16./RE
1170. 00 IF(RE.GT.2000.)FF=0.046/RE*.2
1180. 00 FL=(32.2*(141.)*(PIN-PFLA)*D)/(2.*FF*GG*VL)
1190. 00 FVL=TL-FL
1200. 00 IF(FVL.LT.0.)THEN
1210. 00 G2=G
1220. 00 IF((FVL/TL).LT.-0.01)GOTO 41
1230. 00 GOTO 1000
1240. 00
1250. 00 ***** CALC. 2-PHASE FRICTION FACTOR
1260. 00 23 ZF=VISCEN(1,TFLA,XW)
1270. 00 ZG=VISCEN(3,TFLA,XW)
1280. 00 ZU=ZG*XIN*.7F*(1.-XIN)
1290. 00 RE=3600.*G*D/ZU
1300. 00 FF=0.775/SQRT(RE)*EXP((1.-XIN*.25)/2.4)
1310. 00 ***** EVALUATE DENSITY-PRESSURE INTEGRAL
1320. 00 Y=BSIMP(DDENFA,P2PH,POUT,0.002,10,H1,GN,XM)
1330. 00 ***** CALCULATE MASS FLOW RATE
1340. 00 R01=1./VL
1350. 00 R02=DDENFA(XM,POUT,H1,GN)
1360. 00 G2=SQRT(4636.8*Y/(2.*FF*FVL/D+ALOG(R01/R02)))
1370. 00 GD=G-G2
1380. 00 IF(ABS(GD/G).LT..005)THEN
1390. 00 G2=0.5*(G+G2)
1400. 00 GOTO 1000
1410. 00
1420. 00 END IF
1430. 00 IF(11.EQ.1)GOTO 24
1440. 00 IF(ABS(GD1-GD).LT..001.AND.(G-GMIN).LT.0.01)THEN
1450. 00 G2=G
1460. 00 GOTO 1000
1470. 00
1480. 00 END IF
1490. 00 G3=G-GD*(G1-G)/(GD1-GD)
1500. 00 GOTO 25
1510. 00 24 G3=0.1*G+0.9*G2
1520. 00 25 G1=G
1530. 00 GD1=GD
1540. 00 IF(G3.LT.GMIN)G3=G-(G-GMIN)*(GMIN-G3)/(G-G3)
1550. 00 IF(G3.GT.GMAX)G3=G+(GMAX-G)*(G3-GMAX)/(G3-G)
1560. 00 G=G3
1570. 00 30 CONTINUE
1580. 00 G2=G
1590. 00 PRINT 31,G,GD
1600. 00 31 FORMAT(' CAPIL DOES NOT CONVERGE,R6,GD=',2(1PE14.6))
1610. 00 GOTO 1000
1620. 00 32 CMAX=G
1630. 00 G=0.5*(GMIN+CMAX)
1640. 00 40 CONTINUE
1650. 00 41 PRINT 42,GD
1660. 00 42 FORMAT(' CAPIL DID NOT CONVERGE, GD =',F10.4)
1670. 00 1000 XMASS=3600.*G2*AREA*FLOAT(NC)
1680. 00 P2=POUT
1690. 00 44 FORMAT(' EXPANSION DEVICE: ' /
1700. 00 @' INPUT - P12 =',1PE9.3,3X,'H12 =',1PE9.3,3X,'P13 =',1PE9.3)
1710. 00 PRINT 46, POUT,XMASS
1720. 00 46 FORMAT(' OUTPUT - POUT =',1PE9.3,3X,'XMASS =',1PE9.3)
1730. 00 RETURN

```


***** CAPIL *****

1700. 00 50 XMASS=0.
1710. 00 PRINT 52,P1,P2
1720. 00 52 FORMAT(//, '***ERROR IN CALLING CAPIL, P1 LE.P2***', //
1730. 00 &4X, 'P1=', IPE11.3, ' P2=', IPE11.3, ' XMASS=0.0')
1740. 00 RETURN
1750. 00 END

END ELT. ERRORS: NONE. TIME: 0.236 SEC. IMAGE COUNT: 178

@HDG,P ***** CHOKE ***** .L,0

DATE 072184

***** CHOKER *****

```

@ELT,L DD.CHOKER
ELT 8R1 S74Q1C 07/21/84 15:54:57 (0)
10. 00 FUNCTION CHOKER(XW,PF,PEVAP,H0,GG)
20. 00 C
30. 00 C**** PURPOSE:
40. 00 C TO CALCULATE THE THERMODYNAMIC CRITICAL PRESSURE
50. 00 C OF NON-AZEOTROPIC BINARY MIXTURE IN TWO-PHASE FANNO FLOW
60. 00 C
70. 00 C**** NOTE: EVAPORATOR PRESSURE IS RETURNED AS CHOKING PRESSURE
80. 00 C IF CHOKING PRESSURE IS SMALLER THEN PRESSURE
90. 00 C IN THE EVAPORATOR
100. 00 C
110. 00 C**** INPUT DATA:
120. 00 C GG = G*G/(64.4*778.104) (BTU*LBM/FT**3)
130. 00 C WHERE G - REFRIG. MASS FLOW (LBM)/(SEC*FT**2))
140. 00 C H0 - REFRIG. TOTAL ENTHALPY (RTU/LBM)
150. 00 C PEVAP - EVAPORATOR PRESSURE (PSIA)
160. 00 C PF - FLASH PRESSURE (PSIA)
170. 00 C XW - COMPOSITION (WEIGHT FRACTION OF LESS VOLATILE COMPONENT)
180. 00 C
190. 00 C**** OUTPUT DATA:
200. 00 C CHOKER - CRITICAL PRESSURE (PSIA)
210. 00 C
220. 00 C**** SUBPROGRAMS CALLED BY CHOKER:
230. 00 C DFANNO
240. 00 C
250. 00 C
260. 00 C
270. 00 C
280. 00 C
290. 00 C
300. 00 C
310. 00 C
320. 00 C
330. 00 C
340. 00 C
350. 00 C
360. 00 C
370. 00 C
380. 00 C
390. 00 C
400. 00 C
410. 00 C
420. 00 C
430. 00 C
440. 00 C
450. 00 C
460. 00 C
470. 00 C
480. 00 C
490. 00 C
500. 00 C
510. 00 C
520. 00 C
530. 00 C
540. 00 C
550. 00 C
560. 00 C
570. 00 C
580. 00 C
590. 00 C
600. 00 C
610. 00 C
620. 00 C

COMMON/ RDATA2/W1,W2,TC1,TC2
DIMENSION P(3),S(3),DS(3),LN(2),LP(2)

IF(PF.GT.PEVAP)GOTO 10
PRINT 666,PF,PEVAP
666 FORMAT(' ERROR IN CALLING CHOKER, PF,PEVAP=',2(1PE15.5))
CHOKER=PEVAP
RETURN

C 10 XW=XW/(W2/W1*(1.-XW)+XW)
C
C**** CHECK, IF CHOKING PRESSURE IS BELOW PEVAP
P(1)=PEVAP
S(1)=DFANNO(XM,P(1),H0,GG)
SS=DFANNO(XM,P(1)-.1,H0,GG)
DS(1)=S(1)-SS
IF(DS(1).LT.0.)GOTO 1000

C**** SELECT PRESSURE STEP
100=8
DPEND=PF-PEVAP
DP=40.
DO 12 I=1,6
DP=0.5*DP
100=100-I
IF(OPEND.GT.DP)GOTO 14
12 CONTINUE
14 CONTINUE

C**** SEARCH FOR SONIC AND SUBSONIC POINT
DO 20 I=1,20

```

***** CHOKO *****

```

630. 00 P(2)=P(1)+20.
640. 00 IF(P(2).GT.PF) THEN
650. 00 P(1)=PF
660. 00 S(1)=DFANNO(XM,P(1),HO,GG)
670. 00 DS(1)=S(1)-DFANNO(XM,P(1)-.1,HO,GG)
680. 00 IF(DS(1).GT.O.)GOTO 1000
690. 00 P(2)=P(1)-DP
700. 00 S(2)=DFANNO(XM,P(2),HO,GG)
710. 00 DS(2)=S(2)-DFANNO(XM,P(2)-.1,HO,GG)
720. 00 GOTO 30
730. 00 END IF
740. 00 S(2)=DFANNO(XM,P(2),HO,GG)
750. 00 SS=DFANNO(XM,P(2)-.1,HO,GG)
760. 00 DS(2)=S(2)-SS
770. 00 IF(DS(2).LT.O.)GOTO 30
780. 00 P(1)=P(2)
790. 00 S(1)=S(2)
800. 00 DS(1)=DS(2)
810. 00 20 CONTINUE
820. 00
830. 00 C
840. 00 30 CONTINUE
850. 00 DO 50 ID=1,100
860. 00 P(3)=.5*(P(1)+P(2))
870. 00 S(3)=DFANNO(XM,P(3),HO,GG)
880. 00 SS=DFANNO(XM,P(3)-0.1,HO,GG)
890. 00 DS(3)=S(3)-SS
900. 00 IF(DS(3).EQ.O.)GOTO 1010
910. 00 C**** FIND TWO OF THE SAME SLOPE, SELECT ONE WITH BIGGER S
920. 00 NDS=0
930. 00 NDS=0
940. 00 DO 35 I=1,3
950. 00 IF(DS(1).GT.O.)GOTO 32
960. 00 NDS=NDS+1
970. 00 LN(NDS)=I
980. 00 GOTO 35
990. 00 32 NDS=NDS+1
1000. 00 LP(NDS)=I
1010. 00 35 CONTINUE
1020. 00 C**** ASSIGN VALUE OF SINGLE DS POINT FOR FUTURE USE
1030. 00 IF(NNDS.EQ.1)GOTO 40
1040. 00 IT0=LN(1)
1050. 00 IT1=LP(1)
1060. 00 IT2=LP(2)
1070. 00 GOTO 45
1080. 00 40 IT0=LP(1)
1090. 00 IT1=LN(1)
1100. 00 IT2=LN(2)
1110. 00 PS=P(IT0)
1120. 00 SS=S(IT0)
1130. 00 DSS=DS(IT0)
1140. 00 C**** PICK UP THE POINT WITH BIGGER ENTROPY
1150. 00 IBIG=IT2
1160. 00 IF(S(IT1).GT.S(IT2))IBIG=IT1
1170. 00 C**** ASSIGN VALUES TO POINTS 1 AND 2
1180. 00 P(1)=P(IBIG)
1190. 00 S(1)=S(IBIG)
1200. 00 DS(1)=DS(IBIG)
1210. 00 P(2)=PS
1220. 00
1230. 00
1240. 00
1250. 00
1260. 00

```

CHOKE

```

1270.      S(2)=SS
1280.      DS(2)=DSS
1290.      50 CONTINUE
1300.      C
1310.      DO 60 N=1,2
1320.      IL=1
1330.      IF(S(2).LT.S(1))IL=2
1340.      IF(S(3).LT.S(IL))IL=3
1350.      DO 55 I=1,2
1360.      IF(I.NE.IL)GO TO 55
1370.      P(1)=P(3)
1380.      S(1)=S(IL)
1390.      55 CONTINUE
1400.      P(3)=0.5*(P(1)+P(2))
1410.      S(3)=DFANN(XM,P(3),H0,GG)
1420.      60 CONTINUE
1430.      IL=1
1440.      IF(S(2).GT.S(1))IL=2
1450.      IF(S(3).GT.S(IL))IL=3
1460.      CHOKE=P(IL)
1470.      RETURN
1480.      C
1490.      1000 CHOKE=P(1)
1500.      RETURN
1510.      C
1520.      1010 CHOKE=P(3)-.05
1530.      RETURN
1540.      END
1550.
1560.

```

END ELT. ERRORS: NONE. TIME: 0.182 SEC. IMAGE COUNT: 142

@HDG,P ***** COMPARE ***** .L,0

***** COMPAR *****

@ELT, L DD. COMPAR

ELT 8R1 S74QIC 07/21/64 15:54:57 (0)

SUBROUTINE COMPAR

10. 00 C

20. 00 C

30. 00 C

40. 00 C

50. 00 C

60. 00 C

70. 00 C

80. 00 C

90. 00 C

100. 00 C

110. 00 C

120. 00 C

130. 00 C

140. 00 C

150. 00 C

160. 00 C

170. 00 C

180. 00 C

190. 00 C

200. 00 C

210. 00 C

220. 00 C

230. 00 C

240. 00 C

250. 00 C

260. 00 C

270. 00 C

280. 00 C

290. 00 C

300. 00 C

310. 00 C

320. 00 C

330. 00 C

340. 00 C

350. 00 C

360. 00 C

370. 00 C

380. 00 C

390. 00 C

400. 00 C

410. 00 C

420. 00 C

430. 00 C

440. 00 C

450. 00 C

460. 00 C

470. 00 C

480. 00 C

490. 00 C

500. 00 C

510. 00 C

520. 00 C

530. 00 C

540. 00 C

550. 00 C

560. 00 C

C**** PURPOSE:

TO DETERMINE PERFORMANCE PARAMETERS OF A HERMETIC COMPRESSOR
WORKING WITH A NON-AZEOTROPIC MIXTURE FROM TLST UNDER
ONE OPERATING CONDITION

C**** INPUT DATA:

* STANDARD ELECTRIC MOTOR CHARACTERISTICS:

EMETA(K) - COMPRESSOR MOTOR EFFICIENCY IN FRACTION AT FRACTION
OF FULL LOAD SPECIFIED BY EMOFT(K) (-)

EMOFT(K) - COMPRESSOR MOTOR FULL LOAD FRACTION (-)

ENRPM(L) - COEFFICIENT FOR COMPRESSOR MOTOR RPM CALCULATION (-)

* COMPRESSOR DESIGN DATA:

ELEFUL - COMPRESSOR MOTOR ENERGY INPUT RATE AT FULL LOAD (KW)

SWPVOL - TOTAL COMPRESSOR SWEEP VOLUME PER REVOLUTION (IN**3)

* TEST DATA AVAILABILITY:

ILONG = 0 FOR SHORT TEST DATA AVAILABLE

= 1 FOR LONG TEST DATA AVAILABLE

* SHORT TEST DATA INPUT (ILONG=0):

ELEIPT - COMPRESSOR MOTOR ENERGY INPUT RATE AT TEST CONDITION (KW)

P3,T3 - REFRIG. PRESSURE & TEMPERATURE AT COMPRESSOR CAN INLET (PSIA),(F)

P8,T8 - REFRIG. PRESSURE & TEMPERATURE AT COMPRESSOR CAN OUTLET (PSIA),(F)

RMASS - REFRIG. MASS FLOW RATE AT TEST (LBM/H)

RPMCP - COMPRESSOR NUMBER OF REVOLUTIONS PER MINUTE AT TEST (1/MIN)

IF NOT MEASURED RPMCP=0.

TGA - AMBIENT AIR TEMPERATURE (F)

XW - WEIGHT COMPOSITION (FRACTION OF LESS VOLATILE COMPONENT)

* LONG TEST DATA INPUT (ILONG=1):

AS FOR ILONG=0 PLUS

P3,T3 - PRESSURE & TEMPERATURE INSIDE COMPRESSOR CAN (PSIA),(F)

P4,T4 - PRESSURE & TEMPERATURE IN CYLINDER AT SUCTION (PSIA),(F)

P5,T5 - PRESSURE & TEMPERATURE IN CYLINDER AT DISCHARGE (PSIA),(F)

P6,T6 - PRESSURE & TEMPERATURE AT DISCHARGE MANIFOLD (PSIA),(F)

TCAN - COMPRESSOR CAN TEMPERATURE (F)

C**** OUTPUT DATA:

CLRCEFF - CLEARANCE VOLUME, IN FRACTION OF STROKE VOLUME (-)

CPC34 - PRESSURE DROP PARAMETER AT COMPRESSOR CAN INLET
(LBF*H**2/LBM*IN**2*FT**3)

CPC45 - PRESSURE DROP PARAMETER AT COMPRESSOR SUCTION VALVE
(LBF*H**2/LBM*IN**2*FT**3)

CPC67 - PRESSURE DROP PARAMETER AT COMPRESSOR DELIVERY VALVE
(LBF*H**2/LBM*IN**2*FT**3)

CPC78 - PRESSURE DROP PARAMETER AT COMPRESSOR CAN EXIT
(LBF*H**2/LBM*IN**2*FT**3)

QCCQA - PARAMETER FOR CAN WALL-AMBIENT AIR HEAT TRANSFER (FT**2)

QCC4C - PARAMETER FOR CAN WALL-REFRIG. VAPOR HEAT TRANSFER (FT**2)

QCC45 - SUCTION VALVE HEAT TRANSFER PARAMETER (FT**2)

QCC67 - DELIVERY VALVE HEAT TRANSFER PARAMETER (FT**2)

QCC78 - CAN EXIT HEAT TRANSFER PARAMETER (FT**2)

EFFYM - COMPRESSOR MOTOR EFFICIENCY AT TEST (-)

EFFYV - COMPRESSOR VOLUMETRIC EFFICIENCY AT TEST (-)

EMETA(K) - COMPRESSOR MOTOR EFFICIENCY IN FRACTION AT FRACTION
OF FULL LOAD SPECIFIED BY EMOFT(K) (-)

EMRPM(L) - COEFFICIENT FOR COMPRESSOR MOTOR RPM CALCULATION (-)

```

570. 00 C
580. 00 C
590. 00 C
600. 00 C
610. 00 C
620. 00 C
630. 00 C
640. 00 C
650. 00 C
660. 00 C
670. 00 C
680. 00 C
690. 00 C
700. 00 C
710. 00 C
720. 00 C
730. 00 C
740. 00 C
750. 00 C
760. 00 C
770. 00 C
780. 00 C
790. 00 C
800. 00 C
810. 00 C
820. 00 C
830. 00 C
840. 00 C
850. 00 C
860. 00 C
870. 00 C
880. 00 C
890. 00 C
900. 00 C
910. 00 C
920. 00 C
930. 00 C
940. 00 C
950. 00 C
960. 00 C
970. 00 C
980. 00 C
990. 00 C
1000. 00 C
1010. 00 C
1020. 00 C
1030. 00 C
1040. 00 C
1050. 00 C
1060. 00 C
1070. 00 C
1080. 00 C
1090. 00 C
1100. 00 C
1110. 00 C
1120. 00 C
1130. 00 C
1140. 00 C

ETAPLY - COMPRESSOR POLYTROPIC EFFICIENCY (-)
GAMA - REFRIG. AVE. SPECIFIC HEAT RATIO AT COMPRESSION (-)
RPM - COMPRESSOR MOTOR RPM (1/MIN)
V5 - REFRIG. SPECIFIC VOLUME AT SUCTION (FT**3/LBM)
WM - REFRIG. MOLECULAR WEIGHT (G/MOL)
X11 - MOLAR COMPOSITION (FRACTION OF LESS VOLATILE COMPONENT)

C**** SUBPROGRAMS CALLED BY COMPAR:
DEWTEM,ENTROP,HCVCP,HPIN,SPIN,VISCON,VOLITI

C**** NOTE: NUMBERS AT DIFFERENT SYMBOLS INDICATING LOCATION USED IN THIS SUBROUTINE
C**** CORRESPOND TO NUMBERS GREATER BY 1 USED IN MAIN PROGRAM & PROGRAM DOCUMENTATION

COMMON/RDATA2/W1,W2,TC1,TC2
COMMON/RDATA3/XW,XM,VM
COMMON/COND3/P2,T2,H2,XQ2
COMMON/COND47/P3,T3,H3,XQ3,P4,T4,H4,XQ4,S4,P5,T5,H5,XQ5,
@ P6,T6,H6,XQ6
COMMON/COND8/P7,T7,H7,XQ7
COMMON/COND9/P8,T8,H8,XQ8
COMMON/COMP/CPC23,CQC3C,CQCCOA,CPC34,CQC34,
& EMTA(11),EMOPT(11),EMRPM(6),ELEFUL,SWPVOL,
& ETAPLY,CLREFF,CPC56,CQC56,CPC67,CQC67
DATA NO,N1,N3,N5/O,1,3,5/

C
C
5 CONTINUE
WRITE(6,290)
READ(5,777)XW
XW=1.-XW
XM=XW/(P2/W1*(1.-XW)+XW)
WM=W1*(1.-XM)+W2*XM
WRITE(6,300)
READ(5,295)ILONG
IF(ILONG.EQ.1)GOTO 20
C**** INPUT SHORT FORM COMPRESSOR DATA
WRITE(6,301)
READ(5,777)ELEFUL,ELEIPT,RPMCP,SWPVOL,RMASS,TOA
EBTUI=3414.*ELEIPT
EBTUF=3414.*ELEFUL
WRITE(6,302)
READ(5,777)T2,P2
WRITE(6,303)
READ(5,777)T7,P7

C
TK2=(T2+459.67)/1.8 @POINT 2
PA2=P2/14.6959
CALL VOLITI(N1,TK2,PA2,XM,VM2)
CALL HCVCP(N1,TK2,VM2,XM,I12,CP2,CV2)
TK7=(T7+459.67)/1.8 @POINT 7
PA7=P7/14.6959
CALL VOLITI(N1,TK7,PA7,XM,VM7)
CALL HCVCP(N3,TK7,VM7,XM,H7,CP7,CV7)
V7=VM7*16.01346/WM
QCAN=EBTUI-RMASS*(H7-H2)
T6=T7+20.
P6=P7+1.
TK6=(T6+459.67)/1.8 @POINT 6

```

COMPAR

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1150. 00 PA3=P6/14.6959
1160. 00 CALL VOLIT1(N1,TK6,PA5,XM,VM5)
1170. 00 CALL HCVCP(N3,TK6,VM6,XM,H6,CV5,CP6)
1180. 00 V5=VM6*16.01816/VM
1190. 00 P5=P6+20.
1200. 00 H5=H6+0.2*(H6-H7)
1210. 00 TK5=TK6
1220. 00 PA5=P5/14.6959
1230. 00 ACC=0.001
1240. 00 CALL SPIN(N1,H5,PA5,XM,ACC,TK5,X05,XL,XV,VL,VM5)
1250. 00 T5=TK5*1.8-459.67
1260. 00 S5=ENTROP(TK5,VM5,XM)
1270. 00 CALL HCVCP(N5,TK5,VM5,XM,DUM,CV5,CP5)
1280. 00 GA5=CF5/CV5
1290. 00 GG5=(GA5-1.)/GA5
1300. 00 P3=P2-0.1
1310. 00 P4=P3-3.
1320. 00 PA4=P4/14.6959
1330. 00 CALL DEWTEM(NO,PA4,XM,TKD4,XL)
1340. 00 CALL VOLIT1(N1,TKD4,PA4,XM,VD4)
1350. 00 SD4=ENTROP(TKD4,VD4,XM)
1360. 00 S4=S5+0.1
1370. 00 DS4=0.1
1380. 00 P5P4=P5/P4
1390. 00 P4P5=P4/P5
1400. 00 INT=5
1410. 00 JNT=11
1420. 00 EMFRN=ELEIPT/ELEFUL
1430. 00 DO 16 I=1,INT
1440. 00 DO 12 J=1,JNT
1450. 00 S4=S4-DS4
1460. 00 IF(S4.GT.SD4)GOTO 10
1470. 00 INT=INT+1
1480. 00 JNT=2*JNT-1
1490. 00 S4=S4+DS4
1500. 00 DS4=0.5*DS4
1510. 00 GOTO 16
1520. 00 10 CONTINUE
1530. 00 TK4=TK5*P4P5*GG5
1540. 00 IF(TK4.LT.TKD4)TK4=TKD4+0.01
1550. 00 ACC=0.000001
1560. 00 CALL SPIN(N1,S4,PA4,XM,ACC,TK4,VM4)
1570. 00 CALL HCVCP(N3,TK4,VM4,XM,H4,CV4,CP4)
1580. 00 GA4=CF4/CV4
1590. 00 TK5S=TK5
1600. 00 ACC=0.001
1610. 00 CALL SPIN(N1,S4,PA5,XM,ACC,TK5S,VM5S)
1620. 00 CALL HCVCP(N3,TK5S,VM5S,XM,H5S,CV5S,CP5S)
1630. 00 GA5S=CF5S/CV5S
1640. 00 GA=0.5*(GA4+GA5S)
1650. 00 GG=(GG4-1.)/GG4
1660. 00 GN=GG/0.76
1670. 00 CETA=(P5P4*GG-1.)/(0.76*(P5P4*GN-1.))
1680. 00 H5S=H4+(H5S-H4)/CETA
1690. 00 IF(H5S.LE.H5)GOTO 14
1700. 00 12 CONTINUE
1710. 00 14 IF(1.EQ.5)GOTO 16
1720. 00 S4=S4+DS4

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@POINTS 3, 4 & 5


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***** COMPAR *****
1730. 00
1740. 00
1750. 00
1760. 00
1770. 00
1780. 00
1790. 00
1800. 00
1810. 00
1820. 00
1830. 00
1840. 00
1850. 00
1860. 00
1870. 00
1880. 00
1890. 00
1900. 00
1910. 00
1920. 00
1930. 00
1940. 00
1950. 00
1960. 00
1970. 00
1980. 00
1990. 00
2000. 00
2010. 00
2020. 00
2030. 00
2040. 00
2050. 00
2060. 00
2070. 00
2080. 00
2090. 00
2100. 00
2110. 00
2120. 00
2130. 00
2140. 00
2150. 00
2160. 00
2170. 00
2180. 00
2190. 00
2200. 00
2210. 00
2220. 00
2230. 00
2240. 00
2250. 00
2260. 00
2270. 00
2280. 00
2290. 00
2300. 00

DS4=0.1*DS4
16 CONTINUE
   CBTUI=RMASS*(H5-H4)
   H3=H2+(H6-H7)+(EBTUI-CBTUI-QCAN)/RMASS
   TK3=TK4
   PA3=P3/14.6959
   ACC=0.001
   CALL FPIN(N1,H3,PA3,XM,ACC,TK3,XQ,XL,XV,VL,VM3)
   V3=VM3*16.01846/WM
   T3=TK3*1.8-459.67
   CALL HCVCP(N5,TK3,VM3,XM,DUM,CV3,CP3)
   TCAN=(TOA+2.*T3)/3.
   T4=TK4*1.8-459.67
   GOTO 22

C**** ENTER LONG FORM COMPRESSOR DATA
20 WRITE(6,304)
   READ(5,777)ELEFUL,ELEIPT,RMCP,SWPVOL,RMASS,TCAN,TOA
   I=2
   WRITE(6,305)
   WRITE(6,306)I
   READ(5,777)T2,P2
   I=I+1
   WRITE(6,306)I
   READ(5,777)T3,P3
   I=I+1
   WRITE(6,306)I
   READ(5,777)T4,P4
   I=I+1
   WRITE(6,306)I
   READ(5,777)T5,P5
   I=I+1
   WRITE(6,306)I
   READ(5,777)T6,P6
   I=I+1
   WRITE(6,306)I
   READ(5,777)T7,P7

C
C**** CALCULATE REFRIGERANT STATE
TK2=(T2+459.67)/1.8
TK3=(T3+459.67)/1.8
TK4=(T4+459.67)/1.8
TK5=(T5+459.67)/1.8
TK6=(T6+459.67)/1.8
TK7=(T7+459.67)/1.8
PA2=P2/14.6959
PA3=P3/14.6959
PA4=P4/14.6959
PA5=P5/14.6959
PA6=P6/14.6959
PA7=P7/14.6959
CALL VOLIT1(N1,TK2,PA2,XM,VM2)
CALL VOLIT1(N1,TK3,PA3,XM,VM3)
CALL VOLIT1(N1,TK4,PA4,XM,VM4)
CALL VOLIT1(N1,TK5,PA5,XM,VM5)
CALL VOLIT1(N1,TK6,PA6,XM,VM6)
CALL VOLIT1(N1,TK7,PA7,XM,VM7)
CALL HCVCP(N3,TK2,VM2,XM,H2,CV2,CF2)
CALL HCVCP(N3,TK3,VM3,XM,H3,CV3,CP3)

```



```

*****
2310. 00
2320. 00
2330. 00
2340. 00
2350. 00
2360. 00
2370. 00
2380. 00
2390. 00
2400. 00
2410. 00
2420. 00
2430. 00
2440. 00
2450. 00
2460. 00
2470. 00
2480. 00
2490. 00
2500. 00
2510. 00
2520. 00
2530. 00
2540. 00
2550. 00
2560. 00
2570. 00
2580. 00
2590. 00
2600. 00
2610. 00
2620. 00
2630. 00
2640. 00
2650. 00
2660. 00
2670. 00
2680. 00
2690. 00
2700. 00
2710. 00
2720. 00
2730. 00
2740. 00
2750. 00
2760. 00
2770. 00
2780. 00
2790. 00
2800. 00
2810. 00
2820. 00
2830. 00
2840. 00
2850. 00
2860. 00
2870. 00
2880. 00

CALL HCVCP(N3,TK4,VM4,XM,H4,CV4,CP4)
CALL HCVCP(N3,TK5,VM5,XM,H5,CV5,CP5)
CALL HCVCP(N3,TK6,VM6,XM,H6,CV6,CP6)
CALL HCVCP(N3,TK7,VM7,XM,H7,CV7,CP7)
S4=ENTROP(TK4,VM4,XM)
S5=ENTROP(TK5,VM5,XM)
C**** CALCULATE COMPRESSOR PERFORMANCE PARAMETERS
22 S2=ENTROP(TK2,VM2,XM)
S3=ENTROP(TK3,VM3,XM)
S6=ENTROP(TK6,VM6,XM)
S7=ENTROP(TK7,VM7,XM)
SIPV=SWPVL/1728.
GA4=CP4/CV4
GA5=CP5/CV5
GA=0.5*(GA4+GA5)
DUM=(GA-1.)/GA
TK5S=TK4*P5P4**DUM
S5S=S4
ACC=0.000001
CALL SPIN(N1,S5S,PA5,XM,ACC,TK5S,VM5S)
T5S=TK5S*1.8-459.67
CALL HCVCP(N3,TK5S,VM5S,XM,H5S,CV5S,CP5S)
GA5S=CP5S/CV5S
AM2=VISCOS(3,T2,XW)
AM3=VISCOS(3,T3,XW)
AM4=VISCOS(3,T4,XW)
AM5=VISCOS(3,T5,XW)
AM6=VISCOS(3,T6,XW)
AM7=VISCOS(3,T7,XW)
AK2=VISCOS(4,T2,XW)
AK3=VISCOS(4,T3,XW)
AK4=VISCOS(4,T4,XW)
AK5=VISCOS(4,T5,XW)
AK6=VISCOS(4,T6,XW)
AK7=VISCOS(4,T7,XW)
WRITC(6,310)
WRITE(6,297)T2,P2,H2,S2
WRITE(6,297)T3,P3,H3,S3
WRITE(6,297)T4,P4,H4,S4
WRITE(6,297)T5,P5,H5,S5
WRITE(6,297)T6,P6,H6,S6
WRITE(6,297)T7,P7,H7,S7
EMFRN=ELEIPT/ELEFUL
EBTUI=3414.*ELEIPT
EBTUF=3414.*ELEFUL
C**** DETERMINE CAN HEAT LOSS COEFFICIENTS CQC3C AND CQCCOA
QCAN=EBTUI-RMASS*(H7-H2)
CQCCOA=QCAN/(TCAN-TOA)**1.233
CQC3C=QCAN*AM3**0.467/(RMASS**0.6*CP3**0.323*AK3**0.667*
& (T3-TCAN))
V3=VM3*16.01846/WM
CPC23=1./(V3*RMASS**2)
CPC2C=(P2-P3)*CPC23
C**** DETERMINE SUCTION VALVE HEAT TRANSFER COEFFICIENT
QSUCV=RMASS*(H4-H3)
T34=0.5*(T3+T4)
AM34=0.5*(AM3+AM4)

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```

***** COMPAR *****
2890. 00 CP34=0.5*(CP3+CP4)
2900. 00 AK34=0.5*(AK3+AK4)
2910. 00 CQC34=A/134**0.467/(RMASS**0.8*CP34**0.333
2920. 00 & *AK34**0.667*(T5-T3))
2930. 00 CQC34=QSUCV*CQC34
2940. 00 V4=VM4*16.01846/W4
2950. 00 CPC34=1./(V4*RMASS**2)
2960. 00 CFC34=(P3-P4)*CPC34
2970. 00 C***** DETERMINE COMPRESSION AND MOTOR COEFFICIENTS
2980. 00 RPMEM=0.
2990. 00 ETAEM=0.
3000. 00 EMFFF=RMASS*(H5-H4)/(0.96*EBTUF)
3010. 00 DO 23 I=1,6
3020. 00 J=1-1
3030. 00 AA=EMFFF**J
3040. 00 RPMEM=RPMEM+AA*EMRPM(I)
3050. 00 23 CONTINUE
3060. 00 J=12
3070. 00 DO 25 I=1,11
3080. 00 J=J-1
3090. 00 IF(EMFFF.LE.EMOPT(J))GOTO 24
3100. 00 ETAEM=EMETA(J)+(EMETA(J+1)-EMETA(J))*(EMFFF-EMOPT(J))
3110. 00 & /(EMOPT(J+1)-EMOPT(J))
3120. 00 GOTO 28
3130. 00 24 IF(1.NE.11)GOTO 25
3140. 00 ETAEM=EMFFF*EMETA(J)/EMOPT(J)
3150. 00 25 CONTINUE
3160. 00 28 CONTINUE
3170. 00 RPMRT=1.
3180. 00 IF(RPMCP.GE.100.)RPMRT=RPMCP/RPMEM
3190. 00 CPOPT=RMASS*(H5-H4)
3200. 00 CPIPT=CPOPT/0.96
3210. 00 ETART=CPIPT
3220. 00 ETART=ETART/(ETAEM*EBTUF)
3230. 00 DO 30 I=1,11
3240. 00 IF(1.GT.6)GOTO 30
3250. 00 LMRPM(I)=RPMRT*EMRPM(I)
3260. 00 EMETA(I)=ETART*EMETA(I)
3270. 00 CETA=(H5-H4)/(H5-H4)
3280. 00 ETAPLY=1.1
3290. 00 DE=0.1
3300. 00 P5P4=P5/P4
3310. 00 GGA=0.5*(GA4+GA5S)
3320. 00 GG=(GGA-1.)/GGA
3330. 00 P5P4=P5P4*GG
3340. 00 DO 36 I=1,3
3350. 00 DO 32 J=1,11
3360. 00 POLY=1./ETAPLY
3370. 00 CC=(P5P4-1.)/(ETAPLY*(P5P4**POLY-1.))
3380. 00 IF(CC.LE.CETA)GOTO 34
3390. 00 32 ETAPLY=ETAPLY-DE
3400. 00 34 IF(1.EQ.3)GOTO 38
3410. 00 ETAPLY=ETAPLY+DE
3420. 00 DE=0.1*DE
3430. 00 36 CONTINUE
3440. 00 38 CONTINUE
3450. 00 CRPM=0.
3460. 00 DO 40 I=1,6

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***** COMPAR *****
3470.
3480.
3490.
3500.
3510.
3520.
3530.
3540.
3550.
3560.
3570.
3580.
3590.
3600.
3610.
3620.
3630.
3640.
3650.
3660.
3670.
3680.
3690.
3700.
3710.
3720.
3730.
3740.
3750.
3760.
3770.
3780.
3790.
3800.
3810.
3820.
3830.
3840.
3850.
3860.
3870.
3880.
3890.
3900.
3910.
3920.
3930.
3940.
3950.
3960.
3970.
3980.
3990.
4000.
4010.
4020.
4030.
4040.

J=1-1
AA=EMFF**J
40 CCRM=CRM1+AA*CMRPM(1)
TVOL=GO.*CRM*SWPV
AVOL=RMASS*V4
VETA=AVOL/TVOL
AN=ETAPLY*CGA
AN=AN/(1.+AN-GGA)
ANR=1./AN
CLREFF=1.-1.0417*VETA
CLREFF=CLREFF/((P5/P4)**ANR-1.)
WRITE(6,323)
WRITE(6,297)(EMETA(1),I=1,5)
WRITE(6,297)(EMETA(1),I=6,11)
WRITE(6,324)
WRITE(6,297)(EMRPM(1),I=1,6)
WRITE(6,325)
WRITE(6,297)ELEFUL,SWPVOL,ETAPLY,CLREFF
WRITE(6,307)
WRITE(6,297)CRM,ETAEM,VETA,V4,GGA,SWPV
C**** DETERMINE DISCHARGE VALVE HEAT TRANSFER & PRESSURE COEFFICIENT
ODISV=RMASS*(H5-H6)
IF(ODISV.LE.0.0)ODISV=0.
T56=0.5*(T5+T6)
AM56=0.5*(AM5+AM6)
AK56=0.5*(AK5+AK6)
CP56=0.5*(CP5+CP6)
COC56=AM56**0.467/(RMASS**0.8*CP56**0.333*AK56**0.667*
& (T5-T4))
COC56=COC56*ODISV
V6=VMG*16.01846/MM
CPC56=1./(V6*RMASS**2)
CPC56=CPC56*(F5-P6)
C**** DETERMINE DISCHARGE LINE HEAT TRANSFER COEFFICIENT
AM67=0.5*(AM6+AM7)
AK67=0.5*(AK6+AK7)
CP67=0.5*(CP6+CP7)
DD1=AM67**0.467*CP6**0.333*AK6**0.667
DD2=1.5*AM3**0.467*CP67**0.333*AK67**0.667
ODISC=RMASS*(H6-H7)
IF(ODISC.LE.0.01)GOTO 42
ALOC67=(T7-T6)/ALOG((T7-T3)/(T6-T3))
COC67=ODISC*(DD1+DD2)/(ALOC67*RMASS**0.8)
COC67=COC67/((CPC3*CP67)**0.333*(AK3*AK67)**0.667)
GOTO 44
42 COC67=0.
V7=VM7*16.01846/MM
44 CPC67=1./(0.5*(V6+V7)*RMASS**1.8*AM67**0.2)
CPC67=CPC67*(P6-P7)
WRITE(6,327)
WRITE(6,297)CPC23,CPC34,CFC56,CPC67
WRITE(6,326)
WRITE(6,297)COC3C,COC30A,COC34,COC56,COC67
WRITE(6,330)
READ(5,777)ICON
IF(ICON.NE.999)GOTO 5
C 290 FORMAT(/' WEIGHT COMPOSITION OF MIXTURE IN FRACTION

```

```

***** COMPAR *****
4050. 00
4060. 00
4070. 00
4080. 00
4090. 00
4100. 00
4110. 00
4120. 00
4130. 00
4140. 00
4150. 00
4160. 00
4170. 00
4180. 00
4190. 00
4200. 00
4210. 00
4220. 00
4230. 00
4240. 00
4250. 00
4260. 00
4270. 00
4280. 00
4290. 00

@ / ' OF MORE VOLATILE COMPONENT, XW= '
295 FORMAT(110)
297 FORMAT(6(1PE11.3))
298 FORMAT(4(1PE11.3),2X,'ISENTRIOPIC 6')
300 FORMAT(///2X,'DETERMINE COMPRESSOR PERFORMANCE PARAMETERS'//
& 2X,'ENTER: ILONG EQUAL TO 0 OR 1 FOR SHORT OR LONG TEST DATA=')
301 FORMAT(/2X,'ENTER:ELEFUL,ELEIPT,RPMCP,SWFVOL,RMASS,TOA=')
302 FORMAT(/2X,'ENTER:T3,P3=')
303 FORMAT(/2X,'ENTER:T8,P8=')
304 FORMAT(/2X,'ENTER:ELEFUL,ELEIPT,RPMCP,SWFVOL,RMASS,TCAN,TOA: '
& /)
305 FORMAT(/2X,'ENTER T AND P AT DIFFERENT STATIONS: '//)
306 FORMAT(2X,'STATION',I2,':')
307 FORMAT(/2X,'R01',8X,'EFFVM',6X,'EFFV',6X,'V5',
& 9X,'GAMA',6X,'SWPV')
310 FORMAT(/2X,'T',10X,'P',10X,'H',10X,'S')
323 FORMAT(/2X,'EMTA(11):')
324 FORMAT(/2X,'EMRM(6):')
325 FORMAT(/2X,'ELEFUL',5X,'SWPVOL',5X,'ETAPLY',5X,'CLREFF')
327 FORMAT(/2X,'CPC34',6X,'CPC45',6X,'CPC67',6X,'CPC78')
326 FORMAT(/2X,'CQC4C',6X,'CQCCOA',5X,'CQC45',6X,'CQC67',6X,'CQC78')
330 FORMAT(/' PUT ICON=999 TO STOP, ICON=?')
777 FORMAT ( )
      RETURN
      END

```

END ELT. ERRORS: NONE. TIME: 0.518 SEC. IMAGE COUNT: 429

@HDG,P ***** COMPRE ***** .L,0

265

```

***** COMPRE *****
1160.
1170.
1180.
1190.
1200.
1210.
1220.
1230.
1240.
1250.
1260.
1270.
1280.
1290.
1300.
1310.
1320.
1330.
1340.
1350.
1360.
1370.
1380.
1390.
1400.
1410.
1420.
1430.
1460.
1470.
1480.
1490.
1500.
1510.
1520.
1530.
1540.
1550.
1560.
1570.
1580.
1590.
1600.
1610.
1620.
1630.
1640.
1650.
1660.
1670.
1680.
1690.
1700.
1710.
1720.
1730.
1740.
1750.

BTUFO=3413.*ELETUL
SMPV=SMPV(L/1728.
P4=P2-2.
H4=H2+9.
TK4=TK2
TK5=(TG5+459.67)/1.8
CALL DEVPRE(N0,TK5,XM,PA5,XL)
P5=PA5*14.6959
SLOPE=0.
*****
***** ITERATE THERMODYNAMIC STATE OF REFRIGERANT
***** WITHIN COMPRESSOR CAN
*****
***** START ENTHALPY LOOP
DO 50 MH=1,15
ROB=0.
*****
***** START PRESSURE LOOP
DO 50 MP=1,10
PA4=P4/14.6959
CC
ACC=0.0005
CALL HPPROP(N1,H4,PA4,XM,ACC,TK4,XQ4,XML4,XMV4,VML4,VMV4,
@ VM4,V4,CP4,CV4,AM4,AK4)
T4=TK4*1.8-459.67
GA4=CP4/CV4
IF (XQ4.LT.1.) THEN
S4=(1.-XQ4)*ENTROP(TK4,VML4,XML4)
S4=S4+XQ4*ENTROP(TK4,VMV4,XMV4)
ELSE
S4=ENTROP(TK4,VM4,XM)
END IF
***** CALCULATE POINT 5
F5P4=F5/P4
T5S=(T4+460)*P5P4**0.15-460.
TK5S=(T5S+460.)/1.8
ACC=0.00002
CALL SPIN(N1,S4,PA5,XM,ACC,TK5S,VM5S)
CALL HCVCP(N3,TK5S,VM5S,XM,H5S,CV5S,CP5S)
GA5S=CP5S/CV5S
GA45=0.5*(GA4+GA5S)
GA45=(GA45-1.)/GA45
/NA45=GCA45/ETAPLY
AN=(1.+ETAPLY*GA45-GA45)/(ETAPLY*GA45)
ETAV=0.95*(1.-CLREF)*P5P4**AN-1.))
ETAC=(P5P4*GA45-1.)/(P5P4*GA45-1.)
ETAC=ETAC/ETAPLY
UCPIPT=(H5S-H4)/ETAC
H5=H4+UCPIPT
T5=T4+2.*UCPIPT/(CP4+CP5S)
TK5=(T5+459.67)/1.8
ACC=0.0005
CALL HPPROP(N1,H5,PA5,XM,ACC,TK5,XQ5,XML5,VM5V,VML5,VMV5,
@ VM5,V5,CP5,CV5,AM5,AK5)
T5=TK5*1.8-459.67
***** CALCULATE REFRIGERANT MASS FLOW RATE
FCPT=-0.0998
DOPT=0.10
DO 16 IC=1,4

```

***** COMPRE *****

```

1760.      DO 12 JC=1,11
1770.      FOPT=FOPT+DOPT
1780.      CIPRM=0.
1790.      DO 10 KC=1,6
1800.      LC=KC-1
1810.      AA=FOPT*LC
1820.      10 CIPRM=CIPRM+AA*ENRPM(KC)
1830.      RMAS=GO.*CIPRM*ETAV*SWPV/V4
1840.      WE=RTUCFO*FOPT
1850.      WC=0.96*WE
1860.      RMACP=WC/UCPIPT
1870.      IF(RM/CP.GE.RMASS)GOTO 14
1880.      12 CONTINUE
1890.      14 IF(IC.EQ.4)GOTO 16
1900.      FOPT=FOPT-DOPT
1910.      DOPT=0.1*DOPT
1920.      16 CONTINUE
1930.      RMAS2=RMAS*RMAS
1940.      RMAS2=RMAS2*.8
1950.      JJ=12
1960.      DO 20 II=1,11
1970.      JJ=JJ-1
1980.      IF(FOPT.LE.EMOPT(JJ))GOTO 18
1990.      ETAE=EMETA(JJ)+(EMETA(JJ+1)-EMETA(JJ))*
2000.      & (FOPT-EMOPT(JJ))/(EMOPT(JJ+1)-EMOPT(JJ))
2010.      GOTO 22
2020.      18 IF(II.NE.11)GOTO 20
2030.      ETAE=FOPT*EMETA(JJ)/EMOPT(JJ)
2040.      20 CONTINUE
2050.      22 E1=WC/(0.96*ETAE)
2060.      IF(IP.GT.1.OR.MH.GT.1)GOTO 24
2070.      C**** CALCULATE POINTS 3,6,7
2080.      T3=T4
2090.      P3=P4
2100.      V3=V4
2110.      VM3=VM4
2120.      AM3=AM2
2130.      AK3=AK2
2140.      CP3=CP2
2150.      T6=T5
2160.      P6=P5
2170.      V6=V5
2180.      VM6=VM5
2190.      AM6=AM5
2200.      AK6=AK5
2210.      CP6=CP5
2220.      T7=T5
2230.      P7=P5
2240.      V7=V5
2250.      VM7=VM5
2260.      AM7=AM5
2270.      AK7=AK5
2280.      CP7=CP5
2290.      24 CONTINUE
2300.      DO 40 K=1,12
2310.      IF(K.EQ.1)GOTO 26
2320.      AM3=VISCON(N3,T3,XW)
2330.      AK3=VISCON(N4,T3,XW)

```

@POINT 3

C C M P R E

```

2340. 00 TK3=(T3+459.67)/1.8
2350. 00 CALL HCVCP(N5,TK3,VM3,VM,DUM,CV3,CP3)
2360. 00
2370. 00 AM6=V1SCON(N3,T6,XW) @POINT 6
2380. 00 AK6=V1SCON(N4,T6,XW)
2390. 00 TK6=(T6+459.67)/1.8
2400. 00 CALL HCVCP(N5,TK6,VM6,VM,DUM,CV6,CP6)
2410. 00
2420. 00 AM7=V1SCON(N3,T7,XW) @POINT 7
2430. 00 AK7=V1SCON(N4,T7,XW)
2440. 00 TK7=(T7+459.67)/1.8
2450. 00 CALL HCVCP(N5,TK7,VM7,VM,DUM,CV7,CP7)
2460. 00
26 CP56=0.5*(CP3+CP4)
2470. 00 CP67=0.5*(CP6+CP7)
2480. 00 AM34=C.5*(AMC+AM14)
2490. 00 AM56=0.5*(AM5+AM6)
2500. 00 AM67=0.5*(AM6+AM7)
2510. 00 AK34=0.5*(AK3+AK4)
2520. 00 AK56=0.5*(AK5+AK6)
2530. 00 AK67=0.5*(AK6+AK7)
2540. 00 V67=0.5*(V6+V7)
2550. 00 QM3=E1-WC
2560. 00 Q34=CQC34*RMAS58*CP34**0.333*AK34**0.667
2570. 00 &*(T5-T3)/(AM34**0.467)
2580. 00 Q56=CQC56*RMAS58*CP56**0.333*AK56**0.667
2590. 00 &*(T6-T4)/(AM56**0.447)
2600. 00 Q67=RMAS58*(CP3*CP67)**0.333*(AK3*AK67)**0.667
2610. 00 Q67=Q67/(CP3**0.333*AK3**0.667*AM157**0.467+
2620. 00 &1.5*CP67**0.333*AK67**0.667*AM3**0.157)
2630. 00 Q67=CQC67*Q67
2640. 00 IF(ABS(T6-T7).GT.0.01)GOTO 28
2650. 00 T673=0.5*(T6+T7)-T3
2660. 00 GOTO 30
2670. 00
28 T673=(T6-T7)/ALOG((T6-T3)/(T7-T3))
2690. 00
30 Q67=Q67*T573
AAX=CQC3C*RMAS58*CP3**0.333*AK3**0.667/AM3**0.467
BX=CQC3C/A
TC=(4.*T3+TA)/5.
SAA=1.
D0 32 ITR=1,20
IF(T3-LT,TC)SAA=-1.
QQ=SAA*AAX*(T3-TC)-BBX*ABS(TC-TA)**1.333
DQDT=SAA*(-AAX-1.333*BBX*ABS(TC-TA)**0.333)
TCC=TC-QQ/DQDT
IF(ABS(TCC-TC).LT.0.001)GOTO 34
TC=TCC
32 CONTINUE
34 TC=TCC
CCA=AAX*(T3-TC)
H3=H4-Q34/RMASS
H6=H5-Q56/RMASS
H7=H5-Q67/RMASS
H2A=H7-(E1-Q3A)/RMASS
P3=P4+CP34*V4*RMAS52
P2A=P3+CP34*V3*RMAS52
P6=P5-CP36*V6*RMAS52
P7=P6-CP36*V67*AM67**0.2*RMAS58*RMAS58
2910. 00

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***** CUMPRE *****

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2920. 00 T3=T4-(H4-H3)/CP34
2930. 00 T6=T5+(H5-H4)/CP56
2940. 00 T7=T6+(H7-H6)/CP67
2950. 00
2960. 00 CC
2970. 00 ACC=0.001
2980. 00 TK3=(T3+459.67)/1.8
2990. 00 PA3=P3/14.6959
3000. 00 CALL HPIN(N1,H3,PA3,XM,ACC,TK3,XQ,XL,XV,VL,VM3)
3010. 00 T3=TK3*1.8-459.67
3020. 00 V3=VM3*16.01846/MM
3030. 00 PA6=P6/14.6959
3040. 00 TK6=(T6+459.67)/1.8
3050. 00
3060. 00 C
3070. 00 CALL HPIN(N1,H6,PA6,XM1,ACC,TK6,XQ,XL,XV,VL,VM6)
3080. 00 T6=TK6*1.8-459.67
3090. 00
3100. 00 C
3110. 00 V6=V16*16.01846/MM
3120. 00 PA7=P7/14.6959
3130. 00 TK7=(T7+459.67)/1.8
3140. 00 CALL HPIN(N1,H7,PA7,XM,ACC,TK7,XQ,XL,XV,VL,VM7)
3150. 00 T7=TK7*1.8-459.67
3160. 00 V7=V17*16.01346/MM
3170. 00 IF(K.EQ.1)GOTO 36
3180. 00 IF(ABS(H2B-H2A).LT.ACC.AND.ABS(H7B-H7).LT.ACC)GOTO 42
3190. 00
3200. 00 36 H2B=H2A
3210. 00 H7B=H7
3220. 00 40 CONTINUE
3230. 00 WRITE(6,202)H2A,H2B
3240. 00 202 FORMAT(' LOOP 4 DID NOT CONVERGE, H2A,H2B=',2F8.3)
3250. 00 42 CONTINUE
3260. 00 H2A=(H2A+H2B)/2.
3270. 00 H7=(H7+H7B)/2.
3280. 00 P2A=P3+CP323*V3*RMAS2
3290. 00 ROA=P2-P2A
3300. 00 SAA=0.002
3310. 00 IF(ABS(H2-H2A).LT.5.)SAA=0.001
3320. 00 IF(ABS(ROA).LT.SAA)GOTO 52
3330. 00 IF(ABS(ROB).GT.0)GOTO 44
3340. 00 IF(SLOPE.NE..0)GOTO 45
3350. 00 P4B=P4
3360. 00 P4=P4+ROA
3370. 00 ROB=ROA
3380. 00 GOTO 50
3390. 00 44 SLOPE=(P4B-P4)/(ROB-ROA)
3400. 00 IF(ABS(ROB).LT.ABS(ROA))GOTO 46
3410. 00 45 ROB=ROA
3420. 00 P4B=P4
3430. 00 P4=P4B-ROB*SLOPE
3440. 00 50 CONTINUE
3450. 00 WRITE(6,204)
3460. 00 C*** END OF PRESURE LOOP
3470. 00 C*****
3480. 00 52 CONTINUE
3490. 00 PHA=H2-H2A
3500. 00 IF(ABS(RHA).LT.0.005)GOTO 62
3510. 00 IF(MH.GT.1)GOTO 54
3520. 00 P4Y=P4
3530. 00 P4=P4+0.1
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***** COMPRE *****
3520. 00 H4B=H4
3530. 00 H4=H4+2.
3540. 00 IF(RHA,LT.0.0)H4=H4-4.
3550. 00 IF(RHA,LT.0.0)P4=P4-0.2
3560. 00 RHB=RHA
3570. 00 GOTO 60
3580. 00
3590. 00 54 SLOPH=(H4B-H4)/(RHB-RHA)
3600. 00 SLOPP=(P4Y-P4)/(RHB-RHA)
3610. 00 IF(ABS(RHB).LT.AES(RHA))GOTO 56
3620. 00 RHB=RHA
3630. 00 H4B=H4
3640. 00 P4Y=P4
3650. 00 56 H4=H4B-RHB*SLOPH
3660. 00 P4=P4Y-RHB*SLOPP
3670. 00 60 CONTINUE
3680. 00 WRITE(6,206)RHA
3690. 00 ***** END OF ENTHALPY LOOP *****
3700. 00 ***** END OF ITERATION PROCESS *****
3710. 00 ***** REFRIG. STATE INSIDE COMPRESSOR CAN *****
3720. 00 ***** AND REFRIGERANT MASS FLOW RATE ARE KNOWN *****
3730. 00 ***** CALC. REFRIG. STATE AT 4-WAY VALVE AND CONNECTING TUBING *****
3740. 00
3750. 00 62 CALL MVAL4(XW,RMASS,V2,AM2,AK2,CP2,V7,AM7,AK7,CP7,
3760. 00 & V1,AM1,AK1,CP1,V8,AM8,AK8,CP8)
3770. 00 IF(NSYS.EQ.2)GOTO 64
3780. 00 CALL PIPE(N0,XW,TRA,RMASS,T8,P8,V8,H8,AM8,AK8,CP8,
3790. 00 & T9,P9,H9,XQ9,RL,RD,RK1,RD1,RK2,RD2,O.,O.)
3800. 00 CALL PIPE(N1,XW,TA,RMASS,T1,P1,V1,H1,AM1,AK1,CP1,
3810. 00 & T0,P0,H0,XQ0,YL,YD,YK1,YD1,YK2,YD2,O.,O.)
3820. 00 GOTO 66
3830. 00 64 CALL PIPE(N1,XW,TRA,RMASS,T1,P1,V1,H1,AM1,AK1,CP1,
3840. 00 & T0,P0,H0,XQ0,RL,RD,RK1,RD1,RK2,RD2,O.,O.)
3850. 00 CALL PIPE(N0,XW,TA,RMASS,T0,P0,V0,H0,AM0,AK0,CP0,
3860. 00 & T9,P9,H9,XQ9,YL,YD,YK1,YD1,YK2,YD2,O.,O.)
3870. 00 56 EI=EI/3412.7
3880. 00 ***** PRINT RESULTS *****
3890. 00 *****
3900. 00 WRITE(6,207)
3910. 00 WRITE(6,210)
3920. 00 I=1
3930. 00 WRITE(6,212)I,T0,P0,H0,XQ0
3940. 00 I=I+1
3950. 00 WRITE(6,212)I,T1,P1,H1,XQ1
3960. 00 I=I+1
3970. 00 WRITE(6,212)I,T2,P2,H2,XQ2
3980. 00 I=I+1
3990. 00 WRITE(6,212)I,T3,P3,H3,XQ3
4000. 00 I=I+1
4010. 00 WRITE(6,212)I,T4,P4,H4,XQ4
4020. 00 I=I+1
4030. 00 WRITE(6,212)I,T5,P5,H5,XQ5
4040. 00 I=I+1
4050. 00 WRITE(6,212)I,T6,P6,H6,XQ6
4060. 00 I=I+1
4070. 00 WRITE(6,212)I,T7,P7,H7,XQ7
4080. 00 I=I+1
4090. 00
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***** COMPRE *****
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      C
      WRITE(6,212)I,T8,P8,H8,XQ8
      I=I+1
      WRITE(6,212)I,T9,P9,H3,XQ9

      200 FORMAT(/2X,'INPUT DATA TO COMPRESS: '//2X,'P3',9X,'T3',9X,'H3',
      & 9X,'XQ3',8X,'TG6',8X,'TRA',9X,'TOA'/7(IPE11.3))
      204 FORMAT(' COMPRESS*ERROR PRESSURE LOOP')
      205 FORMAT(' COMPRESS DOES NOT CONVERGE, RHA= '1PE16.6)
      207 FORMAT(/2X,'COMPRESSOR ITERATION:')
      208 FORMAT(/2X,'EI',9X,'ETAE',7X,'ETAC',7X,'FTAV',7X,
      & 'CPRM',6X,'RMASS'/6(IPE11.3))
      210 FORMAT(/2X,'I',2X,'T',10X,'P',10X,'H',10X,'X')
      212 FORMAT(13.4(IPE11.3))
      RETURN
      END

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END ELT. ERRORS: NONE. TIME: 0.506 SEC. IMAGE COUNT: 419

CHDG,P ***** CONDHX ***** .L,0

***** CONDHX *****

@ELT,L DD.CONDHX

ELT 8R1 S7401C 07/21/84 15:54:59 (0)

SUBROUTINE CONDIX(I10,RMASS,T1,P1,ATIN,APIN,ARHIN,
T2,P2,I12,X2)

&

AUGUST 1983

C**** SIMULATION OF A CONDENSER WORKING WITH A NON-AZEOTROPIC MIXTURE
C**** AS REFRIGERANTC**** THIS PROGRAM COMPUTES IN TUBE-BY-TUBE SCHEME
C**** PERFORMANCE OF CROSS-FLOW AIR COOLED CONDENSER
C**** WITH UP TO 130 PLATE FINNED TUBES
C**** PLACED IN UP TO 5 DEPTH ROWS.

C**** INPUT DATA:

100	C	ANAS(110)	- AIR MASS FLOW RATE THROUGH COIL (LBM/H)
150	C	ANGLE(110)	- ANGLE BETWEEN COIL FACE & AIR STREAMLINES (RAD)
160	C	APIN	- AIR INLET PRESSURE (PSIA)
170	C	ARHIN	- AIR INLET RELATIVE HUMIDITY (-)
180	C	ATIN	- AIR INLET TEMPERATURE (F)
190	C	CONST(110)	- CONSTANT FOR AIR SIDE HEAT TRANSFER CORRELATION (-)
200	C	CPOW(110)	- CONSTANT FOR AIR SIDE HEAT TRANSFER CORRELATION (-)
210	C	DI(110)	- INNER DIAMETER OF TUBES (FT)
220	C	DO(110)	- OUTER DIAMETER OF TUBES (FT)
230	C	DPCH(110)	- TUBE DEPTH PITCH (FT)
240	C	DT(110)	- FIN TIP DIAMETER (FT)
250	C	FLOW(110,M)	- FRACTION OF COIL TOTAL REFRIG. MASS FLOW PASSING THROUGH TUBE M (-)
260	C	FMK(110)	- FIN MATERIAL THERMAL CONDUCTIVITY (BTU/FT*H*F)
270	C	FPCH(110)	- FIN PITCH (FT)
280	C	FTK(110)	- FIN THICKNESS (FT)
290	C	DEPTH(110,M)	- DEPTH ROW OF A TUBE M (-)
300	C	IFROM(110,M)	- NUMBER OF TUBE FROM WHICH TUBE M RECEIVES REFRIG. WHEN COIL WORKS AS EVAPORATOR (-)
310	C	I10	= 1 FOR INDOOR COIL (-)
320	C	IMER(110)	= 2 FOR OUTDOOR COIL (-)
330	C	IST(110)	- NUMBER OF MERGING TUBES (-)
340	C	ISTART(110,L)	- NUMBER OF TUBES REFRIG. FLOWS INTO COIL WORKING AS CONDENSER (-)
350	C	MERGE(110,K,1)	- NUMBER OF TUBE REFRIG. FLOWS INTO COIL WORKING AS CONDENSER, FOUND AS L SUCH TUBE (-)
360	C	MERGE(110,K,2)	- NUMBER OF TUBE FOUND AS K MERGING TUBE (-)
370	C	NDEP(110)	- NUMBER OF TUBE ROW DEPTHS (-)
380	C	NROW(110)	- NUMBER OF TUBES PER ROW (-)
390	C	NSECT(110)	- NUMBER OF REPEATING SECTIONS OF COIL (-)
400	C	NIPSP(1)	- NUMBER OF TUBES PER SECTION (-)
410	C	NTUBS(110,1)	- NUMBER OF TUBES IN ROW 1 OF EACH SECTION (-)
420	C	P1	- REFRIGERANT PRESSURE AT CONDENSER INLET (PSIA)
430	C	RMASS	- TOTAL REFRIG. MASS FLOW RATE THROUGH COIL (LBM/H)
440	C	RPCH(110)	- PITCH BETWEEN TUBES OF THE SAME DEPTH (FT)
450	C	TMK(110)	- TUBE MATERIAL THERMAL CONDUCTIVITY (BTU/FT*H*F)
460	C	T1	- REFRIGERANT TEMPERATURE AT CONDENSER INLET (F)
470	C	WIDTH(110)	- COIL WIDTH (FT)
480	C	XM	- MOLAR COMPOSITION (FRACTION OF LESS VOLATILE COMPONENT)
490	C	XW	- WEIGHT COMPOSITION
500	C		
510	C		
520	C		
530	C		
540	C		
550	C		
560	C		

	(FRACTION OF LESS VOLATILE COMPONENT)	
-	MIXTURE MOLECULAR WEIGHT (G/MOL)	(G/MOL)
-	MOLECULAR WEIGHT OF MORE VOLATILE COMPONENT	(G/MOL)
-	MOLECULAR WEIGHT OF LESS VOLATILE COMPONENT	(G/MOL)
C****	OUTPUT DATA:	
H2	REFRIGERANT ENTHALPY AT CONDENSER OUTLET	(BTU/LBM)
PRM(110,1,1)	REFRIG. PRESSURE AT I TUBE INLET	(PSIA)
PRM(110,2,1)	REFRIG. PRESSURE AT I TUBE OUTLET	(PSIA)
P2	REFRIGERANT PRESSURE AT CONDENSER OUTLET	(PSIA)
TRM(110,1,1)	REFRIG. TEMP. AT I TUBE INLET	(F)
TRM(110,2,1)	REFRIG. TEMP. AT I TUBE OUTLET	(F)
T2	REFRIGERANT TEMPERATURE AT CONDENSER OUTLET	(F)
XRM(110,1,1)	REFRIG. QUALITY AT I TUBE INLET	(-)
XRM(110,2,1)	REFRIG. QUALITY AT I TUBE OUTLET	(-)
XTUBE(110,J)	FRACTION OF TUBE J WITH SUPERHEATED VAPOR (WHEN 2-PHASE FLOW IS IN REST OF TUBE)	(-)
C	OR	
-	FRACTION OF TUBE J WITH 2-PHASE FLOW	
(WHEN SUPCOOLED LIQUID IS IN REST OF TUBE)		(-)
-	LOCKHART-MARTINELLI PARAMETER FOR REFRIG. IN TUBE K	(-)
X2	REFRIGERANT QUALITY AT CONDENSER OUTLET	(-)
VGM(110,1,1)	SPEC. VOLUME OF SATURATED REFRIG. VAPOR AT SAT. TEMP. OF I TUBE INLET	(FT**3/LBM)
VGM(110,2,1)	SPEC. VOLUME OF SATURATED REFRIG. VAPOR AT SAT. TEMP. OF I TUBE OUTLET	(FT**3/LBM)
VLM(110,1,1)	REFRIG. SPEC. VOLUME AT I TUBE INLET	(FT**3/LBM)
VLM(110,2,1)	REFRIG. SPEC. VOLUME AT I TUBE OUTLET	(FT**3/LBM)
C****	SUBPROGRAMS CALLED BY CONDX:	
AIRHT,AIRPR,BURTEM,DEVTE1,DYNDP1,DPDYN2,EDUBTE,ESVOL,FEELIQ,		
FINEFF,HCVCP,HFIN,HTCCCN,SPHPD1,SPHTC,VISCOB,VOLIT1		
C		
COMMON/RDATA2/W1,W2,TG1,TC2		
COMMON/RDATA3/XW,XM,WM		
COMMON/HPHX/NDEP(2),NROW(2),DI(2),DO(2),DT(2),RPCH(2),DPCH(2),		
&WDTH(2),FPCH(2),FTK(2),FMK(2),TMK(2),AMAS(2),ANGLE(2),		
&CONST(2),CPOW(2),NTUB(2,5),JFROM(2,130),NSECT(2),NTPS(2)		
COMMON/MERG/MERGE(2,20,2),IMER(2),ISTART(2,30),IST(2),		
&IDEPTH(2,130),FLOW(2,130),JFROM(2,130),KFEED(2,130,3),		
&KSTART(2,20),KST(2)		
COMMON/MASS/TRM(2,2,130),PRM(2,2,130),XPM(2,2,130),		
&VLM(2,2,130),VGM(2,2,130),XTUBE(2,130),XTT(2,130)		
DIMENSION TAIR(2,6),AIRN(5),JIR(2,130),INC(20),IEND(10),MY(130),		
@ DTR(130),DPR(130),DHR(130),DNR(130),CPRS(130),HCO(5),FFEE(5)		
DIMENSION XLS(2,130),XVS(2,130),VLS(2,130),VVS(2,130)		
DIMENSION TAIR1(2,130),AMS1(5)		
DATA NO,N1,N3,N4,N5/O,1,3,4,5/		
C		
WRITE(6,500)T1,P1,ATIN,ARHIN,RHASS		
C		
C****	DATA PREPARATION	
NNDCP=NDEP(110)		
DD1=DI(110)		
DRO=DQ(110)		
DDT=DT(110)		
RPPCH=RPCH(110)		
00		
570.		
580.		
590.		
600.		
610.		
620.		
630.		
640.		
650.		
660.		
670.		
680.		
690.		
700.		
710.		
720.		
730.		
740.		
750.		
760.		
770.		
780.		
790.		
800.		
810.		
820.		
830.		
840.		
850.		
860.		
870.		
880.		
890.		
900.		
910.		
920.		
930.		
940.		
950.		
960.		
970.		
980.		
990.		
0000.		
0010.		
0020.		
0030.		
0040.		
0050.		
0060.		
0070.		
0080.		
0090.		
0100.		
0110.		
0120.		
0130.		
0140.		

CONDHX

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1150. 00 DDPCH=DPCH(110)
1160. 00 WIDTH=WIDTH(110)
1170. 00 FFPCH=FFPCH(110)
1180. 00 FFTK=FTK(110)
1190. 00 FFMK=FMK(110)
1200. 00 TTMK=TMK(110)
1210. 00 AAMAS=AMAS(110)
1220. 00 AANCLF=ANGLE(110)
1230. 00 CCONST=CONST(110)
1240. 00 CCPOW=CPCW(110)
1250. 00 API=3.1415927*DD1*WIDTH
1260. 00 AP0=3.1415927*DD0*WIDTH
1270. 00 API=0.5*(API+AP0)
1280. 00 AP0=AP0*(FFPCH-FFTK)/FFPCH
1290. 00 AF=1.570796*(DD1+DD0)*(DD1-DD0)
1300. 00 AF=AF*WIDTH/FFPCH
1310. 00 A0=AP0+AF
1320. 00 HD=5000.
1330. 00 HP=2.*TTMK/(DD0-DD1)
1340. 00 C***** FIND INLET STATE FROM PRESSURE AND TEMPERATURE
1350. 00 PA1=P1/14.6959
1360. 00 TK1=(T1+459.67)/1.8
1370. 00 CALL VOLIT1(N1,TK1,PA1,XM,VM1)
1380. 00 V1=16.01846*VM1/WM
1390. 00 CALL HCVCP(N1,TK1,VM1,XM,H1,CV,CP)
1400. 00 X1=1.
1410. 00 C***** ESTIMATE CHANGE OF AIR TEMPERATURE
1420. 00 PA2=PA1
1430. 00 TK2=EBUOTE(PA2,XM)
1440. 00 VM2=ESVOL(N0,TK2,PA2,XM)
1450. 00 CALL HCVCP(N1,TK2,VM2,XM,H2,CV,CP)
1460. 00 CALL AIRPR(1,ATIN,APIN,ARHIN,WAIR,CPAIR,RAIR,
1470. 00 &AMAIR,AKAIR)
1480. 00 TAIR(1,1)=ATIN
1490. 00 DTAIR=RMASS*(H1-H2)/(CPAIR*AMAS*NNDEP)
1500. 00 DO101=1,NNDEP
1510. 00 J=I+1
1520. 00 10 TAIR(1,J)=TAIR(1,1)+DTAIR
1530. 00 DO 11 I=1,NTPS(110)
1540. 00 ICT=IDEPTH(110,I)
1550. 00 TAIRI(1,I)=ATIN+(ICT-1.)*DTAIR
1560. 00 11 TAIRI(2,I)=TAIRI(1,I)+DTAIR
1570. 00 C***** EVALUATE TWO-PHASE SPEC. HEAT
1580. 00 CALL DEWTEM(N0,PA1,XM,TKD,XL)
1590. 00 CALL VOLIT1(N1,TKD,PA1,XM,VV)
1600. 00 CALL HCVCP(N1,TKD,VV,XM,H3,CV,CP)
1610. 00 CPR=(HGV-H2)/(1.8*(TKD-TK2))
1620. 00 DO 12 I=1,NTPS(110)
1630. 00 VLM(110,2,I)=0.0
1640. 00 VGM(110,2,I)=0.0
1650. 00 DTR(I)=0.
1660. 00 DPR(I)=0.
1670. 00 DHR(I)=0.
1680. 00 DXR(I)=0.
1690. 00 CPRS(I)=CPR
1700. 00 12 HY(I)=0
1710. 00 NIST=IST(110)
1720. 00 NIMER=IMEF(110)

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***** CNDHX *****
1730. 00
1740. 00
1750. 00
1760. 00
1770. 00
1780. 00
1790. 00
1800. 00
1810. 00
1820. 00
1830. 00
1840. 00
1850. 00
1860. 00
1870. 00
1880. 00
1890. 00
1900. 00
1910. 00
1920. 00
1930. 00
1940. 00
1950. 00
1960. 00
1970. 00
1980. 00
1990. 00
2000. 00
2010. 00
2020. 00
2030. 00
2040. 00
2050. 00
2060. 00
2070. 00
2080. 00
2090. 00
2100. 00
2110. 00
2120. 00
2130. 00
2140. 00
2150. 00
2160. 00
2170. 00
2180. 00
2190. 00
2200. 00
2210. 00
2220. 00
2230. 00
2240. 00
2250. 00
2260. 00
2270. 00
2280. 00
2290. 00
2300. 00

DO 13 I=1,NIMER
J=MERGE(110,I,1)
13 MY(J)=1
ACC=0.005
*****
***** START MAIN LOOP *****
*****
DO 100 IAIR=1,10
CXZ PRINT 805,IAIR
805 FORMAT(' IAIR=',I4)
CAA IF(IAIR.EQ.1)PRINT 803,TAIR(1,1),TAIR(1,2),TAIR(1,3),TAIR(1,4)
803 FORMAT(' TAIR 1,2,3,4=',4F10.3)
DO 14 I=1,NNDP
NT=NSECT(110)*NTUB(110,I)
AMS=AMAS/NT
TAAV=0.5*(TAIR(1,1)+TAIR(1,I+1))
CALL AIRPR(2,TAAV,AFIN,RIA,WAIP,CFA,RA,AMA,AKA)
NNRQW=NTUB(110,I)
HCO(1)=AIRHT(AMAS,CFA,AMA,AKA,DDO,DDT,NNRQW,WWIDTH,
FRRPCH,FFPCH,FFTK,CCONST,CCPOW,ANGLE)
FFEE(1)=FINEFF(110,DDT,DDO,FFTK,FFMK,HCO(1))
/MSI(1)=AMS
CPAS(1)=CPA
14 AIRN(1)=0.
DO 16 IA=1,NIMER
I=MERGE(110,IA,1)
HR(1,I)=0.
PRM(110,1,I)=0.
IMC(1A)=MERGE(110,IA,2)
16 CONTINUE
IEN=0
***** FIND TUBE REFRIG. FLOWS INTO CONDENSER *****
DO 86 NUMB=1,NIST
I=1START(110,NUMB)
TRM(110,1,I)=T1
PRM(110,1,I)=P1
HR(1,I)=H1
XRM(110,1,I)=1.
VGM(110,1,I)=V1
TRI=T1
PRI=P1
HRI=H1
XRI=1.
GOTO 34
***** FIND NEXT TUBE *****
18 CONTINUE
I=IFRM(110,JJ)
IF(I.EQ.0)GOTO 85
IFCMY(1).EQ.1)GOTO 22
TRI=TRM(110,2,JJ)
PRI=PRM(110,2,JJ)
HRI=HR(2,JJ)
XRI=XRM(110,2,JJ)
VLM(110,1,I)=VLM(110,2,JJ)
VGM(110,1,I)=VGM(110,2,JJ)
XLS(1,I)=XLS(2,JJ)
XVS(1,I)=XVS(2,JJ)
VLS(1,I)=VLS(2,JJ)

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*****
CONDHX *****
2310. 00 VVS(1,1)=VVS(2,JJ)
2320. 00 GOTO 32
2330. 00 C**** REFRIG. MERGES IN TUBE 1. FIND INLET STATE
2340. 00 22 HR(1,1)=HR(1,1)+HR(2,JJ)*FLOW(110,JJ)
2350. 00 PRM(110,1,1)=PRM(110,1,1)+PRM(110,2,JJ)*FLOW(110,JJ)
2360. 00 DO 23 IP=1,NIMER
2370. 00 23 IF(MERGE(110,IP,1).EQ.1)GOTO 24
2380. 00 24 IMC(IP)=MC(IP)-1
2390. 00 IF(IMC(IP).NE.0)GOTO 86
2400. 00 HR(1,1)=HR(1,1)/FLOW(110,1)
2410. 00 PRM(110,1,1)=PRM(110,1,1)/FLOW(110,1)
2420. 00 HRI=HR(1,1)
2430. 00 PRI=PRM(110,1,1)
2440. 00 PARI=PRI/14.6959
2450. 00 CALL HP111(M1,HRI,PARI,XM,ACC,TKRI,XRI,XL,XV,VL,VV)
2460. 00 TRI=TKRI*1.8-459.67
2470. 00 IF(XRI.EQ.0.)THEN
2480. 00 VLM(110,1,1)=16.01846*VL/WM
2490. 00 ELSE
2500. 00 IF(XRI.LT.1.)THEN
2510. 00 WMV=W1*(1.-XV)+W2*XV
2520. 00 VGM(110,1,1)=16.01846*VV/WMV
2530. 00 WML=W1*(1.-XL)+W2*XL
2540. 00 VLM(110,1,1)=16.01846*VL/WML
2550. 00 XLS(1,1)=XL
2560. 00 XVS(1,1)=XV
2570. 00 VLS(1,1)=VL
2580. 00 VVS(1,1)=VV
2590. 00 ELSE
2600. 00 VGM(110,1,1)=16.01846*VV/WM
2610. 00 END IF
2620. 00 END IF
2630. 00 32 CONTINUE
2640. 00 TRM(110,1,1)=TRI
2650. 00 PRM(110,1,1)=PRI
2660. 00 HR(1,1)=HRI
2670. 00 XRM(110,1,1)=XRI
2680. 00
2690. 00 C
2700. 00 C**** TUBE 1 SELECTION FOR CALCULATION DONE
2710. 00 C**** COMPUTE HEAT TRANSFER & REFRIG. PRESSURE DROP FOR TUBE 1
2720. 00 C**** FIND REFRIG. STATE AT OUTLET
2730. 00 34 CONTINUE
2740. 00 TKRI=(TRI+459.67)/1.8
2750. 00 RMS=RMASS*FLOW(110,1)
2760. 00 TRE=TRI-DTR(1)
2770. 00 PRE=PRI-DPR(1)
2780. 00 HRE=HRI-DHR(1)
2790. 00 XRE=XRI-DXR(1)
2800. 00 PRAV=0.5*(PRI+PRE)
2810. 00 PARAV=PARAV/14.6959
2820. 00 ICT=IDEPTH(110,1)
2830. 00 AMS=AMSI(1CT)
2840. 00 CPA=CPAS(1CT)
2850. 00 TAI=TAIR(1,1)
2860. 00 IF((TAI-.1).GE.TRI)TAI=TRI-.0.1
2870. 00 PRINT 806,1,ICT,XRI,PRI,TRI,HRI,TAI
2880. 00 806 FORIAT(' 1,ICT,XRI,PRI,TRI,HRI,TAI=',214,F6.2,5F6.1)
IF(XRI.EQ.1.)GOTO 42
CDE
CDE

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***** CONDHX *****

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2890. IF(XRI.GT.0.0001)GOTO 58
2900. C
2910. C**** CASE 1
2920. C**** SUBCOOLED LIQUID AT TUBE 1 INLET
2930. DO 36 IS=1,5
2940. TRAV=0.5*(TRI+TRE)
2950. TKRAV=(TRAV+459.67)/1.8
2960. ANR=VISCON(1,TRAV,XW)
2970. AKR=VISCON(2,TRAV,XW)
2980. CALL VOLIT1(N0,TKRAV,VL,XI,H,CV,CPR)
2990. CALL HCVCP(N5,TKRAV,VL,XI,H,CV,CPR)
3000. HI=SPHTC(CPR,AKR,AKR,RMS,DDI)
3010. UAO=OVLHTC(AQ,API,APN,APQ,AF,HI,HD,HP,HCC(1CT),FFEE(1CT))
3020. QQ=CPA*AMS*(1.-EXP(-AQ*UAQ/(CPA*AMS)))
3030. Q1=1.-EXP(-QQ/(CPR*RMS))
3040. Q1=CPR*RMS*(TRI-TAI)*Q1
3050. HRE=HRI-Q1/RMS
3060. TRE=TRI+(HRE-HRI)/CPR
3070. IF(TRE.GT.TAI)GOTO 38
3080. IF(TRE.GT.TAI)GOTO 38
3090. 36 CONTINUE
3100. 38 XRE=0.
3110. AL1=WIDTH+20.*DDI
3120. VL=VL*16.01845/WM
3130. PRE=PRI-SPIDP1(RMS,AL1,DDI,VL,AMR)
3140. TAE=TAI+Q1/(CPA*AMS)
3150. GOTO 78
3160.
3170. C
3180. C**** CASE 2 & 3
3190. C**** SUPERHEATED VAPOR AT TUBE 1 INLET
3200. 42 CONTINUE
3210. DO 46 ISUP=1,6
3220. CXZ PRINT 807,PARAV,TKRAV
3230. 807 FORMAT(' PARAV,TKRAV=',2F8.2)
3240. TRAV=0.5*(TRI+TRE)
3250. TKRAV=(TRAV+459.67)/1.8
3260. CALL VOLIT1(N1,TKRAV,PARAV,XM,VV)
3270. CALL HCVCP(N5,TKRAV,VV,XM,H,CV,CPR)
3280. ANR=VISCON(N3,TRAV,XW)
3290. AKR=VISCON(N4,TRAV,XW)
3300. HI=SPHTC(CPR,AKR,AKR,RMS,DDI)
3310. UAO=OVLHTC(AQ,API,APM,APC,AF,HI,HD,HP,HCC(1CT),FFEE(1CT))
3320. QQ=CPA*AMS*(1.-EXP(-AQ*UAQ/(CPA*AMS)))
3330. Q1=1.-EXP(-QQ/(CPR*RMS))
3340. Q1=CPR*RMS*(TRI-TAI)*Q1
3350. HRE=HRI-Q1/RMS
3360. TRE=TRI-(HRI-HRE)/CPR
3370. IF(TSUP.EQ.1)GOTO 44
3380. HDIF=HRE1-HRE
3390. IF(ABS(HDIF).LT.0.01)GOTO 48
3400. 44 HRE1=HRE
3410. 46 CONTINUE
3420. WRITE(6,650)1,HDIF
3430. 650 FORMAT(' CASE 2 TUBE DOES NOT CONVERGE, 1,HDIF=',14,1PE14.4)
3440. 48 CONTINUE
3450. CXZ PRINT 808,HRE
3460. 808 FORMAT(' HRE=',F8.2)
      CALL DEWTEH(N0,PARAV,XM,TKD,XLD)
      CALL VOLIT1(N1,TKD,PARAV,XM,VVD)

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CONDHX *****
3470. 00
3480. 00
3490. 00
3500. 00
3510. 00
3520. 00
3530. 00
3540. 00
3550. 00
3560. 00
3570. 00
3580. 00
3590. 00
3600. 00
3610. 00
3620. 00
3630. 00
3640. 00
3650. 00
3660. 00
3670. 00
3680. 00
3690. 00
3700. 00
3710. 00
3720. 00
3730. 00
3740. 00
3750. 00
3760. 00
3770. 00
3780. 00
3790. 00
3800. 00
3810. 00
3820. 00
3830. 00
3840. 00
3850. 00
3860. 00
3870. 00
3880. 00
3890. 00
3900. 00
3910. 00
3920. 00
3930. 00
3940. 00
3950. 00
3960. 00
3970. 00
3980. 00
3990. 00
4000. 00
4010. 00
4020. 00
4030. 00
4040. 00

      CALL HCVCP(N1,TKD,VVD,XM,HGV,CV,CP)
      PRINT 809,HGV
      809 FORMAT(' HGV=',F8.2)
      IF(HRE.GE.HGV) THEN
        XRE=1.
        Q2=0.
        XSUP=1.
        VW=VVD*16.01846/WM
        GOTO 52
      END IF

C
C**** CASE 3
C**** 2-PHASE FLOW AT OUTLET OF A TUBE WITH SUPERHEATED VAPOR AT INLET
C**** COMPUTE HEAT TRANSFERED BY PART OF A TUBE WITH SUPERHEATED VAPOR
      TKRAV=0.5*(TKRI+TKD)
      TRAV=TKRAV*1.8-459.67
      TD=TKD*1.8-459.67
      VMAV=0.5*(VVD+VCM(110,1,1)*WM/16.01846)
      VW=VMAV*16.01846/WM
      CALL HCVCP(N5,TKRAV,VMAV,XM,H,CV,CPR)
      AMR=VISCOR(3,TRAV,XW)
      AKR=VISCOR(2,TRAV,XW)
      HI=SPHTC(CPR,AMR,AKR,RMS,DDI)
      UAO=OVLHTC(AO,API,APM,APG,AF,HI,HD,HP,HCO(1CT),FFEE(1CT))
      Q1=RHS*(HRI-HGV)
      QQ=CPR*AMS*(1.-EXP(-AO*UAO/(CPR*AMS)))
      XSUP=-CPR*RMS*ALOG(1.-Q1/(RMS*CPR*(TRI-TAI)))/QQ
      PRINT 811,XSUP
      811 FORMAT(' XSUP=',F6.2)
      IF(XSUP.GT.1.) THEN
        HRE=HGV
        Q2=0.
        XRE=1.
        XSUP=1.
        GOTO 52
      END IF

C**** COMPUTE HEAT TRANSFERED BY PART OF A TUBE WITH 2-PHASE
      TKRE=(TRE+459.67)/1.8
      DO 50 IQ=1,6
        CPR=CPRS(I)
        IF(TRE.GT.TD) TRE=TD-2.
        TRAV=0.5*(TD+TRE)
        HI=HTCCONC(TRAV,PEAV,RMS,DDI)
        UAO=OVLHTC(AO,API,APM,APG,AF,HI,HD,HP,HCO(1CT),FFEE(1CT))
        QQ=CPR*AMS*(1.-EXP(-AO*UAO/(CPR*AMS)))
        PRINT 813,CPR
        813 FORMAT(' CPR=',1PE15.6)
        Q2=(1.-XSUP)*CPR*RMS*(TD-TAI)*(1.-EXP(-QQ/(CPR*RMS)))
        HRE=HGV-Q2/RMS
        CALL HPIN(N1,HRE,PARAV,XM,ACC,TKRE,XRE,XL,XV,VL,VV)
        TRE=TKRE*1.8-459.67
        PRINT 812,HGV,HRE,TD,TRE,XRE
        812 FORMAT(' HGV,HRE,TD,TRE,XRE=',4F12.6,F6.2)
        IF(TRE.EQ.TD) THEN
          IF(IQ.GT.1) GOTO 52
          CPRS(I)=2.*CPR
          GOTO 49
        END IF

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***** CONDHX *****
4050. CPRS(1)=(HGV-HRE)/(TD-TRE)
4060. IF(10.EQ.1)GOTO 49
4070. HDIF=HRE-HRE1
4080. IF(ABS(HDIF).LT.0.01)GOTO 52
4090. 49 HRE1=HRE
4100. 50 CONTINUE
4110. WRITE(6,650)1,HDIF
4120. 650 FORMAT(' CASE 3 TUBE DOES NOT CONVERGE, 1,HDIF=',14,1F14.4)
4130. 52 Q=Q1+Q2
4140. XRAV=0.5*(1.+XRE)
4150. TAE=TA1+Q/(CPA*MTS)
4160. 810 FORMAT(' XSUP=',F8.2)
4170. CVXCDE PRINT 810,XSUP
4180. C
4190. C**** CALCULATE FRICTIONAL PRESSURE DROP IN A TUBE
4200. IF(XPAV.GT.99)THEN @SUPERHEATED VAPOR IN A TUBE ONLY
4210. AL1=WIDTH+20.*DDI
4220. DP1=SPHDP1(RMS,AL1,DDI,VW,AMR)
4230. DP2=0.
4240. ELSE @SUPERHEATED VAPOR & 2-PHASE IN A TUBE
4250. AL1=XSUP*WIDTH
4260. DP1=SPHDP1(RMS,AL1,DDI,VW,AMR)
4270. XL=0.5*(XLD+XL)
4280. XLW=XL/(W1/W2*(1.-XL)+XL)
4290. WML=W1*(1.-XL)+W2*XL
4300. TKRAV=(TRAV+459.67)/1.8
4310. PRINT 814,TKRAV,PARAV
4320. 814 FORMAT(' TKRAV,PARAV=',2F10.3)
4330. CALL VOLIT1(NC,TKRAV,PARAV,XL,VL)
4340. VLW=VL*16.01846/WML
4350. AMR=VISC(N1,TRAV,XLW)
4360. CVV PRINT 819,AMR
4370. 819 FORMAT(' AMR=',1F15.5)
4380. AL2=(1.-XSUP)*WIDTH+20.*DDI
4390. RMSX=(1.-XRAV)*RMS
4400. DP2=SPHDP1(RMSX,AL2,DDI,VLW,AMR)
4410. XV=0.5*(XM+XV)
4420. VV=0.5*(VVD+VV)
4430. CVV PRINT 818,XRAV,TRAV,XL,XV,VL,VV
4440. 818 FORMAT(' XRAV,TRAV,XL,XV,VL,VV=',F4.2,F7.2,2F5.2,2F10.4)
4450. FFE2=FEELIQ(XRAV,TRAV,XL,XV,VL,VV,XIT(110,1))
4460. DP2=DP2*FFE2
4470. END IF
4480. CVV PRINT 815,DP1,DP2
4490. 815 FORMAT(' DP1,DP2=',2F10.3)
4500. PRE=PRE-DP1-DP2
4510. GOTO 78
4520. C
4530. C**** CASE 4 & 5
4540. C**** 2-PHASE FLOW AT TUBE I INLET
4550. 58 LXRE=0
4560. X2FH=1.
4570. TKRE=(TRE+459.67)/1.8
4580. DC 63 ITPI=1.6
4590. CPR=CPRS(1)
4600. TRAV=0.5*(TRI+TRE)
4610. XRAV=0.5*(XRI+XRE)
4620. HI=HTCCON(TRAV,PREAV,RMS,DDI)

```



```

*****
CONDHX *****
00 UAO=CVLHTG(AO,API,APM,APQ,AF,H,HD,HP,HCO(IGT),FFEE(IGT))
00 QQ=CPA*AMS*(1.-EXP(-AO*UAO/(CPA*AMS)))
00 Q1=CPR*RMS*(TRI-TAI)*(1.-EXP(-QO/(CPR*RMS)))
00 HRE=IRI-Q1/RMS
00 IF(1.EQ.72)PRINT 821,HRE
00 821 FORMAT(' ',HRE=' ',F9.3)
00 CALL H7IN(N1,HRE,PARAV,XM,ACC,TKRE,XRE,XL,XV,VL,VV)
00 TRE=TKRE*1.8-459.67
00 IF(XRE.EQ.0.)THEN
00 IF(IXRE.EQ.1)GOTO 65
00 TKB=TKRI
00 CALL RUBTEM(N1,PARAV,XM,TKB,XVB)
00 CALL VOLITI(N0,TKB,PARAV,XM,VLB)
00 CALL HCVCP(N1,TKB,VLB,XM,HGL,CV,CP)
00 CALL VOLITI(N1,TKB,PARAV,XVB,VVB)
00 TB=TKB*1.8-459.67
00 TRE=TB
00 CPRS(1)=(HRI-HGL)/(TRI-TB)
00 IXRE=1
00 GOTO 62
00 END IF
00 IXRE=0
00 IF(TRI.EQ.TRE)THEN
00 IF(1TPH.GT.1)GOTO 68
00 CPRS(1)=2.*CPR
00 GOTO 62
00 END IF
00 CPRS(1)=(HRI-HRE)/(TRI-TRE)
00 IF(1TPH.EQ.1)GOTO 62
00 HDIF=HRE1-HRE
00 Q2=0.
00 IF(ABS(HDIF).LT.0.01)GOTO 68
00 62 HRE1=HRE
00 63 CONTINUE
00 WRITE(6,670)1,HDIF
00 670 FORMAT(' CASE 4 TUBE DOES NOT CONVERGE, 1,HDIF=',14,1PE14.4)
00 GOTO 68
00 65 CONTINUE
C
C**** CASE 5
C**** SURFOOLED LIQUID AT OUTLET OF TUBE WITH 2-PHASE FLOW AT INLET
C**** COMPUTE HEAT TRANSFERRED BY 2-PHASE PART OF A TUBE
00 Q1=RMS*(IRI-HGL)
00 X2PH=-CPR*RMS*ALOG(1.-Q1/(RMS*CPR*(TRI-TAI)))/QO
00 IF(X2PH.GE.1.)THEN
00 HRE=H9L
00 Q2=0.
00 X2PH=1.
00 GOTO 68
00 END IF
C**** COMPUTE HEAT TRANSFERRED BY A SINGLE-PHASE PART OF A TUBE
00 DO 67 1TPH=1,6
00 IF(1TPH.GT.1)TRE=TB-1.
00 TRAV=0.5*(TRE+TB)
00 TKRAV=(TRAV+459.67)/1.8
00 CALL VOLITI(IIO,TKRAV,PARAV,XM,VL)
00 CALL HCVCP(N5,TKRAV,VL,XM,H,CV,CPR)
00 AMR=VISCN(1,TRAV,XW)
00 5200.

```

```

*****
CONDHX *****
5210. 00 AKR=VISCOCN(2,TRAV,XV)
5220. 00 HI=SPHTC(CPR,AM3,AKR,RMS,DD1)
5230. 00 UAC=OVLHTC(AQ,API,APM,APD,AF,HI,HD,HP,HCO(ICT),FEE(ICT))
5240. 00 QQ=CFA*AMS*(1.-EXP(-AQ*UAC/(CPA*AMS)))
5250. 00 Q2=CPR*RMS*(TD-TAI)*(1.-EXP(-QO/(CPR*RMS)))*(1.-X2PH)
5260. 00 HRE=HGL-Q2/RMS
5270. 00 IF(ITPH.EQ.1)GOTO 56
5280. 00 HDIF=HGL-HRE
5290. 00 IF(ABS(HDIF).LT.0.01)GOTO 68
5300. 00 66 HRE1=HRE
5310. 00 67 CONTINUE
5320. 00 68 CONTINUE
5330. 00 CCC IF(1.EQ.72)PRINT 320,HRE
5340. 00 CCC 820 FORMAT(' HRE=',F8.2)
5350. 00 Q=Q1+Q2
5360. 00 TAE=TAI+Q/(CPA*AMS)
5370. 00 TARI(2,1)=TAE
5380. 00 C**** COMPUTE FRICTIONAL PRESSURE DROP
5390. 00 XTUDE(110,1)=X2PH
5400. 00 IF(X2PH.LT.1.)THEN
5410. 00 XML=0.5*(XLS(1,1)+XM)
5420. 00 XNV=0.5*(XVS(1,1)+XVB)
5430. 00 XML=XVL/(W1/W2*(1.-XML)+XML)
5440. 00 TRT=0.5*(TRI+TB)
5450. 00 AMR=VISCOCN(1,TRT,XWL)
5460. 00 VLW=0.5*(VLM(110,1,1)+VLB*16.01846/WM)
5470. 00 AL1=WIDTH*X2PH
5480. 00 RMSX=(1.-XRAV)*RMS
5490. 00 DP1=SFHDP1(RMSX,AL1,DD1,VLW,AMR)
5500. 00 WML=W1*(1.-XML)+W2*XML
5510. 00 VML=0.5*(VLB+VLS(1,1))
5520. 00 VMV=0.5*(VVS(1,1)+VVB)
5530. 00 FEE2=FEELIO(XRAV,TRT,XML,XMV,VML,VMV,XTT(110,1))
5540. 00 DP1=DP1+FEE2
5550. 00 AL2=(1.-X2PH)*WIDTH+20.*DD1
5560. 00 VLW=VL*16.01846/WM
5570. 00 DP2=SFHDP1(RMS,AL2,DD1,VLW,AMR)
5580. 00 ELSE @2-PHASE FLOW IN TUBE ONLY
5590. 00 WML=W1*(1.-XL)+W2*XL
5600. 00 VLW=0.5*(VLM(110,1,1)+16.01846*VL/WML)
5610. 00 XL=0.5*(XLS(1,1)+XL)
5620. 00 XWL=XL/(W1/W2*(1.-XL)+XL)
5630. 00 AMR=VISCOCN(1,TRAV,XWL)
5640. 00 AL1=WIDTH+20.*DD1
5650. 00 RMSX=(1.-XRAV)*RMS
5660. 00 DP1=SFHDP1(RMSX,AL1,DD1,VLW,AMR)
5670. 00 VL=0.5*(VL+VLS(1,1))
5680. 00 VV=0.5*(VV+VVS(1,1))
5690. 00 FEE2=FEELIO(XRAV,TRAV,XL,XV,VL,VV,XTT(110,1))
5700. 00 DP1=DP1+FEE2
5710. 00 DP2=0.
5720. 00 END IF
5730. 00 PRE=FRI-DP1-DP2
5740. 00 73 CONTINUE
5750. 00 C****FOLLOWING STATEMENTS FOR CALC. OF DYNAMIC PRESSURE DROP ARE
5760. 00 CC SKIPPED. THEY PROVIDE LITTLE CORRECTION FOR PRESSURE DROP
5770. 00 CC AT EXPENSE OF A LOT OF COMPUTING.
5780. 00 CC IF(XRI.LT.1..AND.XPE.GT.0.1)THEN

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*****
CONDHX *****
5790. CC DIX=DYNDP2(PRI,TRI,PRE,TRE,RMS,DDI)
5800. CC ELSE
5810. CC DIX=DYNDP1(PRI,TRI,PRE,HRE,RMS,DDI)
5820. CC END IF
5830. CC PRE=PRE-DIX
5840. CC
5850. C *****
5860. C ENTHALPY & PRESSURE AT TUBE 1 OUTLET ARE KNOWN
5870. C *****
5880. C FIND TEMP., QUALITY & SPEC. VOLUME
5890. C TKRE=(TRE+459.67)/1.8
5900. C PARE=PRE/14.6959
5910. C CALL HP1N1,HRE,PARE,XM,ACC,TKRE,XRE,XL,XV,VL,VV)
5920. C TRM(110,2,1)=TKRE*1.8-459.67
5930. C PRI(110,2,1)=PRE
5940. C HRE(2,1)=HRE
5950. C XRM(110,2,1)=XRE
5960. C DTR(1)=TRI-TRM(110,2,1)
5970. C DPR(1)=PRI-PRE
5980. C DXR(1)=XRI-XRE
5990. C IF(XRE.EQ.0.) THEN
6000. C VLM(110,2,1)=16.01846*VL/XM
6010. C ELSE
6020. C IF(XRE.LT.1.) THEN
6030. C WMV=W1*(1.-XV)+W2*XV
6040. C VGM(110,2,1)=16.01846*VV/WMV
6050. C VML=W1*(1.-XL)+W2*XL
6060. C VLM(110,2,1)=16.01846*VL/VML
6070. C XLS(2,1)=XL
6080. C XVS(2,1)=XV
6090. C VLS(2,1)=VL
6100. C VVS(2,1)=VV
6110. C ELSE
6120. C VGM(110,2,1)=16.01846*VV/WM
6130. C END IF
6140. C JJ=1
6150. C TAIR(2,ICT+1)=(TAE+AIRN(1CT)*TAIR(2,1CT+1))/
6160. C &(AIRN(1CT)+1.)
6170. C AIRN(1CT)=AIRN(1CT)+1.
6180. C GO TO 18
6190. C 85 IEN=IEN+1
6200. C IEND(IEN)=JJ
6210. C 86 CONTINUE
6220. C ***** ALL TURNS OF COIL COMPUTED. CHECK IF CONVERGENCE OBTAINED
6230. C H2=0.
6240. C DO 90 I=1,IEN
6250. C IE=IEND(I)
6260. C H2=H2+HR(2,IE)*FLOW(110,IE)
6270. C 90 CONTINUE
6280. C H2=H2+NSFCT(110)
6290. C IF(1AIR.EQ.1) GO TO 92
6300. C H2PH2=H2F-H2
6310. C PRINT 850,H2,H2PH2
6320. C FORNAT(' H2,H2PH2=',2F8.2)
6330. C IF(1AIR.LT.3) GO TO 91
6340. C IF(ABS(H2PH2).LT.0.02.AND.A3S(H2PH2).LT.1) GO TO 202
6350. C 91 H2PH2=H2PH2
6360. C 92 H2F=H2
C

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```

***** CONDHX *****
6370. C**** CONVERGENCE NOT OBTAINED
6380. C**** PREPARE AIR SIDE DATA FOR NEW LOOP
6390. DO 94 I=1,NNDIEP
6400. J=I+1
6410. TAIR(I,J)=TAIR(2,J)
6420. 94 TAIR(2,J)=0.
6430. NTOT=0
6440. NDEEP=NDEP(110)-1
6450. DO 149 ICT=1,NDEEP
6460. NTOT=NTOT+NTUB(110,ICT)
6470. NDIF=NTUB(110,ICT)-NTUB(110,ICT+1)
6480.
6490. C
6500. IF (NDIF.EQ.0) THEN
6510. IA=NTOT+1
6520. IB=NTOT+NTUB(110,ICT+1)
6530. DO 142 I=IA,IB
6540. J=JFROM(110,I)
6550. 142 TAIR(1,I)=TAIR(2,J)
6560. END IF
6570. C
6580. IF (NDIF.LT.0) THEN
6590. IA=NTOT+1
6600. IB=NTOT+NTUB(110,ICT)-1
6610. DO 143 I=IA,IB
6620. J=JFROM(110,I)
6630. 143 TAIR(1,I)=0.5*(TAIR(1,I)+TAIR(2,J))
6640. IA=NTOT-NTUB(110,ICT)+1
6650. IB=NTOT-1
6660. T=0.
6670. DO 144 I=IA,IB
6680. T=T+TAIR(2,I)
6690. T=T*(AMSI(1CT)-AMSI(1CT+1))+TAIR(2,NTOT)*AMSI(1CT)
6700. SEG=(NTUB(110,ICT+1)-NTUB(110,ICT)+1)*AMSI(1CT+1)
6710. T=T/SEG
6720. IA=NTOT+NTUB(110,ICT)
6730. IB=NTOT+NTUB(110,ICT+1)
6740. DO 145 I=IA,IB
6750. 145 TAIR(1,I)=0.5*(TAIR(1,I)+T)
6760. END IF
6770. C
6780. IF (NDIF.GT.0) THEN
6790. IA=NTOT-NDIF+1
6800. TCOR=0.
6810. DO 146 I=IA,NTOT
6820. TCOR=TCOR+TAIR(2,I)
6830. TCOR=TCOR/NTUB(110,ICT+1)
6840. IA=NTOT+1
6850. IB=NTOT+NTUB(110,ICT+1)
6860. DO 147 I=IA,IB
6870. J=JFROM(110,I)
6880. 147 TAIR(1,I)=.5*(TAIR(1,I)+(TAIR(2,J)+TCOR)*AMSI(1CT)/AMSI(1CT+1))
6890. END IF
6900. C
6910. 149 CONTINUE
6920. 100 CONTINUE
6930. C*****
6940. C***** END OF MAIN LOOP
        WRITE(6,505)H2PII2

```



```

***** CONDHX *****
6950. 00 202 CONTINUE
6960. 00 P2=0.
6970. 00 DO 206 I=1,IEN
6980. 00 IE=IEND(1)
6990. 00 P2=P2+PRM(110,2,IE)*FLOW(110,IE)
7000. 00 206 CONTINUE
7010. 00 P2=P2*NSECT(110)
7020. 00 PA2=P2/14.6959
7030. 00 CALL HPIN(NO,H2,PA2,X11,ACC,TK2,X2,XL,XV,VL,VV)
7040. 00 T2=TK2*1.9-459.67
7050. 00 500 FORMAT(/2X,'INPUT DATA TO CONDHX: '//2X,
7060. 00 &'T',10X,'P',10X,'TAIR',7X,'RH',9X,'FMASS',/5(1PE11.3))
7070. 00 WRITE(6,501)T1,P1,I1,X1,T2,P2,H2,X2
7080. 00 501 FORMAT(/2X,'CONDENSER ITERATION: '//
7090. 00 &2X,'T',10X,'P',10X,'H',10X,'X',/4(1PE11.3)/4(1PE11.3))
7100. 00 505 FORMAT(' CONDHX DOES NOT CONVERGE, MAX. ERROR=',1PE11.3,
7110. 00 1' (BTU/LB)')
7120. 00 RETURN
7130. 00 END

```

END ELT. ERRORS: NONE. TIME: 0.835 SEC. IMAGE COUNT: 713

@HDG,P ***** DEBUTE ***** .L,0

```

***** DBUBTE *****
@ELT,L DD,DBUBTE
ELT 8R1 S74Q1C 07/21/84 15:55:00 (0)
10. 00 SUBROUTINE DBUBTE(IG,P,X,T,XV)
20. 00 C
30. 00 REAL*8 P,T,XV
40. 00 C
50. 00 C**** PURPOSE:
60. 00 TO CALCULATE IN DOUBLE PRECISION A BUBBLE POINT
70. 00 TEMPERATURE OF A NON-ATGOTROPIC BINARY MIXTURE FROM GIVEN
80. 00 PRESSURE AND COMPOSITION
90. 00 C
100. 00 C**** INPUT:
110. 00 IG = 0, IF GUESS OF TEMPERATURE IS NOT GIVEN
120. 00 = 1, IF GUESS OF TEMPERATURE IS GIVEN
130. 00 P - PRESSURE (STD ATM)
140. 00 T - GUESS OF TEMPERATURE, OPTIONAL (K)
150. 00 X - MOLAR CONCENTRATION (FRACTION OF LESS VOLATILE COMPONENT)
160. 00 C
170. 00 C**** OUTPUT:
180. 00 T - BINARY MIXTURE TEMPERATURE AT BUBBLE POINT (K)
190. 00 XV - MOLAR CONCENTRATION OF LESS VOLATILE COMPONENT
200. 00 IN SAT. VAPOR IN EQUILIBRIUM WITH LIQUID (-)
210. 00 C
220. 00 C**** SUBPROGRAMS CALLED BY DBUBTE:
230. 00 EBUBTE,DOLITY
240. 00 C
250. 00 REAL*8 SLOPE,TLAST,PLAST,XVLAST,XL,T1,T2,
260. 00 XDIF1,XDIF2,DT,XQ
270. 00 * DATA SLOPE/0.00/,PLAST/0.00/
280. 00 C
290. 00 IF (ABS(P-PLAST)).GT.1.E-4)GOTO 10
300. 00 IF (ABS(X-XLAST)).GT.1.E-3)GOTO 10
310. 00 T=TLAST
320. 00 XV=XVLAST
330. 00 RETURN
340. 00 C
350. 00 10 PS=P
360. 00 IF (IG.NE.1)T=EBUBTE(PS,X)
370. 00 DO 50 I=1,20
380. 00 CALL DOLITY(T,P,X,XQ,XV,XL)
390. 00 T2=T
400. 00 XDIF2=X-XL
410. 00 IF (I.NE.1)GOTO 20
420. 00 15 T1=T2
430. 00 XDIF1=XDIF2
440. 00 IF (SLOPE.NE.0.)THEN
450. 00 DT=XDIF2*SLOPE
460. 00 IF (ABS(DT).GT.10.)DT=SIGN(10.,DT)
470. 00 T=T2-DT
480. 00 GOTO 50
490. 00 END IF
500. 00 T=T2+5.
510. 00 IF (XDIF2.GT.0.)T=T2-5.
520. 00 GOTO 50
530. 00 C
540. 00 20 IF (XDIF1.EQ.XDIF2)GOTO 15
550. 00 SLOPE=(T2-T1)/(XDIF2-XDIF1)
560. 00 IF (DABS(XDIF1).LT.DABS(XDIF2))GOTO 30

```

DATE 072184

```

***** DBUBTE *****
520. 00
530. 00
540. 00
550. 00
560. 00
570. 00
580. 00
590. 00
600. 00
610. 00
620. 00
630. 00
640. 00
650. 00
660. 00

      T1=T2
      XDIF1=XDIF2
30  DT=XDIF1*SLOPE
      IF(DABS(DT).LT.0.000001)GOTO 100
      IF(ABS(DT).GT.10.)DT=SIGN(10.,DT)
      T=T1-DT
50  CONTINUE
      WRITE(6,600)XDIF2
600  FORMAT(' ERROR 600 IN DBUBTE, XDIF2=',1PE12.4)
100  PLAST=P
      XLAST=X
      TLAST=T
      XLAST=XV
      RETURN
      END

END ELT.  ERRORS: NONE.  TIME: 0.131 SEC.  IMAGE COUNT: 71
@HDG,P ***** DDENFA ***** .L,0

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```

@ELT,L DD.DDENFA
ELT 8R1 S74Q1C 07/21/84 15:55:00 (0)
10. 00 DOUBLE PRECISION FUNCTION DDENFA(XM,P,H0,GS)
20. 00
30. 00 C**** PURPOSE:
40. 00 TO CALCULATE IN DOUBLE PRECISION
50. 00 DENSITY OF NON-AZEOTROPIC TWO-PHASE MIXTURE
60. 00 IN THE FANNQ FLOW FOR A GIVEN PRESSURE
70. 00
80. 00 C**** INPUT DATA:
90. 00 H0 - REFRIG. TOTAL ENTHALPY (BTU/LM)
100. 00 GG = G*G/(64.4*778.104) (BTU*LM/FT**6)
110. 00 C WHERE G - REFRIG. MASS FLUX (LM/(SEC*FT**2))
120. 00 P - REFRIG. PRESSURE AT WHICH DENSITY IS DESIRED (PSIA)
130. 00 XM - MOLAR COMPOSITION (FRACTION OF LESS VOLATILE COMPONENT)
140. 00 C**** OUTPUT DATA:
150. 00 DDENFA - DENSITY (BTU/LM R)
160. 00
170. 00 C**** SUBPROGRAMS CALLED BY DDENFA:
180. 00 DQUBTE,DH,DQLITY,DVOL1,EPURTE,VISCON
190. 00
200. 00 REAL*8 DH,DVOL1
210. 00 REAL*8 SLOPE,TKB,XVB,PA,TK2,YQ,XV,XL,VNV,YV,VML,HL,HFG,
220. 00 *VMV,VML,VV,VL,VFG,P,S,C,XQG,DIFXQ2,DIFXQ1,TK1
230. 00 COMMON/RDATA2/W1,W2,TC1,TC2
240. 00 DATA SLOPE/0.00/NO,N1/O,1/
250. 00
260. 00 C
270. 00 IF(P.GT.0.)GO TO 10
280. 00 WRITE(6,600)P
290. 00 600 FORMAT(' ERROR IN CALLING DDENFA, P=',1PE15.5,' PSIA')
300. 00 RETURN
310. 00 C
320. 00 10 PA=P/14.6959
330. 00 CALL DQUBTE(NO,PA,XM,TK3,XVB)
340. 00 TKD=TKB*9.5-20.*(1.55-XM)
350. 00 TK2=0.9*TKB+0.1*TKD
360. 00 DO 100 I=1,20
370. 00 IF(TK2.LT.TKB)THEN
380. 00 TK2=TKB
390. 00 XL=XM
400. 00 XV=XVB
410. 00 XQ=0.00
420. 00 ELSE
430. 00 CALL DQLITY(TK2,PA,XM,XQ,XV,XL)
440. 00 END IF
450. 00 VMV=DVOL1(N1,TK2,PA,XV)
460. 00 HV=DH(TK2,VNV,XV)
470. 00 VML=DVOL1(N0,TK2,PA,XL)
480. 00 HL=DH(TK2,VML,XL)
490. 00 HFG=HV-HL
500. 00 VMV=W1*(1.-XV)+W2*XV
510. 00 VML=W1*(1.-XL)+W2*XL
520. 00 VV=VMV*16.01846/VMV
530. 00 VL=VML*16.01846/VML
540. 00 VFG=VV-VL
550. 00 R=GG*VFG*VFG
560. 00 S=2.*GG*VL*VFG+HFG
570. 00 C=GG*VL*VL+HL-H0

```



```

***** DDENFA *****
570. 00
580. 00
590. 00
600. 00
610. 00
620. 00
630. 00
640. 00
650. 00
660. 00
670. 00
680. 00
690. 00
700. 00
710. 00
720. 00
730. 00
740. 00
750. 00
760. 00
770. 00
780. 00
790. 00
800. 00
810. 00
820. 00
830. 00
840. 00
850. 00
860. 00
870. 00
880. 00
890. 00
900. 00

      C
      XQ=(-S+DSQRT(S*S-4.*R*C))/(2.*R)
      DIFXQ2=XQ-Q
      IF(ABS(XQ).LT.1.E-5.AND.ABS(XQ2).LT.1.E-5)GOTO 105
      IF(XQ.EQ.0.DD.AND.XQ2.LT.0.)GOTO 110
      IF(DABS(DIFXQ2).LT.5.E-8)GOTO 105

      IF(1.GT.1)GOTO 70
      TK1=TK2
      DIFXQ1=DIFXQ2
      IF(SLOPE.NE.0.DD)THEN
        TDEL=DIFXQ2*SLOPE
        TD=ABS(TDEL)
        TD=AMIN1(2.,TD)
        TK2=TK1-SIGN(TD,TDEL)
      ELSE
        TK2=TK1+SIGN(1.,DIFXQ2)
      END IF
      GOTO 100

      C
      70 SLOPE=(TK2-TK1)/(DIFXQ2-DIFXQ1)
      IF(ABS(XQ).LT.1.E-5)GOTO 75
      IF(DABS(DIFXQ1).LT.DABS(DIFXQ2))GOTO 80
      75 TK1=TK2
      DIFXQ1=DIFXQ2
      80 TK2=TK1-DIFXQ1*SLOPE
      100 CONTINUE
      PRINT 602,DIFXQ2,XQ
      602 FORMAT(' DDENFA DOES NOT CONVERGE, DIFXQ2=',1PE13.4,
1' XQ=',1PE13.4)
      105 XQ=0.5*(XQ+XQ2)
      110 CONTINUE
      DDENFA=1./(VL+XQ2*VFG)
      RETURN
      END

END ELT. ERRORS: NONE. TIME: 0.155 SEC. IMAGE COUNT: 90
@HDG,P ***** DDEWTE ***** .L,0

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```

@ELT,L DD.DDEWTE
ELT 8R1 S7401C 07/21/84 15:55:00 (0)
10. 00 C SUBROUTINE DDEWTE(IG,P,X,T,XL)
20. 00 C REAL*8 P,T,XL
30. 00 C
40. 00 C ***** PURPOSE:
50. 00 C TO CALC. IN DOUBLE PRECISION A DEW POINT TEMPERATURE
60. 00 C OF A NON-AZEOTROPIC MIXTURE FROM GIVEN PRESSURE
70. 00 C AND COMPOSITION
80. 00 C
90. 00 C ***** INPUT:
100. 00 C IG = 0, IF GUESS OF TEMPERATURE IS NOT GIVEN
110. 00 C = 1, IF GUESS OF TEMPERATURE IS GIVEN
120. 00 C P - PRESSURE (STD ATM)
130. 00 C T - GUESS OF TEMPERATURE, OPTIONAL (K)
140. 00 C X - MOLAR CONCENTRATION (FRACTION OF LESS VOLATILE COMPONENT)
150. 00 C
160. 00 C ***** OUTPUT:
170. 00 C T - BINARY MIXTURE TEMPERATURE AT DEW POINT (K)
180. 00 C XL - MOLAR CONCENTRATION OF LESS VOLATILE COMPONENT
190. 00 C IN SAT. LIQUID IN EQUILIBRIUM WITH VAPOR (-)
200. 00 C
210. 00 C ***** SUBPROGRAMS CALLED BY DDEWTE:
220. 00 C EBUBTE,DOLITY
230. 00 C
240. 00 C
250. 00 C REAL*8 SLOPE,TLAST,PLAST,XLLAST,XV,T1,T2,
260. 00 C * XDIF1,XDIF2,DT,XQ
270. 00 C
280. 00 C DATA SLOPE/0.D0/,PLAST/0.D0/
290. 00 C
300. 00 C IF(DABS(P-PLAST).GT.1.E-4)GOTO 10
310. 00 C IF(DABS(X-XLAST).GT.1.E-3)GOTO 10
320. 00 C T=TLAST
330. 00 C XL=XLLAST
340. 00 C RETURN
350. 00 C
360. 00 C 10 IF(IG.NE.1)THEN
370. 00 C PS=P
380. 00 C TS=EBUBTE(PS,X)+9.5-20.*( .55-X)
390. 00 C T=TS
400. 00 C END IF
410. 00 C DO 50 I=1,20
420. 00 C CALL DOLITY(T,P,X,XQ,XV,XL)
430. 00 C T2=T
440. 00 C XDIF2=X-XV
450. 00 C IF(1.NE.1)GOTO 20
460. 00 C 15 T'=T2
470. 00 C XDIF1=XDIF2
480. 00 C IF(SLOPE.NE.0.)THEN
490. 00 C DT=XDIF2*SLOPE
500. 00 C IF(DABS(DT).GT.10.)DT=SIGN(10.,DT)
510. 00 C T=T2-DT
520. 00 C GOTO 50
530. 00 C END IF
540. 00 C T=T2+5.
550. 00 C IF(XDIF2.LT.0.)T=T2-5.
560. 00 C GOTO 50

```

DATE 072184

```

***** DDEWTE *****
C
540. 00
550. 00
560. 00
570. 00
580. 00
590. 00
600. 00
610. 00
620. 00
630. 00
640. 00
650. 00
660. 00
670. 00
680. 00
690. 00
700. 00
710. 00
720. 00

20 IF(XDIF1.EQ.XDIF2)GOTO 15
   SLOPE=(T2-T1)/(XDIF2-XDIF1)
   IF(DABS(XDIF1).LT.DABS(XDIF2))GOTO 30
   T1=T2
   XDIF1=XDIF2
30 DT=XDIF1*SLOPE
   IF(DABS(DT).LT.0.000001)GOTO 100
   IF(DABS(DT).GT.10.)DT=SIGN(10.,DT)
   T=T1-DT
50 CONTINUE
   WRITE(6,600)IG,P,X,XDIF2
600 FORMAT(' DDEWTE DID NOT CONVERGE',/' IG,P,X,XDIF2=',14,4F8.3)
100 PLAST=P
   XLAST=X
   TLAST=T
   XLLAST=XL
   RETURN
   END

```

END ELT. ERRORS: NONE. TIME: 0.141 SEC. IMAGE COUNT: 75

CHDG,P ***** DENTRO ***** .L,0

***** DENTRO *****

@ELT, L DD.DENTRO

ELT 8R1 S74Q1C 07/21/84 15:55:01 (0)

DOUBLE PRECISION FUNCTION DENTRO(T,V,X)

00 00 C

10. 00 C

20. 00 C

30. 00 C

40. 00 C

50. 00 C

60. 00 C

70. 00 C

80. 00 C

90. 00 C

100. 00 C

110. 00 C

120. 00 C

130. 00 C

140. 00 C

150. 00 C

160. 00 C

170. 00 C

180. 00 C

190. 00 C

200. 00 C

210. 00 C

220. 00 C

230. 00 C

240. 00 C

250. 00 C

260. 00 C

270. 00 C

280. 00 C

290. 00 C

300. 00 C

310. 00 C

320. 00 C

330. 00 C

340. 00 C

350. 00 C

360. 00 C

370. 00 C

380. 00 C

390. 00 C

400. 00 C

410. 00 C

420. 00 C

430. 00 C

440. 00 C

450. 00 C

460. 00 C

470. 00 C

REAL*8 T,V,X

C**** PURPOSE:

C TO CALCULATE IN DOUBLE PRECISION REFRIGERANT ENTROPY

C

C**** INPUT:

C T - REFRIG. TEMPERATURE (K)

C V - REFRIG. SPEC VOLUME (L/MOL)

C W1 - MOLECULAR WEIGHT OF MORE VOLATILE COMPONENT (G)

C W2 - MOLECULAR WEIGHT OF LESS VOLATILE COMPONENT (G)

C X - MOLAR CONCENTRATION OF A LESS VOLATILE REFRIG. (-)

C * - CONSTANTS A,B,C1,D1, AS PER COMMON STATEMENT /PARAM/

C

C**** OUTPUT:

C DENTRO - ENTROPY (BTU/(LB*F))

C

C**** SUBPROGRAMS CALLED BY DENTRO:

C HPAR

C

REAL*8 W1, CONV, S

COMMON/PARAM/A,B,C1,C2,D1,D2

COMMON/RDATA2/W1,W2

DATA R/0.08206/

C

WM=(1.-X)*W1+X*W2

CONV=453.5924/(1.055056*WM)

IQ=1

TK=T

XM=X

CALL HPAR(IQ,TK,XM)

S=(C1*B-A*D1)/B**2*DLOG((V+R)/V)+A*D1/B/(V*B)

S=S-R*B/A./((V-B/A.)*2*(4.*V-3*B/A.))

S=S-R*T*D1*V/2./((V-B/A.)*3*(2.*V-B/A.))

S=S+(1.-X)*0.32098793924*X*0.401655769501

S=S+R*(X*DLOG(V/0.0632978)+(1.-X)*DLOG(V/0.0776799))

S=0.101325*S

S=S+(0.0116393*(1.-X)+0.013966*X)*DLOG(T/233.15)

S=S+(2.163944E-4*(1.-X)+1.540093E-4*X)*(T-233.15)

S=S-(0.512035E-8*(1.-X)+1.533052E-9*X)*(T**2-233.15**2)

IF(X.EQ.1.D0.OR.X.EQ.0.D0)GOTO 1000

S=S-0.101325*R*(X*DLOG(X)+(1.-X)*DLOG(1.-X))

1000 DENTRO=S*CONV/1.8

RETURN

END

END ELT. ERRORS: NONE. TIME: 0.098 SEC. IMAGE COUNT: 47

@HDS,P ***** DEWPRE ***** .L,0


```

***** DEMPRES *****
@ELT,L DD.DEMPRES
ELT 8R1 S74Q1C 07/21/84 15:55:01 (0)
SUBROUTINE DEMPRES(IG,T,X,P,XL)
10 C
20 C
30 C
40 C
50 C
60 C
70 C
80 C
90 C
100 C
110 C
120 C
130 C
140 C
150 C
160 C
170 C
180 C
190 C
200 C
210 C
220 C
230 C
240 C
250 C
260 C
270 C
280 C
290 C
300 C
310 C
320 C
330 C
340 C
350 C
360 C
370 C
380 C
390 C
400 C
410 C
420 C
430 C
440 C
450 C
460 C
470 C
480 C
490 C
500 C
510 C
520 C
530 C
540 C
550 C
560 C

***** PURPOSE:
TO CALC. DEW POINT PRESSURE OF BINARY MIXTURE
FROM GIVEN TEMPERATURE AND COMPOSITION

***** INPUT:
IG = 0, IF GUESS OF PRESSURE IS NOT GIVEN
= 1, IF GUESS OF PRESSURE IS GIVEN
P - GUESS OF PRESSURE, OPTIONAL (STD ATM)
T - TEMPERATURE (K)
X - MOLAR CONCENTRATION (FRACTION OF LESS VOLATILE COMPONENT)

***** OUTPUT:
P - BINARY MIXTURE PRESSURE AT BUBBLE POINT (STD ATM)
XL - MOLAR CONCENTRATION OF LESS VOLATILE COMPONENT
IN SAT. LIQUID IN EQUILIBRIUM WITH VAPOR (-)

***** SUBPROGRAMS CALLED BY DEMPRES:
DEBUPR, QLITY

DATA SLOPE/0.,TLAST/0./

IF(ABS(T-TLAST).GT.1.E-3)GOTO 10
IF(ABS(X-XLAST).GT.1.E-3)GOTO 10
P=PLAST
XL=XLAST
RETURN

10 IF(IG.NE.1)THEN
P=DEBUPR(T,X)
P=P-(T-255.)*0.16*X*(1.-X)
END IF
DO 50 I=1,20
CALL QLITY(T,P,X,XQ,XV,XL)
P2=P
XDIF2=X-XV
IF(ABS(XDIF2).LT.0.00001)GOTO 100
IF(1.NE.1)GOTO 20
15 P1=P2
XDIF1=XDIF2
IF(SLOPE.NE.0.)THEN
DP=XDIF2*SLOPE
IF(ABS(DP).GT.1.)DP=SIGN(1.,DP)
P=P2-DP
GOTO 50
END IF
P=P2+0.5
IF(XDIF2.GT.0.)P=P2-0.5
GOTO 50

20 IF(XDIF1.EQ.XDIF2)GOTO 15
SLOPE=(P2-P1)/(XDIF2-XDIF1)
IF(ABS(XDIF1).LT.ABS(XDIF2))GOTO 30
P1=P2

```

```

***** DEWPRE *****
570.      00      XDIF1=XDIF2
580.      00      30 DP=XDIF1*SLOPE
590.      00      IF(ABS(DP).GT.1.)DP=SIGN(1.,DP)
600.      00      P=P1-DP
610.      00      50 CONTINUE
620.      00      WRITE(6,600)XDIF2
630.      00      600 FORMAT(' ERROR 600 IN DEWPRE, XDIF2=',1PE12.4)
640.      00      100 TLAST=T
650.      00      XLAST=X
660.      00      PLAST=P
670.      00      XLLAST=XL
680.      00      RETURN
690.      00      END

END ELT.  ERRORS: NONE.  TIME:  0.130 SEC.  IMAGE COUNT: 69
@HDG,P  ***** DEWTEM ***** .L,0

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***** DEWTEM *****
570. 00 T=T1-DT
580. 00 50 CONTINUE
590. 00 WRITE(6,600)IG,P,X,XDIF2
600. 00 600 FORMAT(' DEWTEM DID NOT CONVERGE',/, ' IG,P,X,XDIF2=',I4,4F8.3)
610. 00 100 FLAST=P
620. 00 XLAST=X
630. 00 TLAST=T
640. 00 XLLAST=XL
650. 00 RETURN
660. 00 END

```

END ELT. ERRORS: NONE. TIME: 0.127 SEC. IMAGE COUNT: 66

@HDG,P ***** DFANNG ***** .L,0


```

0ELT,L DD.DFANNO
ELT 8RI S7401C 07/21/84 15:55:02 (O)
10. 00 DOUBLE PRECISION FUNCTION DFANNO(XI,P,HO,GG)
20. 00 C
30. 00 C**** PURPOSE:
40. 00 TO CALCULATE IN DOUBLE PRECISION
50. 00 ENTROPY OF NON-AZEOTROPIC TWO-PHASE MIXTURE
60. 00 IN THE FANNO FLOW FOR A GIVEN PRESSURE
70. 00 C
80. 00 C**** INPUT DATA:
90. 00 HO - REFRIG. TOTAL ENTHALPY (BTU/LM)
100. 00 GG = G/G/(64.4*773.104) (BTU*LM/FT**6)
110. 00 WHERE G - REFRIG. MASS FLUX (LM/(SEC*FT**2))
120. 00 P - REFRIG. PRESSURE AT WHICH ENTROPY IS DESIRED (PSIA)
130. 00 XM - MOLAR COMPOSITION (FRACTION OF LESS VOLATILE COMPONENT)
140. 00 C**** OUTPUT DATA:
150. 00 DFANNO - ENTROPY (BTU/LM R)
160. 00 C
170. 00 C**** SUBPROGRAMS CALLED BY DFANNO:
180. 00 DBUBTE,DENTRO,DH,DQITY,DVQL1,EBUBTE
190. 00 C
200. 00 REAL*8 DENTRO,DH,DVQL1
210. 00 REAL*8 SLOPE,TKB,XVB,PA,TK2,XQ,XV,XL,VMV,HV,VML,HL,HFG,
220. 00 *WMV,WML,VV,VL,VFG,R,S,C,XQG,DIFXQ1,TK1,SL,SV
230. 00 COMMON/RDATA2/W1,W2,TG1,TC2
240. 00 DATA SLOPE/O.DO,NO,N1/O,1/
250. 00 C
260. 00 IF(P.GT.O.)GOTO 10
270. 00 WRITE(6,600)P
280. 00 600 FORMAT(' ERROR IN CALLING DFANNO, P=',1PE15.5,' PSIA')
290. 00 RETURN
300. 00 C
310. 00 10 PA=P/14.6959
320. 00 CALL DBUBTE(NO,PA,XM,TKB,XVB)
330. 00 TKB=TKB+9.5-20.*(1.55-XM)
340. 00 TK2=0.9*TKB+0.1*TKD
350. 00 DO 100 I=1,20
360. 00 IF(TK2.LT.TKB)THEN
370. 00 TK2=TKB
380. 00 XL=XM
390. 00 XV=XVB
400. 00 XQ=O.DO
410. 00 ELSE
420. 00 CALL DQITY(TK2,PA,XM,XQ,XV,XL)
430. 00 END IF
440. 00 VMV=DVOL1(N1,TK2,PA,XV)
450. 00 HV=DH(TK2,VMV,XV)
460. 00 VML=DVOL1(N0,TK2,PA,XL)
470. 00 HL=DH(TK2,VML,XL)
480. 00 HFG=HV-HL
490. 00 WMV=W1*(1.-XV)+W2*XV
500. 00 WML=W1*(1.-XL)+W2*XL
510. 00 VV=VMV*1G.01E46/WMV
520. 00 VL=VML*1G.01E46/WML
530. 00 VFG=VV-VL
540. 00 R=GG*VFG/VFG
550. 00 S=2.*GG*VL*V-TG+HFG
560. 00 C=GG*VL*VL+HL-HO

```

```

*****
570. 00
580. 00
590. 00
600. 00
610. 00
620. 00
630. 00
640. 00
650. 00
660. 00
670. 00
680. 00
690. 00
700. 00
710. 00
720. 00
730. 00
740. 00
750. 00
760. 00
770. 00
780. 00
790. 00
800. 00
810. 00
820. 00
830. 00
840. 00
850. 00
860. 00
870. 00
880. 00
890. 00
900. 00
910. 00

*****
XQQ=(-S+DSQRT(S*S-4.*R*C))/(2.*R)
DIFXQ2=XQQ-XQ
IF(AES(XQ).LT.1.E-5.AND.ABS(XQQ).LT.1.E-5)GOTO 105
IF(XQ EQ.0.D0.AND.XQQ.LT.0.)GOTO 110
IF(DABS(DIFXQ2).LT.5.E-8)GOTO 105

C
IF(1.GT.1)GOTO 70
TK1=TK2
DIFXQ1=DIFXQ2
IF(SLOPE.NE.0.D0)THEN
  TDEL=DIFXQ2*SLOPE
  TD=ABS(TDEL)
  TD=AMIN1(2.,TD)
  TK2=TK1-SIGN(TD,TDEL)
ELSE
  TK2=TK1+SIGN(1,DIFXQ2)
END IF
GOTO 100

C
70 SLOPE=(TK2-TK1)/(DIFXQ2-DIFXQ1)
IF(ABS(XQ).LT.1.E-3)GOTO 75
IF(DABS(DIFXQ1).LT.DABS(DIFXQ2))GOTO 80
75 TK1=TK2
DIFXQ1=DIFXQ2
80 TK2=TK1-DIFXQ1*SLOPE
100 CONTINUE
PRINT 602,DIFXQ2
602 FORMAT(' DFANNO DOES NOT CONVERGE, DIFXQ2=',1PE15.5)
105 XQ=0.5*(XQ+XQQ)
110 CONTINUE
SL=DEINTRO(TK2,VML,XL)
SV=DEINTRO(TK2,VIV,XV)
DFANNO=(1.-XQ)*SL+XQ*SV
RETURN
END

END ELT. ERRORS: NONE. TIME: 0.140 SEC. IMAGE COUNT: 91
@HDG,P ***** DH ***** .L,0

```

```

***** DH *****
@ELT,L DD,DH
ELT 8R1 S74Q1C 07/21/84 15:55:02 (0)
10. 00 C DOUPLE PRECISION FUNCTION DH(T,V,X)
20. 00 C REAL*8 T,V,X
30. 00 C
40. 00 C
50. 00 C ***** PURPOSE:
60. 00 C TO CALCULATE REFRIGERANT ENTHALPY
70. 00 C IN DOUBLE PRECISION
80. 00 C
90. 00 C ***** INPUT:
100. 00 C T - REFRIG. TEMPERATURE (K)
110. 00 C V - REFRIG. SPEC VOLUME (L/MOL)
120. 00 C X - MOLAR CONCENTRATION (FRACTION OF LESS VOLATILE REFRIG.)
130. 00 C W1 - MOLECULAR WEIGHT OF MORE VOLATILE COMPONENT (G/MOL)
140. 00 C W2 - MOLECULAR WEIGHT OF LESS VOLATILE COMPONENT (G/MOL)
150. 00 C * - CONSTANTS AS PER COMJON STATEMENT /PARAM/
160. 00 C A,B,C1,D1 FOR CALC. OF ENTHALPY
170. 00 C
180. 00 C ***** OUTPUT:
190. 00 C DH - ENTHALPY (BTU/LB)
200. 00 C
210. 00 C
220. 00 C COMMON/PARAM/A,B,C1,C2,D1,D2
230. 00 C COMMON/RDATA2/W1,W2
240. 00 C REAL*8 W1OL,CONV,T2,T3
250. 00 C DATA R/0.08206/
260. 00 C
270. 00 C ***** SURPROGRAMS CALLED BY HCVCP:
280. 00 C HPAR
290. 00 C
300. 00 C
310. 00 C TS=T
320. 00 C XS=X
330. 00 C CALL HPAR(1,TS,XS)
340. 00 C W1OL=(1.-X)*W1+X*W2
350. 00 C CONV=453.5924/(1.055036*WMOL)
360. 00 C T2=T*T
370. 00 C T3=T2*T
380. 00 C DH=(C1*B*T-A*D1*T-A*B)/B**2*DLOG((V+B)/V)
390. 00 C DH=DH+(A*D1*T-A*B)/(D*(V+B))
400. 00 C DH=DH+2.*R*T*V*(2.*V-B/4.)/(V-B/4.)*3*(B/4.-D1*T/4.)
410. 00 C DH=0.101325*DH
420. 00 C DH=DH+(1.-X)*(0.0199537*(T-233.15)+1.081972E-04*(T2-233.15**2))
430. 00 C DH=DH-(1.-X)*(5.67469E-06*(T3-233.15**3))
440. 00 C DH=DH+(1.-X)*16.8976630993
450. 00 C DH=DH+X*(0.0222804*(T-233.15)+7.700465E-05*(T2-233.15**2))
460. 00 C DH=DH+X*(1.0222346E-03*(233.15**3-T3)+21.9450206504)
470. 00 C DH=DH*CONV
480. 00 C RETURN
00. 00 C END

```

END ELT. ERRORS: NONE. TIME: 0.093 SEC. IMAGE COUNT: 48

@HDG,P ***** DPDYN1 ***** .L,0

```

***** DPDYN1 *****
@ELT, L DD.DPDYN1
ELT 8R1 S74Q1C 07/21/84 15:55:02 (0)
      10. 00 C
      20. 00 C
      30. 00 C
      40. 00 C
      50. 00 C
      60. 00 C
      70. 00 C
      80. 00 C
      90. 00 C
     100. 00 C
     110. 00 C
     120. 00 C
     130. 00 C
     140. 00 C
     150. 00 C
     160. 00 C
     170. 00 C
     180. 00 C
     190. 00 C
     200. 00 C
     210. 00 C
     220. 00 C
     230. 00 C
     240. 00 C
     250. 00 C
     260. 00 C
     270. 00 C
     280. 00 C
     290. 00 C
     300. 00 C
     301. 00 C
     302. 00 C
     303. 00 C
     304. 00 C
     310. 00 C
     320. 00 C
     330. 00 C
     340. 00 C
     350. 00 C
     360. 00 C
     370. 00 C
     380. 00 C
     390. 00 C
     400. 00 C
     410. 00 C
     420. 00 C

      FUNCTION DPDYN1(P1,H1,P2,H2,RMS,D)
      C
      C**** PURPOSE:
      C      TO CALCULATE DYNAMIC PRESSURE DROP IN A TUBE
      C      FOR A NON-AZEOTROPIC MIXTURE SINGLE-PHASE FLOW
      C
      C**** INPUT DATA:
      C      D - TUBE DIAMETER (FT)
      C      H1 - REFRIG. ENTHALPY AT TUBE INLET (BTU/LBM)
      C      H2 - REFRIG. ENTHALPY AT TUBE OUTLET (BTU/LBM)
      C      P1 - REFRIG. PRESSURE AT TUBE INLET (PSIA)
      C      P2 - REFRIG. PRESSURE AT TUBE OUTLET (PSIA)
      C      RMS - REFRIG. MASS FLOW RATE (LBM/H)
      C      WM - MOLECULAR WEIGHT (G/MOL)
      C      XM - MOLAR COMPOSITION (FRACTION OF LESS VOLATILE COMPONENT)
      C
      C**** OUTPUT DATA:
      C      DPDYN1 - DYNAMIC SINGLE-PHASE PRESSURE DROP (PSI)
      C
      C**** SUBPROGRAMS CALLED BY DPDYN1:
      C      HPIN
      C
      COMMON/RDATA3/XV,XM,WM
      DIMENSION VL(2),VV(2)
      PA=P1/14.6959
      H=H1
      IG=0
      DO 10 I=1,2
      IF(1.EQ.2)THEN
      IG=1
      PA=P2/14.6959
      H=H2
      END IF
      10 CALL HPIN(IG,H,PA,XM,0.005,T,XQ,XL,XV,VL(1),VV(1))
      C
      IF(XQ.GT.0.)THEN
      DV=(VV(2)-VV(1))*16.01846/WM
      ELSE
      DV=(VL(2)-VL(1))*16.01846/WM
      END IF
      G=RMS/(0.7853982*D*D)
      G=G*G/(32.2*144.*3600.*3600.)
      DPDYN1=G*DV
      RETURN
      END

```

END ELT. ERRORS: NONE. TIME: 0.097 SEC. IMAGE COUNT: 46

@HDG, P ***** DPDYN2 ***** .L,0


```

***** DP2YN2 *****
@ELT,L DD.DPDYN2
ELT 8R1 S74Q1C 07/21/84 15:55:03 (0)
10. 00 C FUNCTION DP2YN2(P1,T1,P2,T2,RMS,DI)
20. 00 C
30. 00 C ***** PURPOSE:
40. 00 C TO CALCULATE DYNAMIC PRESSURE DROP IN A TUBE
50. 00 C FOR A TWO-PHASE FLOW OF NON-AZEOTROPIC MIXTURE
60. 00 C
70. 00 C ***** INPUT DATA:
80. 00 C DI - TUBE INNER DIAMETER (FT)
90. 00 C P1 - REFRIG. PRESSURE AT TUBE INLET (PSIA)
100. 00 C P2 - REFRIG. PRESSURE AT TUBE OUTLET (PSIA)
110. 00 C T1 - REFRIG. TEMPERATURE AT TUBE INLET (F)
120. 00 C T2 - REFRIG. TEMPERATURE AT TUBE OUTLET (F)
130. 00 C RMS - REFRIG. MASS FLOW RATE (LBM/H)
140. 00 C
150. 00 C ***** OUTPUT DATA:
160. 00 C DP2YN2 - DYNAMIC TWO-PHASE PRESSURE DROP (PSIA)
170. 00 C
180. 00 C ***** SUBPROGRAMS CALLED BY DP2YN2:
190. 00 C QLITY,VISCON,VOLIT1
200. 00 C
210. 00 C COMMON/RDATA2/W1,W2,TC1,TC2
220. 00 C COMMON/RDATA3/XW,XM,WM
230. 00 C DIMENSION END(2)
240. 00 C
250. 00 C AREA=3.145927*DI*DI/4.
260. 00 C G=RMS/AREA/3600.
270. 00 C PA=P1/14.6959
280. 00 C TK=(T1+459.67)/1.8
290. 00 C T=T1
300. 00 C DO 100 I=1,2
310. 00 C IF(1.EQ.2) THEN
320. 00 C PA=P2/14.6959
330. 00 C TK=(T2+459.67)/1.8
340. 00 C T=T2
350. 00 C END IF
360. 00 C CALL QLITY(TK,PA,XM,XQ,XV,XL)
370. 00 C CALL VOLIT1(O,TK,PA,XL,WM)
380. 00 C WMOL=W1*(1.-XL)+W2*XL
390. 00 C VL=16.01846*WMOL
400. 00 C XWL=XL/(W1/W2*(1.-XL)+XL)
410. 00 C VISL=VISCON(1,T,XWL)
420. 00 C CALL VOLIT1(1,TK,PA,XV,WM)
430. 00 C WMOL=W1*(1.-XV)+W2*XV
440. 00 C VV=16.01846*WMOL
450. 00 C XWV=XV/(W1/W2*(1.-XV)+XV)
460. 00 C VISV=VISCON(3,T,XWV)
470. 00 C XTT=((1.-XQ)/XQ)*0.9*(VISL/VISV)**0.1
480. 00 C VQID=1.+XTT*(VL/VV)**0.5
490. 00 C VQID=(1.+XTT**0.8)**(-.378)
500. 00 C IF(XTT.GT.10.) VQID=0.823-0.157*ALOG(XTT)
510. 00 C END(1)=VV*XQ*XQ/VQID+VL*(1.-XQ)**2/(1.-VQID)
520. 00 C
530. 00 C 100 CONTINUE
540. 00 C DP2YN2=G*(END(2)-END(1))/144./32.2
550. 00 C RETURN
END

```

301

***** DVOL1 *****

DATE 072184

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@ELT,L DD,DVOL1
ELT 8R1 S74Q1C 07/21/84 15:55:03 (0)
10. 00 C DOUBLE PRECISION FUNCTION DVOL1(N,T,P5,X)
20. 00 C REAL*8 T,P5,X
30. 00 C
40. 00 C
50. 00 C ***** PURPOSE:
60. 00 C TO ITERATE REFRIG. MIXTURE R-1301/R-152A SPEC. VOL.
70. 00 C FROM EQUATION OF STATE IN DOUBLE PRECISION
80. 00 C
90. 00 C ***** INPUT:
100. 00 C N - OUTPUT QUALIFIER
110. 00 C = 0, IF SPEC. VOL. OF LIQUID IS REQUIRED
120. 00 C = 1, IF SPEC. VOL. OF VAPOR IS REQUIRED
130. 00 C T - REFRIG. TEMPERATURE (K)
140. 00 C P5 - REFRIG. SAT. PRESSURE (STD ATM)
150. 00 C X - MOLAR COMPOSITION (FRACTION OF LESS VOLATILE COMPONENT)
160. 00 C * - REFRIG. CONSTANTS A & B (SEE CORRECTION STATEMENT /PARAM/)
170. 00 C ***** OUTPUT:
180. 00 C DVOL1 - REFRIG. SPEC. VOLUME (L/MOL)
190. 00 C
200. 00 C ***** SUBPROGRAMS CALLED BY DVOL1:
210. 00 C ESVOL, EQPAR
220. 00 C
230. 00 C
240. 00 C
250. 00 C REAL*8 Y,Y2,Y3,Y4,P,P0,P6,V6,V7,V
260. 00 C COMMON/PARAM/A,B,C1,C2,D1,D2
270. 00 C DATA R/0.08206/
280. 00 C IF(P5.GT.0.)GOTO 10
290. 00 C PRINT 601,P5
300. 00 C 601 FORMAT(' DVOL1 CALLED WITH NEG. PRESSURE, P5=',1PD13.3)
310. 00 C DVOL1=1.D0
320. 00 C GOTO 1000
330. 00 C
340. 00 C
350. 00 C 10 IF(T.LT.220.)PRINT 602,T
360. 00 C 602 FORMAT(' WARNING, DVOL1 CALLED WITH TEMP.= ',1PD13.3)
370. 00 C
380. 00 C 10 TS=T
390. 00 C P5S=P5
400. 00 C XS=X
410. 00 C V=ESVOL(N,TS,P5S,XS)
420. 00 C CALL EQPAR(TS,XS)
430. 00 C DO 100 I=1,15
440. 00 C Y=B/(4.*V)
450. 00 C Y2=Y*Y
460. 00 C Y3=Y2*Y
470. 00 C Y4=Y3*Y
480. 00 C P=(R*T*(1.+Y+Y2-Y3)/(1.-Y)**3-A/(V+B))/V
490. 00 C P6=(P5-P)/P5
500. 00 C IF(DABS(P6).LT.1.D-7)GOTO 200
510. 00 C P0=-R*T/V**2*(1.+4.*Y+4.*Y2-4.*Y3+Y4)/(1.-Y)**4
520. 00 C P=P0+A*(2.*V+B)/(V*(V+B))**2
530. 00 C V6=(P-P5)/P0
540. 00 C V7=V-V6
550. 00 C IF(V7.GT.(B/4.))GOTO 50
560. 00 C V=V-(V-B/4.)/10.
570. 00 C GOTO 100

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DATE 072184

***** DVOL1 *****
580. 00 50 V=V7
590. 00 100 CONTINUE
600. 00 WRITE(C,600)P6
610. 00 600 FORMAT(' ERROR 600 IN DVOL1, P6 = ',1PD16.8)
620. 00 200 DVOL1=V
630. 00 1000 RETURN
640. 00 END

END ELT. ERRORS: NONE. TIME: 0.119 SEC. IMAGE COUNT: 63

@HDS,P ***** ERUEPR ***** .L,0


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***** ERUBPR *****
@ELT,L DD.ERUBPR
ELT 8R1 S74Q1C 07/21/84 15:55:04 (0)
10. 00 FUNCTION ERUBPR(TK,XM)
20. 00 C
30. 00 C
40. 00 C *****PURPOSE:
50. 00 C ESTIMATE BUBBLE POINT PRESSURE OF MIXTURE
60. 00 C FROM GIVEN TEMPERATURE AND COMPOSITION
70. 00 C
80. 00 C ***** INPUT:
90. 00 C TK - TEMPERATURE (K)
100. 00 C XM - COMPOSITION (MOLAR FRACTION OF LESS VOLATILE COMPONENT)
110. 00 C
120. 00 C ***** OUTPUT:
130. 00 C ERUBPR - ESTIMATE OF BUBBLE POINT PRESSURE (STD ATM)
140. 00 C
150. 00 C COMMON/RDATA2/W1,V2,TC1,TC2
160. 00 C COM/TION/ESDATA/A(2,3),B(2,3),C(2,3)
170. 00 C DIMENSION P(2)
180. 00 C
190. 00 C TK2=TK*TK
200. 00 DO 10 I=1,2
210. 00 10 P(I)=EXP(A(I,1)+A(I,2)/TK+A(I,3)/TK2)
220. 00 C ***** WEIGHT COMPOSITION
230. 00 C XW=XM/(V1/V2*(1.-XM)+XM)
240. 00 C ***** INTERPOLATE
250. 00 C ERUBPR=P(1)*(1.-XW)+P(2)*XW
260. 00 C RETURN
270. 00 C

END ELT. ERRORS: NONE. TIME: 0.060 SEC. IMAGE COUNT: 28
@HDG,P ***** ERUBTE ***** .L,0

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***** EBUYTE *****
@ELT,L DD.EBUYTE
ELT 8R1 S74Q1C 07/21/84 15:55:04 (0)
10. 00 FUNCTION EBUYTE(P,X)
20. 00 C
30. 00 C
40. 00 C**** PURPOSE:
50. 00 C TO ESTIMATE BUBBLE POINT TEMP. OF A MIXTURE
60. 00 C FROM GIVEN PRESSURE AND COMPOSITION
70. 00 C
80. 00 C**** INPUT:
90. 00 C P - PRESSURE (STD. ATM)
100. 00 C X - MOLAR COMPOSITION (FRACTION OF LESS VOLATILE COMPONENT)
110. 00 C
120. 00 C**** OUTPUT:
130. 00 C EBUYTE - ESTIMATE OF BUBBLE POINT TEMPERATURE (K)
140. 00 C
150. 00 C
160. 00 C
170. 00 C
180. 00 C
190. 00 C
200. 00 C
210. 00 C
220. 00 C
230. 00 C
240. 00 C
250. 00 C
260. 00 C
270. 00 C
280. 00 C
290. 00 C
300. 00 C
310. 00 C
320. 00 C
330. 00 C
340. 00 C
350. 00 C
360. 00 C
370. 00 C
380. 00 C
390. 00 C

DIMENSION PS(2),XW(2)
COMMON/RDATA2/W1,W2,TC1,TC2
COMMON/ESDATA/A(2,3),B(2,3),C(2,3)

XW(2)=W1/W2*(1.-X)+X
XW(2)=X/XW(2)
XW(1)=1.-XW(2)
T=280.
DO 50 N=1,10
  T2=T*T
  DO 10 I=1,2
    PS(I)=XW(I)*EXP(A(I,1)+A(I,2)/T+A(I,3)/T2)
  PDELTA=PS(1)+PS(2)-P
  IF(ABS(PDELTA).LT.0.1)GOTO 100
  T3=T2*T
  DP=0.
  DO 20 I=1,2
    DP=DP+PS(I)*(-A(I,2)/T2-2.*A(I,3)/T3)
  T=T-PDELTA/DP
50 CONTINUE
WRITE(6,600)PDELTA
600 FORMAT(' ERROR 600 IN EBUYTE, PDELTA=',1PE12.4)
100 EBUYTE=T
RETURN
END

END ELT. ERRORS: NONE. TIME: 0.006 SEC. IMAGE COUNT: 39
@HDG,P ***** ENTROP ***** .L,0

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***** ENTROP *****
@ELT,L DD.ENTROP
ELT 8R1 S74Q1C 07/21/84 15:55:04 (0)
10. 00 FUNCTION ENTROP(T,V,X)
20. 00 C
30. 00 C
40. 00 C***** PURPOSE:
50. 00 C TO CALCULATE REFRIG. ENTROPY
60. 00 C
70. 00 C***** INPUT:
80. 00 C T - REFRIG. TEMPERATURE (K)
90. 00 C V - REFRIG. SPEC VOLUME (L/MOL)
100. 00 C W1 - MOLECULAR WEIGHT OF MORE VOLATILE COMPONENT (G)
110. 00 C W2 - MOLECULAR WEIGHT OF LESS VOLATILE COMPONENT (G)
120. 00 C X - MOLAR CONCENTRATION OF A LESS VOLATILE REFRIG. (-)
130. 00 C * - CONSTANTS A,B,C1,D1, AS PER COMMON STATEMENT /PARAM/
140. 00 C
150. 00 C***** OUTPUT:
160. 00 C S - ENTROPY (BTU/(LB*F))
170. 00 C
180. 00 C***** SUBPROGRAMS CALLED BY ENTROP:
190. 00 C HPAR
200. 00 C
210. 00 COMMON/PARAM/A,B,C1,C2,D1,D2
220. 00 COMMON/RDATA2/W1,W2,TC1,TC2
230. 00 DATA R/O.08206/
240. 00 C
250. 00 WM=(1.-X)*W1+X*W2
260. 00 CONV=453.5924/(1.055056*WM)
270. 00 IQ=1
280. 00 CALL HPAR(IQ,T,X)
290. 00 S=(C1*B*A*D1)/B**2*ALOG((V+B)/V)+A*D1/B/(V+B)
300. 00 S=S-R*B/4./((V-B/4.)*2*(4.*V-3*B/4.))
310. 00 S=S-R*T*D1*V/2./((V-B/4.)*3*(2.*V-B/4.))
320. 00 S=S*(1.-X)*0.320937933924+X*0.401655763501
330. 00 S=S+R*(X*ALOG(V/0.0632978)+(1.-X)*ALOG(V/0.0776799))
340. 00 S=0.101325*S
350. 00 S=S+(0.0116393*(1.-X)+0.013956*X)*ALOG(T/233.15)
360. 00 S=S+(2.163944E-4*(1.-X)+1.540093E-4*X)*(T-233.15)
370. 00 S=S-(8.512035E-8*(1.-X)+1.533352E-9*X)*(T**2-233.15**2)
380. 00 IF(X.EQ.1.OF.X.EQ.0.)GO TO 1000
390. 00 S=S-0.101325*R*(X*ALOG(X)+(1.-X)*ALOG(1.-X))
400. 00 1000 ENTROP=S*CONV/1.8
410. 00 RETURN
420. 00 END

END ELT. ERRORS: NONE. TIME: 0.086 SEC. IMAGE COUNT: 42
@HDS,P ***** ENTROP ***** .L,0

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***** ENTROP2 *****
@ELT,L DD.ENTROP2
ELT 8R1 S74Q1C 07/21/84 15:55:05 (0)
SUBROUTINE ENTROP2(XW,TF,P,S,XQ)
10. C
20. C
30. C
40. C
50. C
60. C
70. C
80. C
90. C
100. C
110. C
120. C
130. C
140. C
150. C
160. C
170. C
180. C
181. C
182. C
190. C
200. C
210. C
220. C
230. C
240. C
250. C
260. C
270. C
280. C
290. C
300. C
310. C
320. C
330. C
340. C

C ***** PURPOSE:
C TO CALC. ENTROPY OF NON-AZEOTROPIC REFRIGERANT

C ***** INPUT:
C P - REFRIG. PRESSURE (PSIA)
C TF - REFRIG. TEMPERATURE (F)
C XW - WEIGHT CONCENTRATION OF A LESS VOLATILE REFRIG. (-)

C ***** OUTPUT:
C S - ENTROPY (BTU/(LB*F))
C XQ - QUALITY (-)

C ***** SUBPROGRAMS CALLED BY ENTROP2:
C ENTROP,QLITY,VOLIT1
C COMMON/RDATA2/W1,W2,TC1,TC2
C XM=XW/(W2/W1*(1.-XW)+XW)
C TK=(TF+453.67)/1.8
C PA=P/14.6959
C CALL QLITY(TK,PA,XM,XQ,XV,XL)
C IF(XQ.LT..0001.OR.XQ.GT..9999)THEN
C   N=1.1*XQ
C   CALL VOLIT1(N,TK,PA,XM,V)
C   S=ENTROP(TK,V,XM)
C ELSE
C   CALL VOLIT1(O,TK,PA,XL,V)
C   S=(1.-XQ)*ENTROP(TK,V,XL)
C   CALL VOLIT1(1,TK,PA,XV,V)
C   S=S+XQ*ENTROP(TK,V,XV)
C END IF
C RETURN
C END

```

END ELT. ERRORS: NONE. TIME: 0.084 SEC. IMAGE COUNT: 36

@HDO,P ***** EQPAR ***** .L,0


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***** EOPAR *****
@ELT,L DD,EOPAR
ELT 8R1 S74Q1C 07/21/84 15:55:05 (0)
10. 00 SUBROUTINE EOPAR(T,X)
20. 00 C
30. 00 C
40. 00 C*** PURPOSE:
50. 00 C TO CALC. REFRIG. PARAMETERS FOR EQUATION OF STATE
60. 00 C
70. 00 C*** INPUT:
80. 00 C REFRIG. CONSTANTS AS LISTED IN THE COMMON STATEMENT /RDATA1/
90. 00 C T - REFRIG. TEMPERATURE (K)
100. 00 C X - MOLAR CONCENTRATION OF A LESS VOLATILE REFRIG. (-)
110. 00 C
120. 00 C*** OUTPUT:
130. 00 C * - CONSTANTS A & B FOR EQUATION OF STATE
140. 00 C
150. 00 C
160. 00 C COMMON/RDATA1/A3,A4,A5,A6,A7,A8,E3,B4,E5,D6,B7,B8,F0,F1
170. 00 C COMMON/PARAM/A,D,C1,C2,D1,D2
180. 00 C COMMON/STORE1/A1,A2,B1,B2,SEG31
190. 00 C DATA TLAST,XLAST/2*0./
200. 00 C
210. 00 IF (ABS(T-TLAST).GT.0.0001)GOTO 10
220. 00 IF (ABS(X-XLAST).GT.0.0001)GOTO 10
230. 00 RETURN
240. 00 C
250. 10 TLAST=T
260. 00 XLAST=X
270. 00 X2=X**2
280. 00 XX=1.-X
290. 00 XX2=XX**2
300. 00 XXX=X*XX
310. 00 T2=T**2
320. 00 B1=B2+D4*T+B5*T2
330. 00 B2=B6+D7*T+B8*T2
340. 00 B=XX2*D1+X2*B2
350. 00 SEG31=(B1*(1./3.)+B2*(1./3.))/2.
360. 00 SEG32=SEG31**2
370. 00 B=B+2.*XX*SEG31*SEG32
380. 00 A1=A3+A4*T+A5*T2
390. 00 A2=A5+A7*T+A8*T2
400. 00 A1A2S=SQRT(A1*A2)
410. 00 SEG1=XXX*A1A2S
420. 00 F=F0+T1*T
430. 00 SEG2=XXX*(1.-F)*A1A2S
440. 00 A=A1*XX2+A2*X2+2.*SEG2
450. 00 RETURN
460. 00 END

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END ELT. ERRORS: NONE. TIME: 0.089 SEC. IMAGE COUNT: 46

@HDG,P ***** ESVAL ***** .L,0

END ELT. ERRORS: NONE. TIME: 0.100 SEC. IMAGE COUNT: 37

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***** EVAPHX *****L,O
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***** EVAPHX *****

@ELT, L DD.EVAPHX

ELT 0R1 S74Q1C 07/21/84 15:55:05 (0)

SUBROUTINE EVAPHX(110, RMASS, T1, P1, ATIN, APIN, ARHIN,
& X1, T2, P2, H2, X2)

FEBRUARY 1984

C***** EVAPORATOR SIMULATION

C***** PERFORMANCE OF CROSS-FLOW AIR HEATED EVAPORATOR

C***** WITH UP TO 130 PLATE FINNED TUBES

C***** PLACED IN UP TO 5 DEPTH ROWS.

C FORWARD SCHEME, TUBE-BY-TUBE LOGIC. INLET TEMPERATURES
OF AIR & REFRIGERANT CALCULATED FOR EACH TUBE INDIVIDUALLY.

C***** INPUT DATA:

C ANAS(110)

C ANGLE(110)

C APIN

C ARHIN

C ATIN

C CONST(110)

C CFOW(110)

C DI(110)

C DO(110)

C DPCH(110)

C DT(110)

C FLOW(110, N)

C FMK(110)

C FPCH(110)

C FTK(110)

C IDEPTH(110, M)

C IFROM(110, M)

C 110

C IMER(110)

C IST(110)

C ISTART(110, L)

C JFROM(110, J)

C KFEED(110, J, N)

C KSTART(110, N)

C KST(110)

C NDEP(110)

C NPOW(110)

C NSECT(110)

C NTPS(110)

- AIR MASS FLOW RATE THROUGH COIL (LBM/H)

- ANGLE BETWEEN COIL FACE & AIR STREAMLINES (RAD)

- AIR INLET PRESSURE (PSIA)

- AIR INLET RELATIVE HUMIDITY (-)

- AIR INLET TEMPERATURE (F)

- CONSTANT FOR AIR SIDE HEAT TRANSFER CORRELATION (-)

- CONSTANT FOR AIR SIDE HEAT TRANSFER CORRELATION (-)

- INNER DIAMETER OF TUBES (FT)

- OUTER DIAMETER OF TUBES (FT)

- TUBE DEPTH PITCH (FT)

- FIN TIP DIAMETER (FT)

- FRACTION OF COIL TOTAL REFRIG. MASS FLOW PASSING
THROUGH TUBE M (-)

- FIN MATERIAL THERMAL CONDUCTIVITY (BTU/FT*H*F)

- FIN PITCH (FT)

- FIN THICKNESS (FT)

- DEPTH ROW OF A TUBE M

- NUMBER OF TUBE M RECEIVES REFRIG. FROM
WHEN COIL WORKS AS EVAPORATOR (-)

= 1 FOR INDOOR COIL (-)

= 2 FOR OUTDOOR COIL (-)

- NUMBER OF MERGING TUBES (-)

- NUMBER OF TUBES REFRIG. FLOWS INTO COIL
WORKING AS CONDENSER (-)

- NUMBER OF TUBE REFRIG. FLOWS INTO COIL
WORKING AS CONDENSER, FOUND AS L SUCH TUBE (-)

- NUMBER OF TUBE POSITIONED AIR UPSTREAM

- NUMBER OF TUBE RECEIVING REFRIGERANT FROM
TUBE J WHEN COIL WORKS AS EVAPORATOR.
NOTE THAT TUBE J CAN FEED UP TO 3 TUBES
(N CAN BE 1, 2 AND 3) (-)

- NUMBER OF TUBE REFRIGERANT FLOWS INTO COIL
WORKING AS EVAPORATOR (-)

- NUMBER OF TUBES REFRIGERANT FLOWS INTO COIL
WORKING AS EVAPORATOR

- NUMBER OF TUBE ROW DEPTHS (-)

- NUMBER OF TUBES PER ROW (-)

- NUMBER OF REPEATING SECTIONS OF COIL (-)

- NUMBER OF TUBES PER SECTION (-)

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570. C NTUB(110,1) - NUMBER OF TUBES IN ROW I OF EACH COIL SECTION (-)
580. C P1 - REFRIGERANT PRESSURE AT EVAPORATOR OUTLET (PSIA)
590. C RMASS - TOTAL REFRIG. MASS FLOW RATE THROUGH COIL (LBM/H)
600. C RPCH(110) - TUBE ROW PITCH (FT)
610. C TMK(110) - TUBE MATERIAL THERMAL CONDUCTIVITY (BTU/FT*H*F)
620. C T1 - REFRIGERANT TEMPERATURE AT EVAPORATOR OUTLET (F)
630. C WIDTH(110) - COIL WIDTH (FT)
640. C
650. C XM - MOLAR COMPOSITION
660. C - (FRACTION OF LESS VOLATILE COMPONENT)
670. C XW - WEIGHT COMPOSITION
680. C - (FRACTION OF LESS VOLATILE COMPONENT)
690. C WM - MIXTURE MOLECULAR WEIGHT (G/MOL)
700. C W1 - MOLECULAR WEIGHT OF MORE VOLATILE COMPONENT (G/MOL)
710. C W2 - MOLECULAR WEIGHT OF LESS VOLATILE COMPONENT (G/MOL)
720. C
730. C ***** OUTPUT DATA:
740. C H2 - REFRIGERANT ENTHALPY AT EVAPORATOR INLET (RTU/LBM)
750. C PRM(2,1,1) - REFRIG. PRESSURE AT I TUBE OUTLET (PSIA)
760. C PRM(2,2,1) - REFRIG. PRESSURE AT I TUBE INLET (PSIA)
770. C P2 - REFRIGERANT PRESSURE AT EVAPORATOR INLET (PSIA)
780. C TRM(2,1,1) - REFRIG. TEMP. AT I TUBE OUTLET (F)
790. C T2 - REFRIG. TEMP. AT I TUBE INLET (F)
800. C XRM(2,1,1) - REFRIGERANT TEMPERATURE AT EVAPORATOR INLET (F)
810. C XRM(2,2,1) - REFRIG. QUALITY AT I TUBE OUTLET (-)
820. C XTUBE(2,J) - REFRIG. QUALITY AT I TUBE INLET (-)
830. C - FRACTION OF TUBE J WITH SUPERHEATED VAPOR (-)
840. C - (WHEN 2-PHASE MIXTURE IS IN REST OF TUBE) (-)
850. C OR
860. C - FRACTION OF TUBE J WITH 2-PHASE MIXTURE (-)
870. C - (WHEN SUBCOOLED LIQUID IS IN REST OF TUBE) (-)
880. C X1 - REFRIGERANT QUALITY AT EVAPORATOR INLET (-)
890. C X2 - REFRIGERANT QUALITY AT EVAPORATOR OUTLET (-)
900. C VGM(2,1,1) - SPEC. VOLUME OF SATURATED REFRIG. VAPOR AT
910. C - SAT. TEMP. OF I TUBE OUTLET (FT**3/LBM)
920. C VGM(2,2,1) - SPEC. VOLUME OF SATURATED REFRIG. VAPOR AT
930. C - SAT. TEMP. OF I TUBE INLET (FT**3/LBM)
940. C VLM(2,1,1) - REFRIG. SPEC VOLUME AT I TUBE OUTLET (FT**3/LBM)
950. C VLM(2,2,1) - REFRIG. SPEC. VOLUME AT I TUBE INLET (FT**3/LBM)
960. C
970. C ***** SUBPROGRAMS CALLED BY EVAPHX:
980. C AIRHT,AIRFR,BURTEM,DEVTEM,DYNDP1,DPDYN2,EBURTE,ESVOL,FEELIQ,
990. C FINEFF,HGVCP,HPIN,HTCV,OVLPET,PXOIN2,Q,ITY,SPIDP1,SPHTC,
1000. C VISCON,VOLIT1
1010. C
1020. C COMMON/ACCURA/NEAT,ICOMP
1030. C COMMON/RDATA2/W1,W2,TC1,TC2
1040. C COMMON/RDATA3/XW,X1,W1
1050. C COMMON/PIPHX/NDEP(2),NROW(2),DI(2),DO(2),DT(2),RPCH(2),DPCH(2),
1060. C &WIDTH(2),FPCH(2),FTK(2),FMK(2),TNK(2),AMAS(2),ANGLE(2),
1070. C &CONST(2),CPDW(2),NTUB(2,5),IFROM(2,130),NDECT(2),NTPS(2)
1080. C COMMON/MERG/MERGE(2,20,2),IMER(2),ISTART(2,20),IST(2),
1090. C &IDEPTH(2,130),FLOW(2,130),JFLOW(2,130),KEED(2,130,3),
1100. C &KSTART(2,20),KST(2)
1110. C COMMON/MASS/TRM(2,2,130),PRM(2,2,130),XRM(2,2,130),
1120. C & VLM(2,2,130),VGM(2,2,130),XTUBE(2,130),XTI(2,130)
1130. C DIMENSION TAIR(2,6),AIRN(5),HR(2,130),HBACK(10,2),
1140. C & OMEGA(2,6),DTJFG(130),MY(130),FREEE(5),
@ DFR(130),CPRS(120),HCO(5),HCOO(5),ILOOK(15)

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***** EVAPHX *****
1150.
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1570.
1580.
1590.
1600.
1610.
1620.
1630.
1640.
1650.
1660.
1670.
1680.
1690.
1700.
1710.
1720.

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```

*****
C
DATA PREPARATION
AAngle=ANGLE(110)
NNDEP=NDER(110)
DD1=DI(110)
DD0=DO(110)
DDT=DT(110)
RRPCH=RPCH(110)
DDPC=DPCH(110)
WWIDTH=WIDTH(110)
FFPCH=FPCH(110)
FFTK=FTK(110)
FFNK=FNK(110)
TTMK=TMK(110)
AIRMAS=AMAS(110)
CCONST=CONST(110)
CCPOW=CPW(110)
API=3.1415927*DD1*WWIDTH
APO=3.1415927*DD0*WWIDTH
AFM=0.5*(API+APO)
APO=APO*(FFPCH-FFTK)/FFPCH
AF=1.570796*(DDT+DD0)*(DDT-DD0)
AF=AF*WWIDTH/FFPCH
AQ=APO*AF
AFLOW=WWIDTH*RRPCH*NROW(110)
AFLOW=AFLOW*(RRPCH-DD0)/RRPCH
AFLOW=AFLOW/FFPCH
AFLOW=39.75*AFLOW
WFLW=AQ/RRPCH
HDEP=5000.
HP=2.*TTMK/(DD0-DD1)
AIRMS=AIRMAS/NROW(110)
***** FIND INLET STATE FROM PRESSURE AND TEMPERATURE
PA1=P1/14.6959
TK1=(T1+459.67)/1.8
CALL QLTY(TK1,PA1,X1,X1,XV,XL)
IF(X1.GT.0.)THEN
  CALL VOLIT1(N0,TK1,PA1,XL,VL)
  CALL HCVCP(N1,TK1,VL,XL,HL,CV,CP)
  CALL VOLIT1(N1,TK1,PA,XV,VV)
  CALL HCVCP(N1,TK1,VV,XV,HV,CV,CP)
  H1=(1.-X1)*HL X1*HV
  WPL=W1*(1.-XL)+W2*XL
  WMV=W1*(1.-XV)+W2*XV
  VLW=16.01846*VL/WMPL
  VVW=16.01846*VV/WMV
  XLW1=XL/(W1/W2*(1.-XL)+XL)
ELSE
  CALL VOLIT1(N0,TK1,PA1,XM,VL)
  CALL HCVCP(N1,TK1,VL,XM,HL,CV,CP)
  VLW=16.01846*VL/WM
  VVW=0.
  XLW1=XW
END IF
C

```

```

***** EVAPHX *****
1730.
1740.
1750.
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1770.
1780.
1790.
1800.
1810.
1820.
1830.
1840.
1850.
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1880.
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1900.
1910.
1920.
1930.
1940.
1950.
1960.
1970.
1980.
1990.
2000.
2010.
2020.
2030.
2040.
2050.
2060.
2070.
2080.
2090.
2100.
2110.
2120.
2130.
2140.
2150.
2160.
2170.
2180.
2190.
2200.
2210.
2220.
2230.
2240.
2250.
2260.
2270.
2280.
2290.
2300.

C
DO 4 IS=1,KST(110)
I=KSTART(110,IS)
TRM(110,1,1)=TI
PRM(110,1,1)=PI
XRM(110,1,1)=XI
HR(1,1)=HI
VGM(110,1,1)=VW
VLM(110,1,1)=VLW
4 CONTINUE
C**** ESTIMATE CHANGE OF AIR TEMPERATURE
CALL AIRPR(1,ATIN,APIN,ARHIN,WAIR,CPAIR,RAIR,
&AMAIR,AKAIR)
CALL DEITEM(N0,PA1,XM,TKD,XL)
TD2=TKD*1.8-460.
DTAIR=0.85*(ATIN-TD2)/(NNDP-1)
DTAIR=AMAX1(0,DTAIR)
DO 6 I=1,NTUB(110,1)
TAIR(1,1)=ATIN
TAIR(2,1)=TAIR(1,1)-DTAIR
WAIR(1,1)=WAIR
WAIR(2,1)=WAIR
IA=NTUB(110,1)+1
DO 7 I=IA,NTPS(110)
J=JFROM(110,1)
TAIR(1,1)=TAIR(2,J)
TAIR(2,1)=TAIR(1,1)-DTAIR
WAIR(1,1)=WAIR(2,J)
WAIR(2,1)=WAIR(2,J)
7 END IF
CC
OMEGA(1,1)=WAIR
OMEGA(2,1)=WAIR
TAIR(1,1)=ATIN
DO 8 I=1,NNDP
J=I+1
TAIR(1,J)=TAIR(1,1)-DTAIR
OMEGA(1,J)=OMEGA(1,1)
HICE(1)=1.E+30
HFGWT(1)=0.
TKICE(1)=0.
TWAT(1)=0.
TFIP(1)=0.
HBACK(1,2)=999.
C**** EVALUATE 'TWC-PHASE SPEC. HEAT'
TK2=TKD
PA2=PA1
VV=ESVOL(N1,TK2,PA2,XM)
CALL HSCVP(N1,TK2,VV,YM,H2,CV,CP)
CPR=(H2-H1)/(1.8*(TK2-TK1))
DO 12 I=1,NTPS(110)
VLM(110,2,1)=0.0
VGM(110,2,1)=0.0
DPR(1)=0.
CPRS(1)=CPR
DTHFG(1)=0.
12 MY(1)=0
DO 13 I=1,IMER(110)

```

```

***** EVAPHX *****
2310. 00 J=MERGE(110,1,1)
2320. 00 13 MY(J)=1
2330. 00 ACC=0.003
2340. 00 HBACK(1,2)=999.
2350. 00 AFIN=3.14159*(DDT-DDO)*(DDT+DDO)/4.
2360. 00 SEGFIN=DDT**3/24.-DDT*DDT*DDO/16.+DDO**3/48.
2370. 00 SEGFIN=SEGFIN*2./(AFIN*(DDT-DDO))
2380. 00 HTOL=0.003
2390. 00 H2P=0.
2400. 00
2410. 00 C*****
2420. 00 C***** START MAIN LOOP
2430. 00 C*****
2440. 00 C*****
2450. 00 C
2460. 00 IAIN=10
2470. 00 DO 150 IAR=1,IAIRN
2480. 00 C***** CALC. CHANGE OF AIR MASS FLOW RATE DUE TO WATER/FROST ACCUM.
2490. 00 AFEE=0.
2500. 00 DO 16 I=1,NNDP
2510. 00 16 AFEE=AFEE+(FFPCH-FFTK-2.*TKICE(1))
2520. 00 AFEE=AFEE/(NNDP*(FFPCH-FFTK))
2530. 00 AAMAS=AIRNAS*AFEE*0.56*(530./(460.+ATIN))*0.64
2540. 00 C***** CALC. AIR DATA FOR EACH TUBE
2550. 00 DO 18 I=1,NNDP
2560. 00 NT=NSECT(110)*NTUB(110,1)
2570. 00 AMS=AAMAS/NT
2580. 00 AMSI(1)=AMS
2590. 00 TAAV=0.5*(TAIR(1,1)+TAIR(1,1+1))
2600. 00 WAIR=0.5*(OMEGA(1,1)+OMEGA(1,1+1))
2610. 00 CALL AIRPR(2,TAAV,APIN,RHA,WAIR,CPA,RA,AMA,AKA)
2620. 00 FFFTK=FFTK+2.*TKICE(1)
2630. 00 HCOD(1)=AIRHT(AAMAS,CPA,AMA,AKA,DDO,DDT,NT,WIDTH,
2640. 00 8, RFPCH,FFPCH,FFTK,CONST,CCPOW,ANGLE)
2650. 00 HCO(1)=HCO(1)*(1.+HFGWT(1))
2660. 00 FFEE(1)=FINEFF(110,DDT,DDO,FFTK,FFMK,HCO(1))
2670. 00 UD1(1)=HCO(1)*(1.-AF*(1.-FFEE(1))/AO)
2680. 00 CPAS(1)=CPA
2690. 00 18 ATRN(1)=0.
2700. 00 C***** FIND TUBE REFRIG. FLOWS INTO EVAPORATOR
2710. 00 ILN=0
2720. 00 ILNEXT=0
2730. 00 DO 110 NUMB=1,KST(110)
2740. 00 I=KSTART(110,NUMB)
2750. 00 TRI=T1
2760. 00 PRI=P1
2770. 00 HRI=H1
2780. 00 XRI=X1
2790. 00 XLWI=XLW1
2800. 00 GOTO 30
2810. 00 C***** ASSIGN INLET PARAMETERS FOR NEXT TUBE
2820. 00 22 CONTINUE
2830. 00 TRI=TRN(110,2,JJ)
2840. 00 PRI=PRN(110,2,JJ)
2850. 00 HRI=HR(2,JJ)
2860. 00 XRI=XRN(110,2,JJ)
2870. 00 VLM(110,1,1)=VLM(110,2,JJ)
2880. 00 VGM(110,1,1)=VGM(110,2,JJ)
2890. 00 XLWI=XLWS(JJ)

```

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***** EVAPHX *****
2890. 00
2900. 00
2910. 00
2920. 00
2930. 00
2940. 00
2950. 00
2960. 00
2970. 00
2980. 00
2990. 00
3000. 00
3010. 00
3020. 00
3030. 00
3040. 00
3050. 00
3060. 00
3070. 00
3080. 00
3090. 00
3100. 00
3110. 00
3120. 00
3130. 00
3140. 00
3150. 00
3160. 00
3170. 00
3180. 00
3190. 00
3200. 00
3210. 00
3220. 00
3230. 00
3240. 00
3250. 00
3260. 00
3270. 00
3280. 00
3290. 00
3300. 00
3310. 00
3320. 00
3330. 00
3340. 00
3350. 00
3360. 00
3370. 00
3380. 00
3390. 00
3400. 00
3410. 00
3420. 00
3430. 00
3440. 00
3450. 00
3460. 00

C
C
C**** TUBE SELECTION FOR CALCULATION DONE
C**** COMPUTE HEAT TRANSFER & REFRIG. PRESSURE DROP FOR TUBE 1
C**** FIND REFRIG. STATE AT OUTLET
C
30 TRM(110,1,1)=TRI
PRM(110,1,1)=PRI
HR(1,1)=HRI
XRM(110,1,1)=XRI
TRI=TRI
TRIX=TRI
HRI=HRI
HRIX=HRI
XRI=XRI
XRIX=XRI
RMS=RMASS*FLOW(110,1)
XLIQ=.0
XSLUG=.0
XANNUL=.0
XMI ST=.0
TKRI=(TRI+459.67)/1.8
PRE=PRI+DPR(1)
PRAV=0.5*(PRI+PRE)
PARAV=PRAV/14.6959
DP1=0.
DP2=0.
DP3=0.
VIX=(1.-XRI)*VLM(110,1,1)+XRI*VGH(110,1,1)
ICT=IDEPH(110,1)
TAI=TAIR(1,1)
IF(TAI.LT.(TRI+0.1))TAI=TRI+0.1
CPA=CPAS(1CT)
AMS=AMSI(1CT)
OMEGE=WAIR(1,1)
OMEGE=WAIR(2,1)
CODE IF(TAIR.EQ.1)PRINT 444,IAIR,1,XRI,PRI,TRI,HRI,TAI
444 FORMAT(' 1,XRI,PRI,TRI,HRI,TAI=',215,F5.2,4F8.2)
C
IF(XRI.GT.0.)GOTO 45
C
C**** CASE 1 *****
C
C**** INLET QUALITY 0
C
C
CALL RUBTEM(C,PARAV,XM,TKB,XV)
CALL VOLIT1(N0,TKB,PARAV,XM,VLB)
CALL HCVCP(N1,TKB,VLB,XM,HD,CV,CP)
TRE=TKB*1.8-459.67
C
DO 36 IT=1,5
TRAV=0.5*(TRI+TRE)
TKRAV=(TRAV+459.67)/1.8
AMR=VISCON(1,TRAV,XM)
AKR=VISCON(2,TRAV,XM)
CALL VOLIT1(N0,TKRAV,PARAV,XM,VL)
CALL HCVCP(N5,TKRAV,VL,XM,H,CV,CPR)

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***** EVAPHX *****
3470. HI=SPHTC(CPR,AMR,AKR,RMS,DDI)
3480. CALL OVLVET(AO,AFI,AFM,HI,HDEP,HP,HICE(ICT),UD1(ICT),UAO,UPO,UWO)
3490. QQ=CPR*AMS*(1.-EXP(-UAO/(CPR*AMS)))/(CPR*R13)
3500. Q=1.-EXP(-QQ)
3510. Q=CPR*RMS*(TAI-TRI)*Q
3520. HRE=HRI+Q/RMS
3530. TRE=TRI+(HRE-HRI)/CPR
3540. IF(IT.EQ.1)GOTO 34
3550. IF(HRE.GT.HO)THEN
3560. XLIQ=-ALOG(1.-(HB-HRI)/(CPR*(TAI-TRI)))/QO
3570. AL1=XLI*Q*WIDTH
3580. TRII=TKD*1.8-459.67
3590. HRII=HP
3600. XRII=0.
3610. TRII=TRII
3620. HRII=HRII
3630. XRII=0.
3640. VIX=VLB*16.01846/W
3650. GOTO 40
3660.
3670. END IF
3680. HDIF=HREI-HRE
3690. XRE=0.
3700. XLIQ=0.
3710. IF(ABS(HDIF).LT.HTOL)GOTO 38
3720.
3730. 34 HREI=HRE
3740. 36 CONTINUE
3750. WRITE(6,550)IAIR,I,ICT,XRI,HDIF
3760. 650 FORMAT(' CASE 1 TUBE DOES NOT CONVERGE',/,' IAIR,I,ICT,XRI,HDIF=',
3770. @314,F5.2,F10.5)
3780. 38 CONTINUE
3790.
3800. C
3810. XLYS(I)=XW
3820. AL1=W*IDTH*20.*DDI
3830. VL=VL*16.01846/W
3840. DF1=SFHDP1(RMS,AL1,DDI,VL,AIR)
3850. IF(XLIQ.EQ.0.)GOTO 100
3860. 45 IF(XRI.GT.0.1)GOTO 58
3870.
3880. C
3890. C**** CASE 2 *****
3900. C
3910. C**** INLET QUALITY 0.0 - 0.1
3920. C
3930. CALL PUBTEM(NO,PARAV,XM,TKB,XV)
3940. TB=TKB*1.8-459.67
3950. CALL VOLIT1(NO,TKR,FARAV,XI1,VLB)
3960. CALL HCVCP(N5,TKB,VL3,XM,H,CV,CPR)
3970. V1SL=V1SCCN(N1,TB,XW)
3980. CONL=V1SCCN(N2,TB,XW)
3990. HI00=SPHTC(CPR,V1SL,CONL,RMS,DDI)
4000. X10=.1
4010. CALL PXGIN2(XM,PARAV,X10,TK10,XL,XV,VL,WV,HI0)
4020. TF10=TK10*1.8-459.67
4030. HI10=HTCEV(TF10,FRAV,RMS,DDI,XTI(HI0,I))
4040. TIRE=TF10
4050. TKIRE=TK10
4060. XRE=0.1
4070.
4080. C
4090. ITE=0
4100.

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```

***** EVAPHX *****
4050. DO 55 IT=1,6
4060. CPR=CPRS(1)
4070. TRAV=0.5*(TR11+TRE)
4080. XRAV=0.5*(XRI1+XRE)
4090. HI=HI10-10.*(1-XRAV)*(HI10-HI00)
4100. CALL OVLMT(AO,API,APM,HI,HIP,HICE(1CT),UD1(1CT),UAO,UPO,UW3)
4110. QQ=CPR*AMS*(1-EXP(-UAO/(CPR*AMS)))/(CPR*RES)
4120. Q=CPR*RMS*(TAI-TR11)*(1-EXP(-QQ*(1-XL1Q)))
4130. HRE=HRI1+Q/RMS
4140. CALL HPIN(N1,HRE,PARAV,XM,ACC,TKRE,XRE,XL,XV,VL,VV)
4150. TRE=TKRE*1.8-459.67
4160. IF(TRE.NE.TR11)GOTO 52
4170. IF(ITE.EQ.0)THEN
4180. ITE=1
4190. CPRS(1)=3.*CPR
4200. GOTO 54
4210. ELSE
4220. GOTO 77
4230. END IF
4240. 52 CPRS(1)=(HRE-HRI1)/(TRE-TR11)
4250. IF(ITE.EQ.1)GOTO 54
4260. IF(HRE.GT.HI0)THEN
4270. XSLUG=-ALOG(1-(HI0-HRI1)/(CPR*(TAI-TR11)))/QQ
4280. AL1=XSLUG*WIDTH
4290. HRI1=HI0
4300. TRI1=TF10
4310. XRI1=.1
4320. GOTO 60
4330. END IF
4340. HDIF=HRE1-HRE
4350. IF(ABS(HDIF).LT.HTOL)GOTO 77
4360. 54 HRE1=HRE
4370. 55 CONTINUE
4380. C
4390. WRITE(6,651)IAIR,I,ICT,XRI,HDIF
4400. 651 FORMAT(' CASE 2 TUBE DOES NOT CONVERGE',/' IAIR,I,ICT,XRI,HDIF=',
4410. @314,F5.2,F10.5)
4420. GOTO 77
4430. C
4440. 58 IF(XRI.GT..9)GOTO 68
4450. C
4460. C**** CASE 3 *****
4470. C
4480. C**** INLET QUALITY 0.1 - 0.9
4490. C
4500. 60 TRE=TR11
4510. TKRE=(TRE+459.67)/1.8
4520. XRE=XRI1
4530. ITE=0
4540. DO 65 IT=1,6
4550. CPR=CPRS(1)
4560. TRAV=0.5*(TR11+TRE)
4570. XRAV=0.5*(XRI1+XRE)
4580. HI=HICEV(TRAV,PARAV,RMS,DD1,XTT(HI0,I))
4590. CALL OVLMT(AO,API,APM,HI,HIP,HICE(1CT),UD1(1CT),UAO,UPO,UW3)
4600. QQ=CPR*AMS*(1-EXP(-UAO/(CPR*AMS)))/(CPR*RES)
4610. Q=CPR*RMS*(TAI-TR11)*(1-EXP(-QQ*(1-XL1Q-XSLUG)))
4620. HRE=HRI1+Q/RMS

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***** EVAPHX *****
4630. CALL HPIN(N1,HRE,PARAV,XM,ACC,TKRE,XRE,XL,XV,VL,VV)
4640. TRE=TKRE*1.8-459.67
4650. IF(TRE.NE.TRII)GOTO 62
4660. IF(ITE.EQ.0)THEN
4670.   ITE=1
4680.   CPR3(1)=3.*CPR
4690.   GOTO 64
4700. ELSE
4710.   GOTO 77
4720. END IF
4730. 62 CPR3(1)=(HRE-HRII)/(TRE-TRII)
4740. IF(ITE.EQ.1)GOTO 54
4750. IF(XRE.GT..9)THEN
4760.   X90=.9
4770.   CALL PXQIN2(XM,PARAV,X90,TK90,XL,XV,VL,VV,H90)
4780.   XAHNUL=-ALOG(1.-(H90-HRII)/(CPR*(TAI-TRII)))/OO
4790.   HRII=H90
4800.   TRII=TK90*1.8-459.67
4810.   XRII=.9
4820.   GOTO 70
4830. END IF
4840. IF(ABS(HRE-HRE1).LT.HTOI)GOTO 77
4850. 64 IF(XRE.GT.0.9)THEN
4860.   X90=0.9
4870.   CALL PXQIN2(XM,PARAV,X90,TK90,XL,XV,VL,VV,H90)
4880.   TRE=TK90*1.8-459.67
4890.   END IF
4900.   HRE1=HRE
4910.   65 CONTINUE
4920. C
4930. WRITE(6,52)IAIR,I,ICT,XRI,HDIF
4940. 52 FORMAT(' CASE 3 TUBE DOES NOT CONVERGE',/' IAIR,I,ICT,XRI,HDIF=',
4950.   @314,F5.2,F10.5)
4960.   GOTO 77
4970. C
4980. 68 IF(XRI.GT..999)GOTO 82
4990. C**** CASE 4 *****
5000. C
5010. C**** INLET QUALITY 0.9 - 1.0
5020. C
5030. 70 X90=.9
5040. CALL PXQIN(XM,PARAV,X90,TK90,XL,XV)
5050. TF90=TK90*1.8-459.67
5060. HI90=HTCEV(TF90,PARAV,RMS,DDI,XTT(110,I))
5070. CALL DEWTEM(N0,PARAV,XM,TKD,XLD)
5080. TD=TKD*1.8-459.67
5090. CALL VOLITI(N1,TKD,PARAV,XM,VVD)
5100. CALL HCVCP(N3,TKD,VVD,XM,HD,CV,CPR)
5110. PRINT 708,TD,HD
5120. 708 FORMAT(' TD,HD=',2F14.5)
5130. VISV=VISCON(N3,TD,XW)
5140. CONV=VISCON(N4,TD,XW)
5150. HI1=SPHTC(CPR,VISV,CONV,RMS,DDI)
5160. TRE=TD
5170. TKRE=TKD
5180. XRE=1.
5190. C
5200. ITE=0

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```

***** EVAPHX *****
5210. 00
5220. 00
5230. 00
5240. 00
5250. 00
5260. 00
5270. 00
5280. 00
5290. 00
5300. 00
5310. 00
5320. 00
5330. 00
5340. 00
5350. 00
5360. 00
5370. 00
5380. 00
5390. 00
5400. 00
5410. 00
5420. 00
5430. 00
5440. 00
5450. 00
5460. 00
5470. 00
5480. 00
5490. 00
5500. 00
5510. 00
5520. 00
5530. 00
5540. 00
5550. 00
5560. 00
5570. 00
5580. 00
5590. 00
5600. 00
5610. 00
5620. 00
5630. 00
5640. 00
5650. 00
5660. 00
5670. 00
5680. 00
5690. 00
5700. 00
5710. 00
5720. 00
5730. 00
5740. 00
5750. 00
5760. 00
5770. 00
5780. 00

CXC
DO 75 IT=1,6
  IF(I.EQ.14)PRINT 710,XRE
710  FORMAT(' XRE=',F12.6)
  CPR=CPRS(I)
  TRAV=0.5*(TRII+TRE)
  XRAV=0.5*(XRII+XRE)
  HI=HI1-10.*(1.-XRAV)*(HI1-HI30)
  IF(I.EQ.14)PRINT 707,HI
707  FORMAT(' HI=',F12.5)
  CALL QVLWET(AO,API,APM,HI,HDEP,HP,HICE(ICT),UDI(ICT),UAQ,UPO,UWO)
  Q=CPR*AMS*(1.-EXP(-UAQ/(CPR*AMS)))/(CPR*RTS)
  Q=CPR*RTS*(TAI-TRII)*(1.-EXP(-QQ*(1.-XLIQ-XSLUG-XANNUL)))
  HRE=HRII+Q/RTS
  IF(I.EQ.14)PRINT 709,HRE,PARAV,TKRE
709  FORMAT(' HRE,PARAV,TKRE=',4F14.6)
  CALL HPIN(N1,HRE,PARAV,XM,ACC,TKRE,XRE,XL,XV,VL,VV)
  TRE=TKRE*1.0-459.67
  IF(I.EQ.14)PRINT 711,HRE,HD,TRII,TRE
711  FORMAT(' HRE,HD,TRII,TRE**=',4F10.5)
  IF(TRE.NE.TRII)GOTO 72
  IF(ITE.EQ.0)THEN
    ITE=1
    CPRS(I)=3.*CPR
    GOTO 74
  ELSE
    GOTO 77
  END IF
72  CPRS(I)=(HRE-HRII)/(TRE-TRII)
  IF(ITE.EQ.1)GOTO 74
  IF(HRE.GT.HD)THEN
    IF(I.EQ.14)PRINT 765,HD,HRII,TAI,TRII
765  FORMAT(' HG,HRII,TAI,TRII=',4F12.4)
    XRE=1.
    XMIST=-ALOG(1.-(HD-HRII)/(CPR*(TAI-TRII)))/QQ
    HRII=HD
    TRII=TD
    XRII=1.
    XV=XM
    WV=VVD
    XL=XLD
    CALL VOLIT1(NQ,TKD,PARAV,XLD,VL)
    XTUBE(110,1)=XSLUG+XANNUL+XMIST
    AL1=XTUBE(110,1)*WIDTH
    TRAV=0.5*(TRIX+TRII)
    XREX=1.
    HREX=HRII
    GOTO 80
  END IF
  HDIF=HRE1-HRE
  IF(ABS(HDIF).LT.HTOL)GOTO 77
74  HRE1=HRE
75  CONTINUE
C
WRITE(6,653)IAIR,I,ICT,XRI,HDIF
653  FORMAT(' CASE 4 TUBE DOES NOT CONVERGE',/' IAIR,I,ICT,XRI,HDIF=',
@314,F5.2,F10.5)
C
C**** COMPUTE FRICTIONAL, 2-PHASE PRESSURE DROP

```



```

***** EVAPHX *****
5790. C
5800.
5810.
5820.
5830.
5840.
5850.
5860.
5870.
5880.
5890.
5900.
5910.
5920.
5930.
5940.
5950.
5960.
5970.
5980.
5990.
6000.
6010.
6020.
6030.
6040.
6050.
6060.
6070.
6080.
6090.
6100.
6110.
6120.
6130.
6140.
6150.
6160.
6170.
6180.
6190.
6200.
6210.
6220.
6230.
6240.
6250.
6260.
6270.
6280.
6290.
6300.
6310.
6320.
6330.
6340.
6350.
6360.

77 AL1=(1.-XLIQ)*WIDTH+20.*DDI
   TRAV=0.5*(TRI1-TRE)
   HREX=XRE
   HREX=HRE
80 XRA/=0.5*(XRI1+XREX)
CXC IF(1.EQ.14)PRINT 720,XRI,XRI1,XRE
720 FORMAT(' XRI,XRI1,XRE=',AFG.2)
VML=W1*(1.-XL)+V2*XL
VLW=VL*16.0134G/WML
WIV=W1*(1.-XV)+V2*XV
VWV=WV*16.0184G/WMV
VMIX=(1.-XREX)*VLW+XREX*VWV
VMIX=0.5*(VMIX+VIX)
XLWS(1)=XL/(W1+W2*(1.-XL)+XL)
XWL=.5*(XLW1+XLWS(1))
VISL=VISCON(1,TRAV,XWL)
CXC IF(1.EQ.14)PRINT 843,HRI,HREX
843 FORMAT(' HRI,HREX=',2F15.6)
IF(HRI.NE.HREX)THEN
  DP2=1.4*EVDP(RMS,HRI,HREX,VMIX,VISL,AL1,DDI)
ELSE
  F=FEELIQ(XRAV,TRAV,XL,XV,VL,VV,XT)
  RMSX=(1.-XRAV)*RMS
  DP2=F*SPHDP1(RMSX,AL1,DDI,VLW,VISL)
END IF
C IF(XRI1.NE.1.)GOTO 100
C *****
C ***** INLET QUALITY 1
C
82 DO 86 IS=1,5
   TRAV=0.5*(TRI1-TRE)
   TKRAV=(TRAV+459.67)/1.8
   AMR=VISCON(3,TRAV,XW)
   AKR=VISCON(4,TRAV,XW)
   CALL VOLIT1(N1,TKRAV,PARAV,XM,VV)
   CALL HCVCP(N5,TKRAV,VV,XM,H,CV,CPR)
   HI=SPHTC(CPR,AMR,AKR,RMS,DDI)
   CALL OVLWET(AO,API,APM,HI,HDEP,HP,HICE/ICT),UD1(1CT),UAO,UPO,UWO)
   QQ=CPA*AMS*(1.-EXP(-UAO/(CPA*AMS)))/(CPR*R4S)
   Q=1.-EXP(-QQ*(1.-XL1Q-XSLUG-XANNUL-X'1ST))
   Q=CPR*RMS*(TAI-TRI)*Q
   HRE=TRI+Q/RMS
CXC IF(1.EQ.14)PRINT 740
740 FORMAT(' HRE=',F10.4)
TRE=TRI1+(HCE-HRI1)/CPR
IF(1S.EQ.1)GOTO 84
HDI1=HRE1-HRE
C IF(ABS(HDI1).LT.HTOL)GOTO 83
84 HRE1=HRE
86 CONTINUE
C
WRITE(6,654)IAIR,1,1CT,XRI,HDIF
654 FORMAT(' CASE 5 TUBE DOES NOT CONVERGE',/,' IAIR,1,1CT,XRI,HDIF=',
@314,F5.2,F10.5)

```

```

***** EVAPHX *****
6370. 00
6380. 00
6390. 00
6400. 00
6410. 00
6420. 00
6430. 00
6440. 00
6450. 00
6460. 00
6470. 00
6480. 00
6490. 00
6500. 00
6510. 00
6520. 00
6530. 00
6540. 00
6550. 00
6560. 00
6570. 00
6580. 00
6590. 00
6600. 00
6610. 00
6620. 00
6630. 00
6640. 00
6650. 00
6660. 00
6670. 00
6680. 00
6690. 00
6700. 00
6710. 00
6720. 00
6730. 00
6740. 00
6750. 00
6760. 00
6770. 00
6780. 00
6790. 00
6800. 00
6810. 00
6820. 00
6830. 00
6840. 00
6850. 00
6860. 00
6870. 00
6880. 00
6890. 00
6900. 00
6910. 00
6920. 00
6930. 00
6940. 00

88 CONTINUE
C
XTUBE(110,1)=1.-XL10-XSL16-XANNUL-XH1ST
AL1=XTUBE(110,1)*WIDTH*20.*DDI
VV=VV*16.01845/MM
AL1=WIDTH*20.*DDI
DF3=SPHDP1(RMS,AL1,DDI,VV,AMR)
750 FORMAT(' PRE=',F10.5)
100 PRE=PRI-DP1-DP2-DP3
PRE=AMAX1(3.,PRE)
CXC
IF(1.EQ.14)PRINT 750,PRE
Q=(HRE-HRI)*RMS
TAE=TAIR1(1,1)-Q/(CFA*AMC)
TAE=TAE+DTI*FG(1)
C*** FOLLOWING STATEMENTS FOR CALC. OF DYNAMIC PRESSURE DROP ARE
CC SKIPPED. THEY PROVIDE LITTLE CORRECTION FOR PRESSURE DROP
CC AT EXPENSE OF A LOT OF COMPUTING.
CC IF(XRI(1,1).AND.XRE(1,1).GT.0.1)THEN
CC DIX=DYNDP2(PRI,TRI,PRE,TRE,RMS,DDI)
CC ELSE
CC DIX=DYNDP1(PRI,HRI,PRE,HRE,RMS,DDI)
CC END IF
CC PRE=PRE+DIX
C
C*****
C*** END OF HEAT TRANSFER AND PRESSURE DROP CALCULATIONS*****
C*** ENTHALPY & PRESSURE AT TUBE I OUTLET ARE KNOWN
C*** FIND TEMP., QUALITY & SPEC. VOLUME
TKRE=(TRE+459.67)/1.8
PARE=PRE/14.6959
CALL HPIN(N1,HRE,PARE,XM,ACC,TKRE,XRE,XL,XV,VL,VV)
TRM(110,2,1)=TKRE*1.8-459.67
PRM(110,2,1)=PRE
HR(2,1)=HRE
XRM(110,2,1)=XRE
DPR(1)=PRI-PRE
IF(XRE.EQ.0.)THEN
VLM(110,2,1)=16.01846*VL/MM
ELSE
IF(XRE.LT.1.)THEN
WMV=W1*(1.-XV)+W2*XV
VGM(110,2,1)=16.01846*VV/MMV
WML=W1*(1.-XL)+W2*XL
VLM(110,2,1)=16.01846*VL/VML
ELSE
VGM(110,2,1)=16.01846*V/VM
END IF
END IF
C
C*** REFRIGERANT SIDE CALCULATIONS FOR TUBE I COMPLETED.
C*** FIND AIR STATE FAST TUBE
OMEGA=OMEGA
WATHFG=0.
IF(Q.LE.0.)THEN
TWATA=TRI
TPIPE=TRI
GOTO 104
END IF

```

```

***** EVAPHX *****
6950. 00
6960. 00
6970. 00
6980. 00
6990. 00
7000. 00
7010. 00
7020. 00
7030. 00
7040. 00
7050. 00
7060. 00
7070. 00
7080. 00
7090. 00
7100. 00
7110. 00
7120. 00
7130. 00
7140. 00
7150. 00
7160. 00
7170. 00
7180. 00
7190. 00
7200. 00
7210. 00
7220. 00
7230. 00
7240. 00
7250. 00
7260. 00
7270. 00
7280. 00
7290. 00
7300. 00
7310. 00
7320. 00
7330. 00
7340. 00
7350. 00
7360. 00
7370. 00
7380. 00
7390. 00
7400. 00
7410. 00
7420. 00
7430. 00
7440. 00
7450. 00
7460. 00
7470. 00
7480. 00
7490. 00
7500. 00
7510. 00
7520. 00

```

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C
OMECHEF=0.
TWATAI=TRI+Q/UWO
TWATAE=TRE+Q/UWO
TWATA=.5*(TWATAI+TWATAE)
TPIPA=0.5*(TRI+TRE)+Q/UPO
CXC IF(1.EQ.14)PRINT 755,TWATA
755 FORMAT(' TWATA=',F12.5)
CALL AIRPR(1,TWATAI,APIN,1.,OMEGWI,CPW,RV,AMW,AKW)
CALL AIRPR(1,TWATAE,APIN,1.,OMEGWE,CPW,RV,AMW,AKW)
IF(OMEGI.LE.OMEGWI)GOTO 104
XCOND=(OMEGI-OMEGWI)/(OMEGWE-OMEGWI)
XCOND=AMIN1(1.,XCOND)
TWATA=TWATAI+0.5*XCOND*(TWATAE-TWATAI)
CALL AIRPR(1,TWATA,APIN,1.,OMEGW,CPW,RV,AMW,AKW)
OMECHEP=(OMEGI-OMEGW)*(1.-EXP(-HICOD(1CT)*APO/(CPA*AMS)))
TTTAIR=TAE+0.5*XCOND*(TAI-TAE)
TFM=TTTAIR-FFEE(1CT)*(TTTAIR-TWATA)
TEND=TWATA+(TFM-TWATA)/SEG*IN
IF(TEND.GT.TAE)TEND=TAE
CALL AIRPR(1,TEND,APIN,1.,OMEGIS,CPW,RV,AMW,AKW)
IF(OMEGIS.LE.OMEGW)THEN
  OMEGFI=OMEGW
  GOTO 101
END IF
DDTS=DDO+(DDT-DDO)*(OMEGI-OMEGW)/(OMEGIS-OMEGW)
DDTS=AMIN1(DDTS,DDT)
IF(DDTS.EQ.DDO)GOTO 103
AFIN=3.14159*(DDTS-DDO)*(DDTS+DDO)/4.
AFS=AFIN*2.*WIDTH/FFPCH
SEG=DDTS*.3/24.-DDTS*DDTS*DDO/16.+DDO*.3/48.
SEG=SEG*2./(AFIN*(DDTS-DDO))
OMEGIS=AMIN1(OMEGIS,OMEGFI)
OMEGFI=OMEGW*(OMEGIS-OMEGW)/SEG
OMECHEF=(OMEGI-OMEGFI)*(1.-EXP(-HICOD(1CT)*AFS/(CPA*AMS)))
101 OMECH=XCOND*(OMECIP+OMECHEF)
103 OMECH=XCOND*(OMECIP+OMECHEF)
C
OMEGC=OMEGI-OMECHEF
CXC IF(1.EQ.14)PRINT 760,OMEGE
760 FORMAT(' OMEGE=',F14.6)
TWATA=.5*(TWATAI+TWATAE)
TPIPA=.5*(TRI+TRE)+Q/UPO
104 TTAIR=0.5*(TAI+TAE)
VFLA=AMAS*(450.+TTAIR)/(AFLOW*(FFPCH-FFIK-2.*TKICE(1CT)))
CALL WATPR(TWATA,TPIPA,VFLA,OMEGI,WATRO,WATK,
&WATM,WATHFG,WATCF)
DTHFG(1)=WATHFG*(OMEGI-OMEGE)/CPA
TAE=TAE+DTHFG(1)
TTAIR(2,1)=TAE
WAIIR(2,1)=OMEGE
AA1=1./(AIRN(1CT)+1.)
AA2=AA1*AIRN(1CT)
TAIR(2,1CT+1)=AA1*TAE+AA2*TAIR(2,1CT+1)
OMEGA(2,1CT+1)=AA1*OMEGE+AA2*OMEGA(2,1CT+1)
TWAT(1CT)=AA1*TWATA+AA2*TWAT(1CT)
TPIP(1CT)=AA1*TPIPA+AA2*TPIP(1CT)
AIRN(1CT)=AIRN(1CT)+1.
C

```

```

7530. 00 C**** SELECT NEXT TURE FOR CALCULATIONS
7540. 00 106 IF(MY(1).EQ.1)THEN
7550. 00   DO 108 N=2,3
7560. 00   NN=KFEED(110,1,N)
7570. 00   IF(NN.EQ.0)GOTO 109
7580. 00   ILN=ILN+1
7590. 00   ILNOK(ILN)=NN
7600. 00   108 CONTINUE
7610. 00   END IF
7620. 00   109 JJ=1
7630. 00   I=KFEED(110,J,1)
7640. 00   IF(I.NE.-1)GOTO 22
7650. 00   IF(ILN.EQ.ILNEXT)GOTO 110
7660. 00   ILNEXT=ILNEXT+1
7670. 00   I=ILNOK(ILNEXT)
7680. 00   JJ=IFROM(110,1)
7690. 00   GOTO 22
7700. 00   110 CONTINUE
7710. 00   C
7720. 00   C**** ALL TUBES OF COIL COMPUTED. CHECK IF CONVERGENCE OBTAINED
7730. 00   C
7740. 00   H2=0.
7750. 00   DO 111 IT=1,IST(110)
7760. 00   I=1START(110,IT)
7770. 00   111 H2=H2+HR(2,1)*FLOW(110,I)
7780. 00   H2=H2*NSECT(110)
7790. 00   H2PH2=H2P-H2
7800. 00   IF(1AIR.LT.3)GOTO 114
7810. 00   112 IF(ABS(H2PH2).LT.0.06)GOTO 160
7820. 00   114 H2P=H2
7830. 00   HBACK(1AIR,1)=H2
7840. 00   HBACK(1AIR,2)=H2PH2
7850. 00   C**** CONVERGENCE NOT OBTAINED
7860. 00   C**** PREPARE AIR SIDE DATA FOR NEW LOOP
7870. 00   DO 124 I=1,NNDP
7880. 00   J=I+1
7890. 00   TAIR(1,J)=0.5*(TAIR(1,J)+TAIR(2,J))
7900. 00   OMEGA(1,J)=OMEGA(2,J)
7910. 00   124 OMEGA(2,J)=0.
7920. 00   DO 140 I=1,NNDP
7930. 00   TWATA=TWAT(1)
7940. 00   TTAIR=0.5*(TAIR(1,I+1)+TAIR(2,I+1))
7950. 00   TAIR(2,I+1)=0.
7960. 00   WVAIR=OMEGA(1,I)
7970. 00   CALL AIRPR(1,TWATA,APIN,1.,WWATER,CCPA,RAA,
7980. 00   &AAMA,AKA)
7990. 00   TPIPE=TPIP(1)
8000. 00   VELA=AMAS*(460.+TTAIR)/AFLOW
8010. 00   VELA=VELA/(FFCH-FFTK-2.*TKICE(1))
8020. 00   CALL WATPR(TWATA,TPIPE,VELA,WVAIR,WATRO,WATK,
8030. 00   &WATH,WATHFG,WATCP)
8040. 00   IF(OMEGA(1,I).GT.WWATER)GOTO 125
8050. 00   HICE(1)=1.E+30
8060. 00   HFGWT(1)=0.
8070. 00   TKICE(1)=0.
8080. 00   GOTO 140
8090. 00   125 IF(OMEGA(1,I+1).LT.OMEGA(1,I))GOTO 126
8100. 00   OMEGA(1,I+1)=OMEGA(1,I)

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EVAPHX

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8110. HICE(1)=1.E+30
8120. HFGWT(1)=0.
8130. TKICE(1)=0.
8140. GOTO 140
8150.
8160. 126 WMAS=AMAS*(OMEGA(1,1)-OMEGA(1,1+1))
8170. HFGWT(1)=WATHG*(OMEGA(1,1)-WWATER)
8180. CALL AIRPR(2,TAIR,APIN,PRRI,WWAIR,CPA,PPA,AAIA,AAKA)
8190. HFGWT(1)=HFGWT(1)/(CPA*(TAIR(1,1)-TWATA))
8200. IF(TWATA.GT.32.)GOTO 128
8210. TKICE(1)=0.125*WMAS/(AO*NR0W(110)*WATRO)
8220. TKMAX=0.5*(FFPCH-FFTK)
8230. IF(TKICE(1).GE.TKMAX)TKICE(1)=0.9*TKMAX
8240. GOTO 132
8250.
8260. 128 WMW=WMAS/VFLW
8270. WMW=WATH*WMW/'WATRC*WATRO)
8280. TKICE(1)=1.449E-03*WMW/'*0.333
8290. HICE(1)=WATK/TKICE(1)
8300. IF(HICE(1).LT.0.)HICE(1)=0.
8310. IF(TKICE(1).LT.0.)TKICE(1)=0.
8320. IF(HFGWT(1).LT.0.)HFGWT(1)=0.
8330. 140 CONTINUE
8340. NTOT=0
8350. NDEEP=NDEP(110)-1
8360. DO 149 ICT=1,NDEEP
8370. NTOT=NTOT+NTUB(110,ICT)
8380. NDIF=NTUB(110,ICT)-NTUB(110,ICT+1)
8390.
8400. IF(NDIF.EQ.0)THEN
8410. IA=NTOT+1
8420. IB=NTOT+NTUB(110,ICT+1)
8430. DO 142 I=1A,IB
8440. J=JFROM(110,I)
8450. TAIRI(1,1)=TAIRI(2,J)
8460. WAIRI(1,1)=WAIRI(2,J)
8470. END IF
8480.
8490. C
8500. IF(NDIF.LT.0)THEN
8510. IA=NTOT+1
8520. IB=NTOT+NTUB(110,ICT)-1
8530. DO 143 I=1A,IB
8540. J=JFROM(110,I)
8550. TAIRI(1,1)=0.5*(TAIRI(1,1)+TAIRI(2,J))
8560. WAIRI(1,1)=0.5*(WAIRI(1,1)+WAIRI(2,J))
8570. IA=NTOT-NTUB(110,ICT)+1
8580. IB=NTOT-1
8590. T=0.
8600. WW=0.
8610. DO 144 I=1A,IB
8620. T=T+TAIRI(2,I)
8630. WW=WW+WAIRI(2,I)
8640. T=T/SEG
8650. WW=WW/SEG
8660. IA=NTOT+NTUB(110,ICT)
8670. IB=NTOT+NTUB(110,ICT+1)
8680. DO 145 I=1A,IB

```

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***** EVAPHX *****
8690. 00
8700. 00
8710. 00
8720. 00
8730. 00
8740. 00
8750. 00
8760. 00
8770. 00
8780. 00
8790. 00
8800. 00
8810. 00
8820. 00
8830. 00
8840. 00
8850. 00
8860. 00
8870. 00
8880. 00
8890. 00
8900. 00
8910. 00
8920. 00
8930. 00
8940. 00
8950. 00
8960. 00
8970. 00
8980. 00
8990. 00
9000. 00
9010. 00
9020. 00
9030. 00
9040. 00
9050. 00
9060. 00
9070. 00
9080. 00
9090. 00
9100. 00
9110. 00
9120. 00
9130. 00
9140. 00
9150. 00
9160. 00
9170. 00
9180. 00
9190. 00
9200. 00
9210. 00
9220. 00
9230. 00
9240. 00
9250. 00
9260. 00

      TAIRI(1,1)=0.5*(TAIRI(1,1)+T)
      WAIRI(1,1)=0.5*(WAIRI(1,1)+WW)
      END IF

      IF (NDIF.GT.0) THEN
        IA=NTOT-NDIF+1
        TCCR=0.
        WCCR=0.
        DO 146 I=1A,NTOT
          TCCR=TCOR+TAIRI(2,I)
          WCCR=WCCR+WAIRI(2,I)
          TCCR=TCOR/NTUB(110,ICT+1)
          WCCR=WCCR/NTUB(110,ICT+1)
          IA=NTOT+1
          IB=NTOT+NTUB(110,ICT+1)
          DO 147 I=1A,IB
            J=JFROM(110,I)
            TAIRI(1,1)=.5*(TAIRI(1,1)+(TAIRI(2,J)+TCCR)*AMSI(1CT)/AMSI(1CT+1))
            WAIRI(1,1)=.5*(WAIRI(1,1)+(WAIRI(2,J)+WCCR)*AMSI(1CT)/AMSI(1CT+1))
          END IF
        END IF
      END IF
      PRINT 858,H2
      858 FORMAT(' H2=',F8.4)
      150 CONTINUE

C *****
C ***** END OF MAIN LOOP *****
C *****

      I=5
      DO 151 IEV=6,IAIRN
        151 IF(AES(HBACK(1,2)).GT.ABC(HBACK(1EV,2))) I=1EV
        H2PH2=HBACK(1,2)
        Q=HBACK(1,1)+H2PH2
        Q=ABS(H2PH2*RMAS5)
        WRITE(6,1021)H2PH2,Q
      160 CONTINUE
      DO 162 I=1,NTPS(110)
        T=TRM(110,1,I)
        TRM(110,1,I)=TRM(110,2,I)
        TRM(110,2,I)=T
        T=PRM(110,1,I)
        PRM(110,1,I)=PRM(110,2,I)
        PRM(110,2,I)=T
        T=XRM(110,1,I)
        XRM(110,1,I)=XRM(110,2,I)
        XRM(110,2,I)=T
        T=VLM(110,1,I)
        VLM(110,1,I)=VLM(110,2,I)
        VLM(110,2,I)=T
        T=VGM(110,1,I)
        VGM(110,1,I)=VGM(110,2,I)
        VGM(110,2,I)=T
      162 VGM(110,2,I)=T
        P2=0.
      DO 164 I=1,IST(110)
        IE=1START(110,I)
        P2=P2+PRM(110,2,IE)*FLOW(110,IE)
      164 CONTINUE

```

***** EVAPHX *****

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9270. 00 P2=P2*NSECT(110)
9280. 00 PA2=P2/14.0959
9290. 00 CALL HPIN(MO,H2,PA2,XM,ACC,TK2,X2,XL,XV,VL,VV)
9300. 00 T2=TK2*1.8-459.67
9310. 00 IF(X2.LT..9999)THEN
9320. 00   TSUP2=0.
9330. 00 ELSE
9340. 00   CALL DEUTEM(N1,PA2,XM,TKD,XL)
9350. 00   TD=TKD*1.8-459.67
9360. 00   TSUP2=T2-TD
9370. 00 END IF
9380. 00 QT=RIASS*(H2-H1)
9390. 00 QL=WATHEG*AAMAS*(OMEGA(1,1)-OMEGA(1,NNDP+1))
9400. 00 QS=QT-QL
9410. 00 PRINT 880,QT,OS
9420. 00 880 FORMAT('QT,OS=',3(1PE11.3))
9430. 00 WRITE(6,1012)T1,P1,H1,X1,T2,P2,H2,X2,TSUP2
9440. 00 900 FORMAT(/2X,'INPUT DATA TO EVAPHX: '//2X,'T',
9450. 00   &10X,'P',10X,'X',10X,'TAIR',7X,'RH',9X,'RIASS'/6(1PE11.3))
9460. 00 1021 FORMAT('EVAPHX DOES NOT CONVERGE',/
9470. 00   &' CONVERGENCE OBTAINED = ',F6.2,' BTU/L3 ',F11.1,' BTU/H')
9480. 00 1012 FORMAT(/2X,'EVAPORATOR ITERATION: '//2X,
9490. 00   &'T',10X,'P',10X,'H',10X,'X',10X,'TSUP',/4(1PE11.3)/5(1PE11.3))
9500. 00 RETURN
9510. 00 END

```

END ELT. ERRORS: NONE. TIME: 1.071 SEC. IMAGE COUNT: 951

@HDG,P ***** EVDP ***** .L,0

```

***** EVDP *****
@ELT,L DD.EVDP
ELT 8R1 S74Q1C 07/21/84 15:55:06 (O)
      10. 00 FUNCTION EVDP(RMS,H1,H2,VMIX,VISL,AL,D)
      20. 00 C
      30. 00 C**** PURPOSE:
      40. 00 C TO COMPUTE FRICTIONAL EVAPORATION PRESSURE DROP
      50. 00 C FOR FLOW IN A TUBE
      60. 00 C
      70. 00 C**** INPUT DATA:
      80. 00 C AL - TUBE LENGTH (FT)
      90. 00 C VISL - LIQUID DYNAMIC VISCOSITY (LBM/H*FT)
     100. 00 C D - TUBE INSIDE DIAMETER (FT)
     110. 00 C RMS - REFRIG. MASS FLOW RATE (LBM/H)
     120. 00 C H1 - REFRIG. ENTHALPY AT TUBE INLET (BTU/LBM)
     130. 00 C H2 - REFRIG. ENTHALPY AT TUBE OUTLET (BTU/LBM)
     140. 00 C VMIX - REFRIG. AVERAGE SPEC. VOLUME IN A TUBE (FT**3/LBM)
     150. 00 C
     160. 00 C**** OUTPUT DATA:
     170. 00 C EVDP - FRICTIONAL EVAPORATION PRESSURE DROP (PSI)
     180. 00 C
     190. 00 AC=1.6654E-11 @32.174*144.*3600.**2
     200. 00 G=RMS/(0.78539816*D*D)
     210. 00 RE=G*D/VISL
     220. 00 AKF=778.26*(H2-H1)/AL
     230. 00 RATIO=RE/AKF
     240. 00 IF(RATIO.LT.1.)RATIO=1.
     250. 00 F=0.0185/RATIO**.25
     260. 00 EVDP=AC*F*AL*G*VMIX/D
     270. 00 RETURN
     280. 00 END

END ELT. ERRORS: NONE. TIME: 0.080 SEC. IMAGE COUNT: 28

@HDS,P ***** FANNO ***** .L,0

```


***** FANNO *****

@ELT, L DD.FANNO

```

ELT 8R1 S74Q1C 07/21/84 15:55:07 (0)
10. 00 FUNCTION FANNO(XM,P,H0,GG)
20. 00 C
30. 00 C**** PURPOSE:
40. 00 TO CALCULATE ENTROPY OF NON-AZENTROPIC
50. 00 TWO-PHASE MIXTURE IN THE FANNO FLOW
60. 00 FOR A GIVEN PRESSURE
70. 00 C
80. 00 C**** INPUT DATA:
90. 00 H0 - REFRIG. TOTAL ENTHALPY (BTU/LM)
100. 00 GG = G*G/(64.4*778.104) (BTU*LM/FT**6)
110. 00 C WHERE G - REFRIG. MASS FLUX (LM/(SEC*FT**2))
120. 00 P - REFRIG. PRESSURE AT WHICH ENTROPY IS DESIRED (PSIA)
130. 00 XM - MOLAR COMPOSITION (FRACTION OF LESS VOLATILE COMPONENT)
140. 00 C**** OUTPUT DATA:
150. 00 FANNO - ENTROPY (BTU/LM R)
160. 00 C
170. 00 C**** SUBPROGRAMS CALLED BY FANNO:
180. 00 EBUTE,ENTROP,HCVCP,QLITY,VOLITI
190. 00 C
200. 00 COMMON/RDATA2/W1,W2,TC1,TC2
210. 00 DATA SLOPE/O./NO,N1/O,1/
220. 00 C
230. 00 IF(P.GT.0.)GOTO 10
240. 00 WRITE(6,600)P
250. 00 600 FORMAT(' ERROR IN CALLING FANNO, P=',1PE15.5,' PSIA')
260. 00 RETURN
270. 00 C
280. 00 10 PA=P/14.6959
290. 00 TKB=EBUTE(PA,XM)
300. 00 TKD=TKB+9.5-20.*(1.55-XM)
310. 00 TB=TKB*1.8-459.67
320. 00 TD=TKD*1.8-459.67
330. 00 TK2=0.5*(TKB+TKD)
340. 00 DO 100 I=1,20
350. 00 CALL QLITY(TK2,PA,XM,XG,XV,XL)
360. 00 CALL VOLITI(N1,TK2,PA,XV,VTV)
370. 00 CALL HCVCP(N1,TK2,VTV,XV,HV,CV,CP)
380. 00 CALL VOLITI(N0,TK2,PA,XL,VML)
390. 00 CALL HCVCP(N1,TK2,VML,XL,HL,CV,CP)
400. 00 HFG=HV-HL
410. 00 WMV=W1*(1.-XV)+W2*XV
420. 00 WML=W1*(1.-XL)+W2*XL
430. 00 VV=WMV*16.01845/WMV
440. 00 VL=VML*15.01845/VML
450. 00 VFG=VV-VL
460. 00 R=GG*VFG*VFG
470. 00 S=2.*GG*VV*VFG+HFG
480. 00 C=GG*VL*VL+HL-HO
490. 00 XQQ=(-S+SQRT(S*S-4.*R*C))/(2.*R)
500. 00 DIFXQ2=XQQ-XQ
510. 00 TF=TK2*1.8-450.
520. 00 PRINT 800,P,TF,XQQ,DIFXQ2
530. 00 800 FORMAT(' P,TF,XQQ,XDIF=',4(1PE15.6))
540. 00 IF(ABS(DIFXQ2).LT.0.000001)GOTO 110
550. 00 C
560. 00 IF(1.GT.1)GOTO 70

```

```

***** FANNO *****
570.      TK1=TK2
580.      DIFXQ1=DIFXQ2
590.      IF(SLOPE.NE.0.) THEN
600.          TDEL=DIFXQ2*SLOPE
610.          TD=ABS(TDEL)
620.          TD=AMIN1(4.,TD)
630.          TK2=TK1-SIGN(TD,TDEL)
640.      ELSE
650.          TK2=TK1+SIGN(1.,DIFXQ2)
660.      END IF
670.      GOTO 100
C
70  SLOPE=(TK2-TK1)/(DIFXQ2-DIFXQ1)
   IF(ABS(DIFXQ1).LT.ABS(DIFXQ2))GOTO 80
   TK1=TK2
   DIFXQ1=DIFXQ2
80  TK2=TK1-DIFXQ1*SLOPE
100 CONTINUE
110 PRINT 602,DIFXQ2
   FORMAT(' FANNO DOES NOT CONVERGE, DIFXQ2=',1PE15.5)
   SL=ENTROP(TK2,VML,XL)
   SV=ENTROP(TK2,VMV,XV)
   FANNO=(1.-XQQ)*SL+XQQ*SV
   RETURN
   END
END ELT.  ERRORS: NONE.  TIME:  0.135 SEC.  IMAGE COUNT: 82
@HDS,P ***** FEELIQ ***** .L,0

```

```

***** FEELIQ *****
@ELT,L DD,FEELIQ
ELT 8R1 S7401C 07/21/84 15:55:07 (0)
FUNCTION FEELIQ(X,T,XML,XMV,VML,VMV,XTT)
10. 00 C
20. 00 C ***** PURPOSE:
30. 00 C COMPUTE FOR A NON-AZEOTROPIC MIXTURE TWO-PHASE FLOW
40. 00 C LOCKHART-MARTINELLI CORRECTION PRESSURE DROP FACTOR
50. 00 C
60. 00 C ***** INPUT DATA:
70. 00 C T - REFRIG. AVERAGE TEMPERATURE (F)
80. 00 C VML - LIQUID SPECIFIC VOLUME (L/MOL)
90. 00 C VMV - VAPOR SPECIFIC VOLUME (L/MOL)
100. 00 C X - REFRIG. AVERAGE QUALITY (-)
110. 00 C XL - LIQUID MOLAR COMPOSITION
120. 00 C XV - VAPOR MOLAR COMPOSITION
130. 00 C W1 - FRACTION OF LESS VOLATILE COMPONENT
140. 00 C W2 - FRACTION OF MORE VOLATILE COMPONENT
150. 00 C W1 - MOLECULAR WEIGHT OF MORE VOLATILE COMPONENT
160. 00 C W2 - MOLECULAR WEIGHT OF LESS VOLATILE COMPONENT
170. 00 C
180. 00 C
190. 00 C ***** OUTPUT DATA:
200. 00 C FEELIQ - PRESSURE DROP CORRECTION FACTOR (-)
210. 00 C XTT - LOCKHART-MARTINELLI PARAMETER (-)
220. 00 C
230. 00 C ***** SUBPROGRAMS CALLED BY FEELIQ:
240. 00 C VISCON
250. 00 C
260. 00 C
270. 00 C COMMON/RDATA2/W1,W2,TC1,TC2
280. 00 C
290. 00 C A0=-.418956
300. 00 C A1=1.47330
310. 00 C A2=.668583
320. 00 C A3=-.321168
330. 00 C A4=0.0408167
340. 00 C XML=XML/(W1/W2*(1.-XML)+XML)
350. 00 C VISL=VISCON(1,T,XML)
360. 00 C XWV=XWV/(W1/W2*(1.-XWV)+XWV)
370. 00 C VISV=VISCON(3,T,XWV)
380. 00 C WML=W1*(1.-XML)+W2*XML
390. 00 C VL=16.01846*VML/VML
400. 00 C VMV=W1*(1.-XWV)+W2*XWV
410. 00 C WV=16.01846*VMV/VMV
420. 00 C XTT=((1.-X)/X)**.9*(VL/WV)**.5*(VISL/VISV)**.1
430. 00 C H4=1./XTT
440. 00 C H1=H4**0.25
450. 00 C H2=H1*H1
460. 00 C H3=H2*H1
470. 00 C FEELIQ=EXP(A0+A1*H1+A2*H2+A3*H3+A4*H4)
***** FEELIQ**2
END

```

END ELT. ERRORS: NONE. TIME: 0.098 SEC. IMAGE COUNT: 48

@HDS,P ***** FGIBBS ***** .L,0

```

***** FGIBBS *****
@ELT,L DD.FGIBBS
ELT 8R1 S74Q1C 07/21/84 15:55:07 (0)
FUNCTION FGIBBS(T,P,X,V)
10. 00 C
20. 00 C
30. 00 C
40. 00 C
50. 00 C
60. 00 C
70. 00 C
80. 00 C
90. 00 C
100. 00 C
110. 00 C
120. 00 C
130. 00 C
140. 00 C
150. 00 C
160. 00 C
170. 00 C
180. 00 C
190. 00 C
200. 00 C
210. 00 C
220. 00 C
230. 00 C
240. 00 C
250. 00 C
260. 00 C
270. 00 C
280. 00 C
290. 00 C
300. 00 C

C**** PURPOSE:
      TO EVALUATE GIBBS FREE ENERGY

C**** INPUT:
      P - REFRIG. PRESSURE (ATM)
      T - REFRIG. TEMPERATURE (K)
      V - REFRIG. SPEC. VOLUME (L/MOL)
      X - MOLAR CONCENTRATION OF A LESS VOLATILE REFRIG. (-)
      * - CONSTANTS A @ B AS PER COMMON STATEMENT PARAM

C**** OUTPUT:
      G - GIBBS FREE ENERGY (KJ)

COMMON/PARAM/A,B,C1,C2,D1,D2
DATA R/0.06206/

G=R*T*ALOG(R*T/(P*V))
G=G-2.*R*T/(V-B/4.)*(V-B/2.)
G=G+A/B*(ALOG(V/(V+B))-R/(V+B))
G=G+R*T/(V-B/4.)*2*(2.*V**3/(V-B/4.)+B**2/16.)
IF(X.EQ.0.OR.X.EQ.1.)GO TO 10
G=G+R*T*(X*ALOG(X)+(1.-X)*ALOG(1.-X))
10 FGIBBS=G
RETURN
END

END ELT. ERRORS: NONE. TIME: 0.081 SEC. IMAGE COUNT: 30
@HDG,P ***** HCVCP ***** .L,0

```



```

***** HCVCP *****
@ELT,L DD.HCVCP
ELT 8R1 S7401C 07/21/84 15:55:07 (0)
SUBROUTINE HCVCP(IQ,T,V,X,H,CV,CP)
10. C
20. C
30. C
40. C
50. C
60. C
70. C
80. C
90. C
100. C
110. C
120. C
130. C
140. C
150. C
160. C
170. C
180. C
190. C
200. C
210. C
220. C
230. C
240. C
250. C
260. C
270. C
280. C
290. C
300. C
310. C
320. C
330. C
340. C
350. C
360. C
370. C
380. C
390. C
400. C
410. C
420. C
430. C
440. C
450. C
460. C
470. C
480. C
490. C
500. C
510. C
520. C
530. C
540. C
550. C
560. C

***** PURPOSE:
    TO CALCULATE REFRIG. THERMODYNAMIC PROPERTIES

***** INPUT:
    IQ - OUTPUT QUALIFIER
        = 1 FOR ENTHALPY ONLY
        = 2 FOR ENTHALPY AND SPEC. HEAT AT CONST. VOL.
        = 3 FOR ENTHALPY, SPEC. HEAT AT CONST. VOL. @
          SPEC. HEAT AT CONST. PRESSURE
        = 4 FOR SPEC. HEAT AT CONST. VOL. ONLY
        = 5 FOR SPEC. HEAT AT CONST. VOL. @ SPEC. HEAT
          AT CONST. PRESSURE
    T - REFRIG. TEMPERATURE (K)
    V - REFRIG. SPEC VOLUME (L/MOL)
    X - MOLAR CONCENTRATION (FRACTION OF LESS VOLATILE REFRIG.)
    W1 - MOLECULAR WEIGHT OF MORE VOLATILE COMPONENT (G/MOL)
    W2 - MOLECULAR WEIGHT OF LESS VOLATILE COMPONENT (G/MOL)
    * - CONSTANTS AS PER COMMON STATEMENT /PARAM/
        A,B,C1,D1 FOR CALC. OF ENTHALPY
        A,B,C1,C2,D1,D2 FOR CALC. OF EITHER OF SPEC. HEAT

***** OUTPUT:
    CP - SPEC. HEAT AT CONST. PRESSURE (BTU/(LB*F))
    CV - SPEC. HEAT AT CONST. VOLUME (BTU/(LB*F))
    H - ENTHALPY (BTU/LB)

COMMON/PARAM/A,B,C1,C2,D1,D2
COMMON/RDATA2/W1,W2
DATA R/O.08206/

***** SUBPROGRAMS CALLED BY HCVCP:
    HPAR
    CALL HPAR(IQ,T,X)
    WMOL=(1.-X)*W1+X*W2
    CONV=453.5924/(1.055036*WMOL)
    T2=T*T
    T3=T2*T
    IF(IQ.GT.3)GOTO 10
***** CALC. OF ENTHALPY
    H=(C1*B*T-A*D1*T-A*B)/B**2*ALOG((V+B)/V)
    H=H+(A*D1*T-A*B)/(B*(V+B))
    H=H+2.*R*T*V*(2.*V-B/4.)/(V-B/4.)*3*(B/4.-D1*T/4.)
    H=0.101325*M
    H=H+(1.-X)*(0.0193537*(T-233.15)+1.091972E-04*(T2-233.15**2))
    H=H-(1.-X)*(5.6746E-08*(T3-233.15**3))
    H=H+(1.-X)*16.8076630893
    H=H*X*(0.0222804*(T-233.15)+7.700465E-05*(T2-233.15**2))
    H=H*X*(1.0222346E-09*(233.15**3-T3)+21.9460206504)
    H=H*CONV
    IF(IQ.EQ.1)GOTO 1000

```

```

***** HCVCP *****
570.      00      10 D12=D1*D1
580.      00      CV=(C2*B**2*T-2.*C1*D1*B*T+2.*A*D12*T-A*D2*B*T)/B**3
590.      00      CV=CV*ALOG((V+B)/V)
600.      00      CV=CV+T/B/(V+B)*((A*D2*B**2.*C1*D1*B-2.*A*D12)/B)
610.      00      CV=CV-T/B/(V+B)*A*D12/(V+B)
620.      00      CV=CV+2.*R*T*V/(V-B/4.)*3*((D2/4.*T+D1/2.)*(B/4.-2.*V)+D12/16.*T)
630.      00      CV=CV+6.*R*T2*V*D12/16.*(B/4.-2.*V)/(V-B/4.)*4
640.      00      CV=0.101325*CV
650.      00      CV=CV+(1.-X)*(0.0116293+2.163944E-04*T-1.702407E-7*T2)
660.      00      CVM=CV+X*(0.013963+1.540093E-04*T-3.066704E-9*T2)
670.      00      CV=CVM*CONV/1.8
680.      00      IF(IQ.EQ.2.OR.IQ.EQ.4)GOTO 1000
690.      00      Y=B/V/4.
700.      00      Y2=Y*Y
710.      00      Y3=Y2*Y
720.      00      Y4=Y3*Y
730.      00      PO=-R/(1.-Y)*4*(1+4.*Y+4.*Y2-4.*Y3+Y4)+A*(2.*V+B)/T/(V*B)**2
740.      00      TO=R/(1.-Y)*3*(1+Y+Y2-Y3+D1*T*(4.+4.*Y-2.*Y2)/V/4./(1.-Y))
750.      00      TO=TO-C1/(V+B)+A*D1/(V+B)**2
760.      00      CP=-0.101325*TO**2/PO
770.      00      CP=CONV*(CP+CVM)/1.8
780.      00      1000 RETURN
790.      00      END

```

END ELT. ERRORS: NONE. TIME: 0.134 SEC. IMAGE COUNT: 79

@HDG,P ***** HPAR ***** .L,0

***** HPAR *****

ELT, L DD, HPAR
ELT 8R1 S74Q1C 07/21/84 15:55:08 (0)

SUBROUTINE HPAR(IQ,T,X)

```

10. 00 C
20. 00 C
30. 00 C
40. 00 C
50. 00 C
60. 00 C
70. 00 C
80. 00 C
90. 00 C
100. 00 C
110. 00 C
120. 00 C
130. 00 C
140. 00 C
150. 00 C
160. 00 C
170. 00 C
180. 00 C
190. 00 C
200. 00 C
210. 00 C
220. 00 C
230. 00 C
240. 00 C
250. 00 C
260. 00 C
270. 00 C
280. 00 C
290. 00 C
300. 00 C
310. 00 C
320. 00 C
330. 00 C
340. 00 C
350. 00 C
360. 00 C
370. 00 C
380. 00 C
390. 00 C
400. 00 C
410. 00 C
420. 00 C
430. 00 C
440. 00 C
450. 00 C
460. 00 C
470. 00 C
480. 00 C
490. 00 C
500. 00 C
510. 00 C
520. 00 C
530. 00 C
540. 00 C
550. 00 C
560. 00 C

```

C ***** PURPOSE:

TO CALC. PARAMETERS FOR CALCULATIONS OF REFRIG.
ENTHALPY, SPEC. HEAT AND ENTROPY
(CONSTANTS A @ B FOR EQ. OF STATE ARE INCLUDED)

C ***** INPUT:

REFRIG. CONSTANTS AS LISTED IN THE COMMON STATEMENT /RDATA1/
IQ - OUTPUT QUALIFIER
= 1, IF CONSTANTS FOR CALC. OF ENTHALPY OR ENTROPY REQUIRED
> 1, IF ALSO CONSTANTS FOR SPEC. HEAT REQUIRED
T - REFRIG. TEMPERATURE (K)
X - MOLAR CONCENTRATION OF A LESS VOLATILE REFRIG. (-)

C ***** OUTPUT:

* - CONSTANTS FOR EVALUATION OF REFRIG.
ENTHALPY, A,B,C1,D1
ENTROPY, A,B,C1,D1
SPEC. HEAT, A,B,C1,C2,D1,D2

C ***** PROGRAMS CALLED BY HPAR:

C EQPAR

COMMON/RDATA1/A3,A4,A5,A6,A7,A8,B3,B4,B5,B6,B7,B8,F0,F1
COMMON/PARAM/A,B,C1,C2,D1,D2
COMMON/STORE1/A1,A2,B1,B2,SEG31
DATA TLAST,XLAST/2*0./

CALL EQPAR(T,X)

IF(ABS(T-TLAST).GT.0.0001)GOTO 10
IF(ABS(X-XLAST).GT.0.0001)GOTO 10
IF(IQ.NE.1)GOTO 10
RETURN

10 TLAST=T

XLAST=X

X2=X**2

XX=1.-X

XX2=XX**2

XXX=XXX

T2=T**2

A1A2S=SQRT(A1*A2)

SEG1=XXX*A1A2S

F=F0+F1*T

SEG2=XXX*(1.-F)*A1A2S

SEG32=SEG31**2

SEG5=B4+2.*B5*T

SEG6=B7+2.*B8*T

SEG7=A4+2.*A5*T

SEG8=A7+2.*A8*T

SEG9=B1*(2./3.)

SEG10=B2*(2./3.)

D1=XX2*SEG5+Y2*SEG6

```

***** HPAR *****
570. 00
580. 00
590. 00
600. 00
610. 00
620. 00
630. 00
640. 00
650. 00
660. 00
670. 00
680. 00
690. 00
700. 00
710. 00
720. 00
730. 00
740. 00
750. 00
760. 00
770. 00
780. 00
790. 00

C
D4=SEG5/SEG9+SEG6/SEG10
D4=D4*XXX*SEG32
D1=D1+D4
C1=XX2*SEG7+X2*SEG8
C1=C1+SEG1*((1.-F)*SEG7/A1)
C1=C1+SEG1*((1.-F)*SEG8/A2)
C1=C1-2.*XXX*F1+A1A2S
IF(IQ.EQ.1)RETURN
D2=SEG5**2/B1** (5./3.)+SEG6**2/B2** (5./3.)
D2=-2./3.*D2+2.*(D5/SEG9-B8/SEG10)
D2=D2*XXX*SEG32
D3=SEG5/SEG9+SEG6/SEG10
D3=D3**2*XXX*SEG31/3.
D2=D2+D3+2.*(XX2*D5+X2*B8)
C2=2.*(XX2*A5+X2*A8)
C2=C2-2.*SEG1*F1*(SEG7/A1+SEG8/A2)
C2=C2+SEG2*(2.*A5/A1+2.*A8/A2)
C2=C2+SEG2*(SEG7*SEG8/A1/A2)
C3=(SEG7/A1)**2*(SEG9/A2)**2
C2=C2-XXX*(1.-F)*(A1*A)**0.5*C3/2.
RETURN
END

```

END ELT. ERRORS: NONE. TIME: 0.125 SEC. IMAGE COUNT: 79

@HDS,P ***** HPIN ***** .L,0


```

***** HPIN *****
0ELT,L DD,HPIN
ELT OR1 S74Q1C 07/21/84 15:55:08 (0)
SUBROUTINE HPIN(IG,H,P,X,ACCUR,T,XQ,XL,XV,VL,VV)
10. C
20. C
30. C
40. C
50. C
60. C
70. C
80. C
90. C
100. C
110. C
120. C
130. C
140. C
150. C
160. C
170. C
180. C
190. C
200. C
210. C
220. C
230. C
240. C
250. C
260. C
270. C
280. C
290. C
300. C
310. C
320. C
330. C
340. C
350. C
360. C
370. C
380. C
390. C
400. C
410. C
420. C
430. C
440. C
450. C
460. C
470. C
480. C
490. C
500. C
510. C
520. C
530. C
540. C
550. C
560. C

***** PURPOSE:
TO CALC. TEMPERATURE OF BINARY MIXTURE FROM GIVEN
ENTHALPY AND PRESSURE

***** INPUT:
- REQUIRED ACCURACY OF CONVEGANCE (BTU/LB)
= 0, IF GUESS OF TEMPERATURE IS NOT GIVEN
= 1, IF GUESS OF TEMPERATURE IS GIVEN
- ENTHALPY (BTU/LB)
- PRESSURE (STD ATM)
- GUESS OF TEMPERATURE, OPTIONAL (K)
- MOLAR CONCENTRATION (FRACTION OF LESS VOLATILE COMPONENT)

***** OUTPUT:
- BINARY MIXTURE TEMPERATURE (K)
- SPEC. VOLUME OF SUBCOOLED OR SAT. LIQUID AT T TEMP.
AND P PRESSURE, IF MIXTURE IS SUBCOOLED
OR IN TWO-PHASE, RESPECTIVELY. (L/MOL)
- SPEC. VOLUME OF SUPERHEATED OR SAT. VAPOR AT T TEMP.
AND P PRESSURE, IF MIXTURE IS SUPERHEATED
OR IN TWO-PHASE, RESPECTIVELY. (L/MOL)
- MOLAR COMPOSITION OF SAT. LIQUID
(FRACTION OF LESS VOLATILE COMPONENT)
- MOLAR COMPOSITION OF SAT. VAPOR (FRACTION OF LESS VOLATILE COMPONENT)
- QUALITY (-)

***** SUBPROGRAMS CALLED BY HPIN:
DBUBTE,DDEWTE,DQITY,ERUBTE,HCVCP,QQITY,VOLITI

REAL*8 TDD,PDD,XQDD,XVDD,XLDD
COMMON/RDATA2/W1,W2,TC1,TC2
DATA SLOPE/0./,PLAST/0./

IF(P.LE.0.)THEN
PRINT S90,P
590 FORMAT(' HPIN CALLED WITH NON-POSSITIVE PRESSURE, PATM=',1PE15.6)
RETURN
END IF

TT=T
IF(ABS(P-PLAST).GT.1.E-4)GOTO 2
IF(ABS(H-HLAST).GT.ACCUR)GOTO 2
IF(ABS(X-XLAST).GT.1.E-4)GOTO 2
T=TLAST
XQ=XQLAST
XL=XLAST
XV=XVLAST
VL=VLLAST
VV=VVLAST
RETURN
2 IF(IG.NE.1)T=ERUBTE(P,X)-3.

```

```

570. 00
580. 00
590. 00
600. 00
610. 00
620. 00
630. 00
640. 00
650. 00
660. 00
670. 00
680. 00
690. 00
700. 00
710. 00
720. 00
730. 00
740. 00
750. 00
760. 00
770. 00
780. 00
790. 00
800. 00
810. 00
820. 00
830. 00
840. 00
850. 00
860. 00
870. 00
880. 00
890. 00
900. 00
910. 00
920. 00
930. 00
940. 00
950. 00
960. 00
970. 00
980. 00
990. 00
1000. 00
1010. 00
1020. 00
1030. 00
1040. 00
1050. 00
1060. 00
1070. 00
1080. 00
1090. 00
1100. 00
1110. 00
1120. 00
1130. 00
1140. 00

XQX=.5
IDD=1.0029-ACCUR
DO 50 I=1,20
IF(I.EQ.1)GOTO 8
IF(XQ.EQ.0.)AND.HDIF2.LT.0.)GOTO 11
IF(XQ.EQ.1.)AND.HDIF2.GT.0.)GOTO 12
8 IF(T.GT.TC1)THEN
  XQ=1.
  GOTO 12
END IF
IF(1DD.EQ.0)THEN
  CALL OLITY(T,P,X,XQ,XV,XL)
ELSE
  TDD=T
  PDD=P
  CALL DOLITY(TDD,PDD,X,XQDD,XVDD,XLDD)
  XQ=XQDD
  XV=XVDD
  XL=XLDD
END IF
IF(XQ.EQ.1.)GOTO 12
11 XX=X
  IF(XQ.NE.0.)XX=XL
  CALL VOLIT1(O,T,P,XX,VL)
  CALL HCVC1(1,T,VL,XX,HL,CV,CP)
  IF(XQ.EQ.0.)GOTO 14
12 XX=X
  IF(XQ.NE.1.)XX=XV
  CALL VOLIT1(1,T,P,XX,VV)
  CALL HCVC1(1,T,VV,XX,HV,CV,CP)
  HH=(1.-XQ)*HL+XQ*HV
  T2=T
  HDIF2=H-HH
  IF(ABS(HDIF2).LT.ACCUR)GOTO 100
  IF(1.NE.1)GOTO 20
15 T1=T2
  HDIF1=HDIF2
  IF(SLOPE.NE.0.)THEN
    DT=HDIF2*SLOPE
    IF(ABS(DT).GT.10.)DT=SIGN(10.,DT)
    T=T2-DT
    GOTO 48
  END IF
  T=T2+8.
  IF(HDIF2.LT.0.)T=T2-8.
  GOTO 48

C
20 IF(XQ.EQ.0.)AND.HDIF2.GT.0.)THEN
  IF(XQX.EQ.0.)GOTO 22
  1DD=1
  TDD=T2
  PDD=P
  CALL NBURTE(1,PDD,X,TDD,XVDD)
  TB=TDD
  CALL VOLIT1(O,TB,P,X,VL)
  CALL HCVC1(1,TB,VL,X,HB,CV,CP)
  IF(H.GT.HB)THEN
    HDIF2=H-HB

```

```

***** HPIN *****
1150. 00
1160. 00
1170. 00
1180. 00
1190. 00
1200. 00
1210. 00
1220. 00
1230. 00
1240. 00
1250. 00
1260. 00
1270. 00
1280. 00
1290. 00
1300. 00
1310. 00
1320. 00
1330. 00
1340. 00
1350. 00
1360. 00
1370. 00
1380. 00
1390. 00
1400. 00
1410. 00
1420. 00
1430. 00
1440. 00
1450. 00
1460. 00
1470. 00
1480. 00
1490. 00
1500. 00
1510. 00
1520. 00
1530. 00
1540. 00
1550. 00
1560. 00
1570. 00
1580. 00
1590. 00
1600. 00
1610. 00
1620. 00
1630. 00
1640. 00
1650. 00
1660. 00

      T2=TB
      ELSE
        HDIF1=H-HB
        T1=TB
      END IF
      GOTO 22
    END IF
    IF(XQ.EQ.1..AND.HDIF2.LT.0.)THEN
      IF(XQX.EQ.1.)GOTO 22
      IDD=1
      TDR=T2
      PDD=P
      CALL NDEWTE(1,PDD,X,TDD,XLDD)
      TD=TDD
      CALL VOLIT1(1,TD,P,X,VV)
      CALL HCVCP(1,TD,VV,X,HV,CV,CP)
      IF(H.LT.HV)THEN
        T2=TD
        HDIF2=H-HV
      ELSE
        T1=TD
        HDIF1=H-HV
      END IF
    END IF
    22 IF(HDIF1.EQ.HDIF2)THEN
      WRITE(6,601)XQ,DT,HDIF2
      GOTO 60
    END IF
    SLOPE=(T2-T1)/(HDIF2-HDIF1)
    IF(ABS(HDIF1).LT.ABS(HDIF2))GOTO 30
    T1=T2
    HDIF1=HDIF2
    30 DT=HDIF1*SLOPE
    IF(ABS(DT).GT.10.)DT=SIGN(10.,DT)
    T=T1-DT
    48 XQX=XQ
    50 CONTINUE
    60 WRITE(6,600)IG,HDIF2,H,P,X,ACCUR,TT,XQ,XQX
    600 FORMAT(' ERROR 600 IN HPIN, IG,HDIF2=',I3,1PE12.4,
      & /, ' H,P,X,ACCUR,TT,XQ,XQX=',7(1PE12.4))
    601 FORMAT(' HPIN DOES NOT CONVERGE ANY FURTHER, XQ,DT,HDIF2=',3F8.4)
    100 PLAST=P
    HLAST=H
    XLAST=X
    TLAST=T
    XQLAST=XQ
    XLAST=XL
    XVLAST=XV
    VLLAST=VL
    VVLAST=VV
    RETURN
  END

```

END ELT. ERRORS: NONE. TIME: 0.221 SEC. IMAGE COUNT: 166

@HDG,P ***** HPPROP ***** .L,0

***** HPPROP *****

```

@ELT,L DD,HPPROP
ELT 8R1 S74Q1C 07/21/84 15:55:08 (0)
SUBROUTINE HPPROP(IG,H,PA,XM,ACC,TK,XQ,XML,XMV,VLM,VVM,
@ VM,V,CP,CV,AM,AK)
C
C**** PURPOSE:
C TO CALC. THERMODYNAMIC AND TRANSPORT AND PROPERTIES
C OF BINARY MIXTURE FROM GIVEN PRESSURE AND ENTHALPY
C
C**** INPUT DATA:
C IG = 0, IF GUESS OF REFRIG. TEMP. IS NOT GIVEN
C = 1, IF GUESS OF REFRIG. TEMP. IS GIVEN
C ACC - ACCURACY OF CONVERGENCE REQUIRED (BTU/LB)
C H - ENTHALPY (BTU/LB)
C PA - PRESSURE (STD. ATM)
C TK - GUESS OF REFRIG. TEMPERATURE, OPTIONAL (K)
C XM - MOLAR COMPOSITION (FRACTION OF LESS VOLATILE COMPONENT)
C
C**** OUTPUT DATA:
C AK - THERMAL CONDUCTIVITY (BTU/(H*FT*F))
C AM - ABSOLUTE VISCOSITY (LB/(H*FT))
C CP - SPEC. HEAT AT CONSTANT PRESSURE (BTU/(LB*F))
C CV - SPEC. HEAT AT CONSTANT VOLUME (BTU/(LB*F))
C TK - TEMPERATURE (K)
C V - SPEC. VOLUME OF MIXTURE (FT**3/LB)
C VLM - SPEC. VOLUME OF SAT. LIQUID AT TK TEMP. AND PA PRESSURE
C VM - SPEC. VOLUME OF MIXTURE (L/MOL)
C VVM - SPEC. VOLUME OF SAT. VAPOR AT TK TEMP. AND PA PRESSURE
C IF MIXTURE IS IN TWO-PHASE (L/MOL)
C XML - COMPOSITION OF SAT. LIQUID AT TK TEMP. AND PA PRESSURE
C (MOLAR FRACTION OF LESS VOLATILE COMPONENT)
C XMV - COMPOSITION OF SAT. VAPOR AT TK TEMP. AND PA PRESSURE
C (MOLAR FRACTION OF LESS VOLATILE COMPONENT)
C XQ - QUALITY (-)
C
C**** SUBPROGRAMS CALLED BY HPPROP:
C HCVCP,HPIN,VISCO
C
COMMON/RDATA2/W1,W2,TC1,TC2
DATA N1,N2,N3,N4,N5/1,2,3,4,5/
C
C
WM=(1.-XM)*W1+XM*W2
XW=XM/(W1/W2*(1.-XM)+XM)
CALL HPIN(IG,H,PA,XM,ACC,TK,XQ,XML,XMV,VLM,VVM)
T=TK*1.8-459.67
IF(XQ.EQ.0.) THEN
VM=VLM
V=VM*16.01845/WM
AM=VISCO(N1,T,XW)
AK=VISCO(N2,T,XW)
CALL HCVCP(N5,TK,VLM,XM,DUM,CV,CP)
GOTO 1000
END IF
C
IF(XQ.NE.1.) THEN
@TWO-PHASE
XL=XML/(W1/W2*(1.-XML)+XML)

```



```

***** HPPROP *****
570.      00      XV=XI*V/(V1/W2*(1.-XIV)+XIV)
580.      00      AM=(1.-XQ)*VISCN(N1,T,XL)
590.      00      AM=AM+XQ*VISCN(N3,T,XV)
600.      00      AK=(1.-XQ)*VISCN(N2,T,XL)
610.      00      AK=AK+XQ*VISCN(N4,T,XV)
620.      00      WML=(1.-XIV)*W1+XIV*W2
630.      00      WMV=(1.-XIV)*W1+XIV*W2
640.      00      V=16.01845*(1.-XQ)*VLM/WML+XQ*VM1/WMV)
650.      00      VM=V*W1/16.01846
660.      00      CALL HCVCP(N5,T,VLM,XML,DUM,CVL,CPL)
670.      00      CALL HCVCP(N5,T,VM1,XIV,D'11,CVV,CPV)
680.      00      CP=(1.-XQ)*CPL+XQ*CPV
690.      00      CV=(1.-XQ)*CVL+XQ*CVV
700.      00      ELSE
710.      00      VM=VM
720.      00      V=VM*13.01846/WM
730.      00      AM=VISCN(N3,T,XV)
740.      00      AK=VISCN(N4,T,XV)
750.      00      CALL HCVCP(N5,TK,VVM,XM,DUM,CV,CP)
760.      00      END IF
770.      00      1000 RETURN
          00      END
          00      @SUPERHEATED VAPOR

```

END ELT. ERRORS: NONE. TIME: 0.161 SEC. IMAGE COUNT: 78

@HDG,P ***** HTCCON ***** .L,0

***** HTCCON *****

@ELT, L DD.HTCCON

ELT 8R1 S7401C 07/21/84 15:55:09 (0)

FUNCTION HTCCON(T,P,RMAS,D)

10.

00 C

20. C**** PURPOSE:

30. C TO COMPUTE CONDENSATION HEAT TRANSFER COEFFICIENT
 40. C FOR NON-AZEOTROPIC MIXTURE FLOW INSIDE A TUBE
 50. C
 60. C
 70. C

C**** INPUT DATA:

80. C D - TUBE DIAMETER (FT)

90. C P - REFRIG. AVERAGE PRESSURE (PSIA)

100. C RMAS - REFRIG. MASS FLOW RATE (LBM/H)

110. C T - REFRIG. AVERAGE TEMPERATURE (F)

120. C XW - MIXTURE WEIGHT COMPOSITION

130. C XM - (FRACTION OF LESS VOLATILE COMPONENT)

140. C W1 - (FRACTION OF LESS VOLATILE COMPONENT)

150. C W2 - MOLECULAR WEIGHT OF MORE VOLATILE COMPONENT (G/MOL)

160. C W2 - MOLECULAR WEIGHT OF LESS VOLATILE COMPONENT (G/MOL)

170. C

180. C

190. C**** SUBPROGRAMS CALLED BY HTCCON:

200. C SUBTEM, DEWTEM, HCVCP, QLITY, SPHTC, VISCON, VOLITI

210. C

220. C COMMON/RDATA2/W1,W2,TC1,TC2

230. C COMMON/RDATA3/XW,XM,WM

240. C

250. C TK=(T+459.67)/1.8

260. C PA=P/14.6959

270. C CALL QLITY(TK,PA,XM,X,XV,XL)

280. C X1=X

290. C IF(X.LT.0.1)X1=0.1

300. C IF(X.GT.0.95)X1=0.95

310. C G=0.7853981*D*D

320. C G=RMAS/G

330. C

340. C CALL VOLITI(0,TK,PA,XL,VL)

350. C CALL HCVCP(5,TK,VL,XL,H,CV,CPL)

360. C WMOL=W1*(1.-XL)+W2*XL

370. C VL=16.01846*VL/WMOL

380. C XL=XL/(W1/W2*(1.-XL)+XL)

390. C VISL=VISCON(1,T,XL)

400. C CONL=VISCON(2,T,XL)

410. C CALL VOLITI(1,TK,PA,XV,VV)

420. C WMOL=W1*(1.-XV)+W2*XV

430. C VV=16.01846*VV/WMOL

440. C XV=XV/(W1/W2*(1.-XV)+XV)

450. C VISV=VISCON(3,T,XV)

460. C PRF=VISL*CPL/CONL

470. C RETP=G*(1.-X1)*D/VISL

480. C XTT=((1.-X1)/X1)*0.9*SQRT(VL/VV)*(VISL/VISV)**0.1

490. C F1=0.15*(1.+2.85*(TT**0.524)/XTT

500. C F2=0.707*PRF*SQRT(RETP)

510. C IF(RETP.GT.50. AND. RETP.LT.1125.) F2=F2*PRF+5.*ALOG(1.+PRF*(

520. C 80.09636*RETP**0.585-1.))

530. C IF(RETP.GE.1125.) F2=F2*PRF*5.*ALOG(1.+5.*PRF)+2.5*ALOG(0.0031*RETP

540. C 8**0.812)

550. C ALF=1.

560. C IF(F1.GT.1.) ALF=1.15

```

***** HTCCON *****
570. HTCCON=CONL*PRF*F1**ALF*RETP**0.9/(D*F2)
580. IF(X.GE.0.1)GOTO10
590. CALL DUBTEM(1,PA,XM,TK,XV)
600. CALL VOLIT1(0,TK,PA,XM,VL)
610. CALL HCVCF(5,TK,VL,XM,H,CV,CPL)
620. TFB=TK*1.8-459.67
630. VISL=VISCON(1,TFB,XW)
640. CONL=VISCON(2,TFB,XW)
650. HL=SPHTC(CPL,VISL,CONL,RMAS,D)
660. HTCCON=HL+10.*X*(HTCCON-HL)
670. GOTO20
680.
690. 10 CONTINUE
700. IF(X.LE.0.95)GOTO20
710. CALL DEWTEM(1,PA,XM,TK,XL)
720. CALL VOLIT1(1,TK,PA,XM,VV)
730. CALL HCVCF(5,TK,VV,XM,H,CV,CPV)
740. TFD=TK*1.8-459.67
750. VISV=VISCON(3,TFD,XW)
760. CONV=VISCON(4,TFD,XW)
770. HV=SPHTC(CPV,VISV,CONV,RMAS,D)
780. HTCCON=HTCCON-20.*(X-0.95)*(HTCCON-HV)
790. 20 RETURN
    END

```

END ELT. ERRORS: NONE. TIME: 0.125 SEC. IMAGE COUNT: 79

@HDG,P ***** HTCEV ***** .L,0

***** HTCEV *****

@ELT, L DD, HTCEV

ELT 8R1 S74Q1C 07/21/84 15:55:09 (0)

```

10. 00 FUNCTION HTCEV(TF, PSIA, RMS, DOI, XTT)
20. 00 C
30. 00 C**** PURPOSE:
40. 00 C    TO COMPUTE EVAPORATION HEAT TRANSFER COEFFICIENT
50. 00 C    FOR NON-AZEOTROPIC MIXTURE FLOW INSIDE A TUBE
60. 00 C
70. 00 C**** INPUT DATA:
80. 00 C    DD1 - TUBE DIAMETER (FT)
90. 00 C    PSIA - REFRIG. AVERAGE PRESSURE (PSIA)
100. 00 C    RMS - REFRIG. MASS FLOW RATE (LBM/H)
110. 00 C    TF - REFRIG. AVERAGE TEMPERATURE (F)
120. 00 C    XM - MIXTURE MOLAR COMPOSITION
130. 00 C    (FRACTION OF LESS VOLATILE COMPONENT)
140. 00 C    W1 - MOLECULAR WEIGHT OF MORE VOLATILE COMPONENT (G/MOL)
150. 00 C    W2 - MOLECULAR WEIGHT OF LESS VOLATILE COMPONENT (G/MOL)
160. 00 C
170. 00 C**** OUTPUT DATA:
180. 00 C    HTCEV - EVAPORATIVE HEAT TRANSFER COEFF. (BTU/(H*F*FT**2))
190. 00 C    XTT - LOCKHART-MARTINELLI PRESSURE DROP PARAMETER
200. 00 C
210. 00 C**** SUBPROGRAMS CALLED BY HTCEV:
220. 00 C    HCVCP, QLITY, SPHTC, VISCON, VOLITI
230. 00 C
240. 00 C    COMMON/RDATA2/W1, W2, TC1, TC2
250. 00 C    COMMON/RDATA3/XW, XM, WM
260. 00 C
270. 00 C    PA=PSIA/14.6959
280. 00 C    TK=(TF+459.67)/1.8
290. 00 C    CALL QLITY(TK, PA, XM, XG, XV, XL)
300. 00 C
310. 00 C    IF(XG.EQ.0..OR.XG.EQ.1.) THEN
320. 00 C      PRINT 10, XG
330. 00 C    10 FORMAT(' ERROR IN CALLING HTCEV, XG=', F4.1)
340. 00 C    HTCEV=200.
350. 00 C    RETURN
360. 00 C    END IF
370. 00 C
380. 00 C    XLW=XL/(W1/W2*(1.-XL)+XL)
390. 00 C    XVW=XV/(W1/W2*(1.-XV)+XV)
400. 00 C    VISL=VISCON(1, TF, XLW)
410. 00 C    CONL=VISCON(2, TF, XLW)
420. 00 C    VISV=VISCON(3, TF, XVW)
430. 00 C    CALL VOLITI(0, TK, PA, XL, VL)
440. 00 C    CALL HCVCP(5, TK, VL, XL, HL, CV, CPL)
450. 00 C    WML=W1*(1.-XL)+W2*XL
460. 00 C    WMV=W1*(1.-XV)+W2*XV
470. 00 C    VLW=VL*16.01846/WML
480. 00 C    VWV=VW*16.01846/WVW
490. 00 C    XTT=((1.-XG)/XG)**.9*(VLW/VWV)**.5*(VISL/VISV)**.1
500. 00 C    HL=SPHTC(CPL, VISL, CONL, RMS, DOI)
510. 00 C    HTCEV=2.12*HL*(1./XTT)**0.865
520. 00 C    HTCEV=3.23*HL*XTT*(-0.3)
530. 00 C    RETURN
540. 00 C    END

```


***** HXCODE *****

@ELT, L DD, HXCODE

ELT 8R1 S74Q1C 07/21/84 15:55:09 (0)

SUBROUTINE HXCODE(110)

```

10. 00 C
20. 00 C
30. 00 C
40. 00 C
50. 00 C
60. 00 C
70. 00 C
80. 00 C
90. 00 C
100. 00 C
110. 00 C
120. 00 C
130. 00 C
140. 00 C
150. 00 C
160. 00 C
170. 00 C
180. 00 C
190. 00 C
200. 00 C
210. 00 C
220. 00 C
230. 00 C
240. 00 C
250. 00 C
260. 00 C
270. 00 C
280. 00 C
290. 00 C
300. 00 C
310. 00 C
320. 00 C
330. 00 C
340. 00 C
350. 00 C
360. 00 C
370. 00 C
380. 00 C
390. 00 C
400. 00 C
410. 00 C
420. 00 C
430. 00 C
440. 00 C
450. 00 C
460. 00 C
470. 00 C
480. 00 C
490. 00 C
500. 00 C
510. 00 C
520. 00 C
530. 00 C
540. 00 C
550. 00 C
560. 00 C

C**** THIS PROGRAM DETERMINES REFRIGERANT & AIR FLOW
C**** DISTRIBUTION FOR HEAT EXCHANGER TUBES

C**** INPUT DATA:
IFROM(110,J) - NUMBER OF TUBE TUBE J RECEIVES REFRIG. FROM
              WHEN COIL WORKS AS EVAPORATOR (-)
110          = 1 FOR INDOOR COIL (-)
              = 2 FOR OUTDOOR COIL (-)
NDP(110)     - NUMBER OF COIL TUBE ROW DEPTHS (-)
NSEC(110)    - NUMBER OF REPEATING SECTIONS (-)
NTUB(110,N)  - NUMBER OF TUBES IN ROW N FOR EACH SECTION
              OF TUBE J (-)

C**** OUTPUT DATA:
FLOW(110,J)  - FRACTION OF COIL TOTAL REFRIG. MASS FLOW
              PASSING THROUGH TUBE J (-)
IDEPH(110,J) - DEPTH ROW OF A TUBE J (-)
IMER(110)    - NUMBER OF MERGING TUBES (-)
IST(110)     - NUMBER OF TUBES REFRIG. FLOWS INTO COIL
              WORKING AS CONDENSER (-)
ISTART(110,L) - NUMBER OF TUBE REFRIG. FLOWS INTO COIL
              WORKING AS CONDENSER, FOUND AS L SUCH TUBE (-)
JFROM(110,J) - NUMBER OF TUBE POSITIONED AIR UPSTREAM
              TUBE J WHEN COIL WORKS AS EVAPORATOR.
              NOTE THAT TUBE J CAN FEED UP TO 3 TUBES
              (N CAN BE 1, 2 AND 3) (-)
KSTART(110,N) - NUMBER OF TUBE REFRIGERANT FLOWS INTO COIL
              WORKING AS EVAPORATOR (-)
KST(110)      - NUMBER OF TUBES REFRIGERANT FLOWS INTO COIL
              WORKING AS EVAPORATOR
MERGE(110,K,1) - NUMBER OF TUBE FOUND AS K MERGING TUBE (-)
MERGE(110,K,2) - NUMBER OF TUBES MERGING INTO TUBE K (-)
NTPS(110)     - NUMBER OF TUBES PER SECTION (-)

COMMON/HPHX/NDP(2),NROW(2),DI(2),DO(2),DT(2),RPGH(2),DPCH(2),
& WIDTH(2),FPCH(2),FTK(2),FMK(2),TTK(2),AMAS(2),ANGLE(2),
& CONST(2),CPOW(2),NTUB(2,5),JFROM(2,130),NSEC(2),NTPS(2),
COMMON/MERGE/MERGE(2,20,2),IMER(2),ISTART(2,20),IST(2),
& IDEPH(2,130),FLOW(2,130),JFROM(2,130),KFEED(2,130,3),
& KSTART(2,20),KST(2)
DIMENSION IMC(20),IDID(130)

C**** FIND NUMBER TUBES PER SECTION
NTPS(110)=0
DO 1 I=1,5
1 NTPS(110)=NTPS(110)+NTUB(110,I)

DO 2 I=1,NTPS(110)
2 FLOW(110,I)=0.
DO 4 I=1,20
4 MERGE(110,I,1)=0

```

```

***** HXCODE *****
570. C**** FIND TUBES REFRIGERANT ENTERS CONDENSER
580. C**** FIND TUBES REFRIGERANT MERGES
590. IS=0
600. IM=0
610. DO 10 J=1, NTPS(110)
620. NM=0
630. DO 6 I=1, NTPS(110)
640. IF(I=FROM(110,1).NE.J)GOTO 6
650. NM=NM+1
660. 6 CONTINUE
670. IF(NM.EQ.0)GOTO 8
680. IF(NM.EQ.1)GOTO 10
690. IM=IM+1
700. MERGE(110,IM,1)=J
710. MERGE(110,IM,2)=NM
720. GOTO 10
730. 8 IS=IS+1
740. ISTART(110,IS)=J
750. 10 CONTINUE
760. IST(110)=IS
770. IMER(110)=IM
780.
790. C
800. C**** FIND REFRIGERANT FLOW DISTRIBUTION
810. DO 12 L=1,IM
820. IMC(L)=MERGE(110,L,2)
830. 12 CONTINUE
840. SECFW=1./FLOAT(NSECT(110))
850. STFLOW=SECFW/FLOAT(IS)
860. DO 20 IB=1,IS
870. I=ISTART(110,IB)
880. FLOW(110,I)=STFLOW
890. J=IFROM(110,I)
900. IF(J.EQ.0)GOTO 20
910. DO 16 IC=1,IM
920. IF(MERGE(110,IC,1).EQ.J)GOTO 18
930. 16 CONTINUE
940. FLOW(110,J)=FLOW(110,I)
950. I=J
960. GOTO 14
970. 18 FLOW(110,J)=FLOW(110,J)+FLOW(110,I)
980. IMC(IC)=IMC(IC)+1
990. IF(IMC(IC).NE.0)GOTO 20
1000. I=J
1010. GOTO 14
1020. 20 CONTINUE
1030. C
1040. C**** FIND DEPTH ROW FOR EACH TUBE
1050. NNDEP=NDP(110)
1060. ILAST=0
1070. DO 22 J=1, NNDEP
1080. IFIRST=ILAST+1
1090. ILAST=IL/ST+NTUR(110,J)
1100. DO 22 I=IFIRST, ILAST
1110. IDPTH(110,I)=J
1120. 22 CONTINUE
1130. C
1140. C**** FIND AIR FLOW DISTRIBUTION
      DO 40 I=1, NTUB(110,1)

```

```

***** HXCODE *****
1150. 00 40 JFROM(110,1)=999
1160. 00 N=1-NTUB(110,1)
1170. 00 DO 44 I=N,NTPS(110)
1180. 00 ICT=IDEPTH(110,1)
1190. 00 J=1-NTUB(110,ICT-1)
1200. 00 ICTJ=IDEPTH(110,J)
1210. 00 IF(ICTJ.EQ.ICT)THEN
1220. 00 IR=ICT-1
1230. 00 J=0
1240. 00 DO 42 K=1,IB
1250. 00 J=J+NTUB(110,K)
1260. 00 42 END IF
1270. 00 JFROM(110,1)=J
1280. 00 44 CONTINUE
1290. 00 C
1300. 00 C**** FIND REFRIG. FLOW PATH IN EVAPORATOR COIL
1310. 00 DO 50 I=1,NTPS(110)
1320. 00 DO 48 IK=1,IMER(110)
1330. 00 IF(1.NE.MERGE(110,IK,1))GOTO 48
1340. 00 IDID(1)=MERGE(110,IK,2)
1350. 00 GOTO 50
1360. 00 48 CONTINUE
1370. 00 IDID(1)=1
1380. 00 50 CONTINUE
1390. 00 C
1400. 00 KS=0
1410. 00 DO 60 IS=1,IST(110)
1420. 00 I=ISTART(110,IS)
1430. 00 KFEED(110,1,1)=-1
1440. 00 54 J=1
1450. 00 I=IFROM(110,J)
1460. 00 IF(1.EQ.0)THEN
1470. 00 KS=KS+1
1480. 00 KSTART(110,KS)=J
1490. 00 KST(110)=KS
1500. 00 GOTO 60
1510. 00 END IF
1520. 00 IF(IDID(1).EQ.0)GOTO 60
1530. 00 N=IDID(1)
1540. 00 KFEED(110,1,N)=J
1550. 00 IDID(1)=IDID(1)-1
1560. 00 GOTO 54
1570. 00 60 CONTINUE
1580. 00 RETURN
1590. 00 END

```

END ELT. ERRORS: NONE. TIME: 0.225 SEC. IMAGE COUNT: 159

CHDG,P ***** MVAL4 ***** .L,0

***** MVAL4 *****

QELT, L DD, MVAL4

ELT 8R1 S74Q1C 07/21/84 15:55:10 (O)

SUBROUTINE MVAL4(XW, RMASS, V2, AM2, AK2, CP2,
 @ V7, AM7, AK7, CP7, V11, AM11, AK11, CP11, V71, AM71, AK71, CP71)

C *****

PURPOSE:
 TO SIMULATE PERFORMANCE OF 4-WAY VALVE
 (BINARY MIXTURE REFRIGERANT)

C *****

INPUT DATA:
 RMASS - REFRIG. MASS FLOW RATE (LB/II)
 XW - COMPOSITION (WEIGHT FRACTION OF LESS VOLATILE COMPONENT)
 * REFRIG. PARAMETERS & PROPERTIES:
 AK2, AK7 - THERMAL CONDUCTIVITY (BTU/H*F*FT)
 AM2, AM7 - DYNAMIC VISCOSITY (LB/H*FT)
 CP2, CP7 - SPECIFIC HEAT AT CONST. PRESSURE (BTU/LB*F)
 H2, H7 - ENTHALPY (BTU/LB)
 P2, P7 - PRESSURE (PSIA)
 T2, T7 - TEMPERATURE (F)
 V2, V7 - SPEC. VOLUME (FT**3/LB)
 WHERE IN ABOVE SYMBOLS NUMBERS DENOTE LOCATION:
 2 4-WAY VALVE LOW PRESSURE INLET
 7 4-WAY VALVE HIGH PRESSURE INLET
 * 4-WAY VALVE PARAMETERS:
 CPDR - PARAMETER FOR PRESSURE DROP (LBF*H**2/LB*IN**2*FT**3)
 CQ - PARAMETER FOR HEAT TRANSFER (FT**2)

C *****

OUTPUT DATA:
 * REFRIG. PARAMETERS & PROPERTIES:
 AK11, AK71 - THERMAL CONDUCTIVITY (BTU/H*F*FT)
 AM11, AM71 - DYNAMIC VISCOSITY (LB/H*FT)
 CP11, CP71 - SPECIFIC HEAT AT CONST. PRESSURE (BTU/LB*F)
 H11, H71 - ENTHALPY (BTU/LB)
 P11, P71 - PRESSURE (PSIA)
 T11, T71 - TEMPERATURE (F)
 X11 - QUALITY (-)
 V11, V71 - SPECIFIC VOLUME (LB/FT**3)
 WHERE IN ABOVE SYMBOLS NUMBERS DENOTE LOCATION:
 11 4-WAY VALVE LOW PRESSURE INLET
 71 4-WAY VALVE HIGH PRESSURE OUTLET

C *****

SUBPROGRAMS CALLED BY MVAL4:
 HPPROP

COMMON/RDATA2/W1, W2, TC1, TC2
 COMMON/COND2/P11, T11, H11, X11, S11
 COMMON/COND3/P2, T2, H2, S2
 COMMON/COND3/P7, T7, H7, S7
 COMMON/COND9/P71, T71, H71, S71
 COMMON/WAY4/CQ, CPDR

C

NO=0

N1=1

ACCUR=0.0003

PDRF=CPDR*RMASS*RMASS

XM=XW/(W2/W1*(1.-XW)+XW)

C

C

C

***** MVAL4 *****

```

580. WM=(1-XM)*W1+XM*W2
590. TK2=(T2+459.67)/1.8
600. PA2=P2/14.6959
610. TK7=(T7+459.67)/1.8
620. PA7=P7/14.6959
630. RMASS8=RMASS**0.8
640. T11=T2
650. V11=V2
660. AM11=AM2
670. AK11=AK2
680. CP11=CP2
690. T71=T7
700. V71=V7
710. AM71=AM7
720. AK71=AK7
730. CP71=CP7

```

C

```

D0 100 I=1,10
VS=(V11+V2)/2.
AMS=(AM11+AM2)/2.
AKS=(AK11+AK2)/2.
CPS=(CP11+CP2)/2.
VH=(V7+V71)/2.
AMH=(AM7+AM71)/2.
AKH=(AK71+AK7)/2.
CPH=(CP7+CP71)/2.
P11=P2+PDRF*VS
PA11=P11/14.6959
P71=P7-PDRF*VH
PA71=P71/14.6959
HCS=RMASS8*AKS**0.667*CPS**0.333/AMS**0.467
HCH=RMASS8*AKH**0.667*CP11**0.333/AM11**0.467
DTA=T71-T11
DTB=T7-T2
IF(ABS(DTA-DTB).LT.0.01)GOTO20
TF=(DTA-DTB)/(ALOG(DTA/DTB))
GOTO 21

```

```

20 TF=T7-T2
21 Q=CQ*TF/(1./HCS+1./HCH)
H71=H7-Q/RMASS
T71=T7-Q/(RMASS*CPH)
TK71=(T71+459.67)/1.8
CALL HPPROP(N1,H71,PA71,X1,ACCUR,TK71,XQ,XL,XV,VL71,VW71,
@ VM71,V71,CP71,CV,AM71,AK71)
T71=TK71*1.8-459.67

```

@HIGH PRESSURE OUTLET

@LOW PRESSURE OUTLET

```

H11=H2-Q/RMASS
T11=T2-Q/(RMASS*CPS)
TK11=(T11+459.67)/1.8
CALL HPPROP(N1,H11,PA11,XM,ACCUR,TK11,X11,XL,XV,VL11,VM11,
@ VM11,V11,CP11,CV,AM11,AK11)
T11=TK11*1.8-459.67
IF(I.EQ.1)GOTO51
IF(ABS(H71-H71A).LT.0.001)GOTO101
H71D=H71A
51 H71A=H71
100 CONTINUE
WRITE(6,600)H71A,H71B

```

C

1320. 00 00 100 WRITE(6,600)H71A,H71B

```
***** MVAL4 *****
1330.      00      600 FORMAT(' MVAL4 DOES NOT CONVERGE, H71A,H71B=',2(1PE16.6))
1340.      00      101 CONTINUE
1350.      00      RETURN
1360.      00      END

END ELT.  ERRORS: NONE.  TIME:  0.161 SEC.  IMAGE COUNT: 118

@HDG,P  ***** OVLWET ***** .L,0
```

DATE 072184

***** OVLWET *****

@ELT,L DD.OVLWET

ELT 8R1 S74Q1C 07/21/84 15:55:10 (0)

SUBROUTINE OVLWET(AO,API,APM,HI,HD,HP,HL,HQ,UAO,UPO,UWO)

C

C****

PURPOSE:

TO COMPUTE OVERALL HEAT TRANSFER COEFFICIENT

FOR A WET FINNED TUBE

C

C****

INPUT DATA:

AO - TOTAL OUTSIDE SURFACE AREA (FT**2)

API - TUBE INSIDE SURFACE AREA (FT**2)

APM - SURFACE AREA BASED ON TUBE MEAN DIAMETER (FT**2)

HD - TUBE INSIDE SURFACE DEPOSIT HEAT TRANSFER COEFF. (BTU/H*F*FT**2)

HI - TUBE INSIDE SURFACE HEAT TRANSFER COEFF. (BTU/H*F*FT**2)

HL - HEAT TRANSFER COEFF. FOR WATER (FROST) LAYER (BTU/H*F*FT**2)

HP - TUBE WALL HEAT TRANSFER COEFFICIENT (BTU/H*F*FT**2)

HQ - AIR-SIDE HEAT TRANSFER COEFF. FOR WET FINNED TUBE (BTU/H*F*FT**2)

C****

OUTPUT DATA:

UAO - OVERALL HEAT TRANSFER COEFF. FOR WET FINNED TUBE (BTU/H*F*FT**2)

UPO - HEAT CONDUCTANCE FROM REFRIGERANT TO TUBE SURFACE (BTU/H*F)

UWO - HEAT CONDUCTANCE FROM REFRIGERANT TO WATER (FROST) SURFACE (BTU/H*F)

C

U=AO/(API*HI)+AO/(API*HD)*AQ/(APM*HP)

UPO=AQ/U

U=U+1./HL

UWO=AQ/U

U=U+1./HQ

UAO=AQ/U

RETURN

END

END ELT. ERRORS: NONE. TIME: 0.073 SEC. IMAGE COUNT: 29

@HDG,P ***** PFLASH ***** .L,0

***** PFLASH *****

```

0ELT,L DD,PFLASH
ELT 8R1 S74QIC 07/21/84 15:55:10 (0)
10. 00 SUBROUTINE PFLASH(XM,H1,GN,TIN,TFLASH,PFLA,VL)
20. 00 C
30. 00 C**** PURPOSE:
40. 00 TO CALCULATE FLASHING PRESSURE & TEMPERATURE
50. 00 FOR GIVEN CONSTANT ENERGY FLOW OF NON-AZEOTROPIC MIXTURE
60. 00 C
70. 00 C**** INPUT DATA:
80. 00 GN = G*G/(64.4*778.104) (BTU*LB/FT**6)
90. 00 WHERE G - REFRIG. MASS FLUX (LB/(SEC*FT**2))
100. 00 H1 - REFRIG TOTAL ENTHALPY (BTU/LB)
110. 00 TIN - PFLASHING TEMPERATURE ( GUESS) (F)
120. 00 C
130. 00 C**** OUTPUT DATA:
140. 00 PFLASH - FLASHING PRESSURE (PSIA)
150. 00 TFLASH - FLASHING TEMPERATURE (F)
160. 00 VL - SPEC. VOLUME OF LIQ. REFRIG. (FT**3/LB)
170. 00 XM - MOLAR COMPOSITION (FRACTION OF LESS VOLATILE COMPONENT)
180. 00 C
190. 00 C**** SUBPROGRAMS CALLED BY PFLASH:
200. 00 BUEPRE,HCVCP,VOLIT1
210. 00 C
220. 00 COMMON/RDATA2/W1,W2,TC1,TC2
230. 00 DATA NO,N1/O,1/
240. 00 WM=W1*(1-XM)+W2*XM
250. 00 TKIN=(TIN-459.67)/1.8
260. 00 DO 3 I=1,10
270. 00 TFIN=TKIN*1.8-459.67
280. 00 CALL BUEPRE(NO,TKIN,XM,PAFL,XV)
290. 00 CALL VOLIT1(NO,TKIN,PAFL,XM,VML)
300. 00 VL=VML*16.01846/WM
310. 00 PAFL3=PAFL+3.
320. 00 CALL VOLIT1(NO,TKIN,PAFL3,XM,VML3)
330. 00 VL3=VML3*16.01846/WM
340. 00 PFL=PAFL*14.6959
350. 00 PFL3=PAFL3*14.6959
360. 00 HIN=H1-GN*VL*VL
370. 00 CALL HCVCP(N1,TKIN,VML,XM,HF,CV,CP)
380. 00 HDIF=HIN-HF
390. 00 IF(ABS(HDIF).LT..0003)GOTO 5
400. 00 IF(1.EQ.1)GOTO 1
410. 00 TK=TKIN-HDIF*(TKINP-TKIN)/(HDIFP-HDIF)
420. 00 GOTO 2
430. 00 1 TK=TKIN+SIGN(1.,HDIF)
440. 00 2 IF(TK.GT.TC1)TK=TC1-2.
450. 00 TKINP=TKIN
460. 00 TKIN=TK
470. 00 HDIFP=HDIF
480. 00 3 CONTINUE
490. 00 PRINT 4,HDIF
500. 00 4 FORMAT(' PFLASH DOES NOT CONVERGE, HDIF=',1PE15.5)
510. 00 5 CONTINUE
520. 00 PFLA=PAFL*14.6959
530. 00 TFLASH=TKIN*1.8-459.67
540. 00 RETURN
550. 00 END

```


***** PIPE *****

DATE 072184

0ELT, L 00. PIPE

ELT 8R1 S74Q1C 07/21/04 15:55:11 (0)

SURFOUTLINE PIPE (IBACK, XW, TAIR, RMASS, T1, P1, V1, H1, AM1, AK1, CP1,
& T2, P2, H2, XQ2, PL, PD1, PK1, PD2, PK2, PD3, PK3, PD4)

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TO COMPUTE PARAMETERS OF SUPERHEATED OR WET REFRIG. VAPOR
(BINARY MIXTURE) FLOWING THROUGH A PIPE FOR OUTLET
OR INLET CONDITIONS KNOWN

***** ASSUMPTIONS:

SINGLE PHASE HEAT TRANSFER INSIDE PIPE
FREE CONVECTION OUTSIDE PIPE

***** INPUT DATA:

AK1 - REFRIG. THERMAL CONDUCTIVITY AT PIPE INLET (OUTLET)
(BTU/IN*F*FT)

AM1 - REFRIG. DYNAMIC VISCOSITY AT PIPE INLET (OUTLET) (LBM/H*F*FT)

CP1 - REFRIG. SPEC. HEAT AT PIPE INLET (OUTLET) (BTU/LB*F)

H1 - REFRIG. ENTHALPY AT PIPE INLET (OUTLET) (BTU/LBM)

IBACK = 0 IF INLET REFRIG. STATE KNOWN, OUTLET IS CALCULATED
= 1 IF OUTLET REFRIG. STATE KNOWN, INLET IS CALCULATED

PD1 - PIPE INSIDE DIAMETER (FT)

PD2 - PIPE OUTSIDE DIAMETER (FT)

PD3 - OUTER DIAMETER OF INSIDE INSULATION (FT)

PD4 - OUTER DIAMETER OF OUTSIDE INSULATION (FT)

PK1 - THERMAL CONDUCTIVITY OF PIPE MATERIAL (BTU/H*F*FT)

PK2 - THERMAL CONDUCTIVITY OF INSIDE INSULATION MATERIAL
(BTU/H*F*FT)

PK3 - THERMAL CONDUCTIVITY OF OUTSIDE INSULATION MATERIAL
(BTU/IN*F*FT)

PL - PIPE LENGTH (FT)

P1 - REFRIG. PRESSURE AT PIPE INLET (OUTLET) (PSIA)

RMASS - REFRIG. MASS FLOW RATE (LBM/H)

TAIR - AMBIENT AIR TEMPERATURE (F)

T1 - REFRIG. TEMPERATURE AT PIPE INLET (OUTLET) (F)

V1 - REFRIG. SPEC. VOLUME AT PIPE INLET (OUTLET) (FT**3/LBM)

XW - MIXTURE COMPOSITION
(WEIGHT FRACTION OF LESS VOLATILE COMPONENT)

***** OUTPUT DATA:

H2 - REFRIG. ENTHALPY AT PIPE OUTLET (INLET) (BTU/LBM)

P2 - REFRIG. PRESSURE AT PIPE OUTLET (INLET) (PSIA)

T2 - REFRIG. TEMPERATURE AT PIPE OUTLET (INLET) (F)

XQ2 - REFRIG. QUALITY AT PIPE OUTLET (INLET) (-)

***** SUBPROGRAMS CALLED BY PIPE:

ESVOL, HCVCP, HPTM, GLITY, SPIDF1, SPHTC, VISCON

COMMON/RDATA2/W1, W2, TC1, TC2

DATA NO, N1, N2, N3, N4, N5/0, 1, 2, 3, 4, 5/

ACCR=0.0005

XM=XW/(W2/W1*(1.-XW)+XW)

YM=(1.-XM)*W1+XM*W2

AREA=3.1415927*PD1*PL

RUP=PD1*ALOG(PD2/PD1)/(2.*PK1)

PD0=PD2

```

570. 00
580. 00
590. 00
600. 00
610. 00
620. 00
630. 00
640. 00
650. 00
660. 00
690. 00
700. 00
710. 00
720. 00
730. 00
740. 00
750. 00
760. 00
770. 00
780. 00
790. 00
800. 00
810. 00
820. 00
830. 00
840. 00
850. 00
860. 00
870. 00
880. 00
890. 00
900. 00
910. 00
920. 00
930. 00
940. 00
950. 00
960. 00
970. 00
980. 00
990. 00
1000. 00
1010. 00
1020. 00
1030. 00
1040. 00
1050. 00
1060. 00
1070. 00
1080. 00
1090. 00
1100. 00
1110. 00
1120. 00
1130. 00
1140. 00
1150. 00
1160. 00

IF(PK2.LT.1.E-08)GOTO 5
RUP=RUP+PD1*ALOG(PD3/PD2)/(2.*PK2)
PD0=PD3
IF(PK3.LT.1.E-08)GOTO 5
RUF=RUP+PD1*ALOG(PD4/PD3)/(2.*PK3)
PD0=PD4
5 CONTINUE
TK1=(T1+459.67)/1.8
PA1=P1/14.6953
CALL QLITY(TK1,PA1,XM,XQ1,XV1,XL1)
T2=T1
TK2=TK1
PA2=PA1
CP2=CP1
AM2=AM1
AK2=AK1
V2=V1
TP=T1
DO 50 I=1,10
T=0.5*(T1+T2)
CP=0.5*(CP1+CP2)
AM=0.5*(AM1+AM2)
AK=0.5*(AK1+AK2)
V=0.5*(V1+V2)
C**** CALC. PRESSURE DROP
IF(XQ1.NE.1..AND.XQ2.NE.1.)THEN
XL=0.5*(XL1+XL2)
XV=0.5*(XV1+XV2)
AML=VISCN(N1,T,XL)
AMV=VISCN(N3,T,XV)
WL=(1.-XL)*W1+XL*W2
PA=0.5*(PA1+PA2)
TK=0.5*(TK1+TK2)
VL=ESVOL(N0,TK,PA,XL)
VL=VL*16.01846/WL
WV=(1.-XV)*W1+XV*W2
VV=FSVOL(N1,TK,PA,XV)
VV=VV*16.01846/WV
XQ=0.5*(XQ1+XQ2)
XTT=(AML/AMV)**.1*(VL/VV)**.5*((1.-XQ)/XQ)**.9
FI=1.+2.35*XTT**.523
RMS=XQ*RMAS
PDROP=FI*SPHDP1(RMS,PL,PD1,VV,AMV)
ELSE
PDROP=SPHDP1(RMASS,PL,PD1,V,AM)
END IF
P2=P1-PDROP
IF(1BACK.EQ.1)P2=P1+PDROP
C**** CALC. HEAT TRANSFER, SINGLE PHASE FLOW ASSUMED
UI=SP'ITC(CP,AM,AK,RMASS,PD1)
UO=0.27*(ABS(TAIR-TP)/PD0)**0.25
RUI=1./UI
RUO=PD1/(PD0*UO)
UOA=1./((RUI+RUP+RUO)
UOI=1./((RUI+RUP)
Q=UOA*AREA*(TAIR-T)
TP=T+Q/(UOI*AREA)
H2=H1+Q/RMASS

```

@TWO PHASE PRESSURE DROP

@SINGLE-PHASE PRESSURE DROP

```

***** PIPE *****
1170. 00
1180. 00
1190. 00
1200. 00
1210. 00
1220. 00
1230. 00
1240. 00
1250. 00
1260. 00
1270. 00
1280. 00
1290. 00
1300. 00
1310. 00
1320. 00
1330. 00
1340. 00
1350. 00
1360. 00
1370. 00
1380. 00
1390. 00
1400. 00
1410. 00
1420. 00
1430. 00
1440. 00
1450. 00
1460. 00
1470. 00
1471. 00
1472. 00
1480. 00
1490. 00
1500. 00

PA2=P2/14.6959
CALL HPIN(N1,H2,PA2,XM,ACCUR,TK2,XQ2,XL2,XV2,VL2,VV2)
T2=TK2*1.8-459.67
IF(XQ2.NE.1.) THEN
  WL=(1.-XL2)*V1+XL2*W2
  WV=(1.-XV2)*V1+XV2*W2
  VL2=VL2*16.01846/WL
  VV2=VV2*16.01846/WV
  V2=(1.-XQ2)*VL2+XQ2*VV2
  XL2W=XL2/(W1/W2*(1.-XL2)+XL2)
  XV2W=XV2/(W1/W2*(1.-XV2)+XV2)
  AM2=(1.-XQ2)*V1SCON(N1,T2,XL2W)
  AM2=AM2+XQ2*V1SCON(N3,T2,XV2W)
  AK2=(1.-XQ2)*V1SCON(N2,T2,XL2W)
  AK2=AK2+XQ2*V1SCON(N4,T2,XV2W)
  VV2=V2*WM/16.01846
  CALL HCVCP(N5,TK2,VV2,XV2,DUM,CV,CP2)
  CP2=XQ2*CP2+(1.-XQ2)*V1SCON(N5,T2,XL2W)
ELSE
  V2=VV2*16.01846/WM
  AM2=V1SCON(N3,T2,XM)
  AK2=V1SCON(N4,T2,XM)
  CALL HCVCP(N5,TK2,VV2,XM,DUM,CV,CP2)
END IF
IF(1.EQ.1)GOTO 40
IF(ABS(H2-H2A).LT.0.003.AND.ABS(P2-F2A).LT.0.005)GOTO 60
40 H2A=H2
P2A=P2
50 CONTINUE
WRITE(5,600)H2,H2A
600 FORMAT(' PIPE DOES NOT CONVERGE, H2, H2A=',2(IPE16.6))
60 CONTINUE
RETURN
END

```

END ELT. ERRORS: NONE. TIME: 0.212 SEC. IMAGE COUNT: 148

@HDG,P ***** PXQIN ***** .L,0

@ELT,L DD,PXQIN

ELT 8R1 S74Q1C 07/21/84 15:55:11 (0)

SUBROUTINE PXQIN(X,P,XQ,T,XL,XV)

10. 00 C

20. 00 C

30. 00 C

40. 00 C

50. 00 C

60. 00 C

70. 00 C

80. 00 C

90. 00 C

100. 00 C

110. 00 C

120. 00 C

130. 00 C

140. 00 C

150. 00 C

160. 00 C

170. 00 C

180. 00 C

190. 00 C

200. 00 C

210. 00 C

220. 00 C

230. 00 C

240. 00 C

250. 00 C

260. 00 C

270. 00 C

280. 00 C

290. 00 C

300. 00 C

310. 00 C

320. 00 C

330. 00 C

340. 00 C

350. 00 C

360. 00 C

370. 00 C

380. 00 C

390. 00 C

400. 00 C

410. 00 C

420. 00 C

430. 00 C

440. 00 C

450. 00 C

460. 00 C

470. 00 C

480. 00 C

490. 00 C

500. 00 C

510. 00 C

520. 00 C

530. 00 C

540. 00 C

550. 00 C

560. 00 C

C**** PURPOSE:

TO CALC. TEMPERATURE OF NON-AZOTROPIC MIXTURE
FORM GIVEN PRESSURE, COMPOSITION AND QUALITY

C**** INPUT:

P - PRESSURE (STD ATM)

X - COMPOSITION (MOLAR FRACTION OF LESS VOLATILE COMPONENT)

XQ - QUALITY (-)

C**** OUTPUT:

T - TEMPERATURE OF MIXTURE (K)

XL - COMPOSITION OF SAT. LIQUID AT TEMP. TK AND PRESSURE PA
(MOLAR FRACTION OF LESS VOLATILE COMPONENT)XV - COMPOSITION OF SAT. VAPOR AT TEMP. TK AND PRESSURE PA
(MOLAR FRACTION OF LESS VOLATILE COMPONENT)

C**** SUBPROGRAMS CALLED BY PXQIN:

DBURTE, DDEWTE, DQLITY, EFUBTE, QLITY

REAL*8 TDD, PDD, XQDD, XLDD, XVDD

DATA SLOPE/0., PLAST/0. /

IF(XQ.LE.0..OR.XQ.GE.1.) THEN

PRINT 666,XQ

666 FORMAT(' ERROR IN CALLING PXQIN, XQ=',1PE16.6)

RETURN

END IF

IF(ABS(P-PLAST).GT.1.E-4) GOTO 10

IF(ABS(X-XLAST).GT.1.E-3) GOTO 10

IF(ABS(XQ-XQLAST).GT.1.E-4) GOTO 10

T=TLAST

XL=XLLAST

XV=XVLAST

RETURN

10 XQDIF1=0.

DT=1.

T=ERUBTE(P,X)-XQ*SLOPE

DO 50 I=1,20

IF(XQ.LT..01.OR.XQ.GT..99) THEN

TDD=T

PDD=P

CALL DQLITY(TDD,PDD,X,XQDD,XVDD,XLDD)

XQN=XQDD

XV=XVDD

XL=XLDD

ELSE

CALL QLITY(T,P,X,XQN,XV,XL)

END IF

IF(XQN.EQ.0..AND.XQDIF1.LT.0.) THEN

***** PXQIN *****

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570. 00 PDD=P
580. 00 TDD=T
590. 00 CALL DBUBTE(1,PDD,X,TDD,XVDD)
600. 00 T=TDD
610. 00 XV=XVDD
620. 00
630. 00 END IF
640. 00 IF(XQN.EQ.1..AND.XQDIF1.GT.0.)THEN
650. 00 PDD=P
660. 00 TDD=T
670. 00 CALL DDEWTE(1,PDD,X,TDD,XLDD)
680. 00 XL=XLDD
690. 00 END IF
700. 00 T2=T
710. 00 XQDIF2=XQ-XQN
720. 00 IF(ABS(DT).LT.0.005.AND.ABS(XQDIF2).LT..0001)GOTO 100
730. 00 IF(1.NE.1)GOTO 20
740. 00 T1=T2
750. 00 XQDIF1=XQDIF2
760. 00 IF(SLOPE.NE.0.)THEN
770. 00 IF(XQN.EQ.1..OR.XQN.EQ.0.)GOTO 15
780. 00 DT=XQDIF2*SLOPE
790. 00 IF(ABS(DT).GT.5.)DT=SIGN(5.,DT)
800. 00 T=T2-DT
810. 00 GOTO 50
820. 00 END IF
830. 00 15 T=T2+5.
840. 00 IF(XQDIF2.LT.0.)T=T2-5.
850. 00 GOTO 50
860. 00
870. 00 C
880. 00 20 IF(XQDIF1.EQ.XQDIF2)GOTO 15
890. 00 SLOPE=(T2-T1)/(XQDIF2-XQDIF1)
900. 00 T1=T2
910. 00 XQDIF1=XQDIF2
920. 00 DT=XQDIF1*SLOPE
930. 00 IF(ABS(DT).GT.10.)DT=SIGN(10.,DT)
940. 00 T=T1-DT
950. 00 50 CONTINUE
960. 00 WRITE(6,500)X,P,XQ,DT,XQDIF2
970. 00 600 FORMAT(' PXQIN DID NOT CONVERGE; X,P,XQ,DT,XQDIF2=',5(1PE12.4))
980. 00 100 PLAST=P
990. 00 XLAST=X
1000. 00 XQLAST=XQ
1010. 00 TLAST=T
1020. 00 XLLAST=XL
1030. 00 XVLAST=XV
1040. 00 RETURN
      END

```

END ELT. ERRORS: NONE. TIME: 0.154 SEC. IMAGE COUNT: 104

@HDG,P ***** PXQIN2 ***** .L,0

```

***** PXQIN2 *****
@ELT,L DD,PXQIN2
ELT 8R1 S74Q1C 07/21/84 15:55:11 (O)
SUBROUTINE PXQIN2(X,P,XQ,T,XL,XV,VL,VV,H)
C
C
C ***** PURPOSE:
C TO CALC. TEMPERATURE & ENTHALPY OF 2-PHASE NON-AZEOTROPIC MIXTURE
C FORM GIVEN PRESSURE, COMPOSITION AND QUALITY
C
C ***** INPUT:
C P - PRESSURE (STD ATM)
C X - COMPOSITION (MOLAR FRACTION OF LESS VOLATILE COMPONENT)
C XQ - QUALITY (-)
C
C ***** OUTPUT:
C H - MIXTURE ENTHALPY (BTU/LB)
C T - TEMPERATURE OF MIXTURE (K)
C XL - COMPOSITION OF SAT. LIQUID AT TEMP. TK AND PRESSURE PA
C (MOLAR FRACTION OF LESS VOLATILE COMPONENT)
C XV - COMPOSITION OF SAT. VAPOR AT TEMP. TK AND PRESSURE PA
C (MOLAR FRACTION OF LESS VOLATILE COMPONENT)
C VL - SPECIFIC VOLUME OF SAT. LIQUID OF XL COMPOSITION
C AT TEMPERATURE TK (L/MOL)
C VV - SPECIFIC VOLUME OF SAT. VAPOR OF XV COMPOSITION
C AT TEMPERATURE TK (L/MOL)
C
C ***** SUBPROGRAMS CALLED BY PXQIN2:
C HCVCP,PXQIN,VOLIT1
C
IF(XQ.LE.0..OR.XQ.GE.1.) THEN
PRINT 666,XQ
666 FORMAT(' ERROR IN CALLING PXQIN2, XQ=',1PE16.6)
RETURN
END IF
C
IF(ABS(P-PLAST).GT.1.E-4)GOTO 10
IF(ABS(X-XLAST).GT.1.E-3)GOTO 10
IF(ABS(XQ-XQLAST).GT.1.E-4)GOTO 10
T=TLAST
XL=XLLAST
XV=XVLAST
VL=VLLAST
VV=VVLAST
H=HLAST
RETURN
C
10 CALL PXQIN(X,P,XQ,T,XL,XV)
CALL VOLIT1(O,T,P,XL,VL)
CALL HCVCP(1,T,VL,XL,HL,CV,CP)
CALL VOLIT1(1,T,P,XV,VV)
CALL HCVCP(1,T,VV,XV,HV,CV,CP)
H=(1.-XQ)*HL+XQ*HV
100 PLAST=P
XLAST=X
XQLAST=XQ
TLAST=T
XLLAST=XL
XVLAST=XV
560.

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***** PXQIN2 *****
570.      00      VLLAST=VL
580.      00      VVLAST=VV
590.      00      HLAST=H
600.      00      RETURN
610.      00      END

END ELT.  ERRORS: NONE.  TIME:  0.124 SEC.  IMAGE COUNT: 61

@HDG,P  ***** QLITY ***** .L,0
```

```

@ELT.L DD.QUALITY
ELT 8R1 S7401C 07/21/84 15:55:12 (0)
SUBROUTINE QLITY(T,P,X,XQ,XV,XL)
C
C
C**** PURPOSE:
C    TO CALCULATE QUALITY OF A BINARY FLUID
C
C**** INPUT:
C    IA - QUALIFIER OF PRECISION OF CALCULATIONS (-)
C    = 0, SIMPLIFIED CORRELATIONS TO BE USED
C    = POSITIVE INTEGER, COMPLETE SET EQUATIONS TO BE USED
C    P - PRESSURE (STD ATM)
C    T - TEMPERATURE (K)
C    X - MOLAR CONCENTRATION OF A LESS VOLATILE
C        COMPONENT (-)
C
C**** OUTPUT:
C    XL - MOLAR CONCENTRATION AT BUBBLE POINT (-)
C    XQ - FLUID QUALITY (-)
C    XV - MOLAR CONCENTRATION AT DEW POINT (-)
C
C**** SUBPROGRAMS CALLED BY QLITY:
C    SATCOM
C
C    COMMON/ACCURA/IA
C
C    IF(IA.EQ.0) THEN
C        T2=T*T
C        P1=10.0522804-2204.5632/T+9636.3313/T2
C        P1=EXP(P1)
C        P2=10.6410518-2642.8904/T+460.87585/T2
C        P2=EXP(P2)
C        GOTO 10
C    END IF
C
C    CALL SATCOM(T,P1,V1,V1,P2,V2,V2)
C    @FLUID IS LIQUID AT ALL CONCENTRATIONS
C
C    IF(P.GE.P1) THEN
C        XQ=0.
C        XL=0.
C        XV=0.
C        GOTO 1000
C    END IF
C
C    IF(P.LE.P2) THEN
C        @FLUID IS VAPOR AT ALL CONCENTRATIONS
C
C        XL=1.
C        XQ=1.
C        XV=1.
C        GOTO 1000
C    END IF
C
C    T2=T*T
C    RP=(P1-P)/(P1-P2)
C    SEG0=12.58741-0.06465226*T+9.56391E-5*T2
C    SEG8=-48.9799+0.289414*T-4.504915E-4*T2
C    SEG9=47.7509-0.273993*T+4.14187E-4*T2
C    SEG=SEG0+SEG8*RP+SEG9*RP**2
C    Z0=-0.34065415-3.337532E-3*T+9.66115E-6*T2
C    Z1=-10.3139754+0.0693956*T-9.9099162E-5*T2
C    Z2=10.035917-0.05527259*T+7.6821554E-5*T2

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***** QLITY *****
540.      00      Z=1.+Z0*(RP-1.)+Z1*(RP**2-1.)+Z2*(RP**3-1.)
550.      00      XL=(1.+SEG)*RP/(1.+SEG*RP) @SAT. LIQUID COMPOSITION
560.      00      XV=XL*Z @SAT. VAPOR COMPOSITION
570.      00      XQ=(XL-X)/(XL-XV)
580.      00      IF(XQ.GE.1.)XQ=1.
590.      00      IF(XQ.LT.0.)XQ=0.
600.      00      1000 RETURN
610.      00      END

END ELT.  ERRORS: NONE.  TIME:  0.125 SEC.  IMAGE COUNT: 64

@HDG,P ***** ROFANN ***** .L,0
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***** ROFANN *****
@ELT, L DD,ROFANN
ELT 8R1 S74Q1C 07/21/84 15:55:12 (0)
10. 00 FUNCTION ROFANN(XM,P,H0,GG)
20. 00
30. 00 C**** PURPOSE:
40. 00 TO CALCULATE DENSITY OF NON-AZEOTROPIC TWO-PHASE MIXTURE
50. 00 FOR ANY PRESSURE ON A FANNO FLOW PATH
60. 00
70. 00 C**** INPUT DATA:
80. 00 HO - REFRIG. TOTAL ENTHALPY (BTU/LB)
90. 00 GG = G*G/(64.4**73.104) (BTU*LB/FT**6)
100. 00 C WHERE G - REFRIG. MASS FLUX (LB/(SEC*FT**2))
110. 00 C P - REFRIG. PRESSURE AT WHICH ENTROPY IS DESIRED (PSIA)
120. 00 C XM - MOLAR COMPOSITION (FRACTION OF LESS VOLATILE COMPONENT)
130. 00 C**** OUTPUT DATA:
140. 00 ROFANN - ENTROPY (BTU/LB R)
150. 00 C
160. 00 C SUBPROGRAMS CALLED BY ROFANN:
170. 00 C EBUITE,ENTROP,HCVCP,QLITY,VOLITI
180. 00 C
190. 00 C COMMON/RDATA2/W1,W2,TC1,TC2
200. 00 C DATA SLOPE/O./NO,N1/O,1/
210. 00
220. 00 IF(P.GT.O.)GOTO 10
230. 00 WRITE(6,600)P
240. 00 600 FORMAT(' ERROR IN CALLING ROFANN, P=',1PE15.5,' PSIA')
250. 00 RETURN
260. 00 C
270. 00 10 PA=P/14.6959
280. 00 TK3=ERUBTE(PA,XM)
290. 00 TKD=TKB+9.5-20.*(1.55-XM)
300. 00 TB=TKB*1.8-459.67
310. 00 TD=TKD*1.8-459.67
320. 00 TK2=0.5*(TKB+TKD)
330. 00 DO 100 I=1,20
340. 00 CALL QLITY(TK2,PA,XM,XQ,XV,XL)
350. 00 CALL VOLITI(N1,TK2,PA,XV,VMV)
360. 00 CALL HCVCP(N1,TK2,VMV,XV,HV,CV,CP)
370. 00 CALL VOLITI(N0,TK2,PA,XL,VML)
380. 00 CALL HCVCP(N1,TK2,VML,XL,HL,CV,CP)
390. 00 HFG=HV-HL
400. 00 VMV=W1*(1.-XV)+W2*XV
410. 00 VML=W1*(1.-XL)+W2*XL
420. 00 VV=VMV*16.01646/WMV
430. 00 VL=VML*16.01646/VML
440. 00 VFG=VV-VL
450. 00 R=GG*VFG*VFG
460. 00 S=2.*GG*VV*VFG+HFG
470. 00 C=GG*VL*VL+HL-H0
480. 00 XQQ=(-S+SQRT(S*S-4.*R*C))/(2.*R)
490. 00 DIFXQ2=XQQ-XQ
500. 00 IF(ABS(DIFXQ2).LT.0.000001)GOTO 110
510. 00 C
520. 00 IF(1.GT.1)GOTO 70
530. 00 TK1=TK2
540. 00 DIFXQ1=DIFXQ2
550. 00 IF(SLOPE.NE.O.)THEN
560. 00 TK2=TK1-DIFXQ2*SLOPE

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***** ROFANN *****
570. 00
580. 00
590. 00
600. 00
610. 00
620. 00
630. 00
640. 00
650. 00
660. 00
670. 00
680. 00
690. 00
700. 00
710. 00
720. 00
730. 00

      ELSE
      TK2=TK1+SIGN(1.,DIFXQ2)
      END IF
      GOTO 100

C
      70 SLOPE=(TK2-TK1)/(DIFXQ2-DIFXQ1)
      IF(ABS(DIFXQ1).LT.ABS(DIFXQ2))GOTO 80
      TK1=TK2
      DIFXQ1=DIFXQ2
      80 TK2=TK1-DIFXQ1*SLOPE
      100 CONTINUE
      PRINT 602,DIFXQ2
      602 FORMAT(' ROFANN DOES NOT CONVERGE, DIFXQ2=',1PE15.5)
      110 CONTINUE
      ROFANN=1./(VL+XQQ*VFG)
      RETURN
      END

END ELT.  ERRORS: NONE.  TIME:  0.129 SEC.  IMAGE COUNT: 73

@HDG,P ***** SATCOM ***** .L,0

```

***** SATCOM *****

@ELT, L DD.SATCOM

ELT 8R1 S74Q1C 07/21/84 15:55:12 (0)

SUBROUTINE SATCOM(T,P1,V1,VL1,P2,V2,VL2)

C

C**** PURPOSE:

TO CALC. FOR PURE MIXTURE COMPONENTS AT SATURATION:

PRESSURE, SPEC. VOLUME OF LIQUID, SPEC. VOLUME OF VAPOR

C

C**** INPUT:

T - TEMPERATURE (K)

C

C**** OUTPUT:

P(1) - SAT. PRESSURE OF MORE VOLATILE REFRIGERANT (ATM)

P(2) - SAT. PRESSURE OF LESS VOLATILE REFRIGERANT (ATM)

V(1) - SPEC. VOLUME OF VAPOR OF MORE VOLATILE REFRIG. AT SATURATION (L/MOL)

V(2) - SPEC. VOLUME OF VAPOR OF LESS VOLATILE REFRIG. AT SATURATION (L/MOL)

VL(1) - SPEC. VOLUME OF LIQUID OF MORE VOLATILE REFRIG. AT SATURATION (L/MOL)

VL(2) - SPEC. VOLUME OF LIQUID OF LESS VOLATILE REFRIG. AT SATURATION (L/MOL)

C

C**** SUBPROGRAMS CALLED BY SATCOM:

FGIBBS, SATLIB, VOLIT

C

COMMON/RDATA2/W1,W2,TC1,TC2

DIMENSION P(2),X(2),V(2),VL(2)

DATA X(1),X(2)/0.,1./

C

CALL SATLIB(T,P(1),V(1),VL(1),P(2),V(2),VL(2))

IF(T.GT.(TC1-10.))GOTO 100

C

DO 100 I=1,2

DO 90 J=1,20

CALL VOLIT(T,P(1),X(1),V(1))

GV=FGIBBS(T,P(1),X(1),V(1))

CALL VOLIT(T,P(1),X(1),VL(1))

GL=FGIBBS(T,P(1),X(1),VL(1))

P7=(GL-GV)/(VL(1)-V(1))

P(1)=P(1)-P7/3.

P7=P7/P(1)

IF(ABS(P7).LT.1E-04)GOTO 100

90

CONTINUE

PRINT 600,I,P(1),P7

600 FORMAT(' ERROR 600 IN SATCOM, I,P,P7=',I3,2(1PE14.6))

100 CONTINUE

P1=P(1)

V1=V(1)

VL1=VL(1)

P2=P(2)

V2=V(2)

VL2=VL(2)

RETURN

END

END ELT. ERRORS: NONE. TIME: 0.113 SEC. IMAGE COUNT: 54


```

***** SATLIB *****
@ELT,L DD.SATLIB
ELT 8R1 S7401C 07/21/84 15:55:13 (0)
SUBROUTINE SATLIB(T,P1,V1,VL1,P2,V2,VL2)
10. 00 C
20. 00 C
30. 00 C
40. 00 C
50. 00 C
60. 00 C
70. 00 C
80. 00 C
90. 00 C
100. 00 C
110. 00 C
120. 00 C
130. 00 C
140. 00 C
150. 00 C
160. 00 C
170. 00 C
180. 00 C
190. 00 C
200. 00 C
210. 00 C
220. 00 C
230. 00 C
240. 00 C
250. 00 C
260. 00 C
270. 00 C
280. 00 C
290. 00 C
300. 00 C
310. 00 C
320. 00 C
330. 00 C
340. 00 C
350. 00 C

***** PURPOSE:
FOR GIVEN T, ESTIMATE SAT. PRESSURE, SPEC. VOLUME OF VAPOR,
AND SPEC. VOLUME OF LIQUID. OF MIXTURE PURE COMPONENTS

***** INPUT:
T - TEMPERATURE (K)

***** OUTPUT:
P1 - SAT. PRESSURE OF MORE VOLATILE COMPONENT (ATM)
P2 - SAT. PRESSURE OF LESS VOLATILE COMPONENT (ATM)
V1 - SPEC. VOLUME OF MORE VOLATILE COMPONENT SAT. VAPOR (L/MOL)
V2 - SPEC. VOLUME OF LESS VOLATILE COMPONENT SAT. VAPOR (L/MOL)
VL1 - SPEC. VOLUME OF MORE VOLATILE COMPONENT SAT. LIQUID (L/MOL)
VL2 - SPEC. VOLUME OF LESS VOLATILE COMPONENT SAT. LIQUID (L/MOL)

COMMON/ESDATA/A(2,3),B(2,3),C(2,3)

T2=T*T
P1=EXP(A(1,1)+A(1,2)/T+A(1,3)/T2)
V1=(B(1,1)+B(1,2)*T+B(1,3)*T2)*T/P1
VL1=C(1,1)+C(1,2)*T+C(1,3)*T2
P2=EXP(A(2,1)+A(2,2)/T+A(2,3)/T2)
V2=(B(2,1)+B(2,2)*T+B(2,3)*T2)*T/P2
VL2=C(2,1)+C(2,2)*T+C(2,3)*T2
RETURN
END

END ELT. ERRORS: NONE. TIME: 0.064 SEC. IMAGE COUNT: 35
CHDG,P ***** SPHDP1 ***** .L,0

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***** SPHDP1 *****
@ELT,L DD.SPHDP1
ELT 8R1 S74Q1C 07/21/84 15:55:13 (0)
10. 00 C FUNCTION SPHDP1(AM,AL,D,VSP,AMU)
20. 00 C
30. 00 C ***** PURPOSE:
40. 00 C TO COMPUTE FRICTIONAL PRESSURE DROP
50. 00 C FOR SINGLE PHASE FLOW IN A TUBE
60. 00 C
70. 00 C ***** INPUT DATA:
80. 00 C AL - TUBE LENGTH (FT)
90. 00 C AM - FLUID MASS FLOW RATE (LBM/H)
100. 00 C AMU - FLUID DYNAMIC VISCOSITY (LBM/H*FT)
110. 00 C D - TUBE DIAMETER (FT)
120. 00 C VSP - FLUID SPECIFIC VOLUME (FT**3/LBM)
130. 00 C
140. 00 ACC=3.3309E-11
150. 00 G=0.78539816*D*D
160. 00 G=AM/G
170. 00 RE=G*D/AMU
180. 00 F=0.046/RE**0.2
190. 00 SPHDP1=ACC*F*VSP*AL*G*G/D
200. 00 RETURN
210. 00 END

```

END ELT. ERRORS: NONE. TIME: 0.059 SEC. IMAGE COUNT: 21

@HDG,P ***** SPHTC ***** .L,0

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***** SPHTC *****
@ELT,L DD.SPHTC
ELT 8R1 S74Q1C 07/21/84 15:55:13 (0)
      10. 00      FUNCTION SPHTC(CP,AM,AK,RMASS,D)
      20. 00
      30. 00      C*****
      40. 00      C***** PURPOSE:
      50. 00      C      TO COMPUTE SINGLE PHASE HEAT TRANSFER COEFFICIENT
      60. 00      C      FOR FLOW INSIDE TUBE
      70. 00      C***** INPUT DATA:
      80. 00      C      AM - FLUID DYNAMIC VISCOSITY (LBM/FT*II)
      90. 00      C      AK - FLUID THERMAL CONDUCTIVITY (BTU/II*F*FT)
      100. 00      C      CP - FLUID SPECIFIC HEAT AT CONST. PRESSURE (BTU/LBM*F)
      110. 00      C      D - TUBE DIAMETER (FT)
      120. 00      C      RMASS - FLUID MASS FLOW RATE (LBM/H)
      130. 00      C
      140. 00      C***** OUTPUT DATA:
      150. 00      C      SPHTC - SINGLE PHASE HEAT TRANSFER COEFF. (BTU/H*F*FT**2)
      160. 00      C
      170. 00      G=RMASS/(0.7853982*D*D)
      180. 00      RE=D*O/AM
      190. 00      IF (RE.GE.2000.)GOTO10
      200. 00      SPHTC=4.36*AK/D
      210. 00      GOTO20
      220. 00      10 PR=(AM*CP/AK)**0.4
      230. 00      RE=RE**0.8
      240. 00      SPHTC=0.023*AK*PR**RE/D
      250. 00      20 RETURN
      260. 00      END

END ELT.  ERRORS: NONE.  TIME: 0.071 SEC.  IMAGE COUNT: 26

@HDG,P ***** SPIN ***** .L,0

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***** SPIN *****

```

@ELT,L DD,SPIN
ELT 8R1 S74Q1C 07/21/84 15:55:13 (0)
10. SUBROUTINE SPIN(IG,S,P,X,ACCUR,T,V)
20. C
30. C
40. C***** PURPOSE:
50. C TO CALC. TEMPERATURE OF BINARY MIXTURE VAPOR
60. C FROM GIVEN ENTROPY AND PRESSURE
70. C
80. C***** INPUT:
90. C ACCUR - ACCURACY OF CONVERGENCE (BTU/LB)
100. C IG = 0, IF GUESS OF TEMPERATURE IS NOT GIVEN
110. C = POSITIVE INTEGER, IF GUESS OF TEMPERATURE IS GIVEN
120. C P - PRESSURE (STD ATM)
130. C S - ENTROPY (BTU/LB*F)
140. C T - GUESS OF TEMPERATURE, OPTIONAL (K)
150. C X - MOLAR CONCENTRATION (FRACTION OF LESS VOLATILE COMPONENT)
160. C
170. C***** OUTPUT:
180. C T - BINARY MIXTURE TEMPERATURE (K)
190. C V - SPEC. VOLUME (L/MOL)
200. C
210. C***** SUBPROGRAMS CALLED BY SPIN:
220. C DEWTEM,ENTROP,ESVOL,VOLIT
230. C
240. C DATA SLOPE/0.,PLAST/0./
250. C
260. C IF(ABS(P-PLAST).GT.1.E-4)GOTO 10
270. C IF(ABS(S-SLAST).GT.ACCUR)GOTO 10
280. C IF(ABS(X-XLAST).GT.1.E-4)GOTO 10
290. C T=TLAST
300. C V=VLAST
310. C RETURN
320. C
330. C 10 N1=1
340. C CALL DEWTEM(0,P,X,TD,XL)
350. C IF(IG.EQ.0)T=TD+20.
360. C DO 50 I=1,15
370. C T=AMAX1(TD,T)
380. C V=ESVOL(N1,T,P,X)
390. C CALL VOLIT(T,P,X,V)
400. C SS=ENTROP(T,V,X)
410. C T2=T
420. C SDIF2=S-SS
430. C IF(ABS(SDIF2).LT.ACCUR)GOTO 100
440. C IF(SDIF1.EQ.SDIF2)THEN
450. C WRITE(6,601)
460. C GOTO 60
470. C
480. C END IF
490. C IF(1.NE.1)GOTO 20
500. C
510. C 15 T1=T2
520. C SDIF1=SDIF2
530. C IF(SLOPE.NE.0.)THEN
540. C DT=SDIF2*SLOPE
550. C IF(ABS(DT).GT.15.)DT=SIGN(15.,DT)
560. C T=T2-DT
570. C GOTO 50
580. C END IF

```


***** SPIN *****

```

570.      00
580.      00
590.      00
600.      00
610.      00
620.      00
630.      00
640.      00
650.      00
660.      00
670.      00
680.      00
690.      00
700.      00
710.      00
720.      00
730.      00
740.      00
750.      00
760.      00
770.      00
780.      00

      C
      T=T2+10.
      IF(SDIF2.LT.0.)T=T2-10.
      GOTO 50

      20 SLOPE=(T2-T1)/(SDIF2-SDIF1)
      IF(ABS(SDIF1).LT.ABS(SDIF2))GOTO 30
      T1=T2
      SDIF1=SDIF2
      30 DT=SDIF1*SLOPE
      IF(ABS(DT).GT.15.)DT=SIGN(15.,DT)
      T=T1-DT
      50 CONTINUE
      CO WRITE(6,600)SDIF2
      600 FORMAT(' ERROR 600 IN SPIN, SDIF2=',1PE12.4)
      601 FORMAT(' SPIN DOES NOT CONVERGE ANY FURTHER=')
      100 PLAST=P
      SLAST=S
      XLAST=X
      TLAST=T
      VLAST=V
      RETURN
      END

```

END ELT. ERRORS: NONE. TIME: 0.142 SEC. IMAGE COUNT: 78

@HDG,P ***** TPROP ***** .L,0

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```

***** TPROP *****
@ELT, L DD, TPROP
ELT 8R1 S74Q1C 07/21/84 15:55:14 (0)
SUBROUTINE TPROP(TF,PSIA,XW,XQ,H,V)
10. C
20. C**** PURPOSE:
30. C    TO CALCULATE NON-AZEOTROPIC MIXTURE THERMODYNAMIC PROPERTIES
40. C    FROM GIVEN TEMPERATURE, PRESSURE AND COMPOSITION
50. C
60. C
70. C**** INPUT:
80. C    TF - TEMPERATURE (F)
90. C    PSIA - PRESSURE (PSIA)
100. C    XW - WEIGHT COMPOSITION (FRACTION OF LESS VOLATILE COMPONENT)
110. C
120. C**** OUTPUT:
130. C    H - ENTHALPY (BTU/LB)
140. C    V - SPEC. VOLUME (FT**3/LB)
150. C    XQ - QUALITY (-)
160. C
170. C
180. C
190. C
200. C
210. C
220. C
230. C
240. C
250. C
260. C
270. C
280. C
290. C
300. C
310. C
320. C
330. C
340. C
350. C
360. C
370. C
380. C
390. C
400. C
410. C
420. C

TK=(TF+459.67)/1.8
PA=PSIA/14.6959
XM=XW/(W2/W1*(1.-XW)+XW)
WM=(1.-XM)*W1+XM*W2
CALL GLITY(TK,PA,XM,XQ,XV,XL)
IF(XQ.NE.0..AND.XQ.NE.1.)GOTO 10
N=1.0001*XQ
CALL VOLIT1(N,TK,PA,XM,V)
CALL HCVCP(1,TK,V,XM,H,CV,CP)
V=V*16.01845/WM
GOTO 1000

10 CALL VOLIT1(0,TK,PA,XL,VL)
CALL HCVCP(1,TK,VL,XL,HL,CV,CP)
CALL VOLIT1(1,TK,PA,XV,VV)
CALL HCVCP(1,TK,VV,XV,HV,CV,CP)
H=(1.-XQ)*HL+XQ*HV
WML=(1.-XL)*W1+XL*W2
WMV=(1.-XV)*W1+XV*W2
VLV=VL*16.01846/WM
VVV=VV*16.01846/WMV
V=(1.-XQ)*VLM+XQ*VVW
1000 RETURN
END

END ELT. ERRORS: NONE. TIME: 0.095 SEC. IMAGE COUNT: 42
@HOG,P ***** TPROP2 ***** .L,0

```



```

***** TXQIN *****
@ELT,L 00.TXQIN
ELT 0R1 S740IC 07/21/84 15:55:14 (0)
SUBROUTINE TXQIN(X,T,XQ,P,XL,XV)
10. C
20. C
30. C
40. C
50. C
60. C
70. C
80. C
90. C
100. C
110. C
120. C
130. C
140. C
150. C
160. C
170. C
180. C
190. C
200. C
210. C
220. C
230. C
240. C
250. C
260. C
270. C
280. C
290. C
300. C
310. C
320. C
330. C
340. C
350. C
360. C
370. C
380. C
390. C
400. C
410. C
420. C
421. C
422. C
430. C
460. C
470. C
480. C
490. C
500. C
510. C
520. C
530. C
540. C
550. C
560. C

C**** PURPOSE:
TO CALC. PRESSURE OF NON-AZEOTROPIC MIXTURE
FORM GIVEN COMPOSITION, TEMPERATURE AND QUALITY

C**** INPUT:
T - TEMPERATURE (F)
X - COMPOSITION (MOLAR FRACTION OF LESS VOLATILE COMPONENT)
XQ - QUALITY (-)

C**** OUTPUT:
P - PRESSURE OF MIXTURE (STD ATM)
XL - COMPOSITION OF SAT. LIQUID AT TEMP. TK AND PRESSURE PA
(MOLAR FRACTION OF LESS VOLATILE COMPONENT)
XV - COMPOSITION OF SAT. VAPOR AT TEMP. TK AND PRESSURE PA
(MOLAR FRACTION OF LESS VOLATILE COMPONENT)

C**** SUBPROGRAMS CALLED BY TXQIN:
EBUBPR,QLITY

DATA SLOPE/0.,TLAST/0./

IF(XQ.LE.0..OR.XQ.GE.1.)THEN
PRINT 677,XQ
677 FORMAT(' ERROR IN CALLING TXQIN, XQ=',1PE16.6)
RETURN
END IF

IF(ABS(T-TLAST).GT.1.E-3)GOTO 10
IF(ABS(X-XLAST).GT.1.E-3)GOTO 10
IF(ABS(XQ-XLAST).GT.1.E-4)GOTO 10
P=PLAST
XL=XLLAST
XV=XVLAST
RETURN

C
10 P=E3('3PR(T,X)-XQ*SLOPE
DO 50 I=1,20
CALL QLITY(T,P,X,XQN,XV,XL)
P2=P
IF(XQN.EQ.0.)CALL BUBPRE(0,T,X,P2,XV)
IF(XQN.EQ.1.)CALL DEWPRE(0,T,X,P2,XL)
XQDIF2=XQ-XQN
IF(1.NE.1)GOTO 20
P1=P2
XQDIF1=XQDIF2
IF(SLOPE.NE.0.)THEN
DP=XQDIF2*SLOPE
IF(ABS(DP).GT.1.)DP=SIGN(1.,DP)
P=P2-DP
GOTO 50
END IF
P=P2-.5
IF(XQDIF2.LT.0.)P=P2+.5

```



```

***** TXQIN *****
570. 00
580. 00
590. 00
600. 00
610. 00
620. 00
630. 00
640. 00
650. 00
660. 00
670. 00
680. 00
690. 00
700. 00
710. 00
720. 00
730. 00
740. 00
750. 00
760. 00
770. 00
780. 00
790. 00
800. 00
810. 00
820. 00
830. 00
840. 00
850. 00
860. 00

C
GOTO 50
20 IF(XQDIF1.NE.XQDIF2)GOTO 25
WRITE(6,610)XQDIF1
610 FORMAT(' TXQIN DOES NOT CONVERGE ANY FURTHER, XQDIF1=',1PE14.5)
GOTO 110

C
25 SLOPE=(P2-P1)/(XQDIF2-XQDIF1)
IF(ABS(XQDIF1).LT.ABS(XQDIF2))GOTO 30
P1=P2
XQDIF1=XQDIF2
30 DP=XQDIF1-SLOPE
IF(ABS(DP).LT.0.002.AND.ABS(XQDIF2).LT.0.0001)GOTO 100
IF(ABS(DP).GT.2.)DP=SIGN(2.,DP)
P=P1-DP
50 CONTINUE
WRITE(6,600)XQDIF2
600 FORMAT(' ERROR 600 IN TXQIN, XQDIF2=',1PE12.4)
100 PLAST=P
XLAST=X
XQLAST=XQ
TLAST=T
XLLAST=XL
XVLAST=XV
110 RETURN
END

```

END ELT. ERRORS: NONE. TIME: 0.135 SEC. IMAGE COUNT: 82

@HDG,P ***** TXQIN2 ***** .L,0

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```
***** TXQIN2 *****
570.          00          VLLAST=VL
580.          00          VVLAST=VW
590.          00          HVLAST=H
600.          00          RETURN
610.          00          END

END ELT.  ERRORS: NONE.  TIME:  0.115 SEC.  IMAGE COUNT:  61

@HDS,P ***** VALVPA ***** .L,0
```

```

***** VALVPA *****
@ELT,L DD,VALVPA
ELT 8R1 S74Q1C 07/21/84 15:55:15 (0)
10. 00 SUBROUTINE VALVPA
20. 00 C
30. 00 C
40. 00 C
50. 00 C
60. 00 C
70. 00 C
80. 00 C
90. 00 C
100. 00 C
110. 00 C
120. 00 C
130. 00 C
140. 00 C
150. 00 C
160. 00 C
170. 00 C
180. 00 C
190. 00 C
200. 00 C
210. 00 C
220. 00 C
230. 00 C
240. 00 C
250. 00 C
260. 00 C
270. 00 C
280. 00 C
290. 00 C
300. 00 C
310. 00 C
320. 00 C
330. 00 C
340. 00 C
350. 00 C
360. 00 C
370. 00 C
380. 00 C
390. 00 C
400. 00 C
410. 00 C
420. 00 C
430. 00 C
440. 00 C
450. 00 C
460. 00 C
470. 00 C
480. 00 C
490. 00 C
500. 00 C
510. 00 C
520. 00 C
530. 00 C
540. 00 C
550. 00 C
560. 00 C

***** PURPOSE:
TO EVALUATE 4-WAY VALVE PERFORMANCE PARAMETERS
FROM TEST UNDER ONE OPERATION CONDITION
(BINARY MIXTURE REFRIGERANT)

***** INPUT DATA:
C [1] REFRIGERANT CONSTANTS
LISTED IN COMMON STATEMENTS RDATA1 & RDATA2
C [2] TEST DATA AS EXPLAINED BELOW:
P2,T2 - REFRIG. PRESSURE & TEMP. AT VALVE LOW PRESSURE INLET
(P2IA),(F)
P3,T3 - REFRIG. PRESSURE & TEMP. AT VALVE LOW PRESSURE OUTLET
(P3IA),(F)
P8,T8 - REFRIG. PRESSURE & TEMP. AT VALVE HIGH PRESSURE INLET
(P8IA),(F)
P9,T9 - REFRIG. PRESSURE & TEMP. AT VALVE HIGH PRESSURE OUTLET
(P9IA),(F)
RMASS - REFRIG. MASS FLOW RATE (LBM/H)
XW - MIXTURE COMPOSITION
(WEIGHT FRACTION OF MORE VOLATILE COMPONENT)

***** OUTPUT DATA:
- PARAMETER FOR 4-WAY VALVE PRESSURE DROP
(LBF*H**2/LBM*IN**2*FT**3)
- PARAMETER FOR PRESSURE DROP ON HIGH PRESSURE SIDE
(LBF*H**2/LBM*IN**2*FT**3)
- PARAMETER FOR PRESSURE DROP ON LOW PRESSURE SIDE
(LBF*H**2/LBM*IN**2*FT**3)
- PARAMETER FOR 4-WAY VALVE HEAT TRANSFER (FT**2)
- HEAT TRANSFER RATE BETWEEN HIGH & LOW PRESSURE REFRIG.
(BTU/H)
- LOW PRESSURE REFRIG. VAPOR HEAT GAIN (BTU/H)
- HIGH PRESSURE REFRIG. VAPOR HEAT LOSS (BTU/H)

***** SUBPROGRAMS CALLED BY VALVPA:
ESVOL,HCVCP,VISCON,VOLIT

COMMON/RDATA1/A3,A4,A5,A6,A7,A8,B3,B4,B5,D6,B7,B8,F0,F1
COMMON/RDATA2/W1,W2,TC1,TC2
DIMENSION TK(4),PA(4),VM(4),H(4),CP(4),AK(4),AM(4)

WRITE(6,19)
15 WRITE(6,20)
READ(5,777)T11,P11,T2,P2
WRITE(6,21)
READ(5,777)T7,P7,T71,P71
WRITE(6,22)
READ(5,777)RMASS,XW
XW=1.-XW
XM=W2/W1*(1.-XW)+XW

```


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***** VALVPA *****

1150. 00 RETURN
1160. 00 END

END ELT. ERRORS: NONE. TIME: 0.128 SEC. IMAGE COUNT: 116

@HDG,P ***** VISCON ***** .L,0

***** VISCON *****

@ELT,L DD.VISCON

ELT 8R1 S74Q1C 07/21/84 15:55:15 (0)

FUNCTION VISCON(IQ,T,XW)

10. 00

20. 00

30. 00

40. 00

50. 00

60. 00

70. 00

80. 00

90. 00

100. 00

110. 00

120. 00

130. 00

140. 00

150. 00

160. 00

170. 00

180. 00

190. 00

200. 00

210. 00

220. 00

230. 00

240. 00

250. 00

260. 00

270. 00

280. 00

290. 00

300. 00

310. 00

320. 00

330. 00

340. 00

350. 00

360. 00

370. 00

380. 00

390. 00

400. 00

410. 00

420. 00

430. 00

440. 00

450. 00

460. 00

470. 00

480. 00

490. 00

500. 00

510. 00

520. 00

530. 00

540. 00

550. 00

560. 00

C**** PURPOSE:

TO CALCULATE ABS. VISCOSITY OR THERMAL CONDUCTIVITY
OF LIQUID OR VAPOR, OR SPEC. HEAT OF SAT. LIQUID
OF R-13B1/R-152A MIXTURE

C**** INPUT:

IQ

- OUTPUT QUALIFIER

= 1, IF LIQUID ABS. VISCOSITY REQUIRED

= 2, IF LIQUID THERMAL CONDUCTIVITY REQUIRED

= 3, IF VAPOR ABS. VISCOSITY REQUIRED

= 4, IF VAPOR THERMAL CONDUCTIVITY REQUIRED

= 5, IF SPEC. HEAT OF SAT. LIQUID REQUIRED

T

XW

- WEIGHT COMPOSITION (FRACTION OF R-152A)

C**** OUTPUT:

VISCON

- LIQ. ABS. VISCOSITY (IQ=1) (LB/(H*FT))

- LIQ. THERMAL CONDUCTIVITY (IQ=2) (BTU/(H*FT*F))

- VAP. ADS. VISCOSITY (IQ=3) (LB/(H*FT))

- VAP. THERMAL CONDUCTIVITY (IQ=4) (BTU/(H*FT*F))

- SAT. LIQUID SPEC. HEAT (IQ=5) (BTU/(LB*F))

DATA W1,W2/148.93,66.05/,TB1,TB2/215.4,248.15/

DATA R/0.125885/

C
GOTO (10,20,30,50)IQ

C**** LIQ. ABS. VISCOSITY

10 TK=(T+459.68)/1.8

TK2=TK*TK

VL1=0.2749422-0.001702569*TK+3.71008313E-G*TK2

VL1=VL1*16.01845/W1

VL2=0.102368715-4.0752759E-4*TK

VL2=VL2+1.0409447E-6*TK2

VL2=VL2*16.01846/W2

VLM=(1.-XW)*VL1+XW*VL2

V11=2419.09*EXP(-4.22529+710.843/TK)*1.E-3

V11=V11*VL1

V12=2419.09*EXP(-4.28224-753.013/TK)*1.E-3

V12=V12*VL2

F1A=(1.-XW)*VL1/VLM

F1B=1.-F1A

SEG=ALOG(VL2/VL1)

ALFAB=-1.7*SEG

ALFAB=0.27*SEG+SQRT(1.3*SFG)

VISCON=F1A*V11*EXP(F1B*ALFAB)+F1B*V12*EXP(F1A*ALFAB)

VISCON=VISCON/VLM

GOTO 1000

C

C**** LIQ. THERMAL CONDUCTIVITY

20 C1=0.035-1.5E-4*T

C2=0.577778*(0.1165-4.97E-4*(T-32.)/1.8)

VISCON=C1+(C2-C1)*XW*(0.72*XW+0.28)

```

***** VISCON *****
570. 00
580. 00
590. 00
600. 00
610. 00
620. 00
630. 00
640. 00
650. 00
660. 00
670. 00
680. 00
690. 00
700. 00
710. 00
720. 00
730. 00
740. 00
750. 00
760. 00
770. 00
780. 00
790. 00
800. 00
810. 00
820. 00
830. 00
840. 00
850. 00
860. 00
870. 00
880. 00
890. 00
900. 00
910. 00
920. 00
930. 00
940. 00
950. 00

C
C**** VAP. ABS. VISCOSITY
30 TK=(T+459.67)/1.8
TK2=TK*TK
X=W2/W1*(1.-XW)+XW
X=XW/X
V11=-0.67329+7.60593E-3*TK-2.81108E-5*TK2
V11=2.41909*(V11+3.4741E-8*TK*TK2)
C2=0.1*(-0.08357+6.32E-4*TK+4.257E-7*TK2)
CP=-7.33704+0.093439*TK-3.61094E-4*TK2+4.80149E-7*TK2*TK
FE=0.115+0.354*CF/R
V12=2.41909+0.03205*C2*W2/FC
IF(10.NE.3)GOTO 40
F12=(1.+SQRT(V11/V12))*(W2/W1)**0.25)**2
F12=F12/SQRT(8+8*W1/W2)
F21=F12*V12*W1/(V11*W2)
VISCON=(1.-X)*V11/(1.-X+X*F12)
VISCON=VISCON+X*V12/(X+(1.-X)*F21)
GOTO 1000

C**** THERMAL CONDUCTIVITY
40 C1=8.2982E-3-5.1971E-5*TK+1.8413E-7*TK2
S1=1.5*TB1
S2=1.5*TB2
S12=0.73*SQRT(S1*S2)
SEG=SQRT(V11*(TK+S1)*(W2/W1)**0.75/(V12*(TK+S2)))
A12=.25*(1.+SEG)**2*(TK+S12)/(TK+S1)
A21=.25*(1.+1./SEG)**2*(TK+S12)/(TK+S2)
VISCON=(1.-X)*C1/(1.-X+X*A12)
VISCON=0.577789*(VISCON+X*C2/((1.-X)*A21+X))
GOTO 1000

C**** SPEC. HEAT OF SAT. LIQUID
50 TK=(T+459.67)/1.8
TK2=TK*TK
C1=-1.42637+0.0278024*TK-1.23753E-4*TK2+1.92012E-7*TK*TK2
C2=-1.78539+0.0252241*TK-8.71975E-5*TK2+1.39262E-7*TK*TK2
VISCON=0.23902*(1.-XW)*C1+XW*C2
1000 RETURN
END

END ELT. ERRORS: NONE. TIME: 0.171 SEC. IMAGE COUNT: 95
@HDBG,P ***** VOLIT1 ***** .L,0

```



```

***** VOLIT1 *****
@ELT,L DD,VOLIT1
ELT 8R1 S74Q1C 07/21/84 15:55:15 (0)
10. 00 SUBROUTINE VOLIT1(N,T,P5,X,V)
20. 00 C
30. 00 C**** PURPOSE:
40. 00 TO ITERATE REFRIG. MIXTURE R-13B1/R-152A SPEC. VOL.
50. 00 FROM EQUATION OF STATE
60. 00 C
70. 00 C**** INPUT:
80. 00 N - OUTPUT QUALIFIER
90. 00 = 0, IF SPEC. VOL. OF LIQUID IS REQUIRED
100. 00 = 1, IF SPEC. VOL. OF VAPOR IS REQUIRED
110. 00 T - REFRIG. TEMPERATURE (K)
120. 00 P5 - REFRIG. SAT. PRESSURE (STD ATM)
130. 00 X - MOLAR COMPOSITION (FRACTION OF LESS VOLATILE COMPONENT)
140. 00 * - REFRIG. CONSTANTS A & B (SEE COMMON STATEMENT /PARAM1/)
150. 00 C**** OUTPUT:
160. 00 V - REFRIG. SPEC. VOLUME (L/MOL)
170. 00 C
180. 00 C**** SUBPROGRAMS CALLED BY VOLIT1:
190. 00 ESVOL, EQPAR
200. 00 C
210. 00 C
220. 00 COMMON/PARAM/A,B,C1,C2,D1,D2
230. 00 DATA R/0.08206/
240. 00 C
250. 00 IF(P5.GT.0.)GOTO 10
260. 00 PRINT 601,P5
270. 00 601 FORMAT(' VOLIT1 CALLED WITH NEG. PRESSURE, P5=',1PE13.3)
280. 00 RETURN
290. 00 C
300. 00 10 IF(T.LT.150.)PRINT 602,T
310. 00 602 FORMAT(' WARNING, VOLIT1 CALLED WITH TEMP.=',1PE11.3)
320. 00 C
330. 00 V=FSVOL(N,T,P5,X)
340. 00 CALL EQPAR(T,X)
350. 00 DO 100 I=1,25
360. 00 Y=B/(4.*V)
370. 00 Y2=Y*Y
380. 00 Y3=Y2*Y
390. 00 Y4=Y3*Y
400. 00 P=(R*T*(1.+Y+Y2-Y3)/(1.-Y)**3-A/(V+B))/V
410. 00 P6=(P5-P)/P5
420. 00 IF(AEG(P5).LT.1.E-04)GOTO 1000
430. 00 P0=-R*T/V**2*(1.+4.*Y+4.*Y2-4.*Y3+Y4)/(1.-Y)**4
440. 00 P0=PC+A*(2.*V+B)/(V*(V+B))**2
450. 00 V6=(P-P5)/P0
460. 00 V7=V-V6
470. 00 IF(V7.GT.(B/4.))GOTO 50
480. 00 V=V-(V-B/4.)/10.
490. 00 GOTO 100
500. 00 50 V=V7
510. 00 100 CONTINUE
520. 00 WRITE(6,600)P5
530. 00 600 FORMAT(' ERROR 600 IN VOLIT1, P6 =',1PE12.4)
540. 00 1000 RETURN
550. 00 END

```

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***** VOLIT1 *****

END ELT. ERRORS: NONE. TIME: 0.121 SEC. IMAGE COUNT: 55

@HDG,P ***** WACCUM ***** .L,0

```

***** WACCUM *****
@ELT, L DD WACCUM
ELT 8R1 S74Q1C 07/21/84 15:55:16 (0)
10. 00 SUBROUTINE WACCUM(T,P,RMASS,WACC,XWA)
20. 00 C
30. 00 C**** PURPOSE:
40. 00 TO CALCULATE MASS OF NON-AZEOTROPIC REFRIGERANT
50. 00 IN AN ACCUMULATOR
60. 00 3/13/84
70. 00 C
80. 00 C**** INPUT DATA:
90. 00 AHGT - ACCUMULATOR HEIGHT (FT)
100. 00 DACC - INNER DIAMETER OF ACCUMULATOR (FT)
110. 00 DHOLE(1) - INNER DIA. OF THE OIL RETURN HOLE (FT)
120. 00 DHOLE(2) - INNER DIA. OF THE UPPER HOLE (FT)
130. 00 DTUBE - INNER DIA. OF ACCUMULATOR TUBE (FT)
140. 00 HDIS - VERTICAL DISTANCE BETWEEN HOLES (FT)
150. 00 P - REFRIG. PRESSURE (PSIA)
160. 00 RMASS - REFRIG. MASS FLOW RATE (LB/H)
170. 00 T - REFRIG. TEMPERATURE (F)
180. 00 C
190. 00 C**** OUTPUT DATA:
200. 00 XWA - WEIGHT COMPOSITION OF MIXTURE IN ACCUMULATOR
210. 00 C (FRACTION OF LESS VOLATILE COMPONENT)
220. 00 WACC - MASS OF REFRIG. IN THE ACCUMULATOR (LB)
230. 00 C
240. 00 C**** SUBPROGRAMS CALLED BY WACCUM:
250. 00 QLITY, VOLIT1
260. 00 C
270. 00 COMMON/RDATA2/W1,W2,TC1,TC2
280. 00 COMMON/RDATA3/XW,XM,WM
290. 00 COMMON/ACCDIM/AHGT,DACC,DHOLE(2),DTUBE,HDIS
300. 00 DIMENSION AHOLE(2)
310. 00 REAL*8 WL,AHOLE,PD,HL(2),HLL,HLDIFA,HLDIFB,HLLA,ZZ,Y1,Y2,
320. 00 & HLLB,Z1,Z2,DP,RM,RMT,DIFF1,DIFF2,SLOPE,ALFA,AT
330. 00 REAL*8 RO,PDYN,AHOLE(2),RMSL
340. 00 DATA PI/3.1415927/
350. 00 C
360. 00 WACC=PI*DACC*DACC/4.
370. 00 PA=P/14.6959
380. 00 TK=(T+459.67)/1.8
390. 00 CALL QLITY(TK,PA,XM,X,XV,XL)
400. 00 XLW=XL/(W1/W2*(1.-XL)+XL)
410. 00 XVW=XV/(W1/W2*(1.-XV)+XV)
420. 00 C
430. 00 IF(X.GT..999)THEN
440. 00 @EVAPOR ONLY IN ACCUMULATOR
450. 00 CALL VOLIT1(1,TK,PA,XM,V)
460. 00 V=V*16.01846/WM
470. 00 WACC=WACC*AHGT/V
480. 00 XWA=XW
490. 00 GOTO 100
500. 00 END IF
510. 00 C
520. 00 CALL VOLIT1(0,TK,PA,XL,V)
530. 00 WML=W1*(1.-XL)+W2*XL
540. 00 RO=WML/(V*16.01846)
550. 00 CALL VOLIT1(1,TK,PA,XV,V)
560. 00 WMV=W1*(1.-XV)+W2*XV
570. 00 V=V*16.01846/WMV

```

***** WACCUM *****

```

570. RMS=RMASS/3600.
580. RMS'=(1.-X)*RMS
590. AHOLE(1)=PI*DHOLE(1)*DHOLE(1)/4.
600. ATUBE=PI*DTUBE*DTUBE/4.
610. AHOL=AHOLE(1)
620. HLL=DHOLE(1)
630. PDYN=0.5*(X*RMS/ATUBE)**2*V
640.
650.
660.
670.
680.
690.
700.
710.
720.
730.
740.
750.
760.
770.
780.
790.
800.
810.
820.
830.
840.
850.
860.
870.
880.
890.
900.
910.
920.
930.
940.
950.
960.
970.
980.
990.
1000.
1010.
1020.
1030.
1040.
1050.
1060.
1070.
1080.
1090.
1100.
1110.
1120.
1130.
1140.

C
DO 10 I=1,20
VL=RMSL/AHOL
PD=VL*VL/(.585*.585*2.*RO)
HL(1)=(PD-PDYN)/(RO*32.2)
IF(1.EQ.1.AND.HL(1).GT.DHOLE(1))GOTO 12
IF(ABS((HL(1)-HLL)/DHOLE(1)).LT.0.01)GOTO 12
IF(1.EQ.1)THEN
HLL=DHOLE(1)/2.
HLLA=DHOLE(1)
HLDIFA=HL(1)-DHOLE(1)
ELSE
HLDIFB=HL(1)-HLL
HLLB=HLL
HLLA=HLLA-HLDIFA*(HLLA-HLLB)/(HLDIFA-HLDIFB)
HLLA=HLLB
HLLA=HLLB
HLDIFA=HLDIFB
IF(HLL.LT.0.)HLL=.9*HLLA
END IF
R=0.5*DHOLE(1)
Z1=HLL-R
Z2=DSQRT(R*R-Z1*Z1)
AT=Z1*Z2
ALFA=2.*ARCOS(ABS(Z1/R))
IF(HLL.GT.R)ALFA=2.*PI-ALFA
AHOL=AHOLE(1)*ALFA/(2.*PI)+AT
10 CONTINUE
WRITE(6,90)
90 FORMAT(' WACCUM LOOP 10 DID NOT CONVERGE')
GOTO 40

C
12 VHGT=AHGT-HL(1)
VHGT=AMAX1(0.,VHGT)
AMASS2=AACC*(HL(1)*RO+VHGT/V)
IF(HL(1).LE.HDIS.OR.HDIS.EQ.0.)GOTO 40
AHOLE(2)=PI*DHOLE(2)*DHOLE(2)/4.
Z1=HL(1)
HL(2)=HL(1)-HDIS
DP=HL(2)*RO*32.2+PDYN
RM=0.585*AHOLE(2)*DSQRT(2.*RO*DP)
DIFF1=RM
VHGT=AHGT-HL(1)
VHGT=AMAX1(0.,VHGT)
AMASS1=AACC*(Z2*RO+VHGT/V)
Z2=HDIS
DO 30 J=1,20
HL(1)=Z2
HL(2)=Z2-HDIS
RMT=0.00
DO 16 I=1,2
DP=HL(1)*RO*32.2+PDYN

```


***** WACCUM *****

```

1150. 00
1160. 00
1170. 00
1180. 00
1190. 00
1200. 00
1210. 00
1220. 00
1230. 00
1240. 00
1250. 00
1260. 00
1270. 00
1280. 00
1290. 00
1300. 00
1310. 00
1320. 00
1330. 00
1340. 00
1350. 00
1360. 00
1370. 00
1380. 00
1390. 00
1400. 00
1410. 00
1420. 00
1430. 00
1440. 00
1450. 00
1460. 00
1470. 00
1480. 00
1490. 00
1500. 00
1510. 00
1520. 00
1530. 00
1540. 00
1550. 00
1560. 00
1570. 00
1580. 00
1590. 00
1600. 00

IF (HL(1).EQ.0.D0)DP=0.D0
AHOL=1.
IF (1.EQ.2) THEN
  IF (HL(2).GT.DHOLE(2))GOTO 15
  IF (HL(2).EQ.0.D0)GOTO 15
  R=.5*DHOLE(2)
  Y1=HL(2)-R
  Y2=DSQRT(R*R-Y1*Y1)
  AT=Y1*Y2
  ALFA=2.*ARCCOS(DABS(Y1/R))
  IF (HL(2).GT.R)ALFA=2.*PI-ALFA
  AHOL=AHOLE(2)*ALFA/(2.*PI)+AT
  AHOL=AHOLE(2)/AHOL
END IF
15 RM=0.585*AHOLE(1)*DSQRT(2.*R0*DP)/AHOL
16 PMT=RMT+RM
DIFF2=RMT-FMSL
VHGT=AHGT-HL(1)
VHGT=AMAX1(0.,VHGT)
AMASS2=AACC*(HL(1)*R0+VHGT/V)
IF (ABS(AMASS1-AMASS2).LT.0.01)GOTO 40
SLOPE=(Z1-Z2)/(DIFF1-DIFF2)
Z2=Z2-DIFF2*SLOPE
IF (ABS(DIFF1).LT.ABS(DIFF2))GOTO 20
AMASS1=AMASS2
DIFF1=DIFF2
Z1=Z2
20 Z2=Z2
Z2=DMAX1(Z2,HDIS)
30 CONTINUE
ERROR=AACC*ABS(Z2-Z1)*R0
WRITE(6,602)ERROR
40 CONTINUE
IF (VHGT.EQ.0.)WRITE(6,600)
WACC=AMASS2
W152A=AACC*(HL(1)*R0+XLW+VHGT/V*XVW)
XWA=W152A/WACC
100 XWB=1.-XWA
WRITE(6,604)XWB,WACC
600 FORMAT(' ACCUMULATOR OVERFILLED')
602 FORMAT(' WACCUM DOES NOT CONVERGE, MAX.ERROR =',1PE10.3,' (LB)')
604 FORMAT(' COMPOSITION OF REFRIG. IN ACCUMULATOR =',F6.3,/,
@ ' (WEIGHT FRACTION OF MORE VOLATILE COMPONENT)',/,
@ ' REFRIG. IN ACCUMULATOR =',F8.3,' LB')
RETURN
END

```

END ELT. ERRORS: NONE. TIME: 0.217 SEC. IMAGE COUNT: 160

@HDG,N

@OFFSEND,S

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U.S. DEPT. OF COMM. BIBLIOGRAPHIC DATA SHEET <i>(See instructions)</i>	1. PUBLICATION OR REPORT NO. NBS/TN-1218	2. Performing Organ. Report No.	3. Publication Date January 1986
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10. SUPPLEMENTARY NOTES <input type="checkbox"/> Document describes a computer program; SF-185, FIPS Software Summary, is attached.			
11. ABSTRACT <i>(A 200-word or less factual summary of most significant information. If document includes a significant bibliography or literature survey, mention it here)</i> <p>An analysis of the vapor compression cycle and the main components of an air-to-air heat pump charged with a binary non-azeotropic mixture has been performed for steady-state operation. The general heat pump simulation model HPBI has been formulated which is based on independent, analytical models of system components and the logic linking them together. The logic of the program requires an iterative solution of refrigerant pressure and enthalpy balances, and refrigerant mixture and individual mixture component mass inventories.</p> <p>The modeling effort emphasis was on the local thermodynamic phenomena which were described by fundamental heat transfer equations and equation of state relationships among material properties. In the compressor model several refrigerant locations were identified and the processes taking place between these locations accounted for all significant heat and pressure losses. Evaporator and condenser models were developed on a tube-by-tube basis where performance of each coil tube is computed separately by considering the cross-flow heat transfer with the external air stream and the appropriate heat and mass transfer relationships. A capillary tube model was formulated with the aid of Fanno flow theory. Equation of state for mixtures is described and equation constants for R13B1/R152a mixture are given.</p> <p>The developed heat pump model was validated by checking computer results against laboratory tests data of one heat pump at two cooling and two heating rating points.</p> <p>Program HPBI can be used to evaluate potentials of non-azeotropic mixtures working in a heat pump. User's Manual and listing of the program is included in the report.</p>			
12. KEY WORDS <i>(Six to twelve entries; alphabetical order; capitalize only proper names; and separate key words by semicolons)</i> air conditioner; capillary tube; coil; compressor; condenser; expansion device; heat pump; modeling; mixture; non-azeotropic refrigerant; vapor modeling cycle			
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